



GO BEYOND



Pratt & Whitney – FAA CLEEN II Consortium

November 7, 2018

Washington, D.C.

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Pratt & Whitney's CLEEN II Technologies

Today's Agenda

Overview of UTC and Pratt & Whitney

Summary of P&W's CLEEN II Efforts

Overview of P&W's CLEEN II Compressor Technologies

Overview of P&W's CLEEN II Turbine Technologies

Overall mission performance and fleet impacts (initial estimates)

Geared Turbofan™ entry into service and applicability of CLEEN II technology to future products



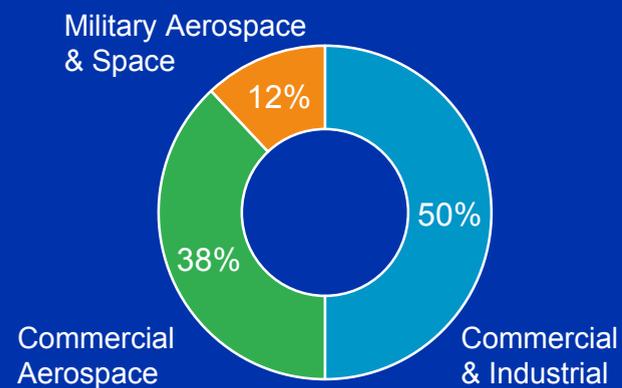
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United Technologies Corp.

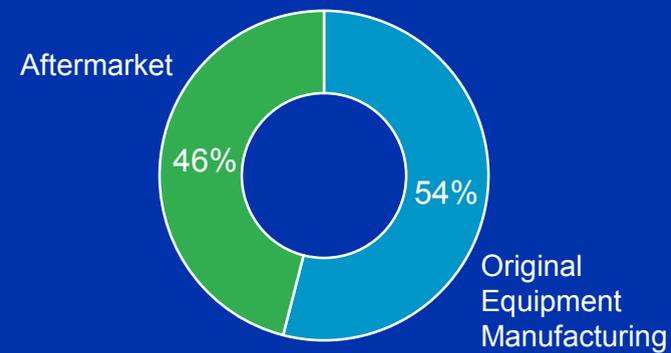


2016 net sales*
\$57.4B

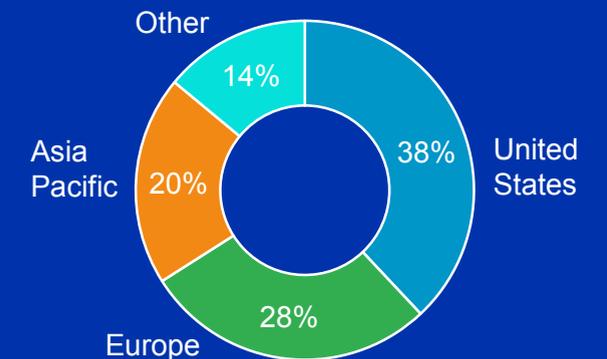
End Markets



Sales By Type



Sales By Geography

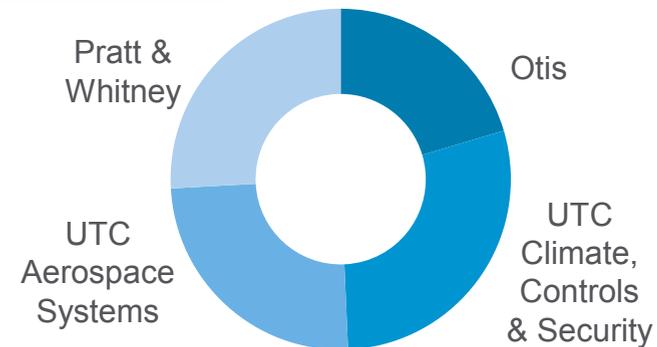


UTC businesses push the limits of innovation



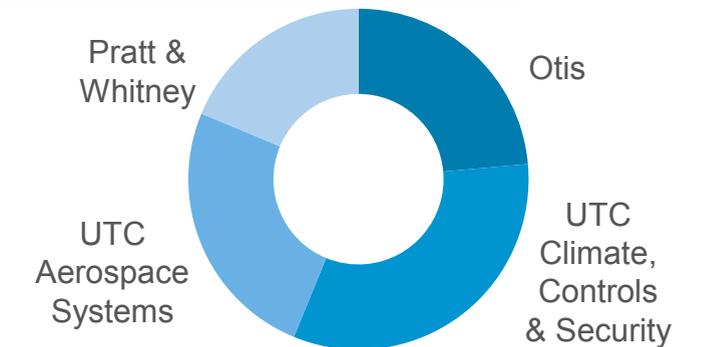
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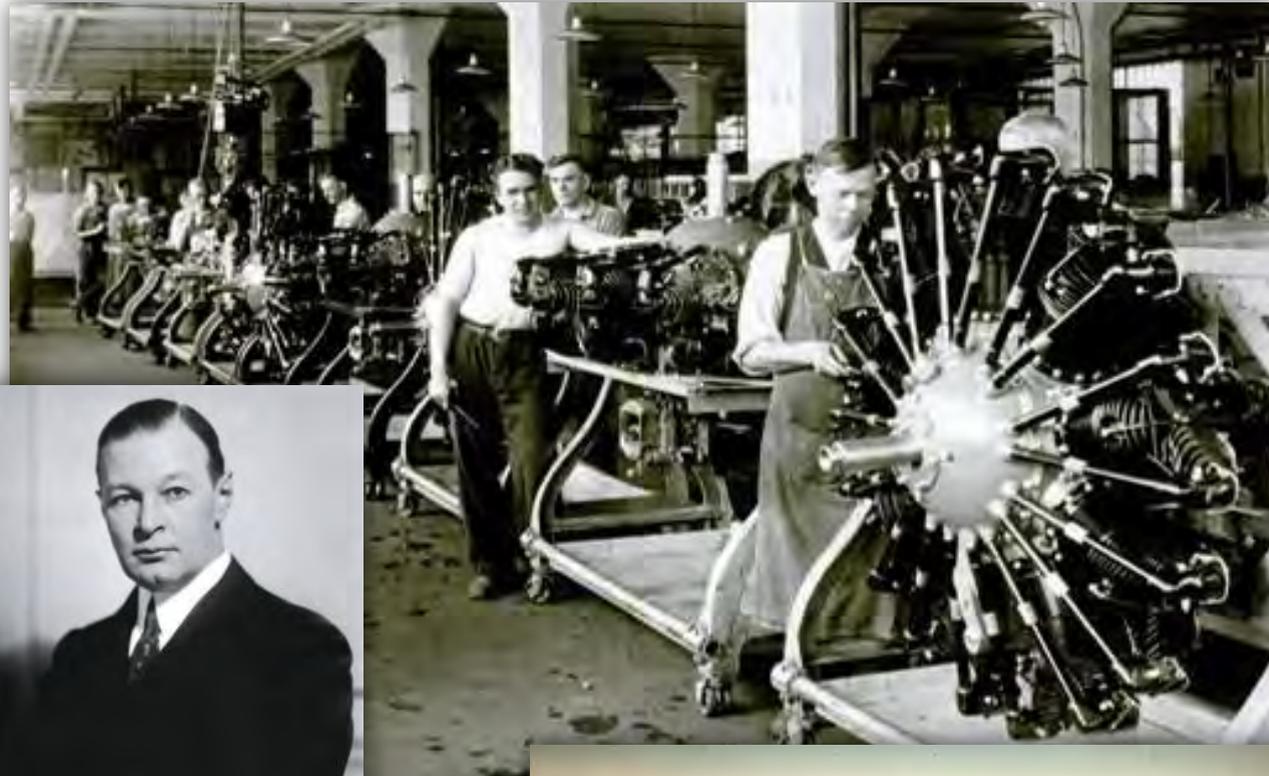


Operating profit*

\$9.4B



Pratt & Whitney's history began in 1925...



Frederick Rentschler



...and continues

Pratt & Whitney segments

50%

Commercial Engines



25%

Military Engines



25%

Gas Turbine Engines for Small Aircraft



35,100 Employees | \$15.1B Sales | \$1.8B EBIT | 11.6% ROS

Pratt & Whitney Commercial Engines



Photo: Boeing

Boeing 757



Photo: Airbus

Airbus 320



Photo: Airbus

Airbus 320neo



Mitsubishi MRJ



Photo: Boeing

Boeing 767



Photo: Airbus

Airbus 330



Photo: Bombardier

Bombardier C Series → A220



Photo: Embraer

Embraer 190/195-E2



Photo: Boeing

Boeing 777



Photo: Airbus

Airbus 380



Irkut MC-21



Photo: Embraer

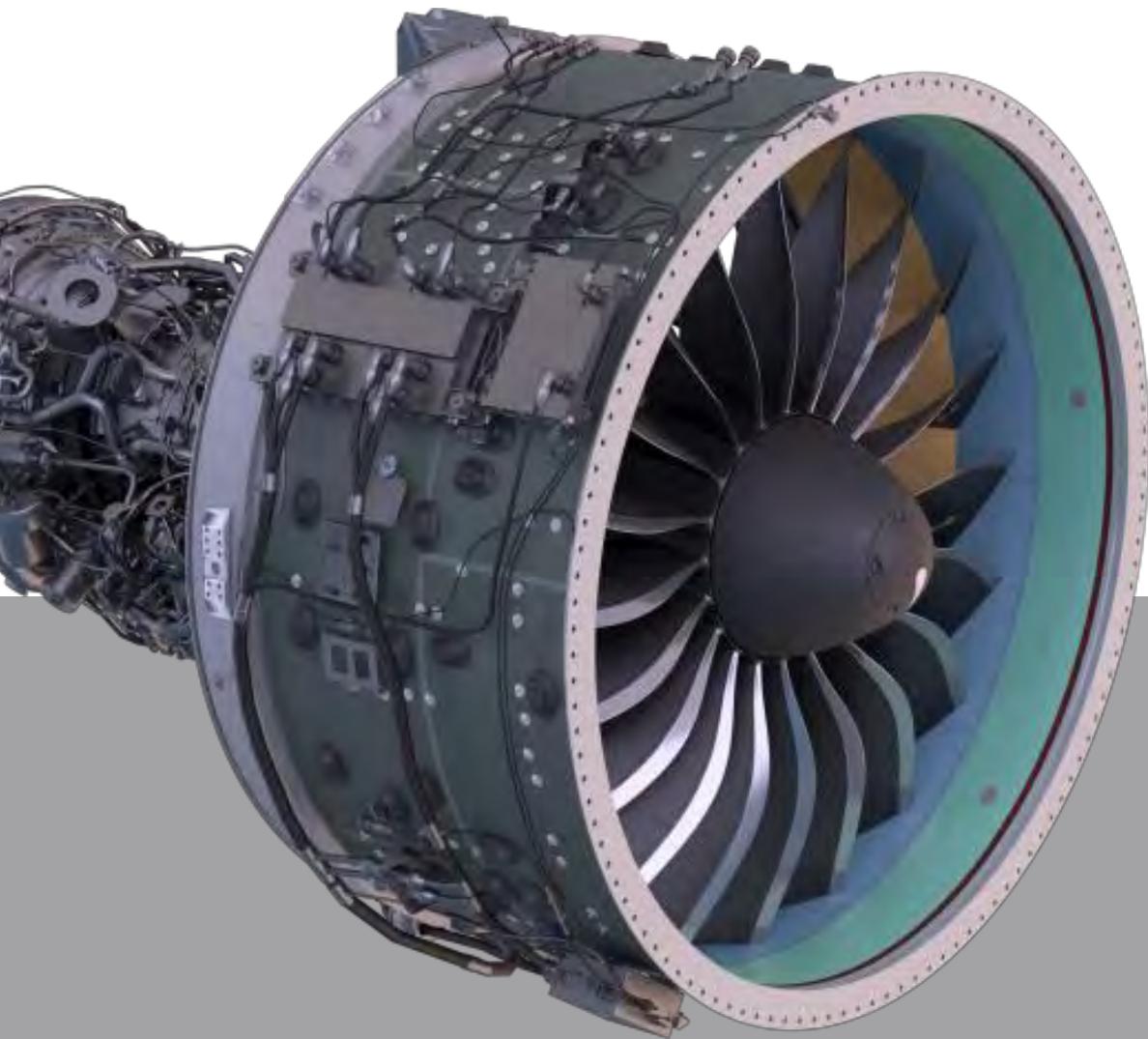
Embraer 175-E2

Mature

Geared Turbofan

Pratt & Whitney GTF™ Engines

Creating the next generation of propulsion



5 AIRCRAFT PLATFORMS

PW1100G-JM A320neo



PW1500G A220



PW1700G & PW1900G E-Jets E2



PW1200G MRJ



PW1400G-JM MC-21



16%

Reduction in
Engine Fuel
Consumption

75%

Reduction in
Noise Footprint

50%

Reduction in
Regulated
Emissions

9000+

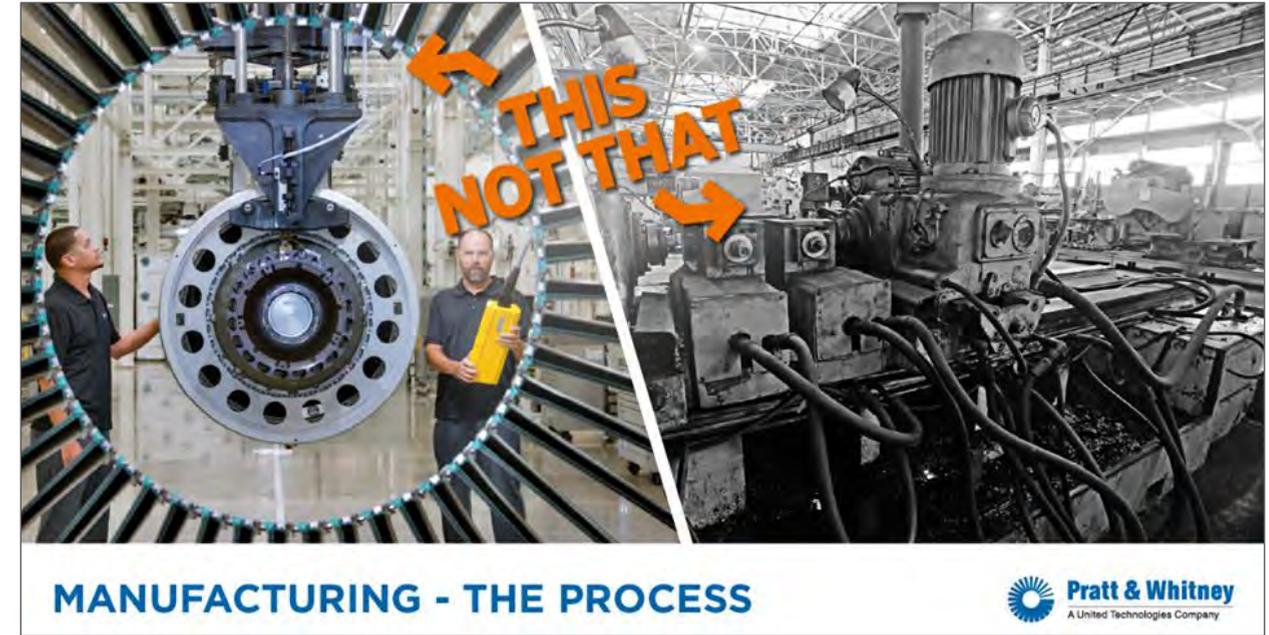
Orders and
Commitments

80+

Customers

ANNUAL OPERATOR SAVINGS >1M PER AIRCRAFT

Pratt & Whitney Global Manufacturing



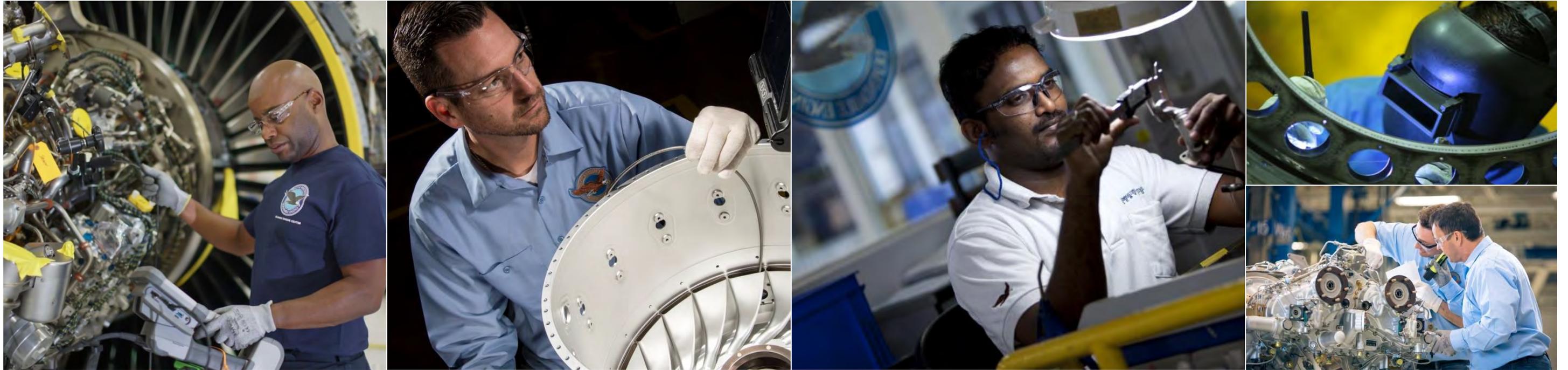
The world's first horizontal assembly lines — up to 50% more productivity

Intelligent manufacturing cells — closed loop machining, robotics, automated inspection

White/blue light quality techniques at sites worldwide — concurrent inspection and manufacturing

State of the art technology to produce advanced aerospace products

Aftermarket and Global Service Network



PRATT & WHITNEY COMMERCIAL ENGINES AFTERMARKET

17 world-class engine overhaul and part repair facilities
5,000 employees with OEM and technical expertise

Upgrades to improve time on wing and fuel efficiency
Maintenance and material solutions — part repair, new
spare parts and serviceable parts

PRATT & WHITNEY CANADA GLOBAL SERVICE NETWORK

Over 30 P&WC-owned and designated service facilities
More than 100 field service representatives worldwide

24/7 Customer First Centre for rapid support
MyP&WC Power
ESP® PurePower PW800

The Workforce is changing

Demographics

25k Total global hires by 2026

20k Total anticipated retirements by 2026

50% Paradigm 4 Parity goal by 2030

Development

\$1.2b Spent to date on Employee Scholar Program

250 Executives through Thayer Leader Development Group

30+ Workforce training programs P&W has invested in

Skillsets

 Job skill portfolio

 Automation

 Agile culture

Hiring and developing the right people with the skills needed for the jobs of today and the jobs of tomorrow

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Improved propulsive efficiency enabled by Geared Turbofan™ architecture

CLEEN II builds upon CLEEN I for overall GTF engine architecture efficiency benefits

Compressor Section – advanced aerodynamics and cavity/airfoil optimization

Turbine Section – advanced aerodynamics and reduced cooling requirements and chargeable flow losses

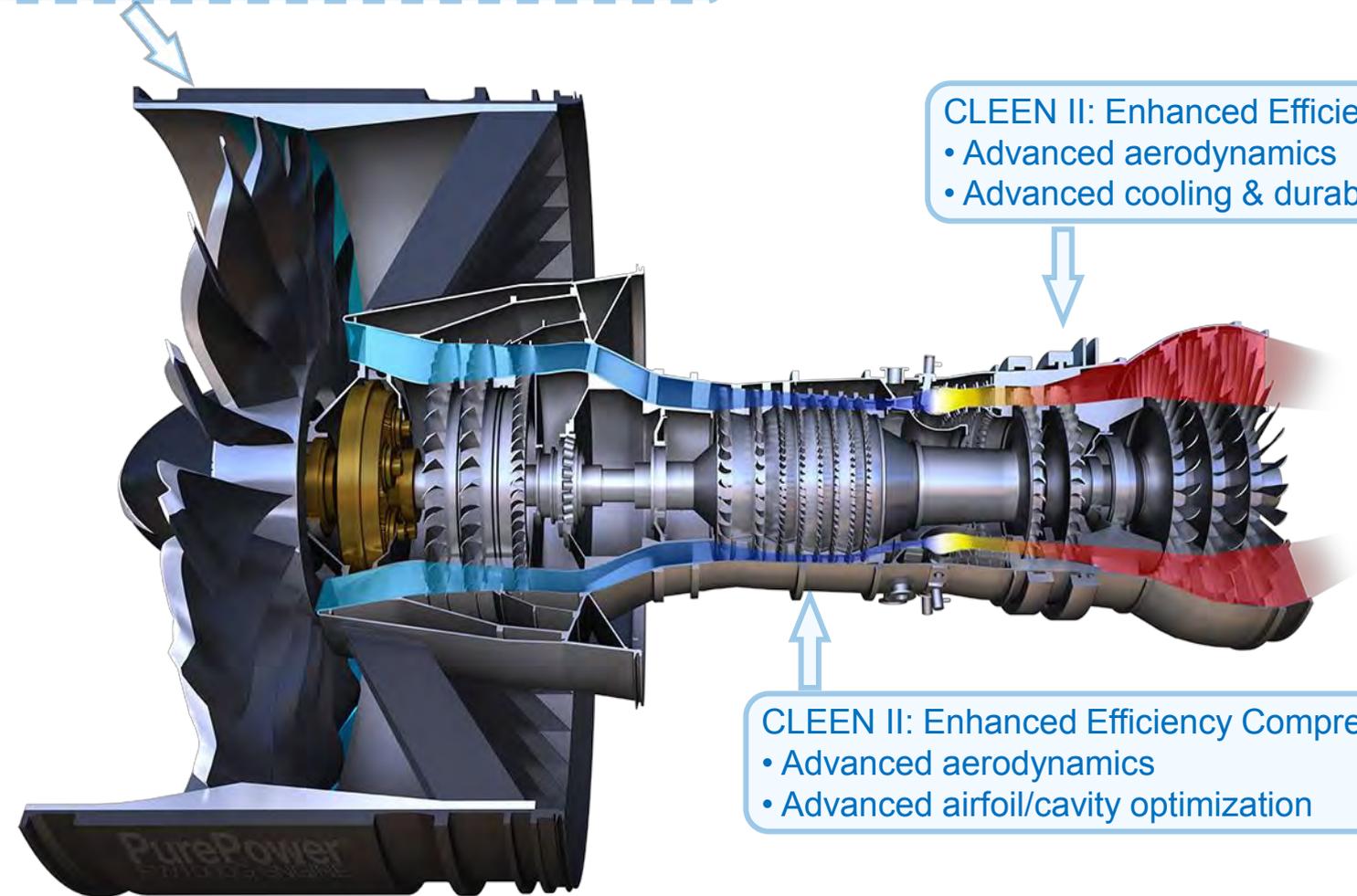
CLEEN II compressor and turbine technologies together improve the thermodynamic efficiency of the GTF architecture.

Leads to a 1.6-2.0% total fuel burn reduction

CLEEN I: Ultra-High Bypass (UHB) Propulsor
(Short Inlet, Low Pressure-Ratio Fan)

CLEEN II: Enhanced Efficiency HPT
• Advanced aerodynamics
• Advanced cooling & durability

CLEEN II: Enhanced Efficiency Compressor
• Advanced aerodynamics
• Advanced airfoil/cavity optimization



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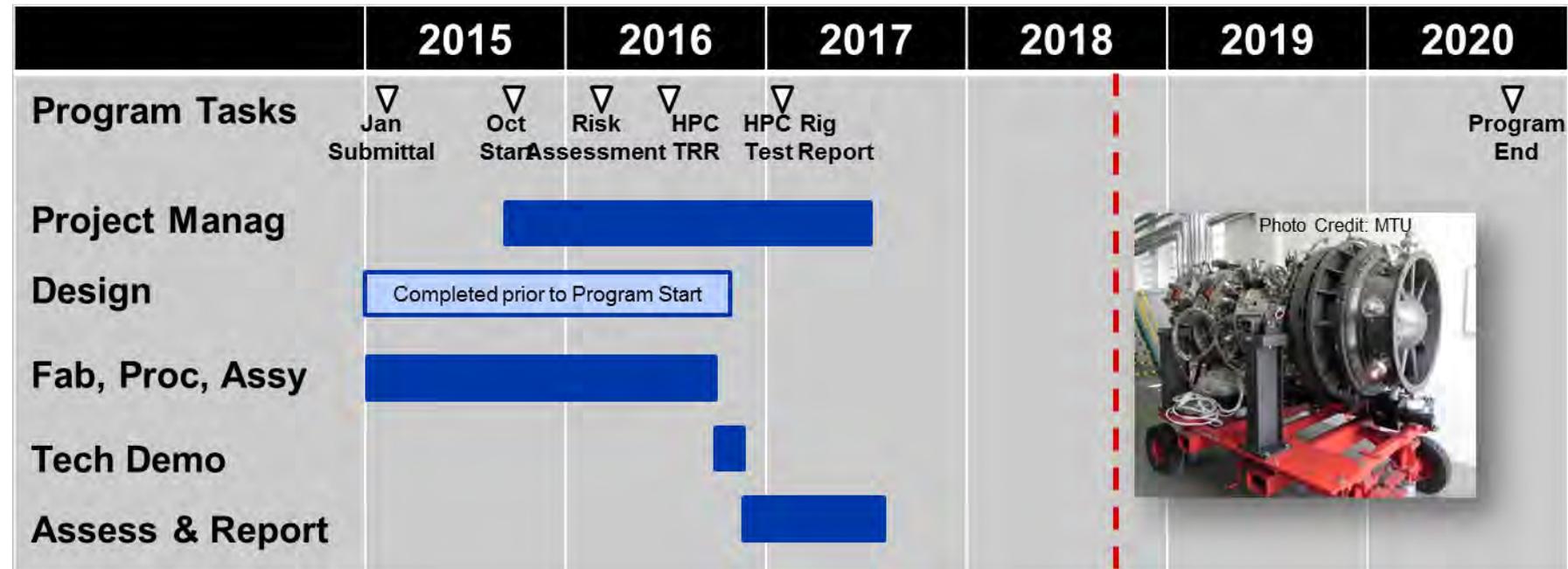
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HPC Technology Status - Schedule

High pressure compressor scope was started prior to the start of CLEEN II

With help from FAA, design was fabricated and brought to test in 2016

Test completed in early 2017 and detailed test report submitted



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HPC Technology Maturation Strategy

Previous investment from P&W brought HPC technologies through Technology Readiness Level (TRL) 4

CFD design and analysis for conceptual design of the technologies

Cascade rigs for correlation of analytical tools

Multistage, subscale rigs for further proving out the technologies in a laboratory environment

Under FAA funding, brought HPC technologies to TRL 5 – “component validation in relative environment”

HPC Technology Maturation Process

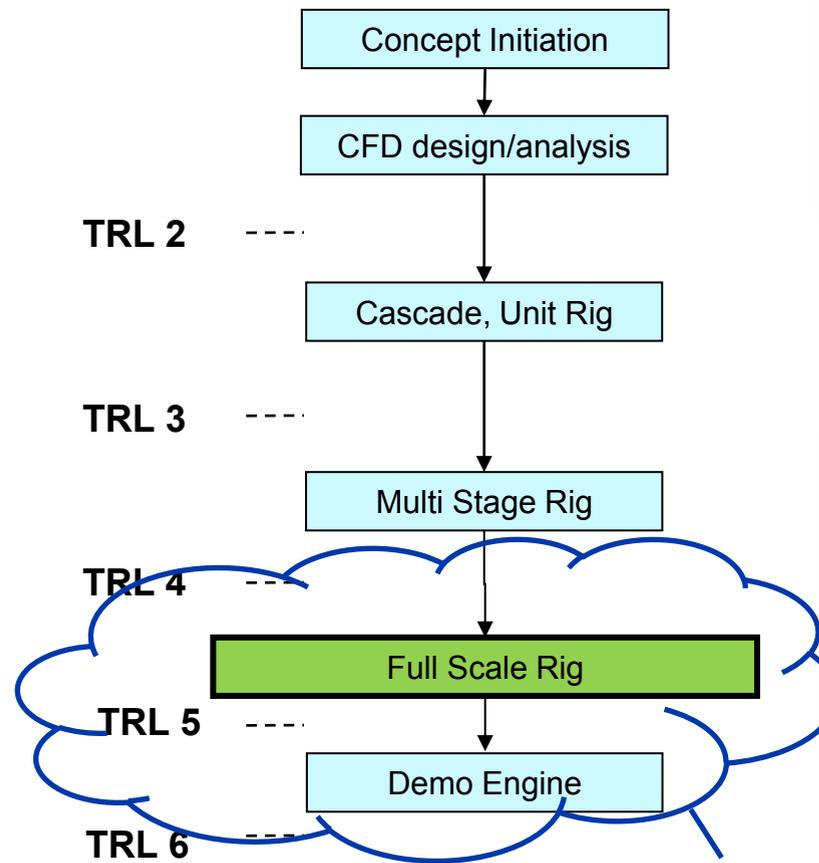


Photo Credit NRC Canada

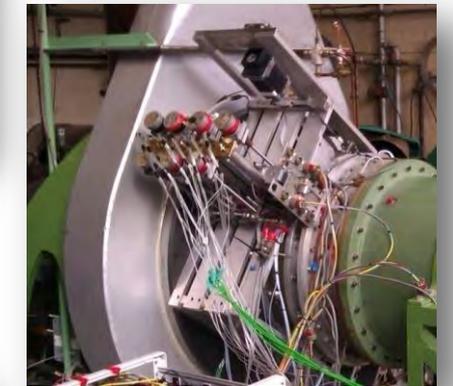


Photo Credit: MTU



CLEEN II Scope

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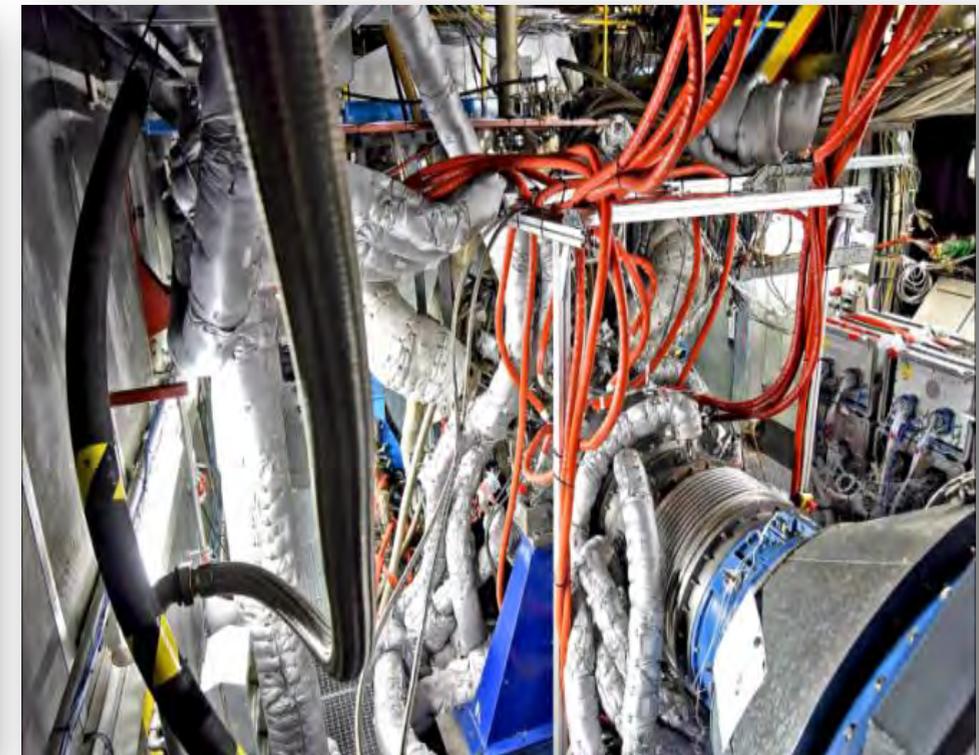
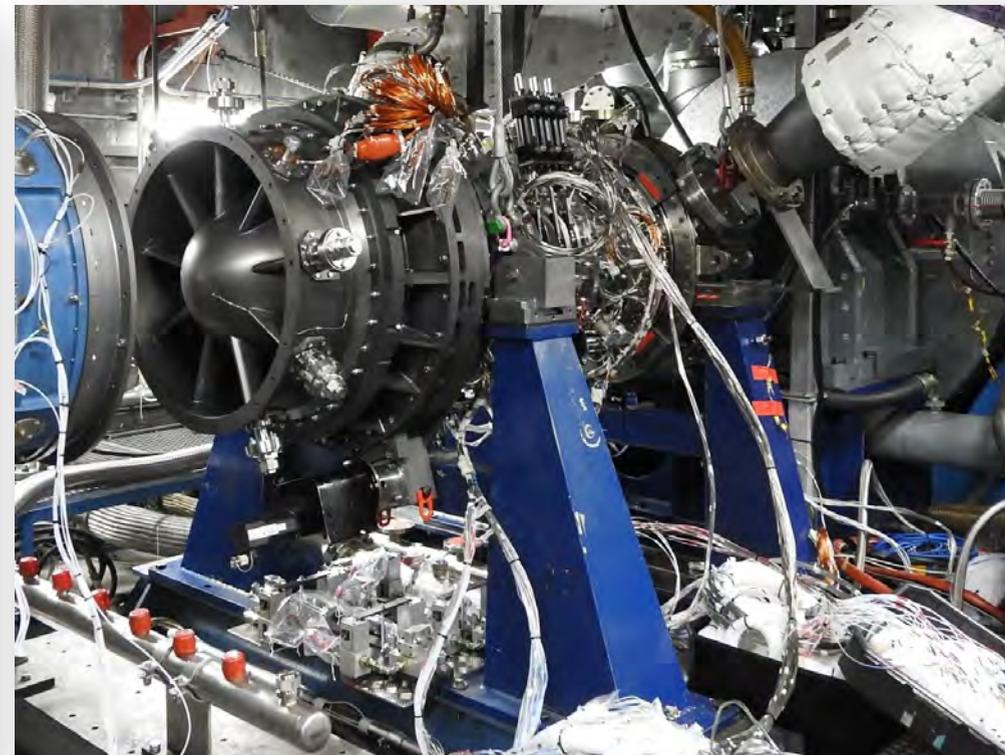
HPC Technology Rig – Build and Installation

HPC Rig assembly completed and module placed on transport cart on July 21, 2016

Module installed into test cell and final instrumentation completed on August 23, 2016

Can you spot the compressor in that last image?

All Photo Credits: MTU



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HPC Technology Rig – Objectives and Testing

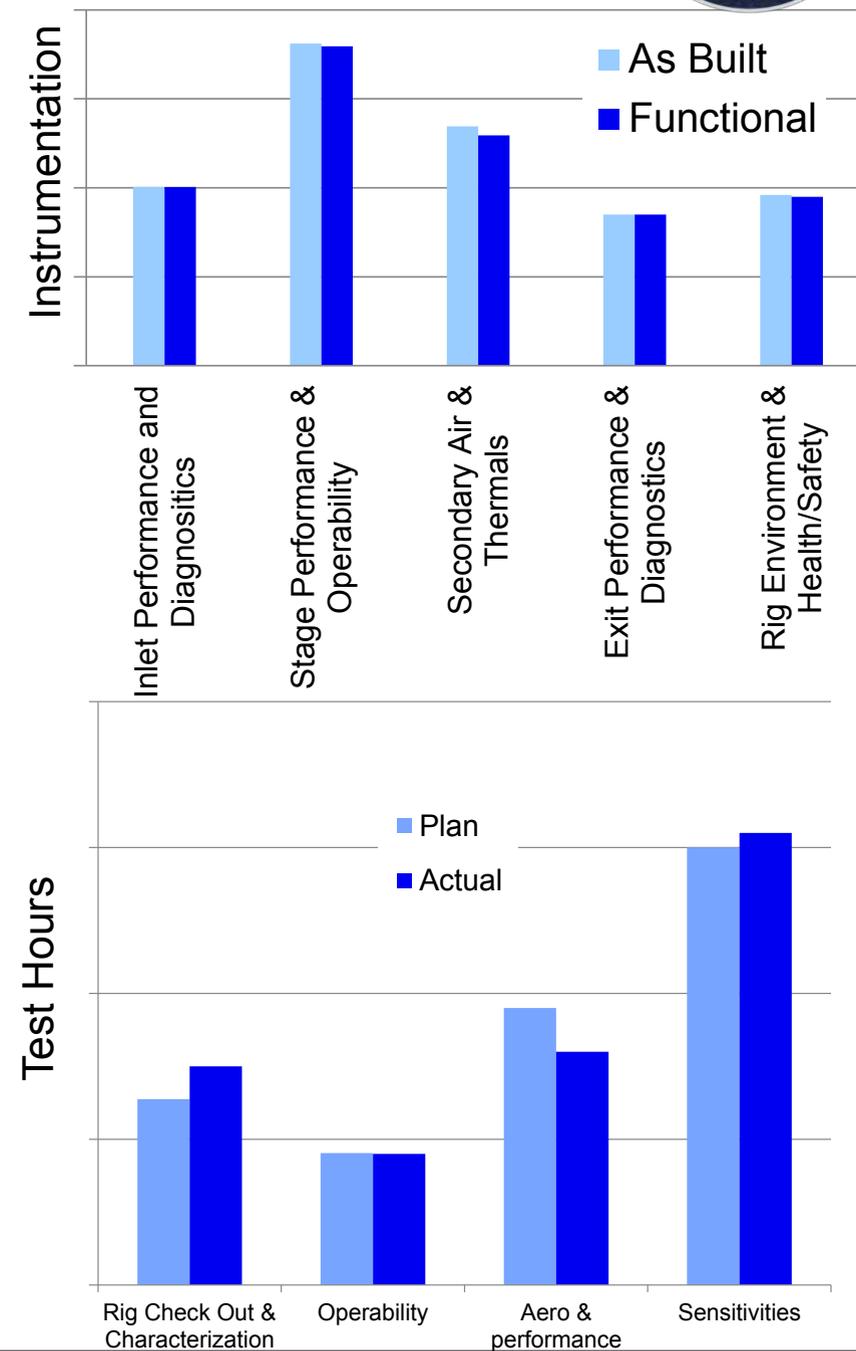
Objective of the rig testing was to validate analytical tools for novel HPC technologies through TRL 5 and to prepare the technologies for engine test

Testing proved out a robust suite of instrumentation

Over 100 hours of testing was conducted

Flexible test schedule allowed for changes to be made on the fly

Example: Less aero/performance testing was completed in favor of expanding sensitivity testing



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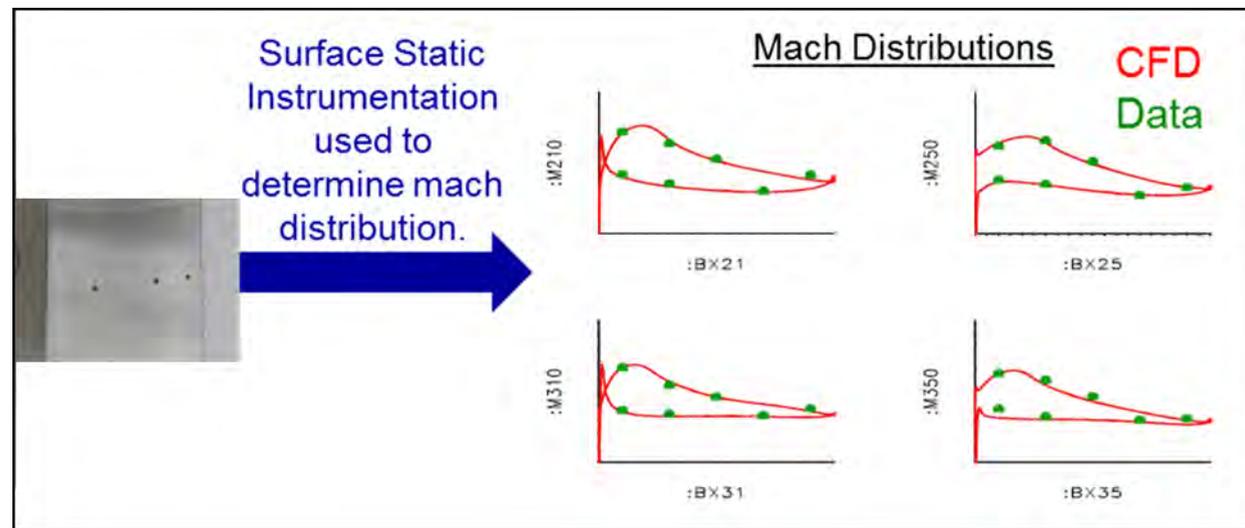
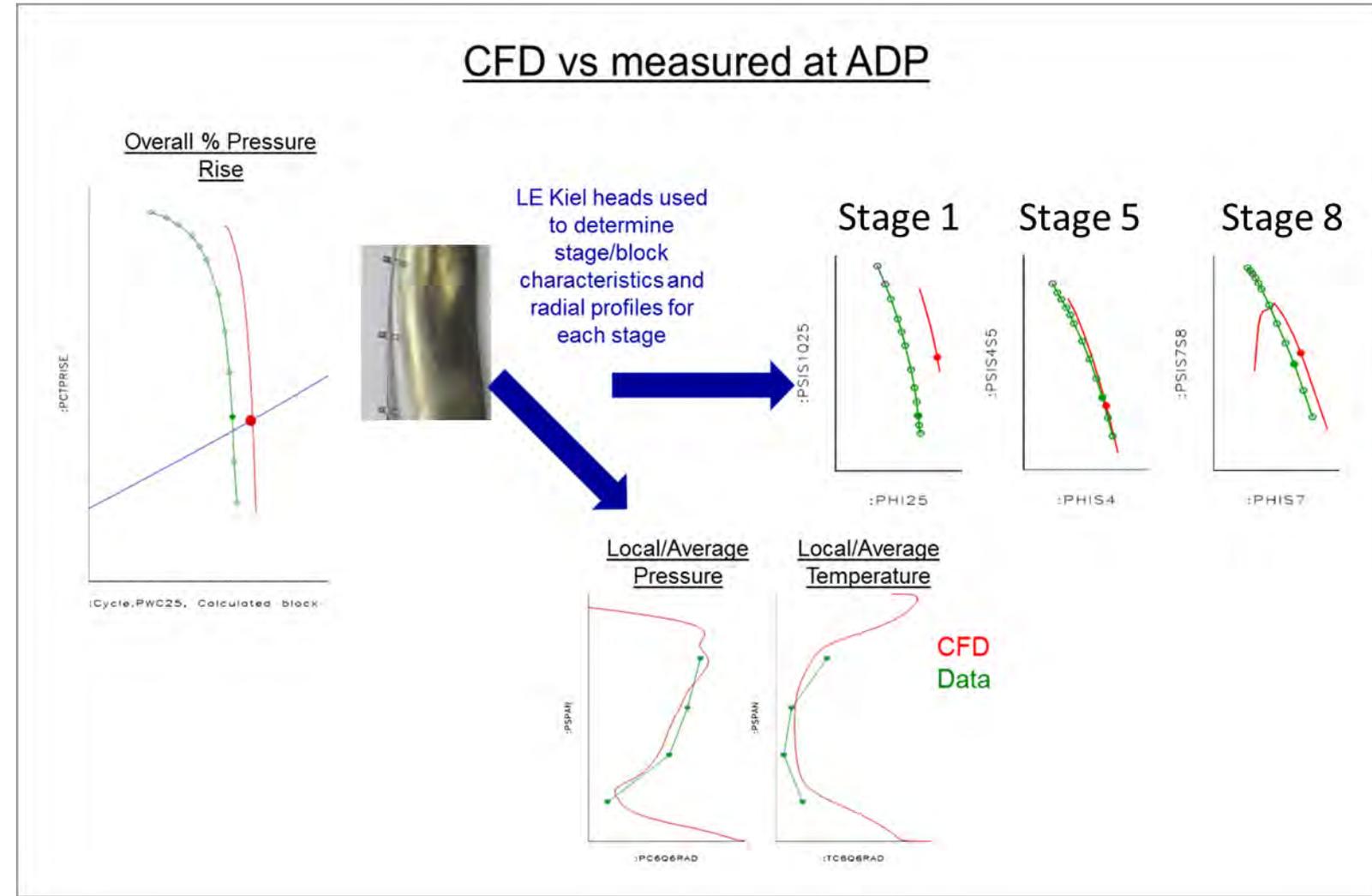
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HPC Technology Rig – Initial Results

Results show good correlation to pre-test Mach distributions and radial profiles

Stage sizes close and will be used to improve modelling

Rig learning across compressor operating space will be used to improve modelling capability



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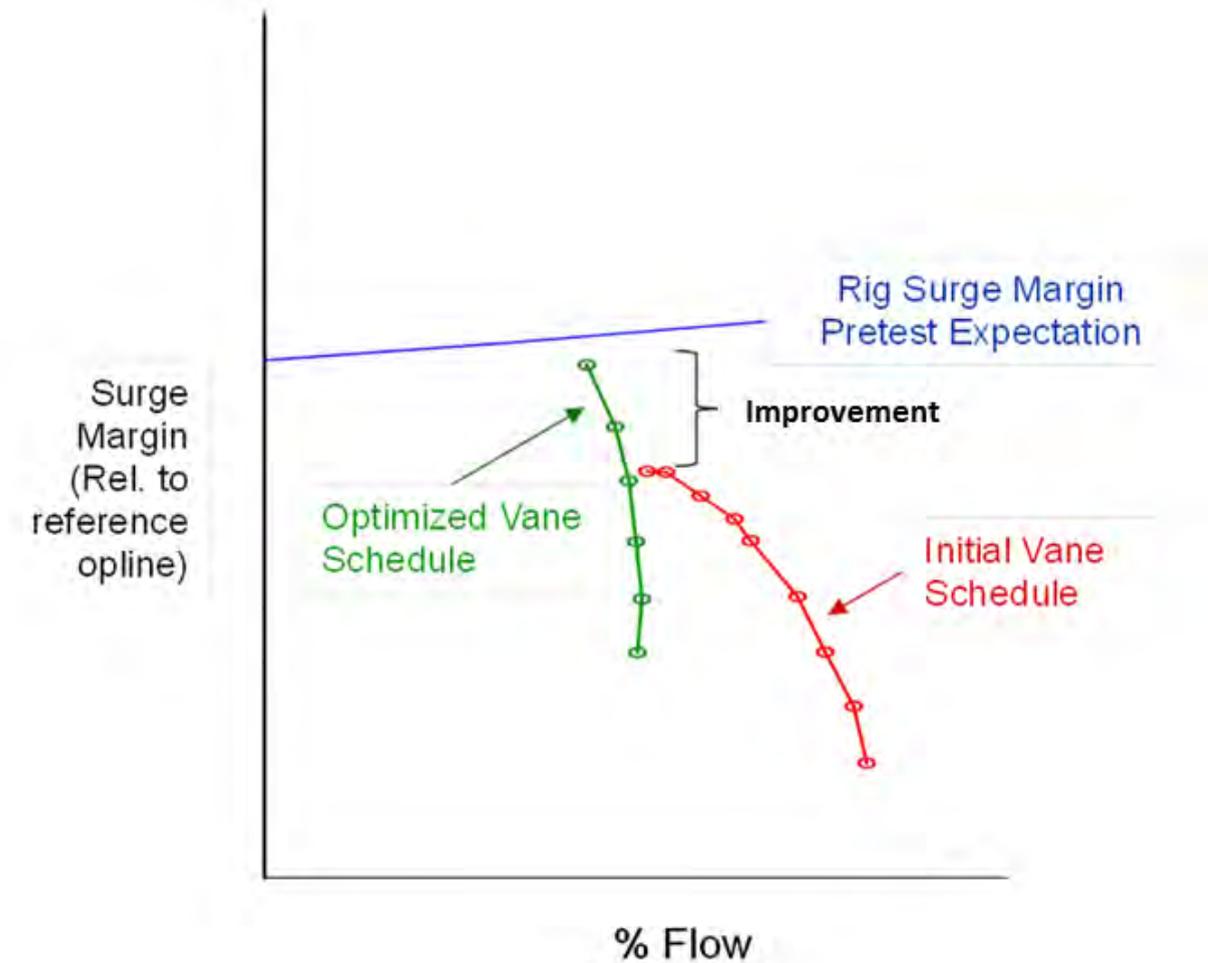
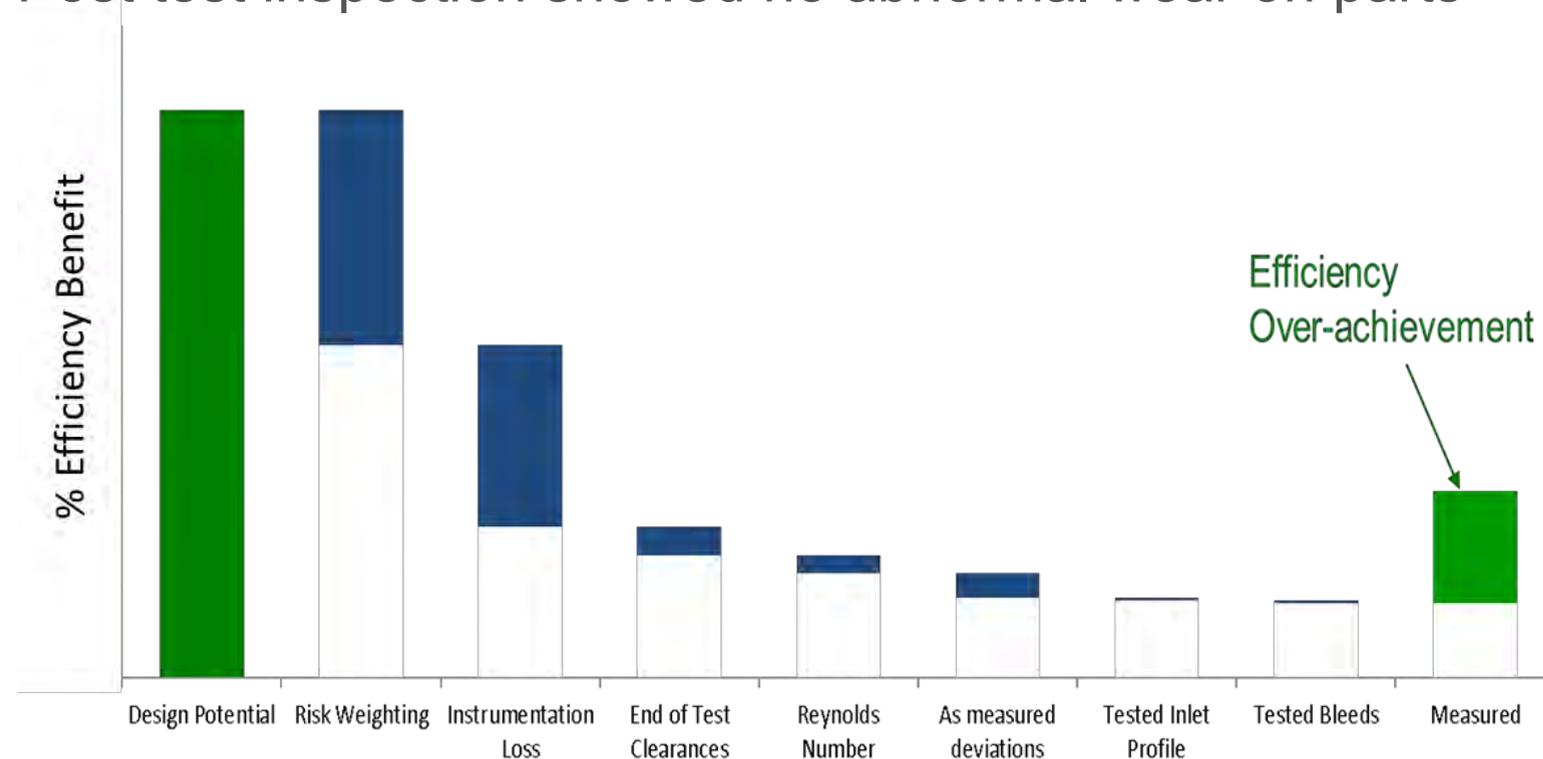
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HPC Technology Rig – Final Results and Learning

Optimized vane schedule achieved by running sensitivities during testing. Data to be used to improve scheduling in future products

Overall efficiency of the rig exceeded pretest predictions

Post test inspection showed no abnormal wear on parts



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HPC Technology Rig – What's next?

Same compressor aero design slated to go to ground test shortly to bring HPC technology to TRL 6 – “Subsystem demonstration in a relative environment”

Tools developed and knowledge gained will be utilized for flight test asset sometime in the future and eventual introduction into GTF product line

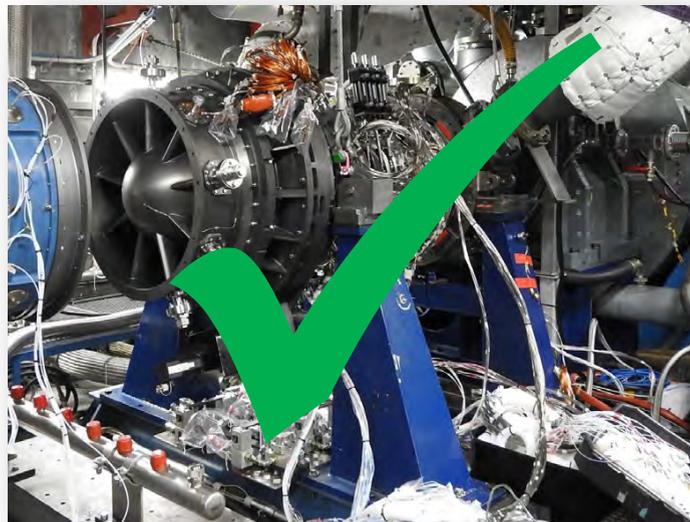


Photo Credit: MTU



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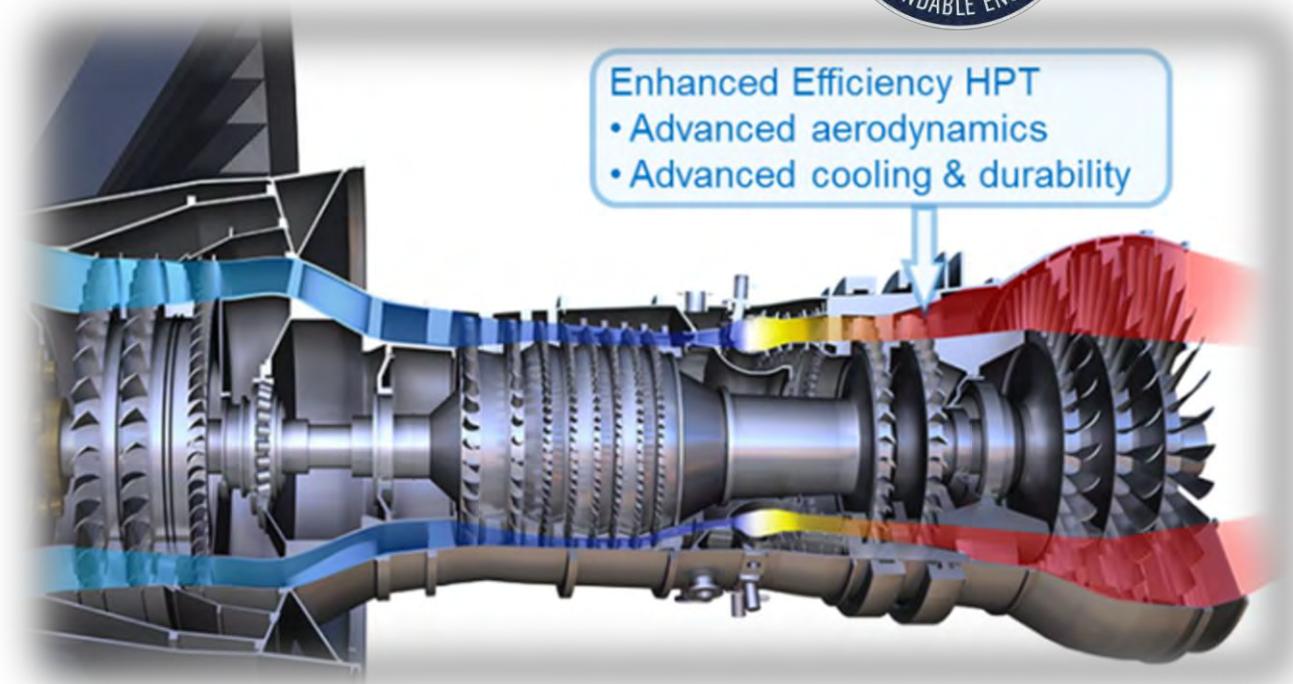
Turbine Aero Efficiency and Durability Technologies

Benefits:

- Improved thermal efficiency
 - ~ 0.8 – 1.0% fuel burn reduction

Risks/Mitigations

- Technology interaction prevents assessment of contribution of individual items
 - Execute additional rig trials to isolate
- Testing delayed due to linear build schedule
 - Procure additional hardware to facilitate parallel build
- New facility debugging
 - Actively working to understand failure modes and backup facility hardware



Objectives: *Demonstrate improved high pressure turbine efficiency via advanced aerodynamic airfoil and durability optimization*

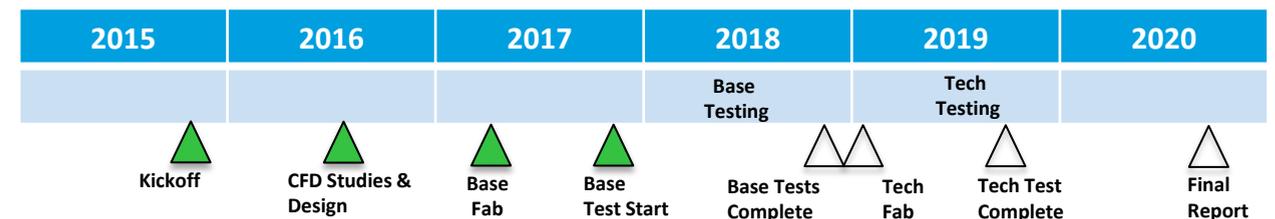
Work Statement:

Continue the TRL advancement of turbine aero-efficiency and durability technologies via CFD studies, detailed design, fabrication, and full-scale rig tests.

Prior Accomplishments:

- Completed cascade rig testing
- Completed START rig traverse assembly
- Kicked off baseline testing

Schedule & Planned Milestones:



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HPT Technology Maturation Strategy

Previous investment from P&W has brought HPT technologies through Technology Readiness Level (TRL) 3

CFD design and analysis for conceptual design of the technologies

Low speed wind tunnel testing for initial learning

Under FAA funding, bringing HPT technologies to TRL 5 for durability and TRL6 for aero technologies

HPT Technology Maturation Process

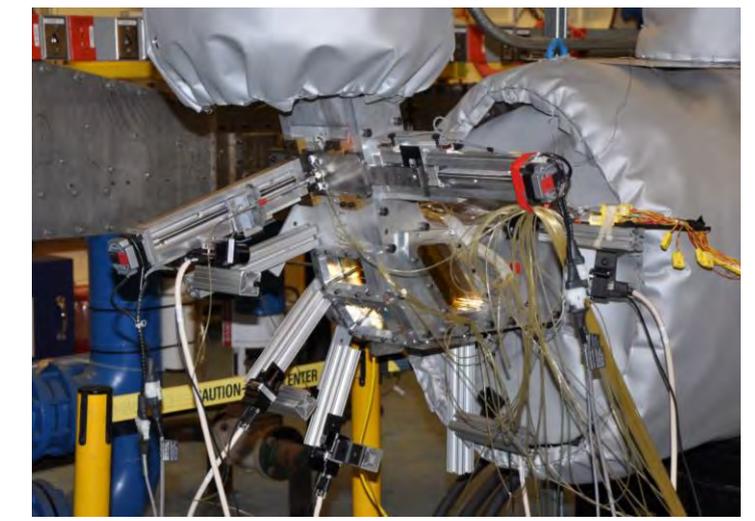
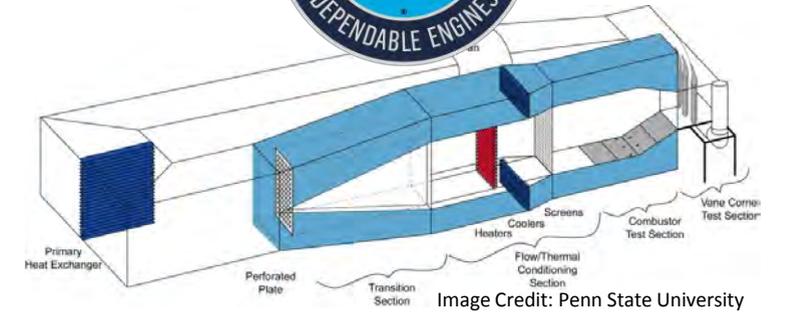
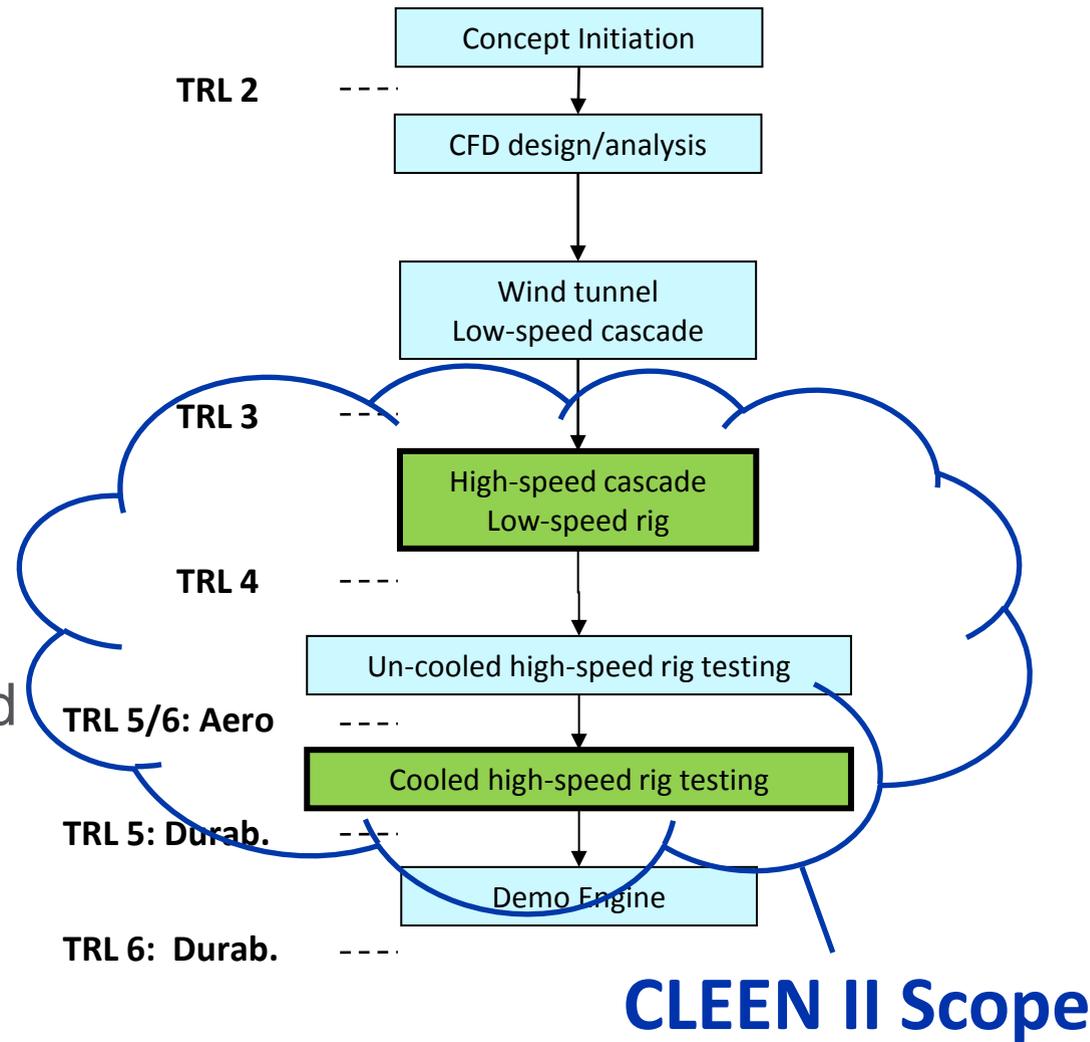


Image Credit: Penn State University

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HPT Technology Status - Schedule

Conceptual design work started prior to CLEEN II contract start

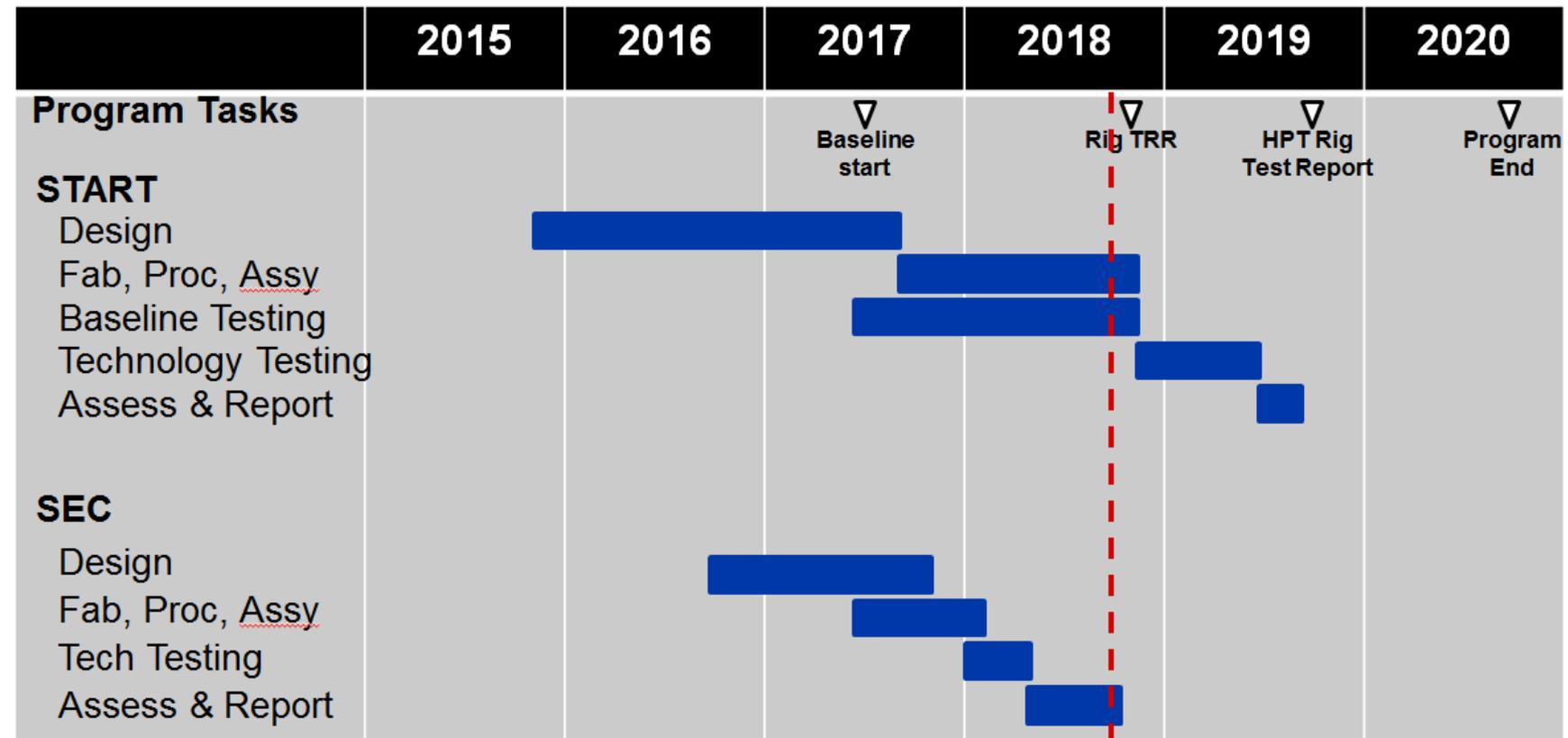
FAA has helped to further mature the HPT technologies beyond TRL 3

HPT scope holding schedule for full-scale hardware Aero/Thermal testing in 2019

Single Element Cascade testing complete

Baseline START testing nearly complete

Currently working hardware fabrication and final rig preparation for technology testing



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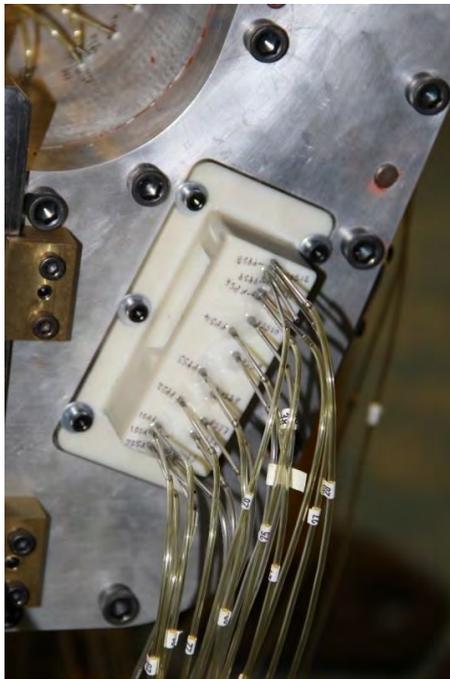
Single Element Cascade (SEC) - Objectives

Test purpose was to measure film cooling effectiveness and aerodynamic loss on high-lift airfoil sections

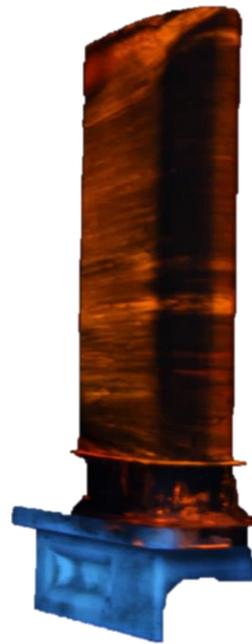
Compare baseline and advanced cooling technology at representative operating conditions

Generate high-quality aero/thermal data to validate Pratt & Whitney's turbine cooling analytical modeling

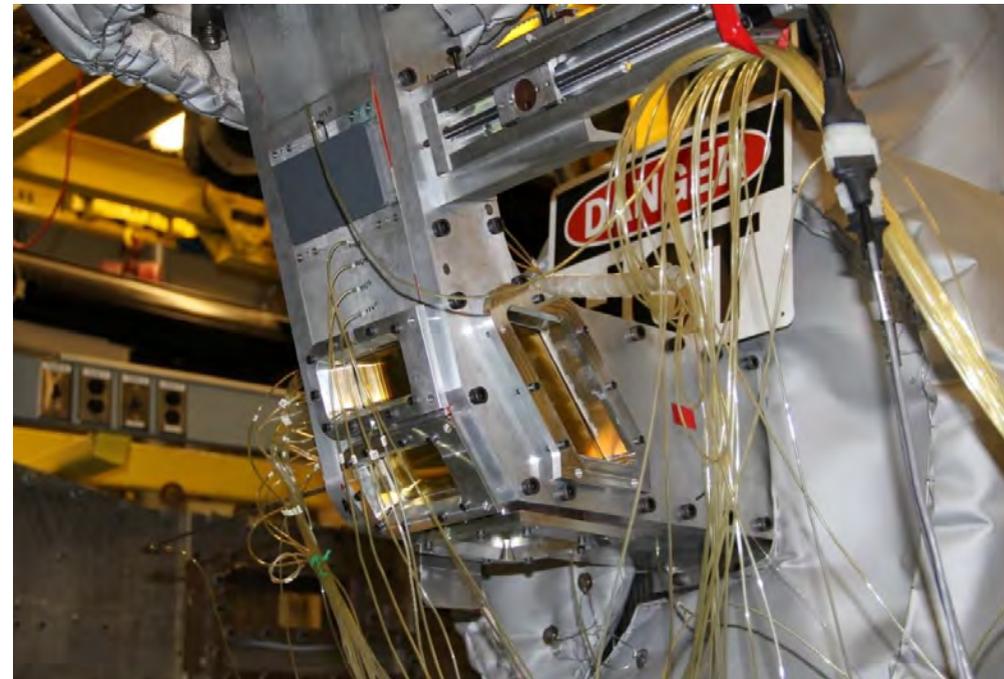
Enhance the understanding of performance losses due to advanced technology cooling architecture



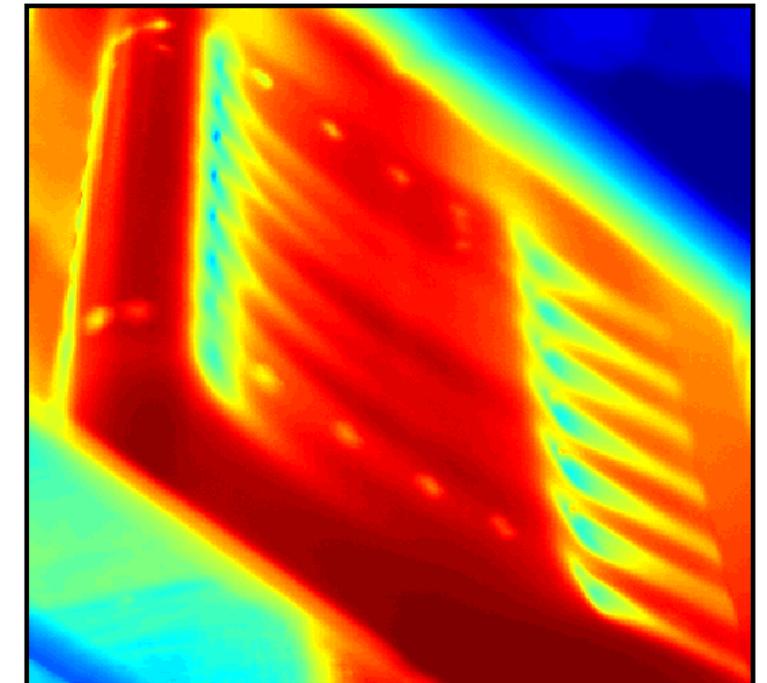
Single Element Cascade Rig Blade



Flow Visualization



Single Element Cascade Rig



Processed IR Image Data

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Single Element Cascade (SEC) - Execution

SEC facility and test article assembly was completed at United Technology's Research Center (UTRC) in March 2018

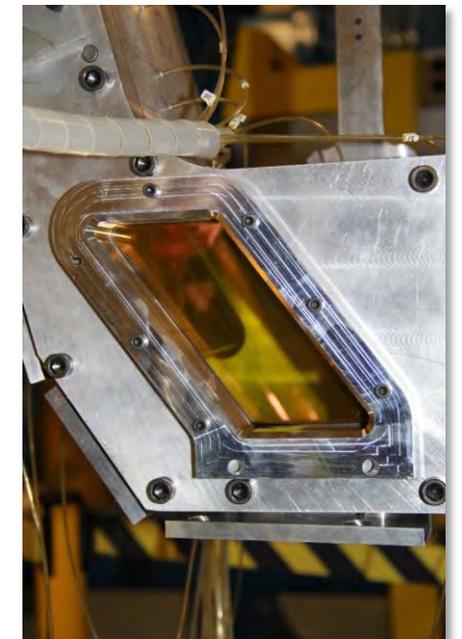
Testing commenced in April and was completed in June 2018

Testing consisted of two major sub-tests:

- Uncooled, baseline testing
- Cooled Aero/thermal testing for the CLEEN II technologies

Data reduction & analysis is currently in-process and expected to be fully complete by the end of the year

Technologies proven-out by the SEC testing allows for future turbine components to reduce cooling air requirements; thus reducing the amount of losses in the engine



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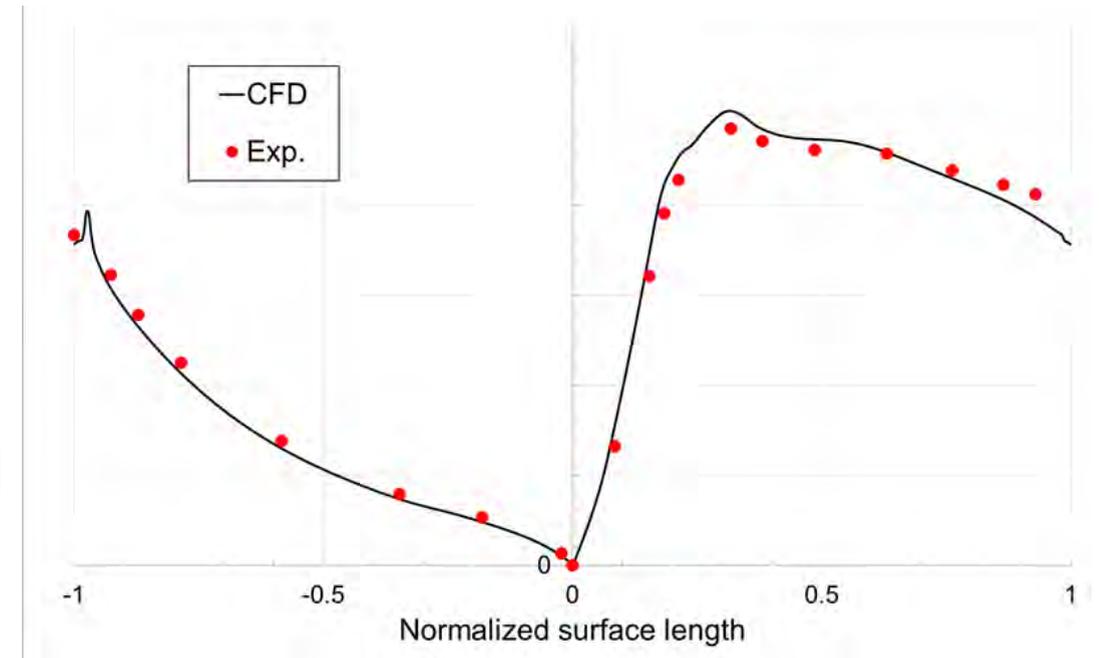
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Single Element Cascade (SEC) - Results

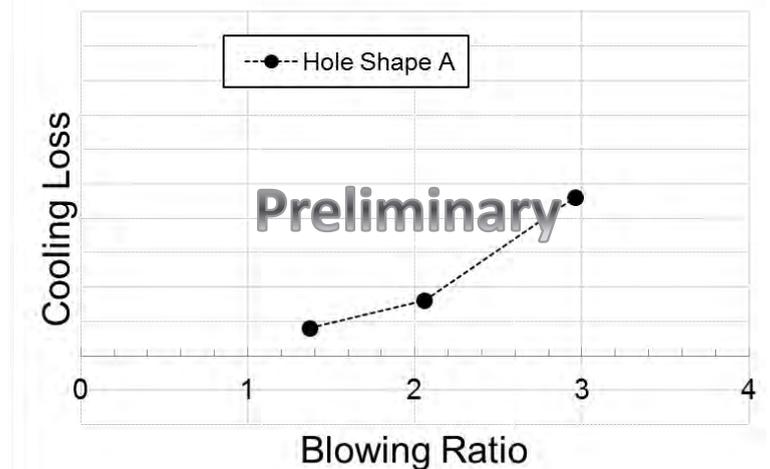
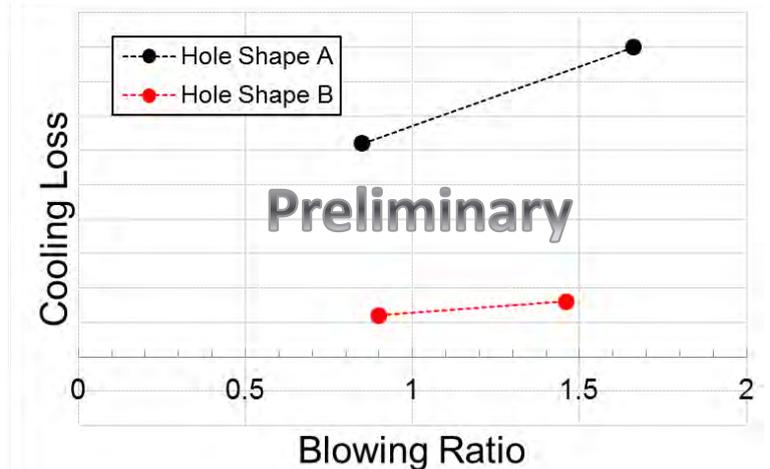
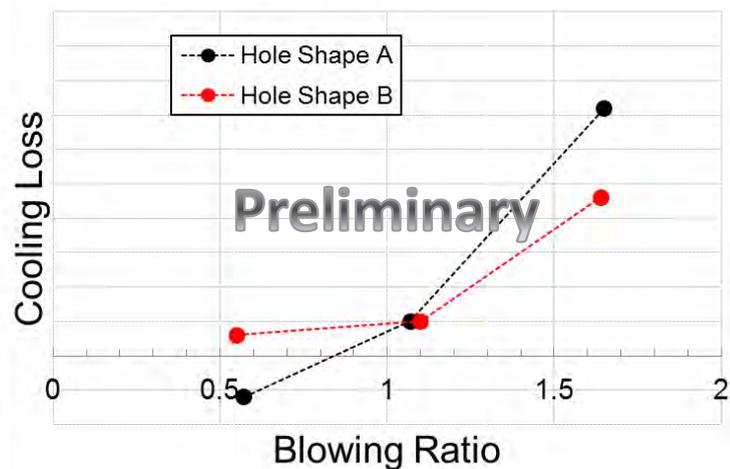
Cascade rig fully instrumented with pressure and temperature probes along with IR camera system

Good correlation of experimental data back to CFD pre-test predictions for surface pressure and Mach number distribution

Cooling hole loss data obtained for various cooling hole shapes and blowing ratios on suction side, leading edge, and pressure side



Detailed analysis in-work



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Penn State START Facility – Background

START = Steady Thermal Aero Research Turbine.

Test section is modeled after Pratt & Whitney's GTF high pressure turbine module

~\$10M combined investment into the facility over the past 5 years

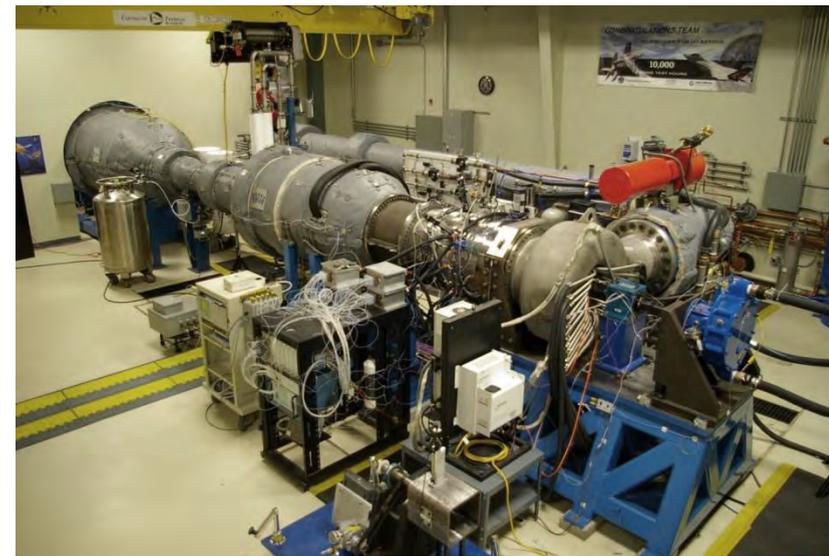
Pratt & Whitney Center of Excellence, World Class Facility



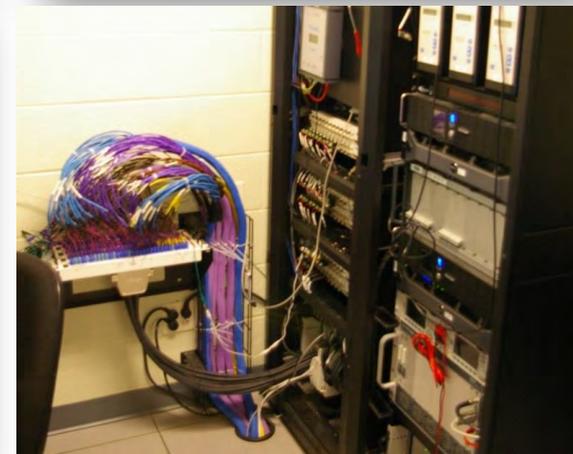
PennState



2013



2018



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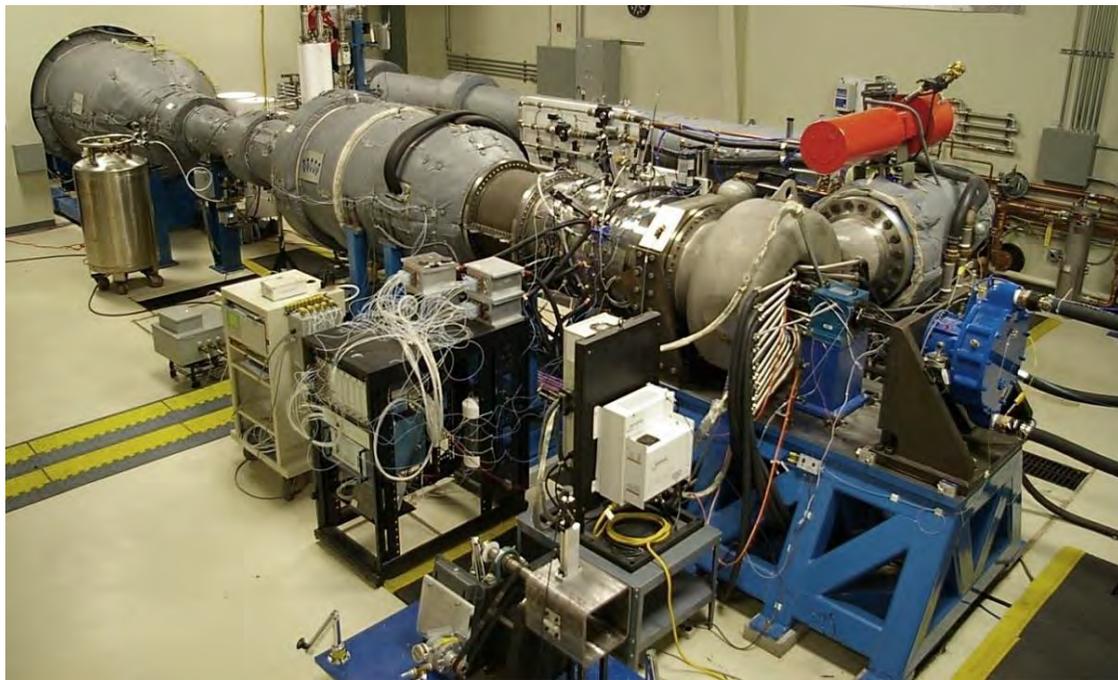
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Penn State START Facility – Objectives

Validate predictions for novel aero/thermal component designs in order to correlate analytical tools for CLEEN II technologies

Compare baseline and advanced aero/thermal technologies at representative operating conditions

Build upon completed SEC testing; verifies full-span 3-dimensional aero



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Penn State START Facility – Execution

Procurement and assembly of “Phase II” START rig facility completed March 2017

START facility shakedown completed August 2017

Cavity Aero testing completed

Testing of baseline GTF technology anticipated by end of 2018

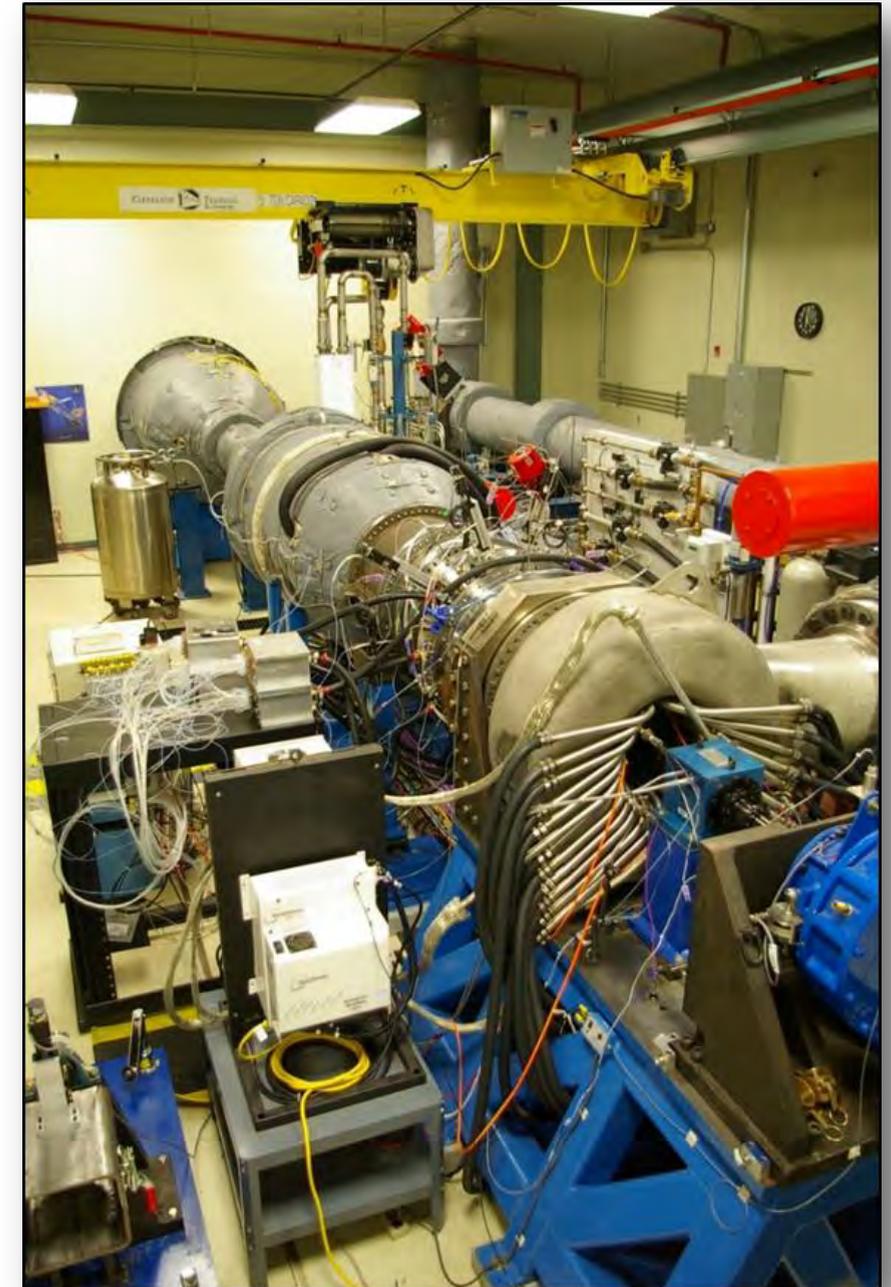
Analytical aero/thermal pre-test predictions anticipated by end of 2018

CLEEN II advanced technology blade aero/thermal instrumentation fabrication in-process

CLEEN II advanced technology blade fabrication in-process

99% of custom rig hardware for CLEEN II technology has been delivered

On schedule for 2019 test completion



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Penn State START Facility – Pretest Predictions

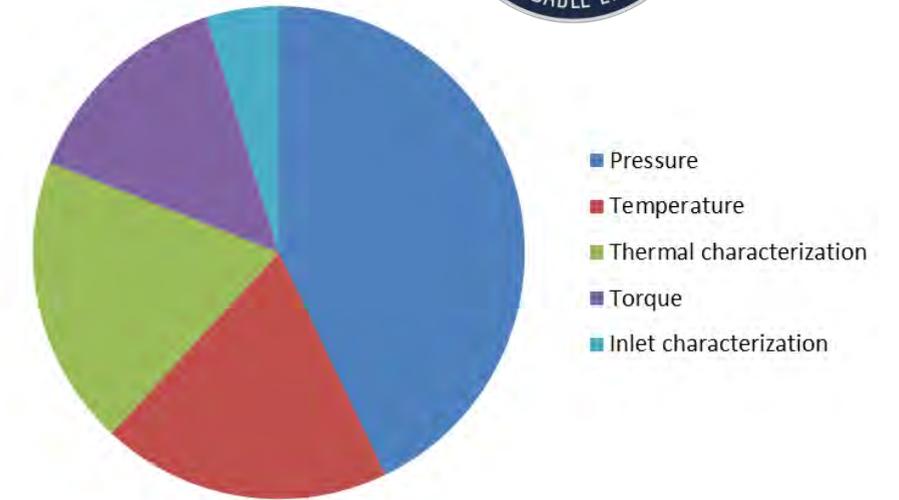
Analytical aero pre-test predictions have been completed and ready for experimental data comparison

Main gas path CFD has been coupled with secondary flow cavity CFD model for test correlation

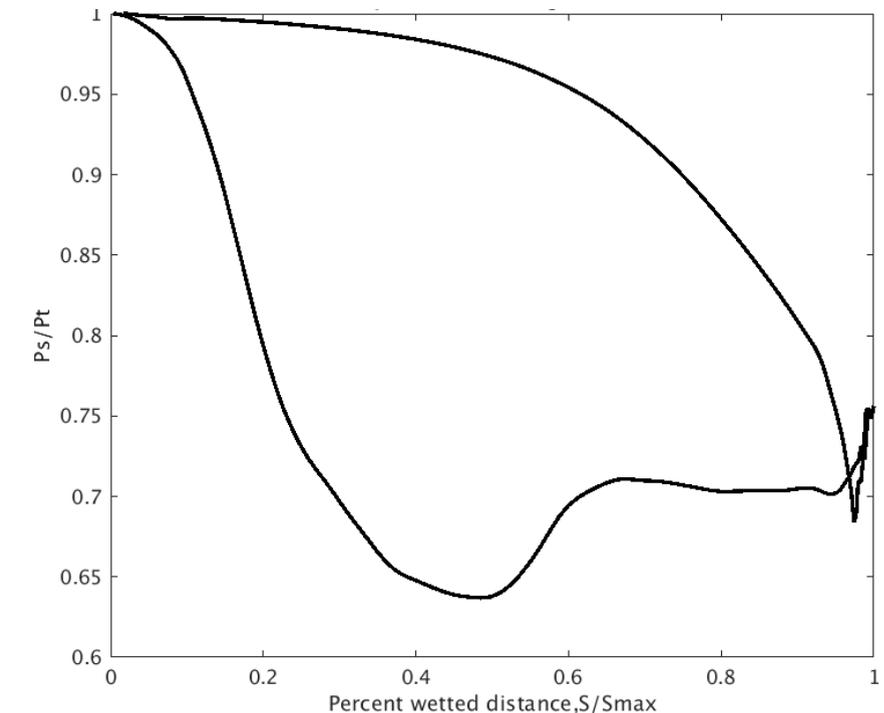
Instrumentation plans are in place for HPT technology testing

80% of the instrumentation required for technology testing has been installed into the START facility

Additional instrumentation specific to technology testing is in-work



CFD Prediction for Pressure Distribution



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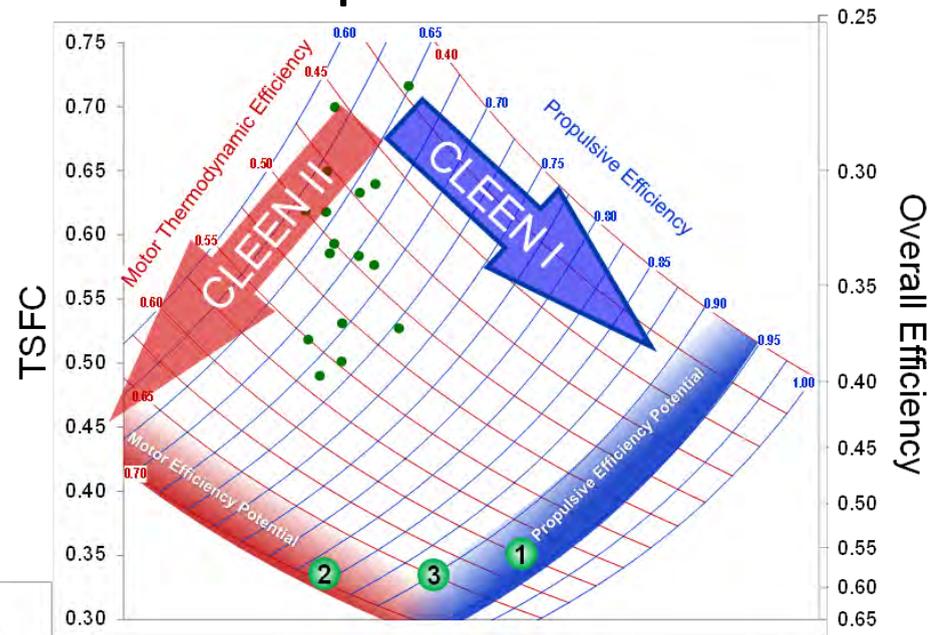
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System Level Impact

CLEEN I increased the propulsive efficiency of the GTF engine with fan technologies

CLEEN II technologies continue to push towards more thermodynamically efficient turbofan engines.

Component Level



Higher component efficiency

Engine / Airframe Level



1.6-2.0% Fuel Burn Reduction

Fleet Level



34-43K gallons of fuel saved per year per plane
A320NEO, 2.0 hour flights, 3,100 annual flight hours

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80+ Customers/8000+ Engine Orders

Including Firm Orders and Options

CLEAN II HPC and HPT technologies to be infused into GTF fleet.

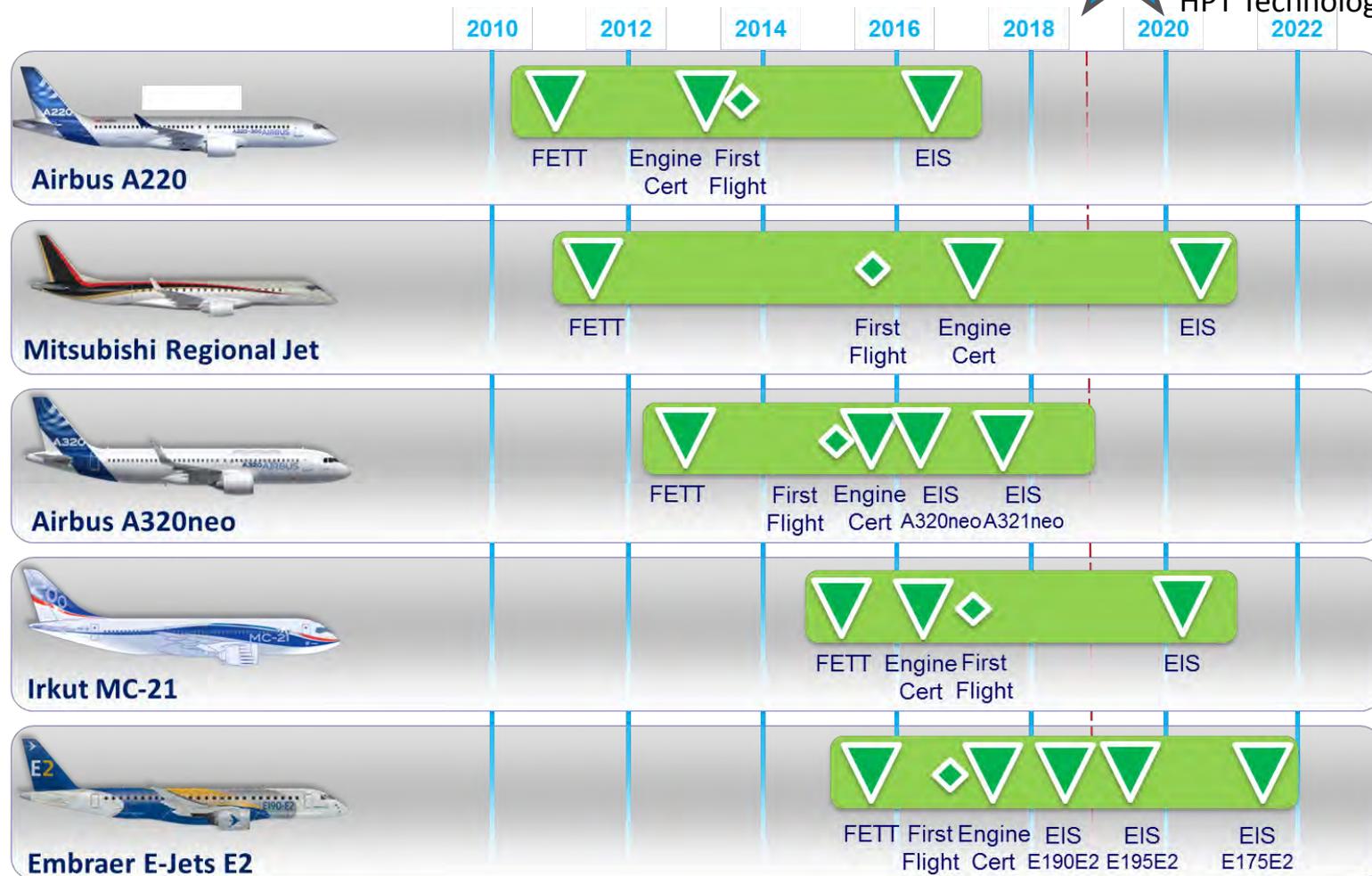


Photo Credit: Airbus



Photo Credit: Widerøe, <https://www.wideroe.no/en/E2>



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Program Summary

High Pressure Compressor scope was completed in 2017, ongoing integration of learning into ground test and flight test assets

High pressure turbine (HPT) Single Element Cascade testing complete, results analysis is underway

HPT START rig baseline testing complete

Industrial phase of CLEEN II technology HPT hardware for START rig continues on-schedule

Queueing-up system level benefits analysis for 2019 and 2020

Future plans are in place for introduction of technologies matured under CLEEN II efforts into Pratt & Whitney's future product offerings



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