AC NO: 91-6A

ATE: 5/24/78



## ADVISORY CIRCULAR

## DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

SUBJECT: WATER, SLUSH, AND SNOW ON THE RUNWAY

- 1. <u>PURPOSE</u>. This advisory circular is issued to provide information, guidelines, and recommendations concerning the operation of turbojet aircraft when water, slush, and snow are on the runway.
- 2. CANCELLATION. AC 91-6, dated January 21, 1965, is canceled.

## 3. BACKGROUND.

- a. Early in the operation of turbojet aircraft, it was determined that adjustment factors should be applied to the takeoff data in order to maintain the aircraft performance requirements as specified in the SR-422 series of the Civil Air Regulations and the Federal Aviation Regulations when water, slush, and/or snow are on the runway. The first test, using a Boeing 707 airplane, with slush depth of 6/10 inch on the runway, showed that retardation of acceleration on takeoff was of such consequence that an offload from the maximum gross weight should be made for a critical field length.
- b. In August 1961, further slush tests were conducted at the National Aviation Facilities Experimental Center (NAFEC) by the Federal Aviation Agency/National Aeronautics and Space Administration using the agency's Convair 880/22M type transport. The test was designed to obtain data regarding the retardation effects of slush and the effects of aquaplaning on the aircraft's takeoff performance, as well as aircraft control problems and damage encountered when operating in a runway slush environment.
- c. The tests at NAFEC were conducted on a slush covered section of a 10,000-foot runway at depths of 0 to 2.0 inches and at velocities of 80 to 160 knots. The retardation forces measured from the deceleration data were considerably greater than those predicted from earlier wheel and tire drag tests and theoretical studies which neglected the factors of slush spray impingement and aquaplaning. Impingement of slush against the aircraft and landing wheels contributed significantly to slush drag forces. At

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velocities above 120 knots, aquaplaning occurred and as a result, drag forces were reduced.

- 4. <u>GUIDELINES AND RECOMMENDATIONS</u>. The following guidelines and recommendations are provided for those situations where an aircraft takes off on runways with standing water, slush, and/or snow:
- a. Takeoffs should not be attempted when standing water, slush, or wet snow greater than 1/2 inch in depth covers an appreciable part of the runway.
- b. The Federal Aviation Regulations concerned with an aircraft's take-off and landing performance and airplane flight manual requirements are predicated on clean, dry runways. Therefore, certain adjustment factors should be applied to the takeoff data when operating in wet snow, slush, or standing water in depths up to 1/2 inch.
- c. Many aircraft manufacturers have provided, in appropriate aircraft manuals, adjustment factors to be applied to the takeoff performance data in the FAA-approved flight manual for varying runway conditions. These adjustment factors may include an increase in the required runway for takeoff, a penalty in runway limited weight, and adjustment to the  $V_1$  speed. The adjustment factors for the runway conditions specified in the applicable operators' manual or flightcrew operating manual should be considered when determining takeoff performance for takeoffs being conducted with water, slush, and/or snow on the runway.
- d. At the present time, there are no validated engineering data available on which to establish accurate adjustment factors; however, a considerable amount of information and experience has been accumulated and is available to operators. The following tables show examples of adjustments currently being used by operators which are consistent with tests conducted by the agency. These tables have been extracted from the manufacturers' airplane operators manual/flightcrew operating manual as examples of the adjustment factors to be applied for takeoff with water, slush, and/or snow on the runway.

					-1~1							
			1/	4 INCH SLU	JSH CORRECT	TONS						
RUNWAY	WEIGHT REDUCTION - LB (KG)							V1 REDUCTION - KTS				
LENGTH	FLAPS	5	FLAPS	3 15	FLAPS 2	20, 25		FLAPS	50 5	FLAPS	20, 25	
FT	S.L.	4000	S.L.	4000	S.L.	4000	L	S.L.	4000	S.L.	4000	
5000	12000 (5500)	12000 (5500)	15000 (7000)	14000 (6500)	16000 (7 <u>50</u> 0)	1500 <b>c</b> (7000)	<u>_</u>	33 .	773	32	31	
7000	15000 (6500)	13000 (6000)	17000 (80 <u>00</u> )	16000 (7500)	119000 (B)	18000 <b>)</b> (8000)	ري	30	29	29	28	
9000	17000 (7500)	15000 (7000)		18000)	(9500)	20000 (9000)	_	29	27	27	27	
11000	18000 (8000)	16000 (7500) <b>€</b>	(10000)	20000 (9000)	24000 (10500)	22000 (10000)	_	28	27	26	26	
36	( <del>200</del> 0)	18000 (8000)	24000 (11000)	22000 (10000)	-	-	_	26	25	-	-	

B-727

FIGURE 1

(See appropriate aircraft flight manual for the correct information.)

B-727

			1	/2 INCH SL	JSH CORREC	TIONS						
RUNWAY		WEIGHT RI	EDUCTION -	LB (KG)				VI REDUCTION - KTS				
LENGTH	FLAP	S 5	FLAP	S 15	FLAPS :	20, 25		FLAPS	5,11	FLAPS	20, 25	
FT	S.L.	4000	S.L.	4000	S.L.	4000 €	7	S.L.	4000	AL.	4000	
5000	20000	20000	22000	21000	24,000	22000	۸١,	۸ ۵۵	200	~~	24	
	(9000)	(9000)	(10000)	(9500)	THUY	(10000)	576	<b>1</b> 29	27	27	26	
7000	22000 (10000)	22000 (10000)	26000 ( <b>±25</b> 80)	24000 (11000)	25000 (12500)	26000 (11500)		26	25	25	25	
	25000	23000	28000	26000	30000	28000	-	20	22	~>	2)	
9000	(11000)	(30500)	(12500)	(12000)	(13500)	(12500)		26	25	25	24	
1	27000/	25000	50000	28000	32000	30000	_	~~	~>	~ /		
Taboar)	(12000)	(22500)	(13500)	(12000)	(14500)	(13500)		26	25	25	24	
1980 J	29000	27000	32000	30000			_				·	
tare	(13500)	(12000)	(14500)	(13500)		_		26	26	_	_	

FIGURE 2

(See appropriate aircraft flight manual for the correct information.)

B-737

				1/4	INCH SLI		TANDING REDUCTION		EPTH		$\overline{}$		
FLAP F		SL	L 4000	SL	4000	SL	5 4000		10 4000	SL SL	1000	SL	25 4000 -
NORMAL F.L. LIMITED GROSS WT 1000 LB	80 90 100 110 115	3400 3500 4500 5700 6100	3200 4300 5400 6700 7300	2300 3400 4500 5500 6100	3200 4100 5100 6700 7300	1800 3400 4500 5200	3000 3700 4900	1600 3300 4400 5200	4100 4100 200	1500 3000 4300 5100	2600 3900 4900 6100	1500 2800 4200 4700	2500 3900 4900 6100

T :	LA INCH SLUSH OR STANDING WATER DEPTH												
	GROSS WEIGHT REDUCTION LB												
FLAP POS					2		5		10	1	.5	25	
PRES	ALT		V4000	SL	4000	SL	4000	SL	4000	SL	4000	SL	4000
NORMAL F.L. LLMINEN PROSS WT 1000 BB	80 90 100 110 115	8400 9400 12200 13200	6600 9000 11200 13900 15000	5300 7300 9300 12100 13000	6600 8600 10900 13500 15000	4800 7100 9200 12100	6400 8400 10700 13400	4500 6800 9000 12000	6000 8400 10400 13200	4300 6300 9000 12000	5600 8300 10400 13000	4200 6300 9000 12000	5900 8300 10400 13000

FIGURE 3

(See appropriate aircraft flight manual for the correct information.)

NOTE: For takeoffs in slush or standing water, reduce the normal runway limited takeoff gross weight by the appropriate amount shown in Figure 3. The weight reduction figures shown in Figure 3 reflect the effect of runway condition on all-engine operating performance. (F.L. = field length)

B-707

	MAXIMUM G	ROSS WEIGHT AT BRA	KE RELEASE FOR VI	= Vmcg - LB (KG)	451
Slush Depth	Airport Altitude	7,000 Ft.	Runway Length 8,000 Ft.	Available ,000 Ft.	90000 Ft.
1/4"	Sea Level	144,000 (65,400)	<b>F133</b>	(166,000 (1166,700)	*
	4000 Ft.	THE	150,000 (68,000)	205,000 (93,000)	263,000 (119,200)
S	A lend	172,000 (78,000)	230,000 (104,300)	*	*
Do	4000 Ft	_	170,000 (77,200)	225,000 (102,000)	*

FIGURE 4

(See appropriate aircraft flight manual for the correct information.)

## \* Not Vmcg Limited

NOTE: See B-707 Operations Manual for  $V_1$  speed reductions.

B-747

	.25 IN (.64C	1) SLUSH	.5 IN (1.3CM)	SLUSH	IŒ	
RUNWAY	WEIGHT	V <sub>l</sub>	WEIGHT	V <sub>l</sub>	WEIGHT	V <sub>1</sub>
LENGTH	REDUCTION	REDUCTION	REDUCTION	REDUCTION	REDUCTION	DREDUCTION
FT	LB(KG)	KTS	LB(KG)	KTS	LB(KG)	KTS
8000	109000 (49,440)	41	73,000 (33,119)	D 41 30 27	151000 (68,498)	41
9000	47000 (21,320)	34	71,000 (32,661)		46400 (20,870)	46
10000	47000 (21,320)	3	71,000 (33,570)		25000 (11,340)	36
1000 1000 1000 1000 1000 1000 1000 100	47000 (21,320) 47009 (21,320) 47000 (21,320)	30 28 28 28	78,000 (35,380) 78,000 (35,380) 78,000 (35,380)	26 25 24	25000 (11,340) 25000 (11,340) 25000 (11,340)	34 32 29

FIGURE 5

(See appropriate aircraft flight manual for the correct information.)

NOTE 1: In all applicable cases,  $V_1$  reductions are based on minimum  $V_1$  equal

to Vmcg.

NOTE 2: Vmcg is based on actual ambient conditions.

DC-10

EFFECT OF SLUSH/WATER ON BALANCED TAKEOFF PERFORMANCE—PRIMARY REVERSERS OPERATIVE

SLUSH OR WATER DEPTH = 1/4 INCH

WEIGHT REDUCTION (1000 LB)

					(2000 .	_,					
FLAP			TAKE	OFF FIELD	LENGTH (FE	ET)					
SETTING	6,000	7,000	8,000	9,000	10,000	11,000	12,000	13,000	14,000		
(DEG)							<u> </u>				
1 .5%	-22	<b>-28</b>	<b>-3</b> 4	<b>-</b> 39	<b>-</b> 43	-46		<b>-</b> 50	<b>-</b> 50		
100	-24 -26	-33 -35	-39 -40	<b>-43</b>	-45 -45	-45 -45	ヘンジ				
50 10° 15° 20°	-30	<b>-</b> 35 <b>-</b> 37	41	-44 -43	747	747	<b>V</b>				
V <sub>1</sub> REDUCTION (KIAS)											
RUNWAY			TAKI	OFF FIELD	LENGTH (FE	EN					
SLOPE	6,000	7,000	8,000	9,000	10,000	27,000	12,000	13,000	14,000		
(%)											
+1%	-22	-24	<b>-</b> 25	-27		-28	<b>-</b> 29	<b>-</b> 30	<b>-31</b>		
0 <b>-</b> 1%	-24 -27	<b>-</b> 26 <b>-</b> 29	<b>-27</b> <b>-3</b> 0	<b>-</b> 29 <b>-</b> 32	1 430	-31 -35	<b>-</b> 32 <b>-</b> 36	-33 -37	-34 -39		
	-2/	-27		-)2				-21			
			CTIICU	CESTIAN CO.	ר/ו שיים מיים	TNCU					
			SLUSH	OK WALER	DEPTH = 1/2 TION (1000)	TMCH					
			~~~	THE RESOU	1101 (1000	ш)					
FLAP			TAK	OF FIELD	LENGTH (FE						
SETTING	6,000	7,000	8,000	9,000	10,000	11,000	12,000	13,000	14,000		
(DEG) 50 100	- 32	ļ	-		<del></del>			ļ			
100	-36 -39	-44	-52 -56	-58 -61	-62 -62	<b>-</b> 64	-64 -62	-64	-64		
150	41	I <>≫~	-57	<b>-60</b>	-61 -	-62 -61	-02				
15° 20°	<del>-45</del>	153/	-58	<b>-</b> 60		_~_					
		17			UCTION (VTA	<u> </u>	<u> </u>	<del>*</del>			
	^	•		v1 120	UCTION (KIA	3)					
RUNWAY			TAK	OFF FIELD	LENGTH (FE	ET)					
SLOPE	6-000	7,000	8.000	9.000	10,000	11.000	12,000	13.000	1/000		

RUNWAY	125	TAKEOFF FIELD LENGTH (FEET)										
SLOPE (%)	5,000	7,000	8,000	9,000	10,000	11,000	12,000	13,000	14,000			
+1%	-17	<b>-</b> 19	-21	-22	-24	<b>-</b> 26	-27	-27	-28			
0 -1%	-19 -21	-21 -23	-23 -25	-24 -27	<b>-</b> 26 <b>-</b> 29	-28 -31	-29 -32	-30 -34	-31 -35			

FIGURE 6 (See appropriate aircraft flight manual for the correct information.)

NOTE: The above performance information charts are only samples and should not be used in computing takeoff data. All performance data should be obtained from the appropriate charts in the manufacturers' operations/flight-crew operating manual for the specific model, engine power, and type aircraft operated.

e. The operations manual of the air carrier and commercial operator or other appropriate documents for general aviation aircraft should include specific instructions for the flightcrew on each type of turbojet aircraft showing the gross weight reduction,  $V_1$  speed adjustments, and/or additional runway length required for the conditions described. These instructions should outline details of the methods to be used in determining runway conditions at departure time.

J. A. FERRARESE Acting Director

Flight Standards Service

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