



**U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION**

National Policy

**ORDER
8260.58D**

Effective Date:

**SUBJ: United States Standard for Performance Based Navigation (PBN) Instrument
Procedure Design**

This order prescribes standardized methods for designing and evaluating Performance Based Navigation (PBN) instrument flight procedures (IFPs) in the United States and its territories.

Performance Based Navigation (PBN) describes an aircraft's capability to navigate using performance standards and is comprised of Area Navigation (RNAV) and Required Navigation Performance (RNP). RNAV enables aircraft to fly on any desired flight path within the coverage of ground or space based navigation aids, the limits of the capability of the self-contained systems, or a combination of both capabilities. RNP is RNAV with the addition of onboard performance monitoring and alerting capability. A defining characteristic of RNP operations is the ability of the aircraft navigation system to monitor the navigation performance it achieves and inform the crew if the requirement is not met during an operation. The performance requirements of PBN are conveyed to the users through navigation specifications.

This order is to be used by all personnel responsible for the preparation, approval, and promulgation of PBN IFPs.

A handwritten signature in black ink, appearing to read "Timothy R. Adams", is positioned above the printed name.

Timothy R. Adams for
Robert M. Ruiz
Acting Executive Director, Flight Standards Service

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Chapter 1. General Information

Section 1-1. Purpose

1-1-1. Purpose of This Order. This order provides guidance for the design and evaluation of Performance Based Navigation (PBN) Instrument Flight Procedures (IFPs). For the purposes of this order, PBN IFPs are those based on Area Navigation (RNAV) or Required Navigation Performance (RNP) to include transitions to an Instrument Landing System (ILS) or Ground Based Augmentation (GBAS) Landing System (GLS) final segment (see appendix D and Order 8260.19, Flight Procedures and Airspace, section 4-6 and chapter 8 paragraph “Equipment Requirements Notes”). This order contains guidance that is pertinent to 14 CFR part [95](#) and [97](#).

1-1-2. Audience. All personnel who are responsible for IFP development and/or evaluation.

1-1-3. Where You Can Find This Order. You can find this Order on the [FAA Web site](#).

1-1-4. What This Order Cancels. Order 8260.58C titled “United States Standard for Performance Based Navigation (PBN) Instrument Procedure Design” dated 09/15/2022.

1-1-5. Explanation of Changes. Significant areas of new direction, guidance, policy, and criteria as follows:

a. General.

- (1) Administrative and editorial changes.
- (2) Changed references to Required Navigation Performance (RNP) values to cross track tolerance (XTT) values.

b. Table of Contents. Updated to reflect the correct paragraph and page numbering.

c. Chapter 1. Added clarification that Obstacle Evaluation Area (OEA) width changes will occur abruptly when switching to A-RNP NavSpec segments.

d. Chapter 3.

- (1) Added requirement for A-RNP NavSpec to be continued from feeder or initial segment until reaching the final segment once initiated.
- (2) Removed requirement for reducing localizer performance (LP) and localizer performance with vertical guidance (LPV) final segment length to accommodate reduced XTT intermediate segments.

e. Chapter 5. Deauthorized A-RNP and RNP AR DP NavSpecs for vector Standard Instrument Departures (SIDs).

f. Appendix A. Added Administrative Information as appendix A and reordered existing appendices.

Section 1-2. Basic Criteria

1-2-1. General.

a. Source directive. Order 8260.3, United States Standard for Terminal Instrument Procedures (TERPS), applies unless otherwise specified by this directive.

b. Related directives. The following orders are referenced in the directive or otherwise related to the development/documentation of PBN procedures:

- (1) Order 8260.19, Flight Procedures and Airspace
- (2) Order 8260.46, Departure Procedures (DP) Program.

c. Scope. This chapter contains only that information common to all PBN IFPs. Criteria which do not have general application are located in the chapters specific to the flight phase, navigation type, or application.

1-2-2. Formulas/Calculations. Formulas in this directive use degree calculations. The examples provided use only two decimal places; therefore, the answers are approximate values. Actual calculations must use the maximum number of values after the decimal place for each step. Rounding of the final result should be for documentation purposes only. For the purposes of this order, the term “r” represents the mean radius of the earth used in spherical calculations and is equal to 20890537 feet. For mathematics convention and other common equation terms see Order 8260.3.

1-2-3. Geospatial Standards. PBN IFPs must be evaluated geodetically (see Order 8260.3).

1-2-4. PBN Concept. The PBN concept is made up of three interrelated elements: the navigation specification, the Navigational Aid (NAVAID) infrastructure, and the navigation application. These three elements define the minimum required aircraft/system performance used as the basis for the obstacle evaluation areas described in this order.

a. Navigation Specification (NavSpec). The NavSpec is a set of aircraft/aircrew certification and operational approval requirements that support a navigation application within a defined airspace concept. NavSpecs which require on-board performance monitoring and alerting are termed RNP specifications. Those that do not require on-board performance monitoring and alerting are termed RNAV specifications. The NavSpec defines the performance required by an RNAV or RNP system as well as any functional requirements (e.g., ability to conduct curved path legs). See current version of AC 90-100, AC 90-101, AC 90-105, and AC 90-107 for definitions. NavSpecs include:

- (1) Performance. The performance required in terms of accuracy, integrity, and continuity.
- (2) Functions. The functions available to achieve the required performance.

(3) Navigation sensors. The navigation sensors that may be used to achieve the required performance.

(4) Procedures. Flight crew and other procedures to achieve the required performance.

b. NAVAID infrastructure. NAVAID infrastructure relates to space or ground-based NAVAIDS addressed in the NavSpec.

c. Navigation application. Also referred to as flight phase. A navigation application is the operation in which a specific NavSpec and associated performance requirements applies (e.g., Air Traffic Service (ATS) route, IFP segment, etc.).

Table 1-2-1. Navigation Accuracy by NavSpec/Flight Phase

Navigation Specification	Flight Phase						
	En route Domestic	STAR/Feeder/TAA	Initial	Intermediate	Final	Missed ¹	Departure
RNAV 2	2.00	2.00					2.00
RNAV 1 ²		1.00					1.00
RNP 2	2.00						
RNP 1 ²		1.00					1.00
RNP APCH ^{2,3}			1.00	1.00	0.30 ⁴ /40m ⁵	1.00	
A-RNP ^{2,6,9}	2.00 or 1.00 ⁸	1.00 or 0.30	0.30	0.30	RNP APCH	0.30	1.00 or 0.30
RNP AR APCH			1.00 - 0.10	1.00 - 0.10	0.30 - 0.10	0.10 - 1.00	
RNP AR DP							0.30 - 1.00
RNP 0.3 ⁷	0.30	0.30	0.30	0.30		0.30	0.30

¹. Missed approach section 2 only. RNP AR APCH sections 1a and 1b.

². STAR/Feeder/TAA, departure, initial, intermediate, and missed section 2. Beyond 30 NM from ARP the effective Cross Track Tolerance (XTT)/Along Track Tolerance (ATT) for the purpose of IFP design is 2.00.

³. RNP APCH part A is enabled by GNSS and baro-VNAV and part B is enabled by SBAS.

⁴. Part A only.

⁵. Part B along-track performance is 40 meters; angular performance requirements apply to lateral.

⁶. A-RNP permits a choice of RNP lateral navigation accuracies in some flight phases. Apply the largest lateral navigation accuracy for the flight phase unless a smaller value is operationally required to achieve the desired ground track.

⁷. For helicopter operations only. RNP APCH applies to the final flight phase. RNP 0.3 is enabled by SBAS.

⁸. Remote Continental use 2.00 (see latest version of AC 90-105 for definition of remote continental).

⁹. Use of A-RNP STAR/Feeder/TAA or departure XTT less than 1.00 requires Flight Standards approval.

1-2-5. Obstacle Evaluation Area (OEA) and Flight Path Construction.

a. Course.

(1) Lateral and longitudinal dimensions. Construct straight-line courses as a geodesic path. Construct parallel and trapezoidal boundary lines as a locus of points measured perpendicular to the geodesic path. Determine OEA lateral boundary relative to course centerline using ellipsoidal calculations.

(2) Vertical dimension (surface elevations). Obstacle clearance evaluation/clearance surfaces are generally defined as either a level or sloping plane relative to a defined origin Mean Sea Level (MSL) elevation. Evaluate the orthometric height above the geoid (MSL) of obstacles and terrain relative to these surfaces.

(3) Alignment. The alignment tolerance is ± 0.03 degrees of all course differentials. The alignment restriction between Track-to-Fix (TF) and Radius-to-Fix (RF) legs is no course change exceeding alignment tolerance. Otherwise, the general alignment restriction at or above Flight Level (FL) 195 is limited to a maximum course change of 70 degrees, while below FL 195 is 90 degrees; however, specific requirements stated in a segment's chapter/section take precedence. When preceded by a Direct-to-Fix (DF) leg, the alignment restriction applies to both the early and late tracks.

Note: When using pre-established fixes, acceptably small differences in fix coordinates can cause legs to exceed alignment tolerance. The alignment tolerance can be exceeded when the calculated ideal fix coordinates round to match the pre-existing fix coordinates (see Order 8260.19, paragraph 8-5-2).

b. Area dimensions. For IFP design purposes, the term Cross Track Tolerance (XTT) is the larger of basic XTT and the effective XTT associated with a leg. The OEAs defined in this order are the minimum size OEAs authorized; larger OEAs may be developed where operationally advantageous.

(1) Length.

(a) Minimum length (fix-to-fix). Generally, minimum leg length is the lesser of $2 \times \text{XTT}$ or 1 Nautical Mile (NM), but where applicable may also be no less than;

1. The sum of the Distance of Turn Anticipation (DTA) for each Fly-by (FB) turn (see formula 1-2-1).

Note: Not applicable for FB turns of 10 degrees or less.

Formula 1-2-1. Distance Turn Anticipation

$$DTA = R \times \tan\left(\frac{\beta}{2}\right)$$

Where:

β = magnitude of heading change in degrees

R = turn radius

Example:

$$DTA = 3.52 \times \tan\left(\frac{75}{2}\right)$$

$$DTA \approx 2.7$$

Note: This formula produces a value with the same unit as R.

2. For TF legs following a Fly-over (FO) turn a length sufficient to allow the aircraft to return to course centerline. Determine using formula 1-2-2.

Formula 1-2-2. TF Minimum Leg Length Following FO Turn

If $\beta < \arccos(\sqrt{3} - 1)$ use:

$$L = R \times \left[\sin(\beta) + 2 \times \sin\left(\arccos\left[\frac{1 + \cos(\beta)}{2}\right]\right) \right] + DTA$$

If $\beta \geq \arccos(\sqrt{3} - 1)$ use:

$$L = R \times [\sin(\beta) + 4 - \sqrt{3} - \sqrt{3} \times \cos(\beta)] + DTA$$

Note: L must not be less than 1 NM

Where:

R = turn radius at the first fix

β = magnitude of heading change at the first fix

DTA = value from formula 1-2-1 for the second fix

Note: When multiple courses precede the FO fix (e.g., Heading-to-Altitude (VA)-DF legs) use the track resulting in the greatest magnitude of heading change at the first fix.

Example:

$$L = 5.6 \times \left[\sin(33) + 2 \times \sin\left(\arccos\left[\frac{1 + \cos(33)}{2}\right]\right) \right] + 2.7$$

$$L \approx 10.16$$

Note: This formula produces a value with the same unit as R.

(b) RF leg length may be calculated using formula 1-2-3. Use in conjunction with formula 1-2-4 when necessary to calculate an RF leg of a specific length.

Formula 1-2-3. RF Leg Length

$$L = \frac{\alpha \times \pi \times R}{180}$$

Where:

α = degrees of arc

R = arc radius

Example:

$$L = \frac{98.9 \times \pi \times 4.2}{180}$$

$$L \approx 7.25$$

Note: This formula produces a value with the same unit as R.

Formula 1-2-4. Degrees of an Arc

$$\alpha = \frac{180 \times L}{\pi \times R}$$

Where:

L = arc length

R = arc radius

Note: This formula works with any unit so long as L & R share the same unit.

Example:

$$\alpha = \frac{180 \times 7.25}{\pi \times 4.2}$$

$$\alpha \approx 98.90^\circ$$

(c) Maximum length (fix-to-fix). Maximum Standard Instrument Approach Procedure (SIAP) segment length is as specified in Order 8260.3, chapter 2. In no case should the length of any PBN leg exceed 500 NM.

(d) Along-Track Tolerance (ATT). OEAs are constructed and evaluated from ATT prior to a leg's initial fix (early) to ATT past its termination fix (late). ATT is equal to the navigation accuracy associated with the flight phase (see table 1-2-1). ATT (same or adjacent legs) can overlap when legs meet minimum length requirements.

1. Except where specified otherwise (e.g., intermediate segment) use the ATT specific for the leg being constructed for both early and late ATT.

2. The baseline used for determining early/late ATT area depends on the leg/turn type.

a TF (no turn) and FO turn. ATT is relative to a line perpendicular to the inbound leg at the fix (see figure 1-2-1).

b FB turn. ATT is relative to the bisector used for OEA construction (see figure 1-2-2).

c RF. ATT is relative to a line drawn radially from the turn center point. Distance is measured from the fix along the RF arc (see figure 1-2-3).

Figure 1-2-1. ATT Straight

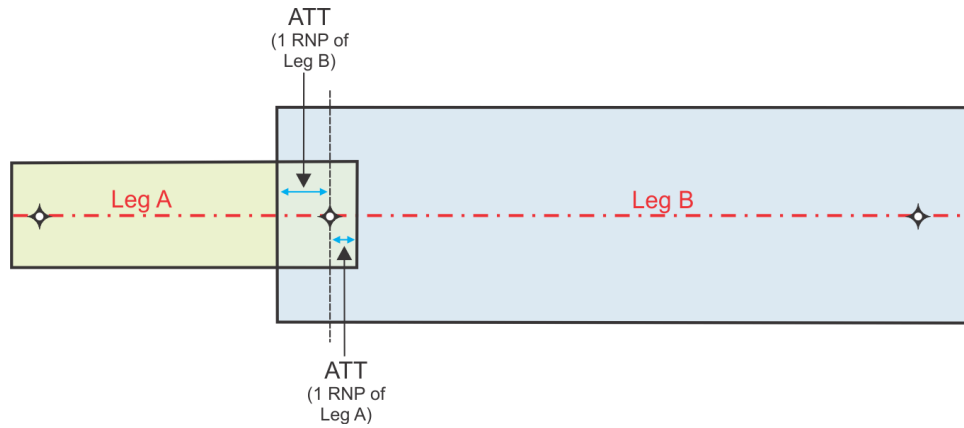


Figure 1-2-2. ATT Turning

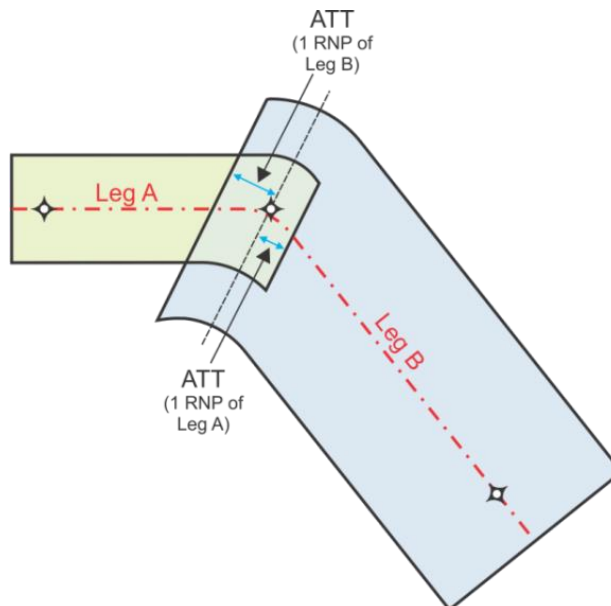
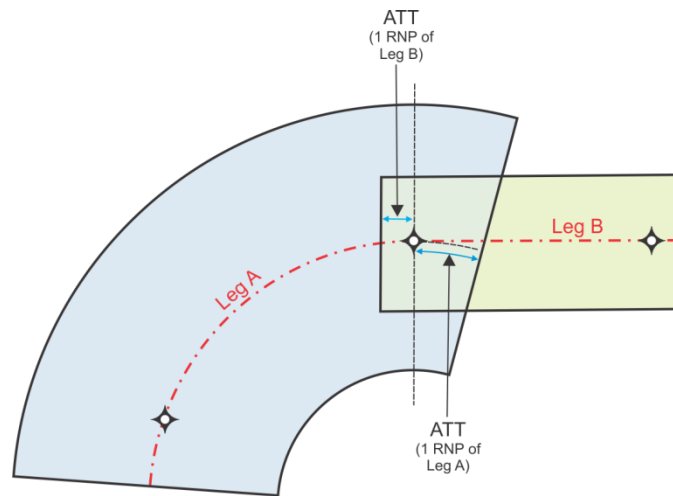


Figure 1-2-3. ATT RF



(2) Width.

(a) Cross-track tolerance (XTT). The XTT is used to determine the OEA lateral dimensions. The XTT is derived from table 1-2-1 navigation accuracy for the defined NavSpec and flight phase. Primary area width is $2 \times \text{XTT}$ either side of course and secondary area width is $1 \times \text{XTT}$ either side of the primary area, if applicable. The XTT for A-RNP and RNP AR is the specified RNP value for the leg being evaluated. Reference to route width is often specified as nautical mile values measured from secondary area edge across the primary area to the secondary edge at the other side. For example, feeder widths may be referred to as “2-4-4-2.”

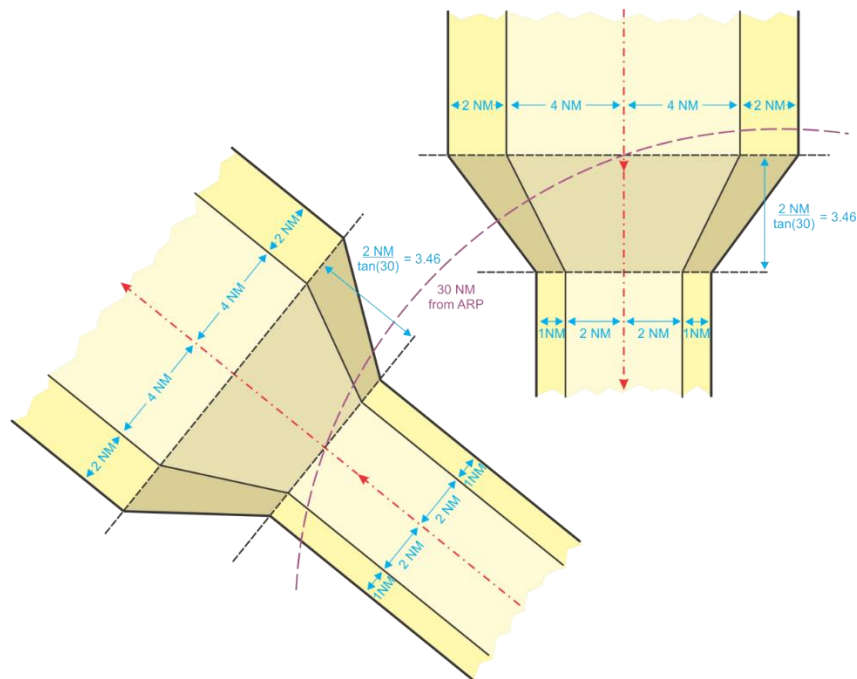
(b) Width changes. RNP AR APCH, A-RNP, and RNP AR DP width changes occur abruptly at the start fix of a leg with a new RNP. If the segment prior to an A-RNP segment is not A-RNP, any applicable width change will occur abruptly. A-RNP width values may only be decreased prior to final and increased on departure and missed approach. For all other IFPs, OEA construction accommodates width changes at the transition to the final approach mode and at the transition to or from the en route mode. See the applicable chapter for width changes at the final approach transition. For the en route transition, legs that cross a point 30 NM from Airport Reference Point (ARP), the OEA primary area width expands or tapers at a rate of 30 degrees relative to course to the appropriate width (see figure 1-2-4 and figure 1-2-5). Calculate the total along-track distance required to complete the transition using formula 1-2-5. Standard Terminal Arrival Route (STAR)/feeder and approach legs designed to cross within 30 NM of the ARP more than once do not change to RNAV 1/RNP 1/A-RNP (terminal mode) until the 30 NM limit is crossed for the last time. A departure or missed approach segment designed to cross a point 30 NM from the ARP more than once changes to RNAV 2/RNP 2/A-RNP (en route mode) and remains expanded when it crosses the boundary the first time. A-RNP width changes for mode transitions at 30 NM from ARP use the 30-degree splay method to/from effective XTT.

1. Non-RF width changes.

a Inbound. The secondary area tapers from en route mode width when the 30-NM point is crossed to terminal mode width abeam the point the primary area is fully tapered. Leg width tapers regardless of fix location within the tapering section unless a turn is associated with the fix. Delay OEA taper until the turn is complete and normal OEA turn construction is possible.

b Outbound. The secondary area expands from terminal mode width when the 30-NM point is crossed to en route mode width abeam the point the primary area is fully expanded. Leg width expands regardless of fix location within the expansion section unless a turn is associated with the fix. Start the OEA expansion so as to achieve full expansion to en route mode width prior to the turn.

Figure 1-2-4. Non-RF Width Change at 30 NM from ARP



2. RF width changes. Calculate the perpendicular distance from the RF leg track centerline to primary and secondary boundaries at any along-track distance (specified as degrees of RF arc) from the point the track crosses the 30-NM point using formula 1-2-6 (see figure 1-2-5). The width transition must be fully contained within a single RF leg.

Formula 1-2-5. RF Width Change Distance at 30 NM from ARP

$$D = \frac{2 \times \text{XTT}_{\text{large}} - 2 \times \text{XTT}_{\text{small}}}{\tan(30)}$$

Where:

$\text{XTT}_{\text{large}}$ = the largest XTT value

$\text{XTT}_{\text{small}}$ = the smallest XTT value

Note: This formula works with any unit so long as all XTT values share the same unit.

Example:

$$D = \frac{2 \times 2 - 2 \times 1}{\tan(30)}$$

$$D \approx 3.46$$

Formula 1-2-6. RF Leg XTT Change

$$\text{Boundary} = \frac{1}{2} \text{Width}_{\text{large}} - \frac{\frac{1}{2} \text{Width}_{\text{small}} \times X_{\text{angle}} \times R \times \pi}{D \times 180}$$

Where:

$\frac{1}{2} \text{Width}_{\text{large}}$ = $2 \times$ largest XTT for primary area or $3 \times$ largest XTT for secondary area

$\frac{1}{2} \text{Width}_{\text{small}}$ = $2 \times$ smallest XTT for secondary or $3 \times$ smallest XTT for secondary area

X_{angle} = angle relative to center point from beginning of taper to point of interest

R = arc radius

D = along track distance required to complete the XTT change

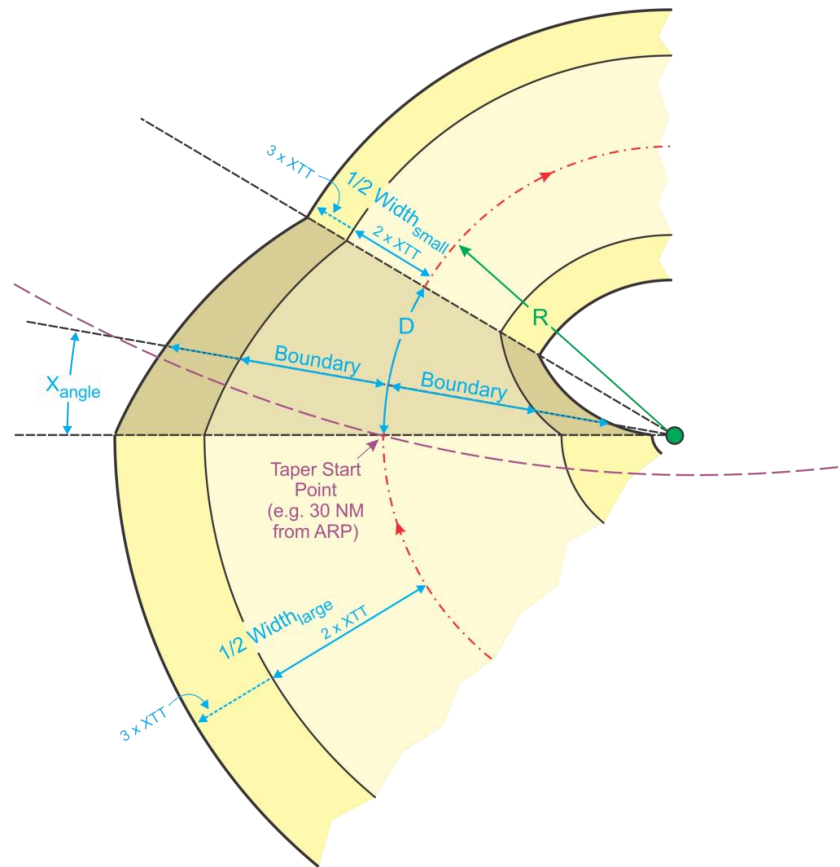
Note: This formula works with any unit so long as all XTT values share the same unit.

Example:

$$\text{Boundary} = 6 - \frac{3 \times 10 \times 6.44 \times \pi}{3.46 \times 180}$$

$$\text{Boundary} \approx 5.03$$

Figure 1-2-5. RF Width Change at 30 NM from ARP



c. Turn parameters. For OEA construction, a turn is indicated when the course change exceeds the alignment tolerance of 0.03 degrees (see paragraph 1-2-5.a(3)).

(1) Altitude. Calculate the assumed altitude at a turn fix/point as specified below. The calculated assumed altitude need not exceed the maximum anticipated use specified by ATC. For other cases (e.g., ATS routes, or STAR or Standard Instrument Departure (SID) legs that precede/follow manual termination legs that make it impractical to determine a starting altitude/point) use the highest anticipated altitude at the turn fix based on specified altitudes, airspace, and anticipated use.

Note: Use of the highest anticipated altitude is required even when obstacle clearance is assured for flyability, path repeatability, and airspace/route separation.

(a) Calculate projected altitude. Except for turns inside the Precise Final Approach Fix (PFAF), projected altitudes are calculated using formula 1-3-8 based on the rise of an assumed vertical path from a specified start point.

1. Final approach segment.

a Turns inside the PFAF. Use the final segment vertical path, i.e., glidepath angle (GPA). Determine the projected altitude by calculating the vertical rise from the Landing Threshold Point (LTP)/Fictitious Threshold Point (FTP) using formula 1-3-4.

b Turns at the PFAF. There is no projected altitude. Use the specified PFAF altitude.

2. Intermediate and initial approach segments. The vertical path is 250 ft/NM [Category (CAT) A-E] or 400 ft/NM (COPTER). Determine the projected altitude by calculating the vertical path rise fix-to-fix from the start point/altitude to the turn fix. For the first leg, the start point is the PFAF at the specified PFAF altitude. For subsequent legs, start at the end fix of the preceding leg at the assumed fix altitude.

3. Feeder segments and STAR runway transitions that terminate on an Instrument Approach Procedure (IAP). The vertical path is 250 ft/NM (CAT A-E) or 400 ft/NM (COPTER). Determine the projected altitude by calculating the vertical path rise fix-to-fix from the start point/altitude to the turn fix.

a Feeder. For the first leg, start at the Initial Approach Fix (IAF) at the assumed IAF altitude. The start point/altitude for subsequent legs is the end fix at the assumed fix altitude of the preceding leg.

b STARs. For the first leg, start at the IAP fix at the assumed fix altitude. For subsequent legs, start at the end fix preceding leg at the assumed fix altitude. If a STAR transition serves more than one IAP, base the STAR on the IAP with the highest assumed altitude.

4. Missed Approach (MA) and departure (prior to transition route). For MA legs, the vertical path is the higher of 250 ft/NM (CAT A-B), 500 ft/NM (CAT C-E), 400 ft/NM (COPTER), or the specified climb gradient (CG) in effect at the point of interest. For departure legs (all CATs), the vertical path is the higher of 500 ft/NM below 10000 MSL and/or 350 ft/NM 10000 MSL and above (CAT A-E), 400 ft/NM (COPTER) or the specified CG in effect at the point of interest. For MA, the start point for the first leg is the Start-of-Climb (SOC) applicable to the approach type (see applicable chapter). For departures, the start point for the first leg is the Departure End of Runway (DER) at DER elevation.

a Turn-at-altitude. To calculate the highest altitude in the turn used for OEA construction, see paragraph 1-3-1.g(3); otherwise use the specified turn altitude. The start point for subsequent legs is the point along the course centerline where the specified turn altitude is reached. Calculate the vertical rise from the start point measured direct to subsequent fix.

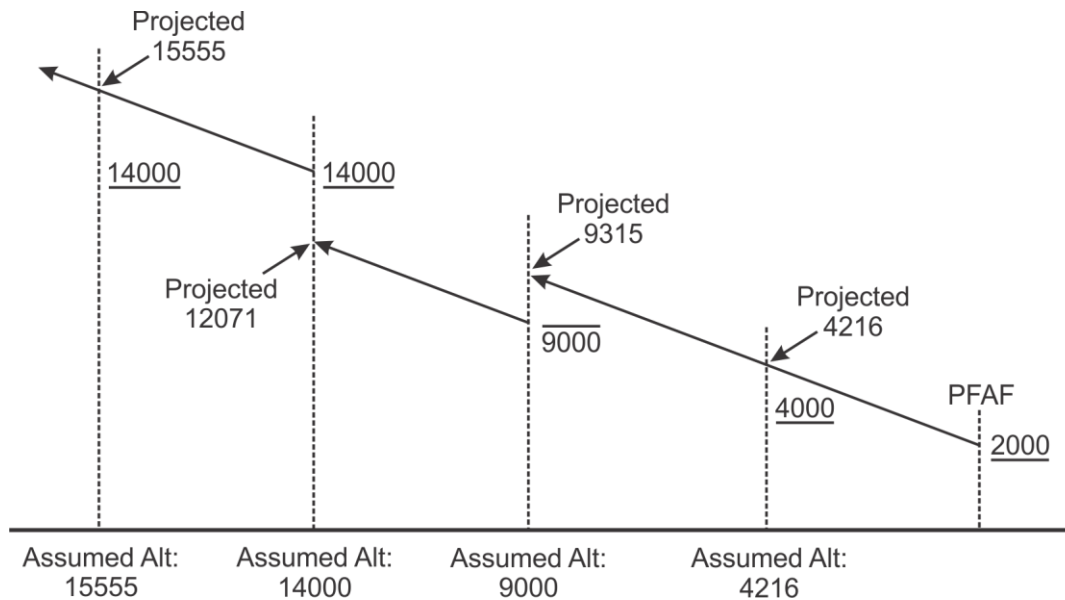
b All other turns. Calculate the projected fix altitude by applying the vertical path rise from the start point/altitude along the course centerline to the subsequent fix/point. For fixes following a DF leg, the course centerline is a direct line from the turn point/fix to the subsequent fix. The start point for subsequent legs is the end fix/point of the preceding leg at the assumed fix/point altitude.

(b) Compare to published altitudes. Compare projected altitude(s) to published altitude(s) to include Minimum En Route IFR Altitudes (MEAs).

(c) Determine assumed altitude (see figure 1-2-6). The assumed altitude used for turn construction is:

1. The higher of the projected altitude and any published minimum or mandatory altitude.
2. The lower of the projected altitude and any maximum altitude, to include the overall maximum altitude applicable to the flight phase (e.g., missed approach clearance limit altitude). The maximum altitude at a fix or for the flight phase can also be specified by Air Traffic Control (ATC).

Figure 1-2-6. Assumed Altitudes for OEA Turn Construction



(2) Airspeed. Locate and use the appropriate Knots Indicated Airspeed (KIAS) from table 1-2-2. Determine the Knots True Airspeed (KTAS) for the turn using formula 1-2-7 based on the assumed altitude at the turn fix/point (see paragraph 1-2-5.c(1)).

Table 1-2-2. Indicated Airspeeds (KIAS)

Flight Phase		Indicated Airspeed by CAT					
		Copter	A	B	C	D	E
At or Above 10000 feet MSL							
En route, STAR/Feeder/TAA, Initial, Intermediate, Missed, Departure		150	180	250	300	300	350
Below 10000 feet MSL							
En route, STAR/Feeder/TAA, Initial, Intermediate		150	150	180	250	250 ¹	310
Final		90	90	120	140	165	250
Missed Approach (MA), Departure		150	110	150	240	265	310
Minimum Airspeed Restriction							
Minimum Airspeed Restriction ²	STAR/Feeder/TAA, Initial, Departure	70	110	140	200 ³	210 ^{3,4}	310
	Intermediate	70	110	140	180	180	310
	Missed Approach	70	100	130	165	185	310
	Final	70 ⁵	Not Authorized				

¹ Consider using 265 KIAS where heavy aircraft routinely exceed 250 KIAS under 14 CFR § 91.117.

² Airspeed restrictions may be established at a charted fix to reduce turn radius, avoid obstacles, accommodate ATC request, etc. Use the fewest number of restrictions possible on the same IFP. Especially avoid consecutive restrictions requiring speed changes of less than 20 KIAS in the same or adjoining segments. Flight Standards or military authority approval is required for missed approach restrictions for other than obstacle avoidance.

³ 250 at or above 10000 feet MSL except for initial and/or STAR termination fix.

⁴ 200 underlying Class B airspace per 14 CFR § 91.117(c).

⁵ Airspeed restrictions between 90 and 70 KIAS are not allowed.

Formula 1-2-7. True Airspeed

$$V_{KTAS} = \frac{V_{KIAS} \times 171233 \times \sqrt{303 - 0.00198 \times \text{alt}}}{(288 - 0.00198 \times \text{alt})^{2.628}}$$

Where:

V_{KIAS} = indicated airspeed for applicable category and segment combination

Example:

$$V_{KTAS} = \frac{165 \times 171233 \times \sqrt{303 - 0.00198 \times 650}}{(288 - 0.00198 \times 650)^{2.628}}$$

$$V_{KTAS} \approx 170.88$$

(3) Maximum bank angle.

(a) Bank limited turns [assumed altitude at turn point/fix < 500 feet above airport elevation (rounded to the nearest foot)]: 3 degrees.

(b) Low altitude transition turns (assumed altitude at turn point/fix from 500 feet above airport elevation (rounded to the nearest foot) up to and including FL 195 (18,000 for helicopters):

1. XTT of the leg being constructed ≥ 1.00 : 25.49 degrees
2. XTT of the leg being constructed < 1.00 : TF turn 20 degrees, RF turn 25.49 degrees
3. Helicopters $< 18,000$ feet: 15 degrees

(c) High altitude transition turns ($> FL 195$): 5 degrees

(d) FB turns only: The lesser of one-half the track change (five degrees minimum) or the applicable low/high altitude transition bank angle. Where operationally beneficial (e.g., to reduce leg length) an exception to the one-half track change rule is authorized at a fix when the XTT for the outbound leg is not less than 1.00, a succeeding turn fix uses the applicable low/high altitude transition bank angle, and/or the succeeding leg meets minimum length requirements.

(4) Tailwind (V_{KTW}). Calculate the appropriate tailwind for the assumed altitude at the turn fix [see paragraph 1-2-5.c(1)] using formula 1-2-8. Where operationally advantageous the 99th percentile wind speed values determined from analysis of a five-year locally measured database may be substituted (see appendix B).

Formula 1-2-8. Assumed Tailwind

$$V_{KTW} = 0.00198 \times \text{alt} + 47$$

Note: If $\text{alt} - \text{apt}_{\text{elev}} \leq 2000$, $V_{KTW} = 30$

Example:

$$\begin{aligned} V_{KTW} &= 0.00198 \times 3400 + 47 \\ V_{KTW} &\approx 53.73 \end{aligned}$$

(5) Turn radius (R). The design turn radius value is based on the turn parameters assumed tailwind (see formula 1-2-8), groundspeed (see formula 1-2-9), assumed altitude at the turn fix/point [see paragraph 1-2-5.c(1)], and bank angle [see paragraph 1-2-5.c(3)]. Calculate R using formula 1-2-10. For RF turn radius, see paragraph 1-2-5.d(3).

Formula 1-2-9. Groundspeed**If alt > FL 195 use:**

$$V_{\text{ground}} = \text{lesser of 570 or } \frac{0.9941 \times \text{alt}}{100} + 287$$

If alt ≤ FL 195 use:

$$V_{\text{ground}} = \text{lesser of 500 or } V_{\text{KTAS}} + V_{\text{KTW}}$$

Where:

 V_{KTAS} = calculated KTAS using alt and applicable KIAS V_{KTW} = calculated or historical tailwind (knots) using alt**Example:**

$$V_{\text{ground}} = \frac{0.9941 \times 26500}{100} + 287$$

$$V_{\text{ground}} \approx 550.44 \text{ knots}$$

Formula 1-2-10. Turn Radius

$$R = \frac{V_{\text{ground}}^2}{\tan(\phi) \times 68625.4}$$

For FB turns, if resulting DTA > 20 NM, then:

$$R = \frac{20}{\tan(\beta \times 0.5)}$$

Where:

 β = magnitude of heading change in degrees V_{ground} = ground speed (knots) ϕ = designed bank angle or 5 degrees where > FL 195**Example:**

$$R = \frac{550.44^2}{\tan(5) \times 68625.4}$$

$$R \approx 50.46 \text{ NM}$$

Formula 1-2-11. RF Bank Angle

$$\phi = \text{atan} \left(\frac{V_{\text{ground}}^2}{R \times 68625.4} \right)$$

Where:

V_{ground} = calculated ground speed (knots) at the highest assumed altitude in the leg

R = turn radius (NM)

Example:

$$\phi = \text{atan} \left(\frac{550.44^2}{50.46 \times 68625.4} \right)$$

$$\phi \approx 5.00^\circ$$

d. Turn construction. When an RNP change is necessary, the construction of the OEA is accomplished using the RNP value of the leg being evaluated extended into the subsequent leg (see figure 1-2-1, figure 1-2-2, and figure 1-2-3).

(1) Turns at FB fixes (see figure 1-2-7). The following steps detail the construction of the minimum OEA adjacent to a single FB turn fix. The centerlines and/or the lateral boundaries may need to be temporarily extended to achieve tangency with the turn arcs. To complete the OEA for a leg, apply the appropriate construction criteria for the remaining fix in the leg and then truncate and/or connect the components such that the most adverse boundary is created. The following steps detail construction of the minimum OEA for a FB turn fix. An OEA larger than the minimum is acceptable where connecting boundary lines becomes problematic due to turns and leg lengths. Leg lengths must comply with paragraph 1-2-5.b(1).

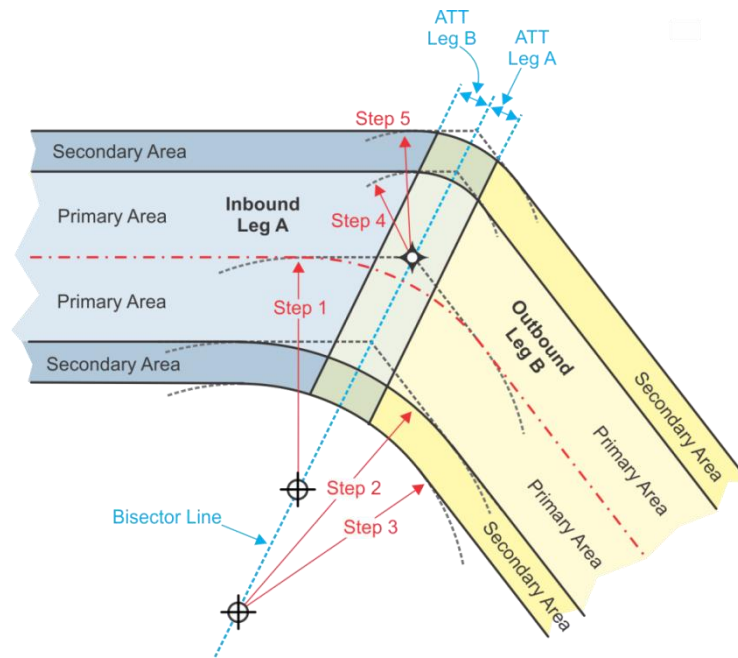
(a) *Step 1.* Establish a line through the turn fix that bisects the turn angle. With the origin on the bisector line, scribe an arc that is tangent to the inbound and outbound centerlines using a radius equal to R.

(b) *Step 2.* Primary area inner boundary. With the origin on the bisector line, scribe an arc that is tangent to the inner primary boundaries of the two legs using a radius equal to $R + \text{XTT}$.

(c) *Step 3.* Secondary area inner boundary (if applicable). Using the origin from *Step 2*, scribe an arc that is tangent to the inner secondary boundaries of the two legs using a radius equal to R.

(d) *Step 4.* Primary area outer boundary. Centered on the turn fix, scribe the primary area outer boundary arc using a radius equal to $2 \times \text{XTT}$.

(e) *Step 5.* Secondary area outer boundary (if applicable). Centered on the turn fix, scribe the secondary area outer boundary arc using a radius equal to $3 \times \text{XTT}$.

Figure 1-2-7. FB Turn Construction

(2) Turns at FO fixes (see figure 1-2-8 and figure 1-2-9).

(a) *Step 1.* FO turn construction incorporates a delay in start of turn to account for pilot reaction time and roll-in time. Calculate the extension distance using formula 1-2-12. FO turn construction is not authorized for A-RNP.

Formula 1-2-12. Reaction & Roll Distance

$$D_{rr} = \frac{V_{KTAS} \times 6}{3600}$$

Where:

V_{KTAS} = calculated KTAS using applicable altitude and KIAS combination

Example:

$$D_{rr} = \frac{253.62 \times 6}{3600}$$

$$D_{rr} \approx 0.42 \text{ NM}$$

(b) *Step 2.* Establish the baseline for construction of the turn expansion area as the line perpendicular to the inbound track at a distance past the turn fix equal to $ATT + D_{rr}$.

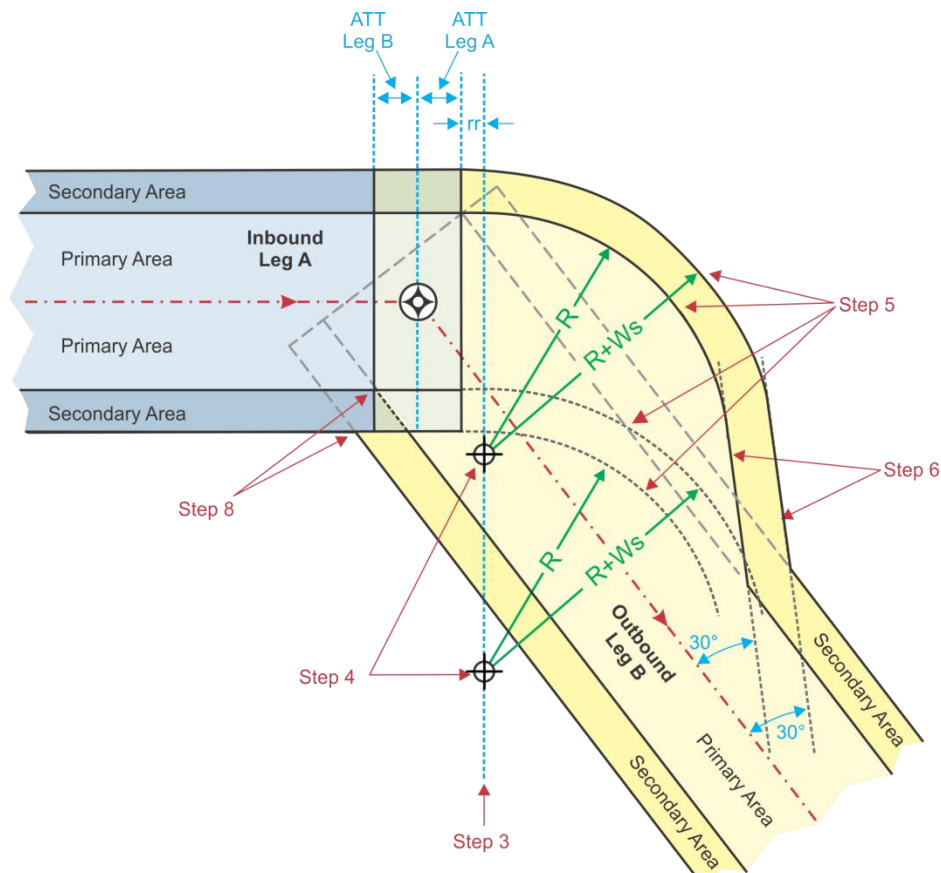
(c) *Step 3.* Determine R based on bank angle intended for design (see paragraph 1-2-5.c(3)).

(d) *Step 4.* On the baseline, locate the center points for the primary and secondary turn boundaries. The first is located at a distance R from the non-turning side primary boundary. The second is located at a distance R from the turning side secondary boundary.

(e) *Step 5.* From these center points construct arcs for the primary boundary of radius R . Complete the secondary boundary by constructing additional arcs of radius $R + \text{secondary area width}$ from the same center points.

(f) *Step 6.* The arcs constructed in *Step 5* are tangent to the outer boundary lines of the inbound leg. Construct lines tangent to the arcs based on the first turn point tapering inward at an angle of 30 degrees relative to the outbound track that joins the arc primary and secondary boundaries with the outbound leg primary and secondary boundaries. If the arcs from the second turn point are inside the tapering lines as shown in figure 1-2-8, then they are disregarded and the expanded area construction is completed. If not, proceed to *Step 7*.

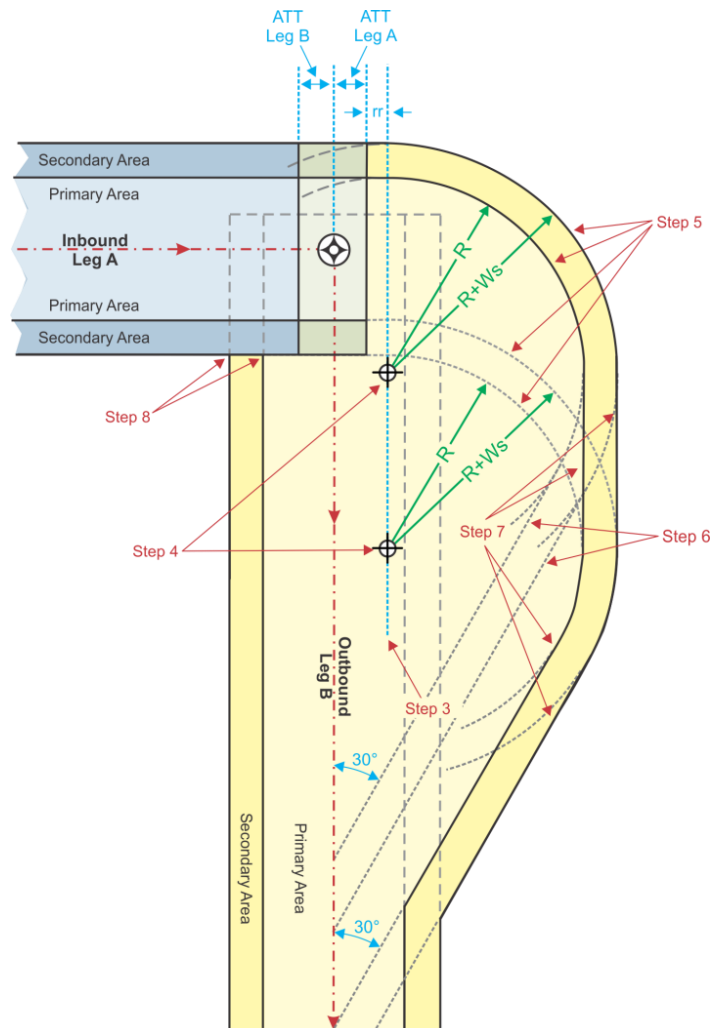
Figure 1-2-8. FO with No Second Arc Expansion



(g) *Step 7.* If both the inner and outer arcs lie outside the tapering lines constructed in *Step 6*, connect the respective inner and outer arcs with tangent lines and then construct the tapering lines from the arcs centered on the second center point.

(h) *Step 8.* The inside turn boundaries are the intersection of the preceding and succeeding leg primary and secondary boundaries.

Figure 1-2-9. FO with Second Arc Expansion



(3) Radius-to-Fix (RF) Turn (see figure 1-2-10 and figure 1-2-11). RF legs are used to control the ground track of a turn where obstructions prevent the design of a FB or FO turn, or to accommodate other operational requirements. Incorporation of an RF leg limits IFP availability to some users; therefore, an RNP APCH that requires RF leg capability to fly the procedure requires Flight Standards approval. The RF leg OEA boundaries are parallel arcs. For OEA construction, the RF start point is the radius extended to early ATT and the RF end point is the radius extended to late ATT.

(a) *Step 1.* Determine the leg R that is required to fit the geometry of the terrain/airspace. Enter this R value into formula 1-2-11 to verify the calculated bank angle (rounded to the nearest whole degree) is ≤ 25 degrees

(b) *Step 2.* Locate the turn center at a perpendicular distance R from the inbound and outbound legs.

(c) *Step 3.* Construct an arc of radius R from the point on centerline tangent to the inbound leg at the turn fix to point on centerline tangent to the outbound leg at the turn fix. (See paragraph 1-2-5.a(3) for alignment tolerance.)

(d) *Step 4.* Primary area outer boundary. Construct an arc of radius $R + \text{primary area half width}$ from a line originating at the turn center extending radially through the RF start point to a line originating at the turn center extending radially through the RF end point.

(e) *Step 5.* Secondary area outer boundary (if applicable). Construct an arc of radius $R + \text{primary area half width} + \text{secondary area width}$ from a line originating at the turn center extending radially through the RF start point to a line originating at the turn center extending radially through the RF end point.

1. Case 1. RF Radius greater than or equal to OEA $\frac{1}{2}$ width (primary and secondary if applicable).

a *Step 6.* Primary area inner boundary. Construct an arc of radius $R - \text{primary area half width}$ from a line originating at the turn center extending radially through the RF start point to a line originating at the turn center extending radially through RF end point.

b *Step 7.* Secondary area inner boundary (if applicable). Construct an arc of radius $R - (\text{primary area half width} + \text{secondary area width})$ from a line originating at the turn center extending radially through the RF start point to a line originating at the turn center extending radially through the RF end point.

2. Case 2. RF Radius less than OEA $\frac{1}{2}$ width (primary and secondary if applicable). This construction is authorized for RNP AR APCH, A-RNP, RNP APCH initial and intermediate segments inside 30 NM from ARP, RNP AR DP, and RNP-1 DP prior to 30 NM from ARP.

a *Step 6.* Construct an angle bisector of the RF leg.

b *Step 7.* Construct a line tangent to the RF leg at the angle bisector.

c *Step 8.* Construct an arc with a radius of $3 \times \text{XTT}$ that is tangent to the start of the RF and the tangent line at the midpoint of the RF. The centerpoint of this arc is centerpoint 1.

d *Step 9.* Construct an arc of radius XTT from centerpoint 1.

e *Step 10.* Construct a line from centerpoint 1 to XTT after the RF end point then radially out from the RF leg centerpoint to the edge of the outer area boundary(s).

f *Step 11.* Construct an arc with a radius of $3 \times \text{XTT}$ that is tangent to the tangent line at the midpoint of the RF and tangent to the end of the RF. The centerpoint of this arc is centerpoint 2.

g *Step 12.* Construct an arc of radius XTT from centerpoint 2.

h *Step 13.* Construct a line from centerpoint 2 to XTT prior to the RF start point then radially out from the RF leg centerpoint to the edge of the outer area boundary(s).

i *Step 14.* Construct a line from centerpoint 1 to centerpoint 2. The area contained within the overlap of the XTT arcs, the line connecting centerpoints 1 and 2 and the RF start and end waypoint ATTs is secondary area. If the XTT arcs don't overlap, the entirety of the inside turn OEA is primary area.

Figure 1-2-10. RF Turn Construction, Case 1

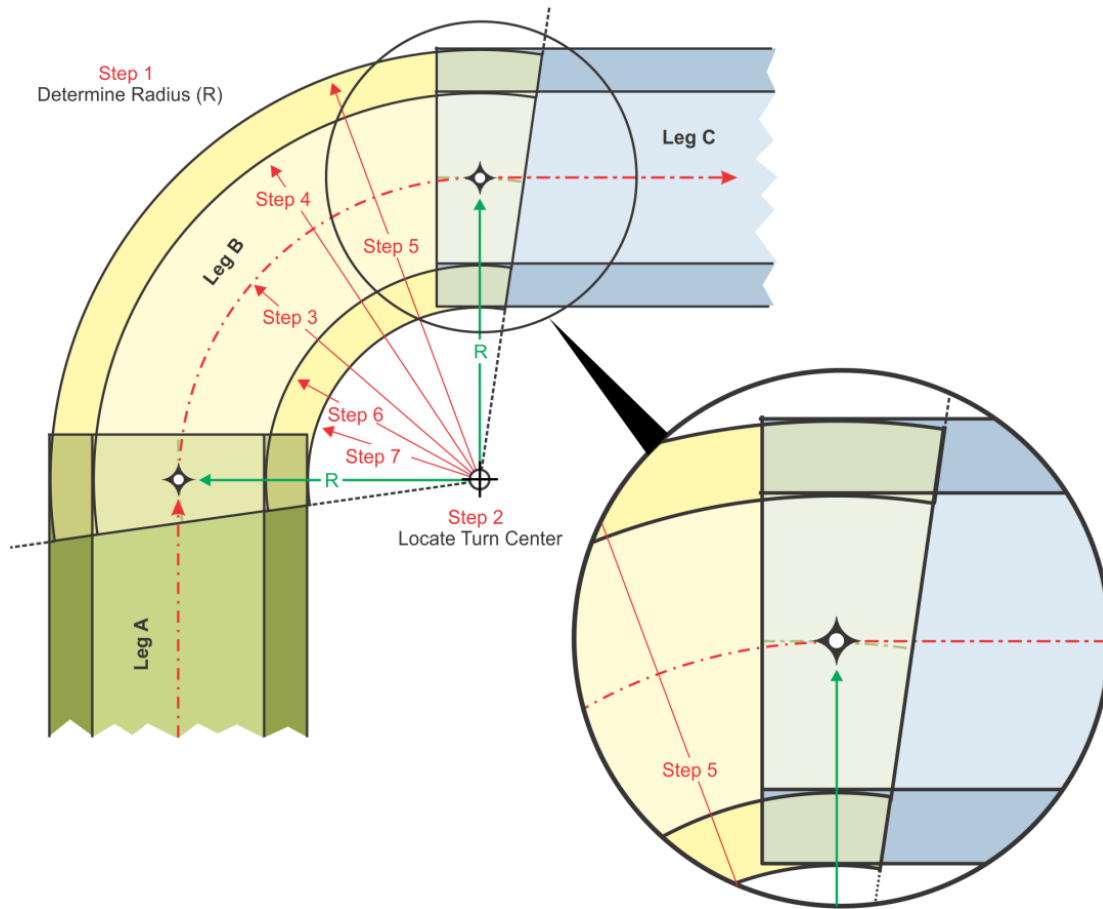
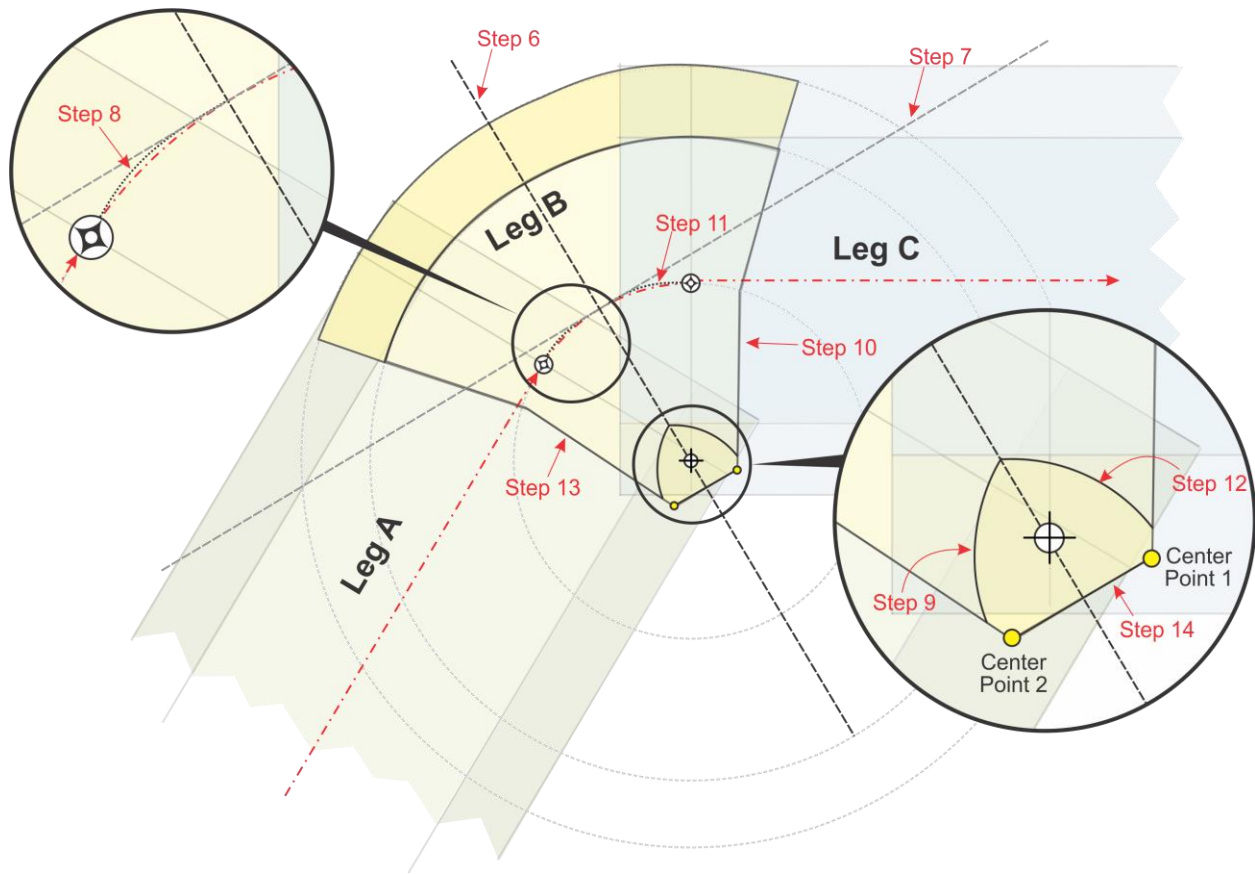


Figure 1-2-11. RF Turn Construction, Case 2



Section 1-3. Common Criteria

1-3-1. Common Criteria. The following common requirements are applicable to PBN IFPs.

a. Descent gradient (DG) calculation. The applicable section of Order 8260.3, chapter 2 applies, except use formula 1-3-1 to calculate descent gradient.

Formula 1-3-1. Descent Gradient

$$DG = \ln \left(\frac{r + \text{alt}_b}{r + \text{alt}_e} \right) \times \frac{r}{D}$$

Where:

Alt_b = beginning altitude (feet)

Alt_e = ending altitude (feet)

D = distance (NM) between the points of interest

Example:

$$DG = \ln \left(\frac{r + 6500}{r + 3200} \right) \times \frac{r}{10.36}$$

$$DG \approx 318.46 \text{ ft/NM}$$

b. Secondary area (when applicable). Obstacles located in secondary areas are subject to less obstacle clearance than those contained in the primary area. Unless otherwise specified, calculate the secondary Required Obstacle Clearance (ROC) using formula 1-3-2. Where an obstacle requires multiple measurements (e.g., an obstacle is equidistant from multiple primary boundary points, it lies along perpendiculars from multiple primary boundary points, etc.), apply the most adverse result from each of the combined primary/secondary measurements. When the secondary area is less than 1 x XTT per paragraph 1-2-5.d(3)(e) and 1-2-5.d(3)(e)2, Ws for formula 1-3-2 is equal to 1 x XTT in feet.

Formula 1-3-2. Secondary ROC

$$\text{ROC}_{\text{secondary}} = (\text{ROC}_{\text{min}} + \text{adj}) \times \left(1 - \frac{d_{\text{primary}}}{w_s}\right) + \text{RASS}$$

Where:

ROC_{min} = min. secondary ROC (feet) at primary boundary (see Order 8260.3, chapter 2)
 adj = ROC adjustments (feet) such as excessive length of final and precipitous terrain
 RASS = remote altimeter setting source adjustment (feet) if applicable
 d_{primary} = dist. (feet) from primary area boundary to obstacle. Zero (0) if not in secondary
 w_s = secondary area width (feet)

Example:

$$\text{ROC}_{\text{secondary}} = (250 + 74.32) \times \left(1 - \frac{405.86}{1822.83}\right) + 91.69$$

$$\text{ROC}_{\text{secondary}} \approx 343.80 \text{ feet}$$

c. ATC turns to join initial and intermediate segments. The first leg of an initial and the first leg of an intermediate segment must be a TF that accommodates a 90-degree intercept angle. Use standard turn parameters at the start fix, except a 25-degree bank angle applies. Where a shorter leg is needed, reduce airspeed in increments of not less than five KIAS until the desired length is achieved (see table 1-2-2).

d. Initial approach segment. Initial approach segments consist of one or more TF or RF legs or a Hold-In-Lieu-of-Procedure Turn (HILPT). Standard length and alignment applies (see paragraph 1-2-5). In some cases, initial segments are optional (see Order 8260.3, paragraph 2-4-1).

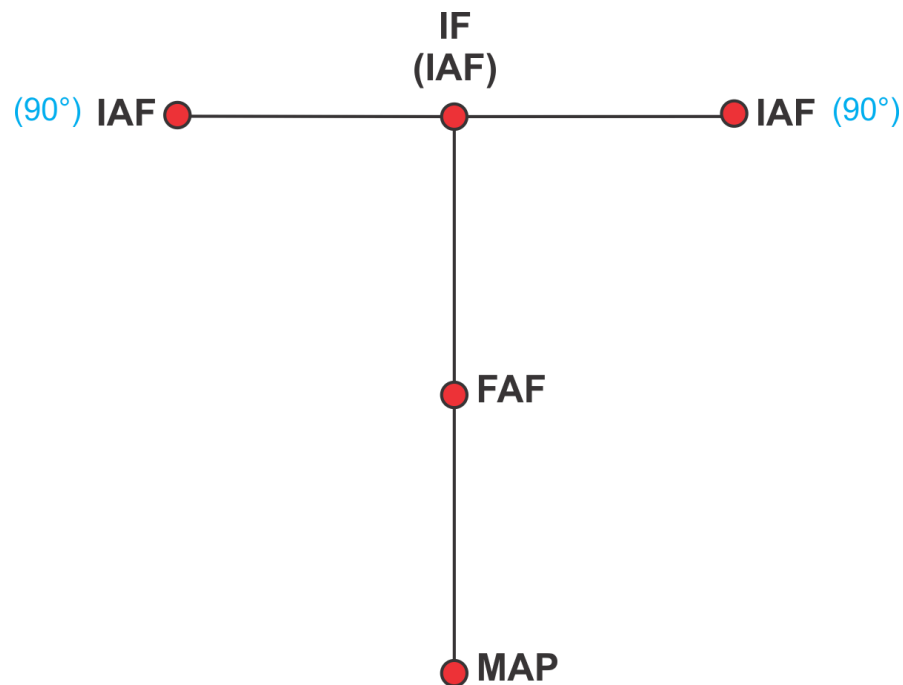
(1) Configuration. The default design incorporates a basic T configuration.

(a) Basic T. Two TF leg initial segments intercept the intermediate segment at 90-degree angles.

(b) Course reversal. Basic T configurations normally include a HILPT course reversal initial established at a fix designated as an Intermediate Fix (IF)/Initial Approach Fix (IAF).

(c) Modified T. Where not practical (e.g., maximum DG exceeded) one or both of the T segments and/or the course reversal may be eliminated.

Figure 1-3-1. Basic T Configuration



(2) Holding pattern. A holding pattern may be incorporated into the initial segment procedure design where an operational benefit can be derived; e.g., arrival holding at an IAF, course reversal pattern at the IF, etc. See Order 8260.3, chapter 16 for RNAV holding pattern construction guidance.

(a) Arrival holding. Ideally, the holding pattern inbound course should be aligned with the subsequent TF leg (tangent to course at the initial fix of the subsequent RF leg). If the pattern is offset from the subsequent TF leg course, the subsequent leg length must accommodate the resulting DTA requirement. The published holding altitude must be at or above the IAF altitude.

(b) Hold-In-Lieu-of-Procedure Turn. Ideally, establish the minimum holding altitude at the minimum IF fix altitude. In any case, the published holding altitude must result in a suitable descent gradient in the intermediate segment. If the pattern is offset from the subsequent TF leg course, the subsequent leg length must accommodate the resulting DTA requirement.

e. Intermediate approach segment. Intermediate approach segments consist of one or more TF or RF legs. Standard length and alignment rules apply (see paragraph 1-2-5) except as follows for the leg ending at the PFAF (the last leg):

(1) Alignment. Use the smallest amount of course change necessary. No course change is default.

(2) Length. For PBN SIAPs, intermediate segment length is independent of the amount of turn at the IF. The default length for CAT A/B is 3 NM. The default length for CAT C/D is 5 NM. The minimum length is as specified in paragraph 1-2-5.b(1). The intermediate segment length may not be greater than 15 NM unless the nominal track of the entire segment is within 15 NM of the ARP. If the intermediate segment exceeds 15 NM and requires a stepdown fix, increase the segment ROC for the segment preceding (as flown) the stepdown fix by five foot per tenth of a track mile between the stepdown fix and the final approach fix. The intermediate segment ROC based on this additive need not exceed 1000 ft. The maximum distance from the stepdown fix to the final approach fix is 15 NM.

f. Final approach segment.

(1) Fixes.

(a) Final segment fixes. Establish stepdown fixes and Visual Decent Points (VDPs) as Along-Track Distance (ATD) fixes relative to the Missed Approach Point (MAP).

(b) Obstacles close to a final approach or stepdown fix. Where turns are designed at the PFAF, Order 8260.3, Chapter 2 “General Criteria” applies except for FB turns at the PFAF the 7:1 (3.5:1 for helicopters) Obstacle Identification Surface (OIS) starts ATT prior to the angle bisector, and extends 1 NM parallel to the final approach centerline.

(2) Final segment vertical path. By default, PBN IFPs should be designed so that all vertical guidance to the runway is based on common threshold crossing height (TCH) and angles. All vertically-guided PBN approaches charted on the same IFP plate must share common TCH and GPA.

(a) Glidepath Angle (GPA)/Vertical Decent Angle (VDA) (see Order 8260.3, section 2-6).

(b) Threshold crossing height. The TCH (nearest whole foot) should accommodate the largest aircraft height group normally expected to use the runway and must not be less than the minimum or exceed the maximum TCH. The TCH is based on achieving an acceptable Wheel Crossing Height (WCH). The WCH is the difference between the published TCH and the approximate glidepath-to-wheel height (see table 1-3-1).

1. The default TCH provides a 30-foot WCH. The minimum WCH is 20 feet and the maximum WCH is 50 feet.

2. Displaced threshold. The TCH over a displaced threshold may result in a WCH of not less than 10 feet provided:

a Pavement equivalent to the strength of the landing runway is present prior to the displaced threshold.

b The calculated height of the glidepath over the beginning of the pavement prior to the displaced threshold is within the minimum/maximum TCH values.

(c) Vertical path calculations.

1. Decision Altitude (DA). Calculate the DA distance from the LTP/FTP using formula 1-3-3. For Localizer Performance with Vertical guidance (LPV)/GLS, see paragraph 1-1-1.

2. PFAF location. To determine the PFAF location, calculate the distance from the LTP/FTP (straight-in aligned procedures) or MAP (circling aligned procedures) to the point the GPA/VDA or design angle intercepts the intermediate segment altitude (see formula 1-3-3). For straight-in procedures the beginning altitude is LTP/FTP + TCH. For circling aligned procedures, the beginning altitude is the airport elevation + 50.

Table 1-3-1. TCH Requirements

Representative Aircraft Type	Glidepath-to- Wheel Height (approximate)	Recommended TCH	Remarks
<u>HEIGHT GROUP 1</u> General Aviation, Small Commuters, Corporate Turbojets, T-38, C-12, C-20, C-21, T-1, Fighter Jets, UC-35, T-3, T-6	10 feet or less	40 feet	General Aviation and Reliever airports, which normally include runways <6000 feet long with reduced widths and/or limited weight bearing capability, limiting larger aircraft use.
<u>HEIGHT GROUP 2</u> F-28, B-737, C-9, DC-9, C-130, T-43, B-2	15 feet	45 feet	Small Hub and Non-hub commercial service airports with limited air carrier service.
<u>HEIGHT GROUP 3</u> B-727/707/720/757, B-52, C-135, C-141, C-17, E-3, P-3, E-8, C-32	20 feet	50 feet	Most primary runways at Medium Hub airports; i.e., runways not normally used by aircraft with glidepath-to-wheel heights > 20 feet.
<u>HEIGHT GROUP 4</u> B-747/767/777, DC-10, A-300, B-1, KC-10, E-4, C-5, VC-25	25 feet	55 feet	Most primary runways at Large Hub airports.

Note: To determine the minimum allowable TCH, add 20 feet to the glidepath-to-wheel height and to determine the maximum allowable TCH, add 50 feet to the glidepath-to-wheel height.

Formula 1-3-3. Distance on Baro Glidepath

$$d_{\text{Baro}} = \ln \left(\frac{r + \text{alt}_e}{r + \text{alt}_b} \right) \times \frac{r}{\tan(\theta)}$$

Where:

alt_e = ending altitude (feet)

alt_b = beginning altitude (feet)

Example:

$$d_{\text{Baro}} = \ln \left(\frac{r + 4500}{r + 1441.6 + 47} \right) \times \frac{r}{\tan(3)}$$

$$d_{\text{Baro}} \approx 57452.70 \text{ feet}$$

3. Glidepath altitude. Calculate the altitude of the glidepath at any distance from the LTP/FTP using formula 1-3-4.

Formula 1-3-4. Altitude on Baro Glidepath

$$Z_{\text{Baro}} = e^{\left(\frac{d \times \tan(\theta)}{r} \right)} \times (r + \text{alt}_b) - r$$

Where:

alt_b = beginning altitude (feet)

d = distance (feet) between the points of interest

Example:

$$Z_{\text{Baro}} = e^{\left(\frac{4708.89 \times \tan(3)}{r} \right)} \times (r + 404.6 + 45) - r$$

$$Z_{\text{Baro}} \approx 696.39 \text{ feet}$$

g. Missed approach and departure.

(1) Sloping Obstacle Clearance Surface (OCS). The OCS slope is dependent on the starting OCS elevation and a specific obstacle of interest. When calculating the height of the sloping surface, a default slope of 40:1 may be used to determine potential controlling obstructions. The actual OCS slope associated with any given obstacle is calculated using formula 1-3-5 and the OCS elevation at a point of interest can be found using formula 1-3-6.

Formula 1-3-5. OCS Slope

$$\text{OCS}_{\text{slope}} = \frac{d_0}{\ln\left(\frac{r + O_{\text{elev}}}{r + \text{OCS}_{\text{start}}}\right)} \times r$$

Where:

O_{elev} = actual or O_{adjusted} obstacle elevation (feet)

$\text{OCS}_{\text{start}}$ = starting altitude (feet) of the OCS

d_0 = dist. (feet) from SOC for missed approach or DER for departure to obstacle (primary) of point abeam obstacle (secondary)

Example:

$$\text{OCS}_{\text{slope}} = \frac{6076.12}{\ln\left(\frac{r + 1152.6}{r + 1000.6}\right)} \times r$$

$$\text{OCS}_{\text{slope}} \approx 39.98$$

Formula 1-3-6. OCS Elevation

$$\text{OCS}_{\text{elev}} = e^{\left(\frac{d_{\text{OCS}}}{r \times \text{OCS}_{\text{slope}}}\right)} \times (r + \text{OCS}_{\text{start}}) - r$$

Where:

d_{OCS} = distance (feet) from $\text{OCS}_{\text{start}}$ point to the point of interest

$\text{OCS}_{\text{start}}$ = starting altitude (feet) of the OCS

Example:

$$\text{OCS}_{\text{elev}} = e^{\left(\frac{2591.8}{r \times 28.5}\right)} \times (r + 1191.75) - r$$

$$\text{OCS}_{\text{elev}} \approx 1282.70 \text{ feet}$$

(2) Climb gradient. Obstacle clearance in a climbing segment is dependent on the aircraft maintaining a minimum CG to a specified altitude. The minimum CG (also referred to as standard CG) is 200 ft/NM (400 ft/NM for helicopter). When one or more obstacle penetrates the default OCS, a single increased CG that provides clearance over all penetrations may be established. A CG greater than 425 ft/NM (600 ft/NM for helicopter) is not authorized for missed approach. Calculate the minimum climb gradient required to achieve the termination altitude at a given distance using formula 1-3-7. The aircraft altitude at a point of interest can be found using formula 1-3-8 and the distance required to achieve an altitude at any CG can be found using formula 1-3-9.

Formula 1-3-7. Required Climb Gradient

$$CG = \ln \left(\frac{r + CG_{\text{term}}}{r + AC_{\text{start}}} \right) \times \frac{r}{D_0}$$

Where:

CG_{term} = climb gradient termination altitude (feet)

AC_{start} = starting altitude (feet) of the aircraft (e.g., DER elevation or SOC)

d_0 = dist. (feet) from SOC for missed approach or DER for departure to obstacle (primary) of point abeam obstacle (secondary)

Example:

$$CG = \ln \left(\frac{r + 2626.52}{r + 1402} \right) \times \frac{r}{3.95}$$

$$CG \approx 309.98 \text{ ft/NM}$$

Formula 1-3-8. Achieved Aircraft Altitude

$$\text{alt}_e = e^{\left(\frac{D \times \text{Vert}_{\text{path}}}{r} \right)} \times (r + \text{alt}_b) - r$$

Where:

D = distance (NM) between the points of interest

$\text{Vert}_{\text{path}}$ = aircraft vertical path (ft/NM)

alt_b = beginning altitude (feet)

Example:

$$\text{alt}_e = e^{\left(\frac{0.76 \times 309.98}{r} \right)} \times (r + 320) - r$$

$$\text{alt}_e = 555.59 \text{ feet}$$

Formula 1-3-9. Climb Distance Between Altitudes

$$D_{\text{climb}} = \ln \left(\frac{r + \text{alt}_e}{r + \text{alt}_b} \right) \times \frac{r}{CG}$$

Where:

alt_e = ending altitude (feet)

alt_b = beginning altitude (feet)

Example:

$$D_{\text{climb}} = \ln \left(\frac{r + 3300}{r + 2105} \right) \times \frac{r}{315}$$

$$D_{\text{climb}} \approx 3.79 \text{ NM}$$

(3) Wind Spiral (WS) parameters. Use the following steps to determine the parameters used in WS construction.

(a) *Step 1.* Calculate KTAS with using the specified/assumed turn altitude/assumed fix altitude and applicable KIAS. Input this value as the ground speed in formula 1-3-10 to find the no-wind turn radius.

(b) formula 1-2-10 *Step 2.* Calculate the Turn Rate (TR) using formula 1-3-10. Maximum TR is three degrees per second.

Formula 1-3-10. Turn Rate (TR)

$$\text{TR} = \text{lesser of } 3 \text{ or } \frac{180 \times V_{\text{KTAS}}}{\pi \times R \times 3600}$$

Where:

V_{KTAS} = KTAS using specified/assumed turn altitude/assumed fix altitude and applicable KIAS

R = calculated no-wind turn radius (NM)

Example:

$$\text{TR} = \frac{180 \times 281.30}{\pi \times 4.30 \times 3600}$$

$$\text{TR} \approx 1.04 \text{ degrees/second}$$

(c) *Step 3.* Determine total magnitude of turn.

1. Turn-at-altitude or turn as soon as practicable. Calculate the total turn magnitude by measuring the arc distance (in degrees) along the no-wind turn radius from the course centerline/late-turn baseline intersection to the direct-to-fix line tangent point.

2. Turn at FO fix. Where the next leg is a DF, use turn-at-altitude or turn as soon as practicable methodology. For TF legs, calculate the total turn magnitude by comparing the inbound and outbound tracks.

(d) *Step 4.* Determine the highest altitude in the turn using the applicable vertical path (see paragraph 1-2-5.c(1)4) starting at the specified/assumed turn altitude/assumed fix altitude plus 1 NM of flight track distance per 12 degrees total turn magnitude.

(e) *Step 5.* Calculate the wind spiral increase (ΔR) relative to the no-wind turn radius for a given incremental turn magnitude using formula 1-3-11. The incremental turn magnitude is an angular measurement from the wind spiral baseline about the no-wind turn radius center point which ranges from zero to the total turn magnitude.

Formula 1-3-11. Wind Spiral ΔR

$$\Delta R = \frac{V_{KTW} \times \beta_{\text{increment}}}{3600 \times TR}$$

Where:

$\beta_{\text{increment}}$ = incremental turn magnitude at point of interest

V_{KTW} = calculated or historical tailwind (knots) using the highest altitude in the turn

TR = calculated turn rate (degrees/second)

Example:

$$\Delta R = \frac{51.55 \times 35}{3600 \times 1.04}$$

$$\Delta R \approx 0.48 \text{ NM}$$

(4) Construction baselines. Missed approach/departure construction baselines are commonly identified by the points that they are comprised of, which can be used to mark the beginning, ending, and/or tie-back point for the OEA. When these lines are used to identify tie-back points for turn construction, the placement of the point will be in alphabetical order from turn-side to non-turn side. For example, a right turning missed approach with lines C-D and L-L' will have points C and L on the right, while a left turning missed approach will have them on the left.

h. A-RNP NavSpec leg usage. Use of A-RNP legs requires chart note, “Use of FD or AP required.” RF turns are the only turn method authorized for turns of greater than 45 degrees when using an RNP value less than 1.00 NM.

Chapter 2. En Route, Standard Terminal Arrivals, Feeder Routes, and Terminal Arrival Areas (TAA)

Section 2-1. En Route

2-1-1. Air Traffic Service (ATS) Routes. ATS routes may be developed to support en route PBN operations. The NavSpec is RNAV 2 (en route flight phase with associated XTT of 2.00). Optional NavSpec may be A-RNP (en route flight phase with associated XTT of 2.00 or 1.00 as appropriate). RNP 0.3 NavSpec may be used for helicopter-only ATS routes (en route flight phase with associated XTT of 0.30). Secondary areas apply.

a. General.

- (1) High altitude routes are for use from FL 180 through FL 450 inclusive.
- (2) Low altitude routes are for use up to but not including 18000 feet MSL.
- (3) Leg/Fix type. Use TF legs only. Fixes may only be FB waypoints or NAVAIDs.

b. OEA construction.

- (1) Area. Except for route termination fixes, apply paragraph 1-2-5. Where applicable, construction is bi-directional.
- (2) Construction at route termination fixes. Order 8260.3 paragraph 14-1-2 applies except primary and secondary boundary arcs are centered ATT past the termination fix.

c. Obstacle clearance. Order 8260.3, section 14-2 applies. Secondary ROC may be calculated as specified in Order 8260.3, section 14-2 or formula 1-3-2 (of this order).

d. Altitudes. Order 8260.3, section 14-3 applies except Minimum Crossing Altitude (MCA) distances are measured from ATT past the fix.

Section 2-2. Standard Terminal Arrival (STAR)

2-2-1. PBN STAR. A PBN STAR may be established to transition from the en route environment to the terminal environment. The NavSpec is RNAV 1 (STAR flight phase with associated XTT of 1.00). Optional NavSpec may be RNP 1 (STAR flight phase with associated XTT of 1.00) or A-RNP (STAR flight phase with associated XTT of 1.00 [or 0.30 with Flight Standards approval]). For STARs that serve multiple airports, use the most conservative distance to determine where the effective XTT is 2.00 (see table 1-2-1 footnote 2). Secondary areas apply, except for A-RNP. Order 8260.3, section 2-2 applies in conjunction with the following PBN specific criteria.

a. General.

(1) Leg type. Use TF or RF legs, except a heading to a manual termination (VM) or from a fix to a manual termination (FM) leg may be established at the STAR termination.

(2) Fixes.

(a) All fixes prior to the termination fix must be FB.

(b) The termination fix may be either FB or FO depending on whether the STAR connects to a SIAP, the attributes of the common fix, and leg type. For this purpose, the first coded fix of the SIAP (i.e., the start of the SIAP “IF” leg) is considered a FB fix.

1. When the termination fix is a common fix followed by an FM or VM leg, the termination fix attribute must be FO and the SIAP fix should be FB.

2. For other cases, common fix attributes must be identical.

3. When the termination is not at a common fix, the STAR must terminate with a VM or FM leg at a FO fix.

Note: When more than one transition/common route terminates at the same fix, more than one of the above can apply simultaneously. In those cases, the waypoint attribute must be the same on each of the common routes/runway transitions.

b. Area. Apply paragraph 1-2-5 for area construction and alignment. Construction and alignment in paragraph 1-2-5 also applies to the intersection between initial routing of the STAR and the ATS route where it begins.

Note: When the STAR terminates at a fix on a SIAP, paragraph 1-2-5 applies between the last leg of a STAR and the first leg of SIAP. When a STAR terminates in a VM or FM leg ensure the fix altitude is at or above the Minimum Vectoring Altitude (MVA)/Minimum IFR Altitudes (MIA).

c. Obstacle clearance. Apply Order 8260.3, paragraph 2-2-1. Alternatively, formula 1-3-2 (of this order) may be used to calculate secondary ROC.

Section 2-3. Feeder Routes

2-3-1. Feeder Route. A feeder route may be developed as an optional transition from the en route environment to the terminal structure (see Order 8260.19, paragraph 4-6-2). Use TF and/or RF legs. The NavSpec is RNAV 1 (feeder flight phase with associated XTT of 1.00). Optional NavSpec may be RNP 1, A-RNP (feeder flight phase with associated XTT of 1.00 or 0.30 as appropriate), or RNP 0.3 for helicopters (feeder flight phase with associated XTT of 0.30). Use an effective XTT of 2.00 where applicable per table 1-2-1 footnote 2. Secondary areas apply, except for A-RNP.

- a. General. Apply paragraph 1-2-5 for course alignment.
- b. Area. Apply paragraph 1-2-5 for area construction.
- c. Obstacle clearance. Apply Order 8260.3, paragraph 2-3-1. Alternatively, formula 1-3-2 (of this order) may be used to calculate secondary ROC.

Section 2-4. Terminal Arrival Area (TAA)

2-4-1. General. A TAA may be established as an optional feeder/arrival transition from the en route structure. It consists of TAA arrival areas developed in conjunction with an approach procedure utilizing RNAV initial segments. In the TAA arrival areas, the NavSpec is RNAV 1 TAA flight phase. Secondary areas are not applied; however, buffers are defined from each area boundary.

2-4-2. Approach Procedure. Design in accordance with the applicable criteria in conjunction with the following guidance:

a. Initial approach segments.

(1) Use TF legs (FB fixes only) in a basic or modified T configuration only (see paragraph 1-3-1.d).

(2) Maximum segment length is 15 NM.

(3) A HILPT initial segment must be established at the IF/IAF at an altitude compatible with the TAA arrival area altitudes (see paragraph 2-4-3.e). The inbound holding course must be aligned with the intermediate segment.

b. Intermediate approach segment. Establish a single dual purpose IF/IAF using TF (FB fixes only) aligned with the Final Approach Segment (FAS).

c. Missed approach segment. Where possible, design missed approach segments to allow a “direct entry” into a missed approach holding pattern (see figure 2-4-1). If the missed approach routing terminates at a base area IAF, align the missed approach holding pattern with the initial inbound course (see figure 2-4-2).

Figure 2-4-1. Default Missed Approach Holding Alignment

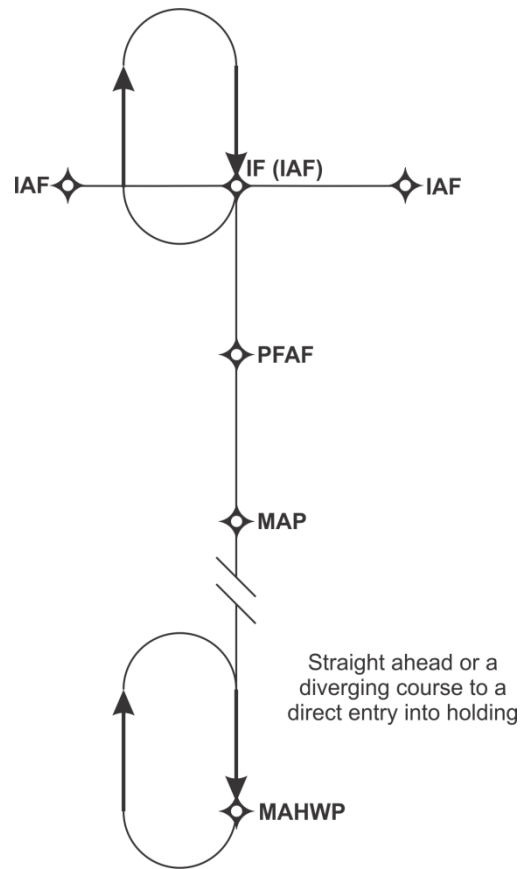
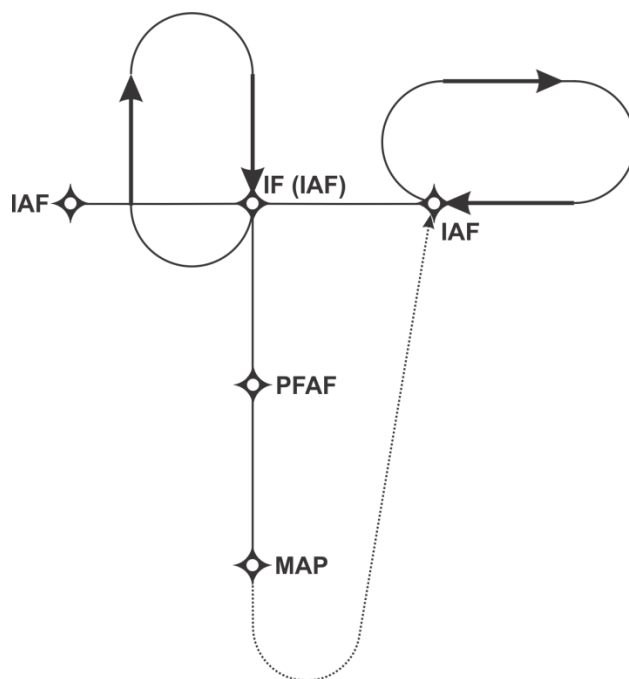
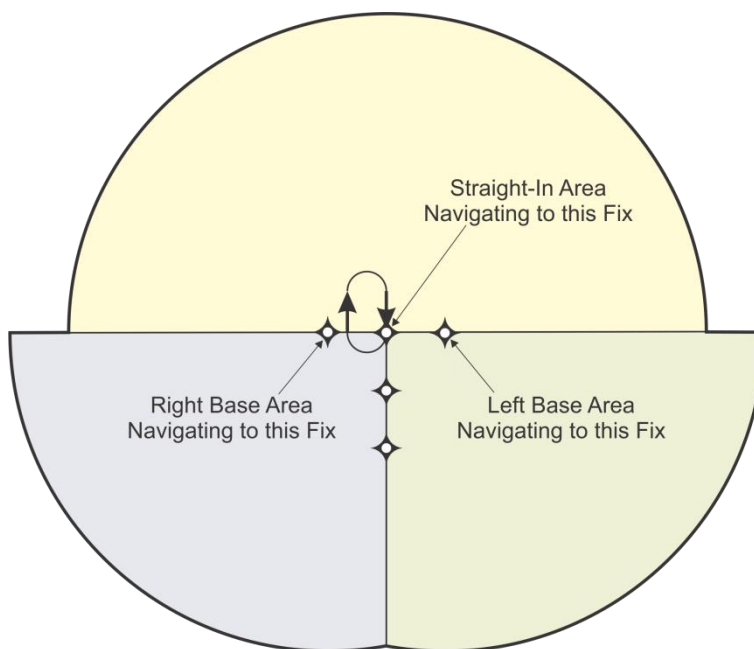


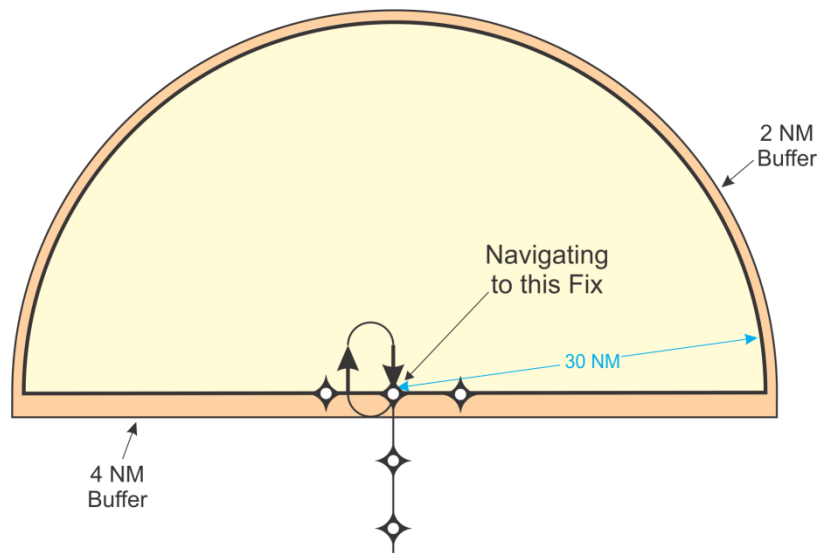
Figure 2-4-2. Missed Approach Holding Alignment at T IAF

2-4-3. Arrival Areas. A default TAA contains a straight-in area, a right base area, and a left base area. Each area is defined relative to a specific IAF (see figure 2-4-3). The boundary between the straight-in area and the base areas is defined by extensions of the T initial segment track centerlines. The boundary between the right and left base areas is an extension of the intermediate segment track centerline.

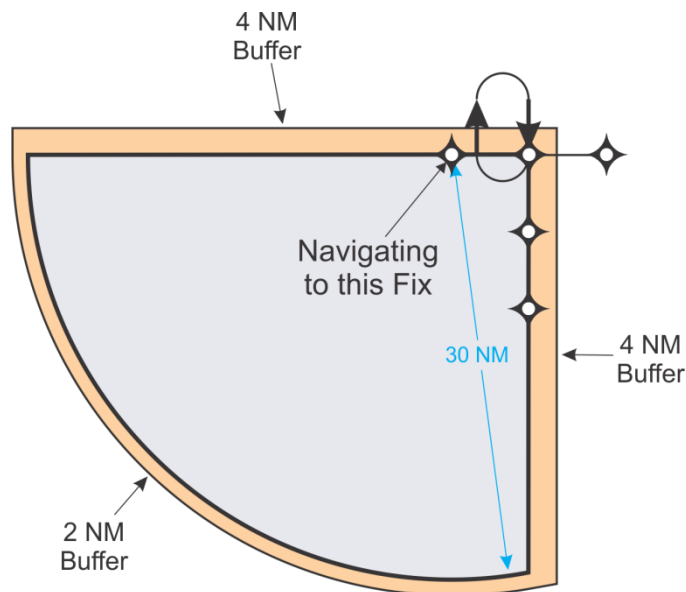
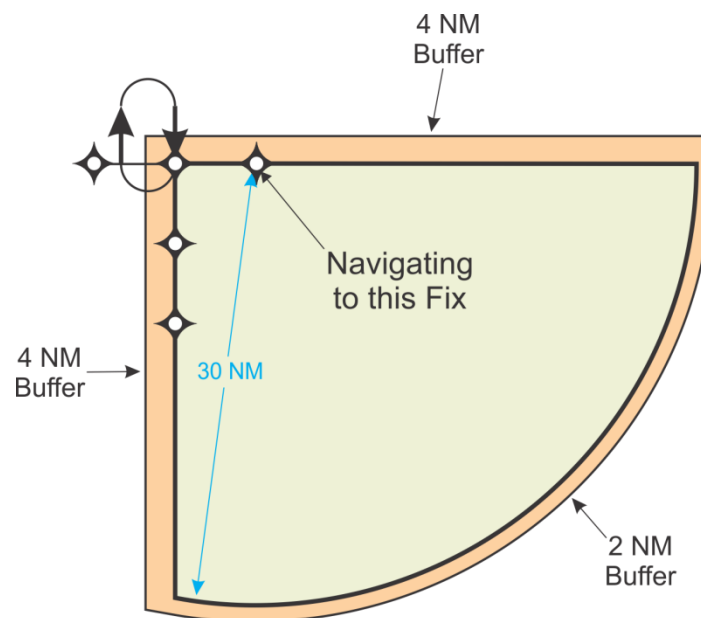
Figure 2-4-3. Default TAA Arrival Areas

a. Straight-in area. Construct the base area side boundary by extending a 30-NM line from the IF/IAF through each T initial segment IAF. Construct the outside boundary by connecting the base area side boundary lines with a 30-NM arc centered on the IF/IAF (see figure 2-4-4). Apply a 4-NM obstacle evaluation buffer on the base area side boundary and a 2-NM buffer on the outside boundary.

Figure 2-4-4. Straight-in Area

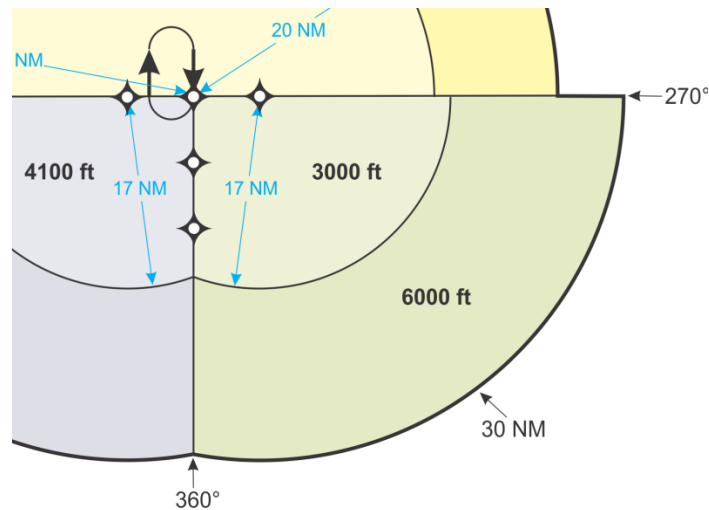


b. Right and left base area. The straight-in side boundaries are the initial segment track centerlines from IAF/IF to each IAF extended outward 30 NM. The boundary between the right and left base areas (inside boundary) is an extension of the intermediate segment track centerline. Construct the outside boundaries as 30-NM arcs centered on each IAF from the straight-in side boundary in the appropriate direction (clockwise or counter-clockwise) to the intersection of the intermediate segment track centerline extended (see figure 2-4-5). Apply a 4-NM obstacle evaluation buffer on the straight-in and inside boundaries and a 2-NM buffer on the outside boundary.

Figure 2-4-5. Right Base Area**Figure 2-4-6. Left Base Area**

c. Area modification. Modifications to the default TAA design may be necessary to achieve an acceptable descent gradient or to meet operational requirements.

(1) **Stepdown arcs.** Areas may be sub-divided using a stepdown arc (see figure 2-4-7). Stepdown arcs are centered on the fix associated with the area and may not be less than 4 NM from the center fix or within 4 NM of the outer boundary. Only one stepdown arc may be established per area/sector. A 2-NM obstacle evaluation buffer applies from the stepdown arc. The altitude at the stepdown arc must not result in an excessive descent gradient in the associated NoPT initial calculated as length plus stepdown arc radius.

Figure 2-4-7. Area Modification; Stepdown Arcs

(2) Radial sectorization. The straight in or extended straight-in area (see paragraph 2-4-3.c) may be divided into up to three sectors defined radially by inbound courses to the IF/IAF (see figure 2-4-8). Radial sectors can be used alone or in conjunction with stepdown arcs (see figure 2-4-9). The angular size of a sector may not be less than 30 degrees, except a sector with a side defined by a base area boundary that also includes a stepdown arc may not be less than 45 degrees. A 4-NM obstacle evaluation buffer applies to radial sector boundaries.

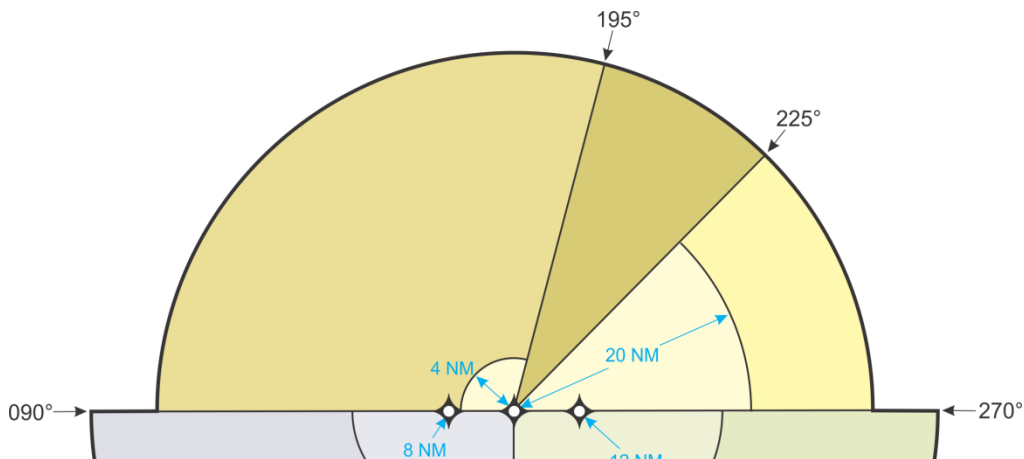
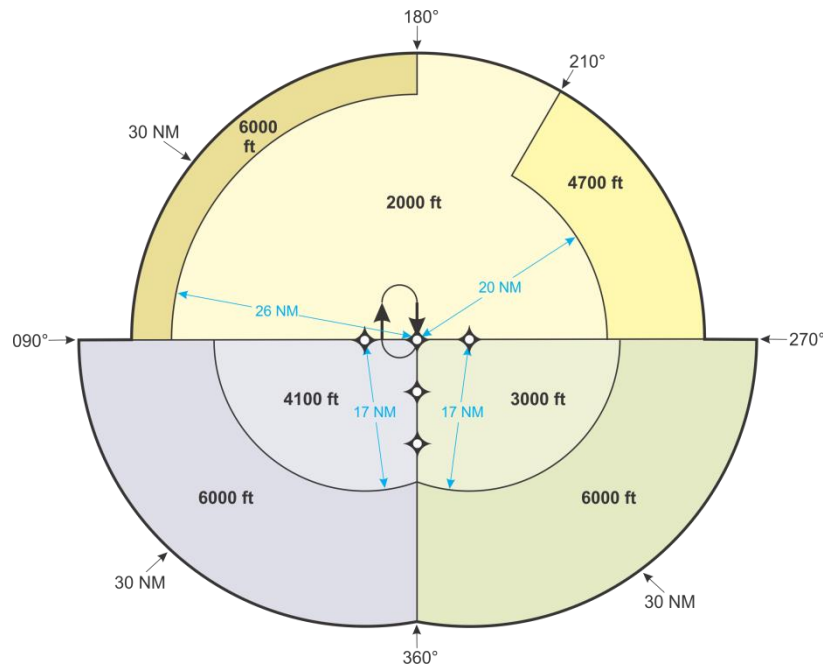
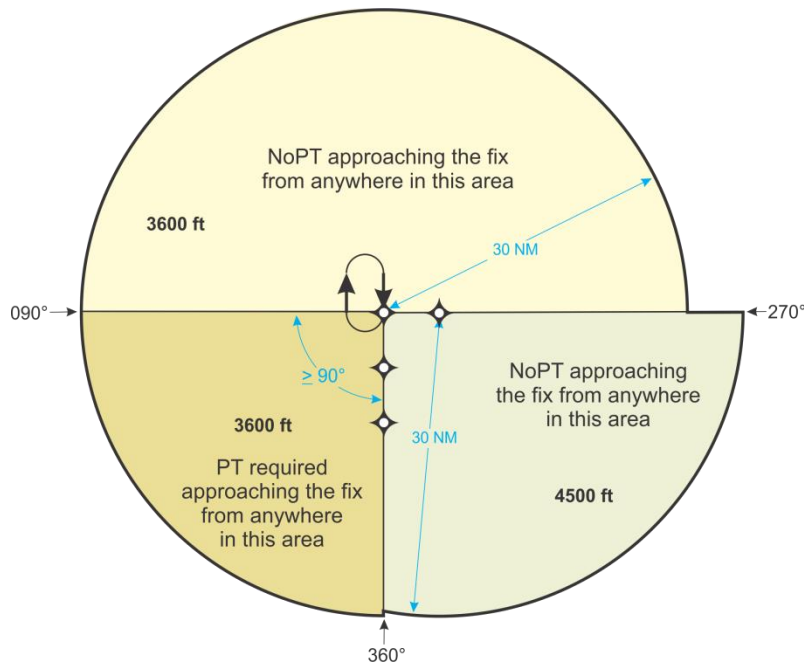
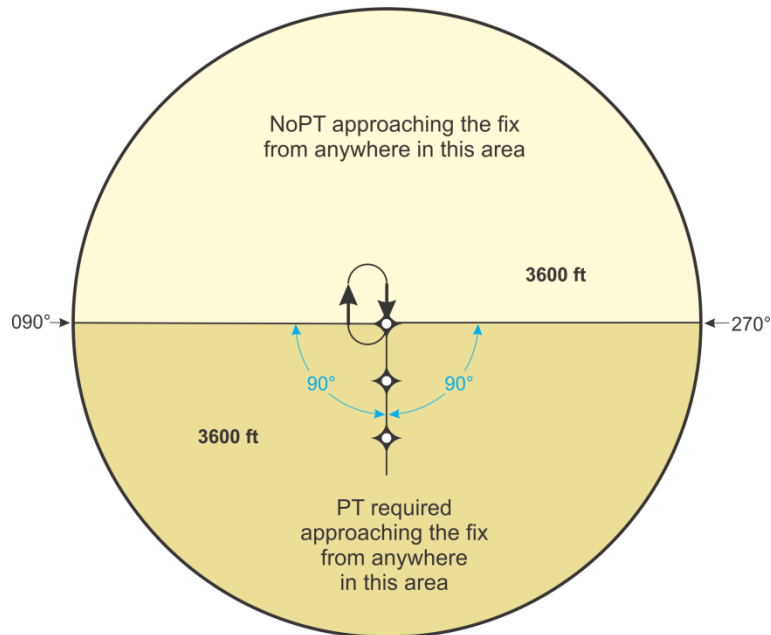
Figure 2-4-8. Area Modification; Straight-in Area Radial Sectorization

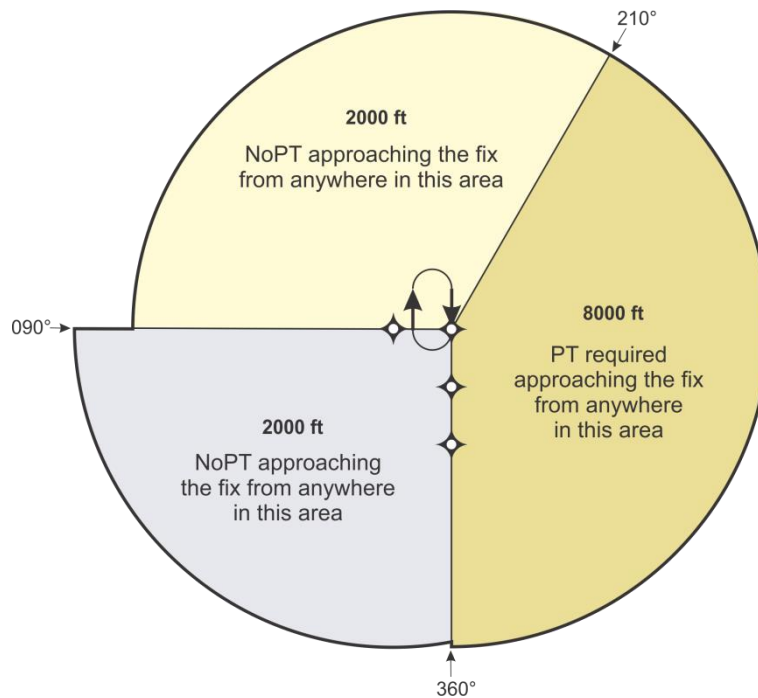
Figure 2-4-9. Area Modification; Radial Sectorization and Stepdown Arcs

(3) Base area elimination. One or both of the basic T initial approach segments and its associated base area may be eliminated (see figure 2-4-10 and figure 2-4-11).

(a) Construction. The straight-in area construction is extended into the area of the eliminated base area(s). Paragraph 2-4-3.b construction applies to a remaining base area. Paragraph 2-4-3.a applies except the outside boundary is a 30-NM arc connecting the base area side boundary line to the intermediate segment track centerline extended. When both base areas are eliminated, the straight-in area is a 30-NM circle centered on the IF/IAF.

Figure 2-4-10. Area Modification; Right Base Area Elimination**Figure 2-4-11. Area Modification; Both Right and Left Base Area Elimination**

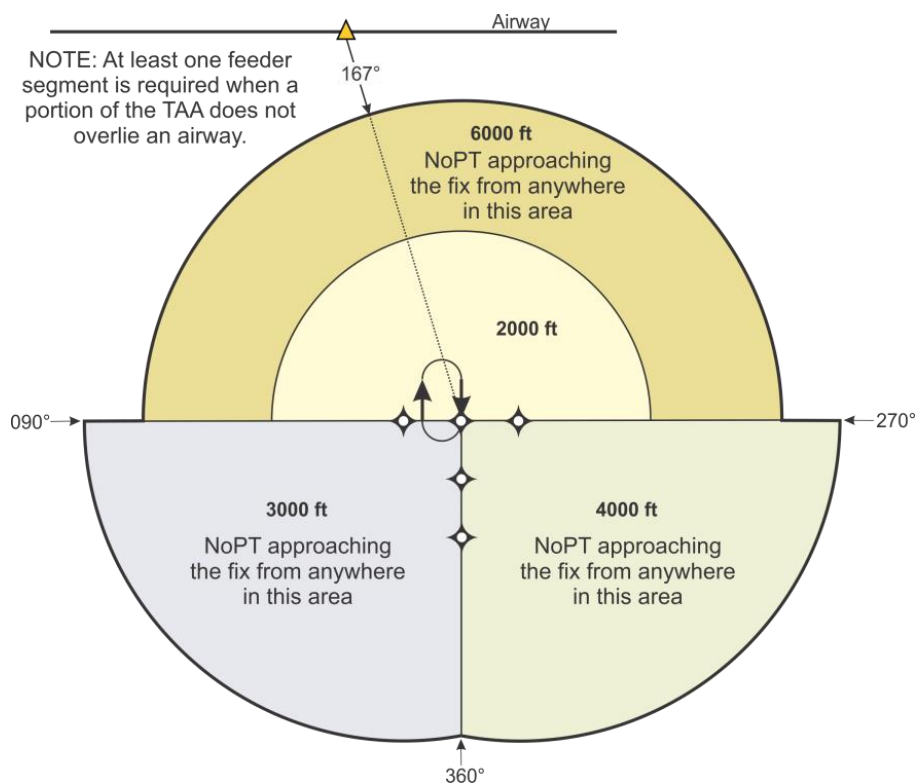
(b) No procedure turn (NoPT) exclusion. The sector associated with an eliminated base area must be excluded from a NoPT designation. A NoPT exclusion area may also be extended from an eliminated base area into the straight-in area when required to meet descent gradient standards.

Figure 2-4-12. Area Modification; NoPT Exclusion

d. Obstacle clearance (see Order 8260.3, paragraph 2-3-1.c).

e. Altitude selection. A minimum altitude is specified for each area and sub-sector. Where possible establish a common IAF altitude for the right/left base and straight-in areas. Where not possible, area altitudes must permit establishment of a common altitude at the IF/IAF. Except for areas/sub-sectors excluded from a NoPT designation, the associated IAF altitude may not be lower than the applicable area/sub-sector altitude. Order 8260.3, paragraph 2-4-5.e applies to area/sub-sectors excluded from a NoPT designation.

2-4-4. Feeder Route. Normally one of the TAA areas (excluding obstacle evaluation buffers) will overlie an airway. Where this is not the case, at least one feeder route must be established from the en route structure to the TAA boundary. Order 8260.3, paragraph 2-3-1 applies except alignment must be a direct course to the IF/IAF or right/left base IAF depending on the location of the feeder fix (see figure 2-4-13).

Figure 2-4-13. TAA with Feeder Route

Chapter 3. RNAV (GPS) Approach

Section 3-1. General Criteria

3-1-1. Applicability. This chapter applies to procedures covered by the RNP APCH NavSpec (LPV, LP, LNAV/VNAV, and LNAV). It also applies to PBN legs that transition to an ILS or GLS final. Incorporation of an RF leg limits IFP availability to some users; therefore, an RNAV (GPS) approach that requires use of an RF leg in the missed approach, intermediate, or in all initials requires Flight Standards approval. See Order 8260.19, chapter 8 paragraph “Minimums” for lines of RNAV minimums.

3-1-2. Feeder Segment. Construct feeder segments as described in chapter 2 using one or more TF or RF legs. The NavSpec is RNAV 1 (feeder flight phase with associated XTT of 1.00). Optional NavSpec may be A-RNP (feeder flight phase with associated XTT of 1.00 or 0.30 as appropriate) or RNP 0.3 for helicopters (feeder flight phase with associated XTT of 0.30). Use an effective XTT of 2.00 where applicable per table 1-2-1 footnote 2. Secondary areas apply, except for A-RNP. If the feeder segment NavSpec is A-RNP, that NavSpec must be continued throughout the transition until reaching the final segment.

3-1-3. Initial. Construct initial segments as described in chapter 1 using one or more TF or RF legs. The NavSpec is RNP APCH (initial flight phase with associated XTT of 1.00). Optional NavSpec may be A-RNP (initial flight phase with associated XTT of 0.30) or RNP 0.3 for helicopters (initial flight phase with associated XTT of 0.30). Use an effective XTT of 2.00 where applicable per table 1-2-1 footnote 2. Secondary areas apply, except for A-RNP. If the initial segment NavSpec is A-RNP, that NavSpec must be continued throughout the transition until reaching the final segment.

3-1-4. Intermediate. Construct intermediate segments as described in chapter 1 using one or more TF or RF legs. The NavSpec is RNP APCH (intermediate flight phase with associated XTT of 1.00). Optional NavSpec may be A-RNP (intermediate flight phase with associated XTT of 0.30) or RNP 0.3 for helicopters (intermediate flight phase with associated XTT of 0.30). Secondary areas apply, except for A-RNP. If the intermediate segment NavSpec is A-RNP, that NavSpec must be continued until reaching the final segment. Paragraph 1-2-5.b(1)(d) applies except the ATT at the PFAF is based on the applicable final approach navigation accuracy from table 1-2-1.

a. RF leg. Except when joining an ILS/GLS/LPV final (see appendix D), an RF leg must end at least 2 NM prior to the PFAF.

b. Alignment (maximum course change at the PFAF). Offset alignment is only authorized when the PFAF is a FB fix.

(1) LNAV and LP. Align the intermediate course within 30 degrees of the final approach course.

(2) LNAV/VNAV. Align the intermediate course within 15 degrees of the final approach course.

c. ILS/GLS/LPV. Apply appendix D for PBN transitions to ILS/GLS/LPV.

d. Area (see figure 3-1-1 and figure 3-1-2).

(1) LNAV and LNAV/VNAV. The intermediate segment primary area connects uniformly from ± 2 XTT at a point 2 NM prior to the PFAF to the primary boundary 1 NM past the PFAF. The secondary boundary connects uniformly from 1 XTT at a point 2 NM prior to the PFAF to the secondary boundary 1 NM past the PFAF.

(2) LP and LPV/GLS. The intermediate segment primary area connects uniformly from ± 2 XTT at a point 2 NM prior to the PFAF to the primary boundary/X OCS outer boundary abeam the PFAF. The secondary boundary connects uniformly from 1 XTT at a point 2 NM prior to the PFAF to the secondary/Y OCS outer boundary abeam the PFAF. When an RF to an LPV/GLS PFAF is used, comply with appendix D and connect relative to the RF track.

Figure 3-1-1. LNAV, LNAV/VNAV Intermediate

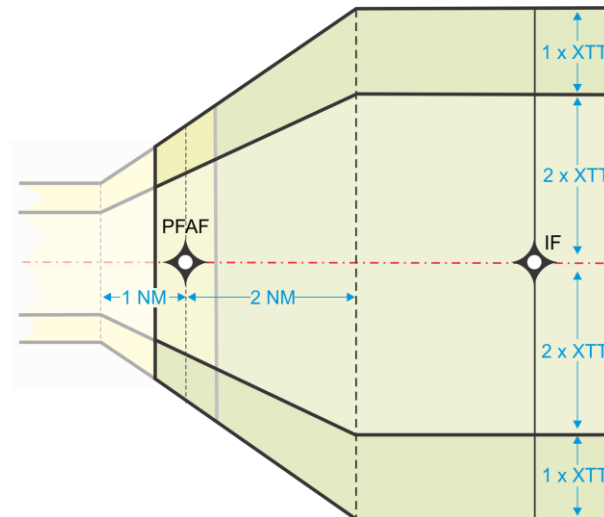
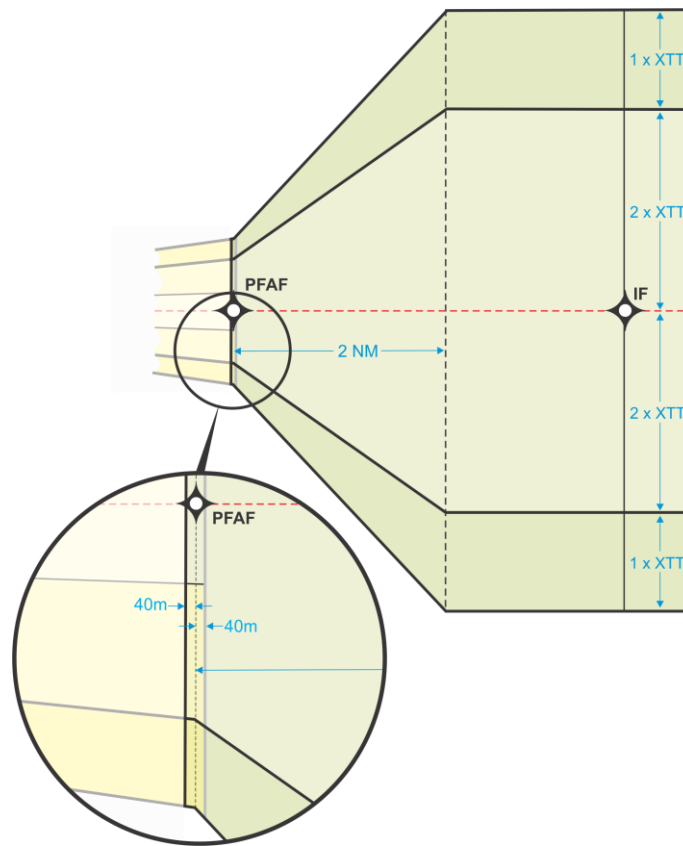


Figure 3-1-2. LP, LPV/GLS Intermediate

(3) Turn at IF or start of aligned leg prior to PFAF. When the standard FB turn construction results in boundaries outside the normal primary/secondary widths at the point 2 NM prior to the PFAF, the inside (turn side) boundaries may be constructed as straight lines from the tangent points on the turning construction arcs to the applicable final segment boundaries (see figure 3-1-3 and figure 3-1-4). Redesign if straight line connections result in an area less than normal primary/secondary area width in the leg prior to PFAF.

Figure 3-1-3. Turn at IF, Example LNAV, LNAV/VNAV

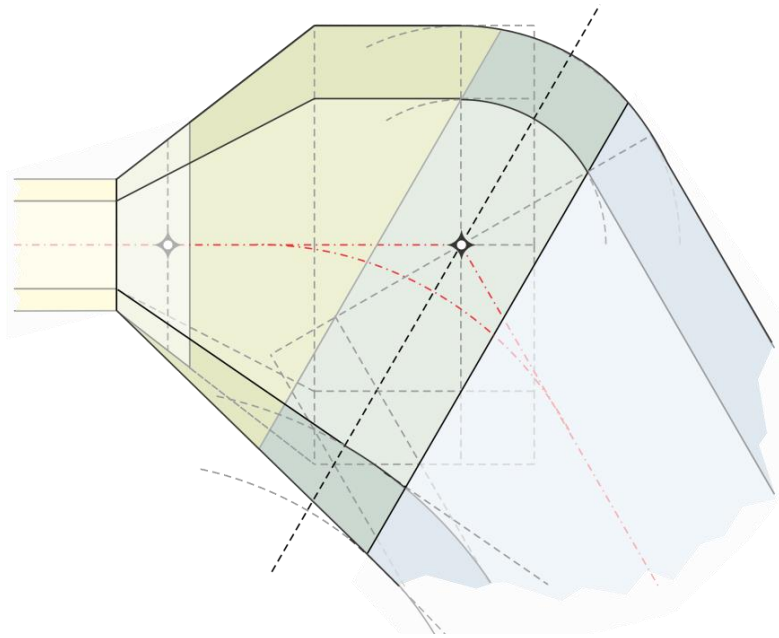
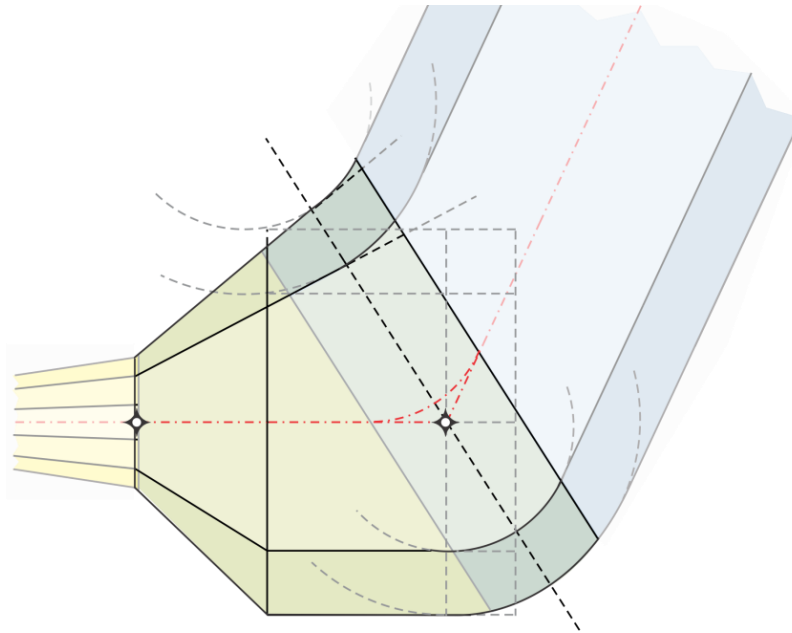


Figure 3-1-4. Turn at IF, Example LP, LPV/GLS



(4) Offset construction. Where the intermediate course is not an extension of the final course, use the following construction techniques.

(a) *Step 1.* Construct line A perpendicular to the intermediate course 2 NM prior to the PFAF.

(b) *Step 2.* Construct line B perpendicular to the intermediate course extended 1 NM past the PFAF.

(c) *Step 3.* Construct the inside turn boundaries by connecting the points of intersection of line A with the turn side intermediate segment boundaries with the intersection of line B with the turn side final segment boundaries.

(d) *Step 4.* This step is dependent on procedure.

1. LNAV and LNAV/VNAV (see figure 3-1-5). Construct arcs centered on the PFAF of 1 NM and 1.3-NM radius on the non-turn side of the fix.

2. LP and LPV/GLS (see figure 3-1-6). Connect lines from the point of intersection of line A and the outside primary and secondary intermediate segment boundaries to the final segment primary and secondary final segment lines at a point perpendicular to the final course at the PFAF.

(e) *Step 5* (applicable to LNAV and LNAV/VNAV Only). Connect lines from the point of intersection of line A and the outside primary and secondary intermediate segment boundaries to tangent points on the arcs constructed by *Step 4*.

(f) *Step 6* (applicable to LNAV and LNAV/VNAV Only). Connect lines tangent to the arcs created in *Step 4* that taper inward at 30 degrees relative to the Final Approach Course (FAC) to intersect the primary and secondary final segment boundaries as appropriate.

Figure 3-1-5. Offset Intermediate, Example LNAV

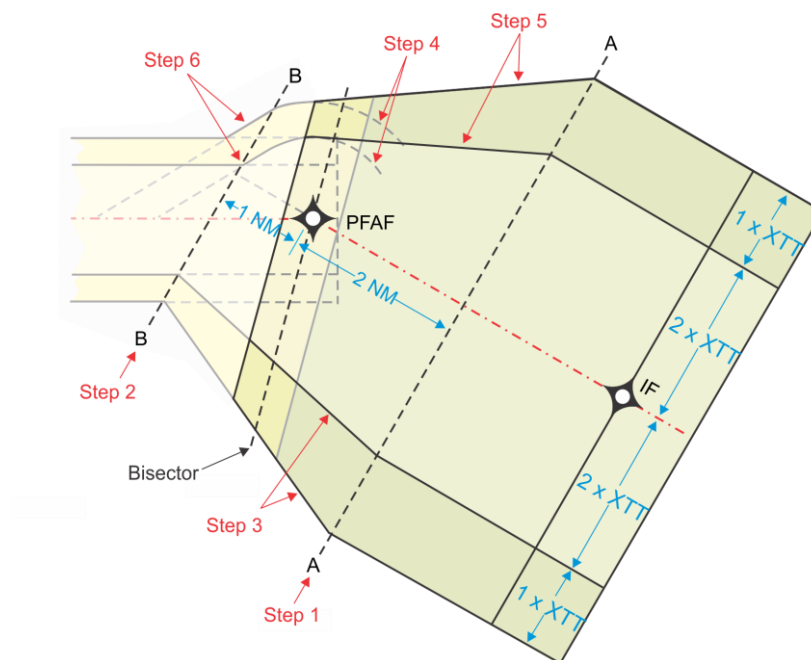
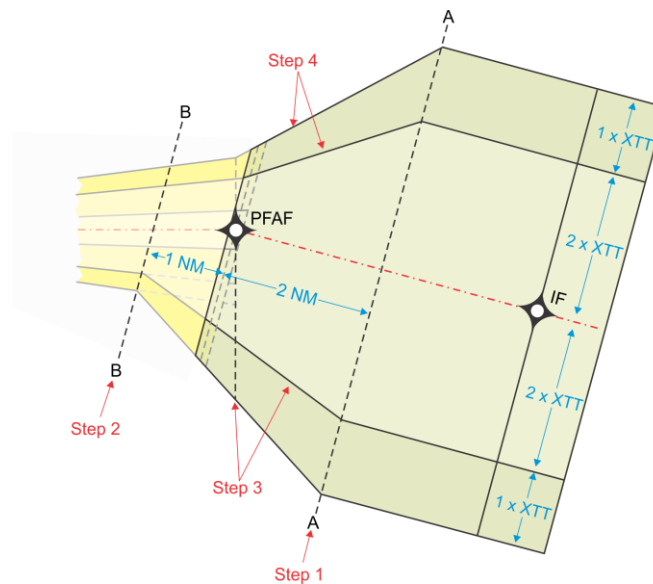


Figure 3-1-6. Offset Intermediate, Example LPV/GLS



3-1-5. Final. Construct final segments as described in section 3-2 using one or more TF legs. RF legs and TF turns are not authorized. The NavSpec is RNP APCH final flight phase (with associated XTT of 0.30 NM for part A [LNAV and LNAV/VNAV] and 40 meters for part B [LP and LPV/GLS]). Secondary areas apply.

- a.** General. Paragraph 1-3-1.f applies except see paragraph 3-4-1 for LPV/GLS exceptions.
- b.** The final segment OEA primary and secondary boundaries follow the intermediate segment OEA primary and secondary boundaries from ATT prior to the PFAF to the applicable taper connection point and includes offset intermediate construction (see figure 3-1-1, figure 3-1-2, figure 3-1-3, figure 3-1-5, and figure 3-1-6).
- c.** FAS data block (see figure 3-1-7). The FAS data block is documentation requirement used to define the lateral and vertical paths unique to SIAPs with LP, LPV, or GLS minimums. FAS data includes the following fields:

(1) Flight Path Alignment Point (FPAP). The FPAP is defined by geodetic coordinates. When the FAC is aligned with the Runway Centerline (RCL), find the FPAP by extending a geodesic line from the LTP toward the DER a distance equal to the greater of the runway length or 9023 feet. For offset procedures, the FPAP is located on the extension of the desired final approach course at a distance from the FTP that provides the appropriate lateral course width. The FPAP is located 9023 feet from the heliport/fictitious heliport (FHP) on the FAC for helicopter procedures.

(2) **Length offset.** The length offset value is the distance between the departure end of runway and the FPAP, rounded to the nearest eight-meter value. It defines the location where lateral sensitivity changes to missed approach sensitivity. The length offset is zero (0) for offset procedures, procedures where the FPAP is located at the departure end of the runway, and

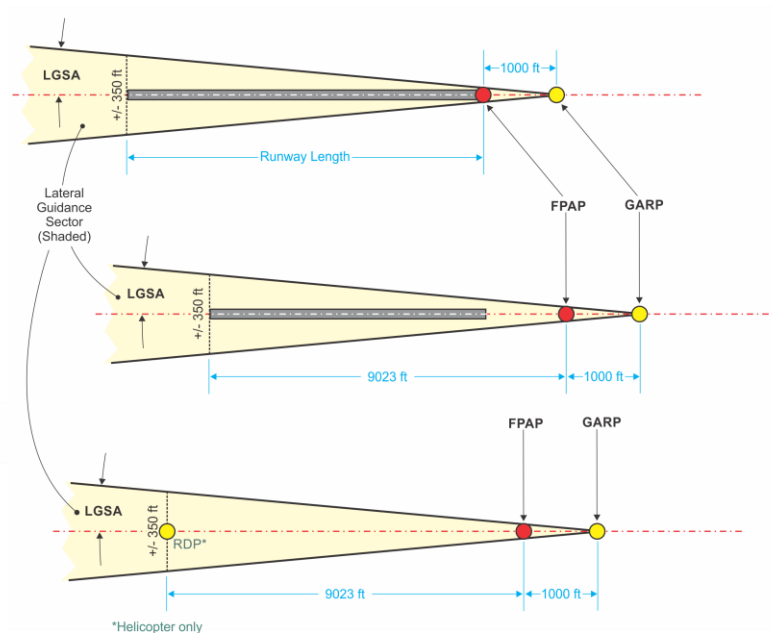
helicopter procedures to heliports that support IFR procedures or for helicopter Point in Space IFR approaches.

(3) Course width at threshold. The course width at threshold is defined by the global navigation satellite (GNSS) azimuth reference point (GARP) and the lateral guidance sector angle (LGSA) using formula 3-1-1. Convert the result to meters and round to the nearest 0.25-meter increment.

(a) GARP. The GARP is a point located 1000 feet beyond the FPAP on the extension of the FAC/RCL geodesic line. This point is used by the airborne system as the origin of the lateral guidance sector.

(b) LGSA. The LGSA originates at the GARP and is the angular dimension of lateral guidance relative to the FAC. The LGSA angle ranges from not more than ± 2 degrees to not less than ± 1.5 degrees.

Figure 3-1-7. FAS Data Block



Formula 3-1-1. Course Width at Threshold

$$\text{Course}_{\text{width}} = \text{greater of 350 feet or } \tan(1.5) \times d_{\text{GARP}}$$

Where:

d_{GARP} = distance (feet) from the LTP/FTP to the GARP

Example:

$$\text{Course}_{\text{width}} = \tan(1.5) \times 13560.66$$

$$\text{Course}_{\text{width}} \approx 355.10 \text{ feet}$$

$$\text{Course}_{\text{width}} \approx 108.23 \text{ meters}$$

$$\text{Course}_{\text{width}} = \text{Rounded to 108.25}$$

Section 3-2. General Non-Vertically Guided Final Segment

3-2-1. General. This section contains obstacle evaluation criteria for LNAV and LP non-vertically guided approach procedures. LP is not applicable if straight-in minimums are not authorized (see paragraph 3-2-2.b and Order 8260.3 paragraph 2-6-2). Do not develop a new LP where an LNAV/VNAV or LPV can be developed for at least one approach CAT, regardless of minima.

3-2-2. Alignment. Optimum non-vertically guided procedure final segment alignment is with the runway centerline extended through the LTP. When published in conjunction with a vertically-guided procedure, alignment must be identical with the vertically guided final segment. TF-TF turns are not allowed in the FAS. The latest point the MAP can be located is LTP/FTP. When an LNAV procedure is combined with an LNAV/VNAV, the MAP must be located at LTP/FTP. Helicopter procedures to heliports must be aligned per Order 8260.3 paragraph 12-2-4.a and are considered straight-in.

a. Straight-in with offset alignment. When the final course must be offset, it may be offset up to 30 degrees (published separately from vertically guided) when the following conditions are met:

- (1) Offset ≤ 5 degrees. Align the course through LTP.
- (2) Offset > 5 degrees and ≤ 10 degrees. The course must cross the runway centerline extended at least 1500 feet prior to LTP (5200 feet maximum).
- (3) Offset > 10 degrees and ≤ 20 degrees. The course must cross the runway centerline extended at least 3000 feet prior to LTP (5200 feet maximum). For offsets > 15 degrees, CAT C/D minimum published visibility 1 SM, minimum height above touchdown (HAT) of 300 feet.
- (4) Offset > 20 to ≤ 30 degrees (CAT A/B only). The course must cross the runway centerline extended at least 4500 feet prior to the LTP (5200 feet maximum).

Note: Where offset alignment as specified above cannot be attained and the final course does not intersect the runway centerline or intersects the centerline more than 5200 feet from LTP, and an operational advantage can be achieved, the final may be aligned to lie laterally within 500 feet of the extended runway centerline at a point 3000 feet outward from LTP. For the purposes of OEA construction, use the Final End Point (FEP) as the FTP (see Order 8260.3, paragraph 2-6-2). This option requires Flight Standards approval.

b. Circling. The optimum final course alignment is to the center of the landing area, but may be to any portion of the usable landing surface. The latest point the MAP can be located is the FEP. For the purposes of OEA construction, use the FEP as the FTP.

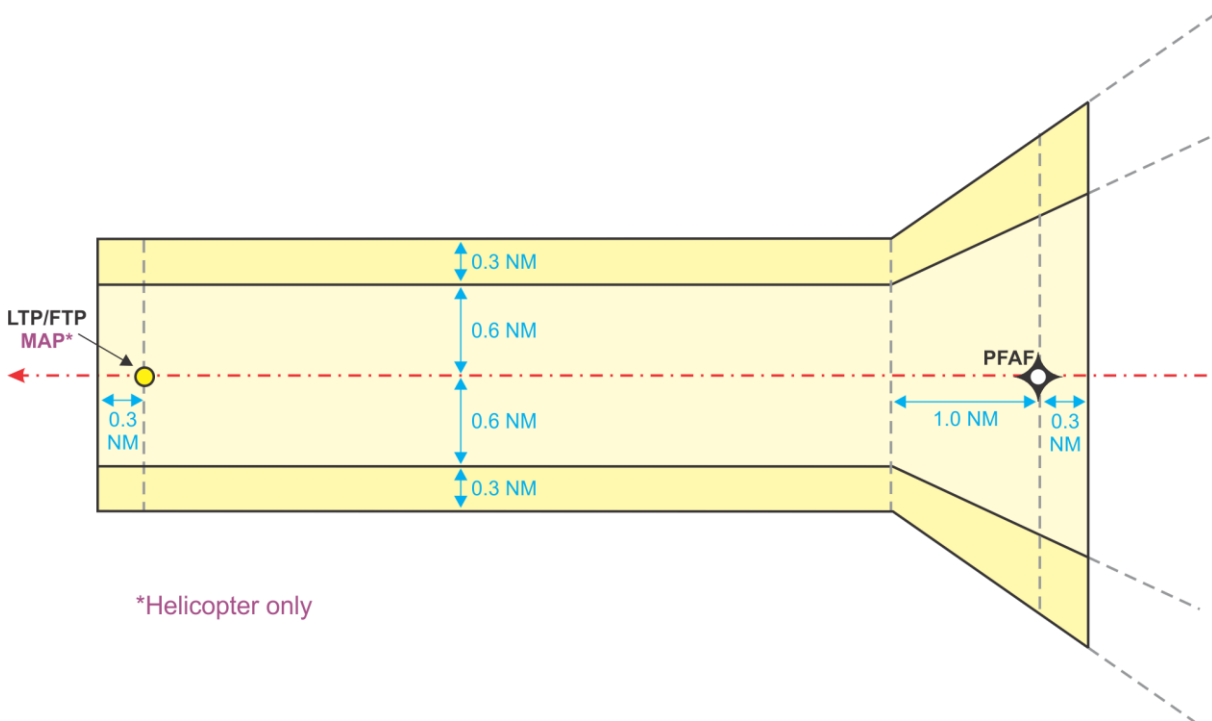
3-2-3. Area.

a. LNAV.

(1) Length. The OEA begins 0.3 NM prior to the PFAF and ends 0.3 NM past the LTP/FTP/(MAP for helicopters). Segment length is the distance from the PFAF location to the LTP/FTP. The maximum length is 10 NM.

(2) Width. The final segment OEA primary and secondary boundaries are coincident with the tapering intermediate segment boundaries from a point 0.3 NM prior to the PFAF to a point 1.0 NM past the PFAF. From this point to 0.3 NM past the LTP/FTP/(MAP for helicopters), the primary OEA boundary is ± 0.6 NM from course centerline. A 0.3-NM secondary area is located on each side of the primary area. Determine the primary and secondary area widths in this tapering area using formula 3-2-1. Helicopter RNP 0.3 segments do not taper and are a continuation of the primary and secondary areas from 0.30 NM prior to the PFAF to 0.30 NM past the MAP.

Figure 3-2-1. LNAV Final Segment OEA



Formula 3-2-1. Tapering Area Width

$$\frac{1}{2} W_p = \frac{1.4 \times D_{\text{taper}}}{3} + 0.6$$

$$W_s = \frac{0.7 \times D_{\text{taper}}}{3} + 0.3$$

Where:

D_{taper} = along track distance (NM) from end of taper in the final segment to the point of interest

Example:

$$\frac{1}{2} W_p = \frac{1.4 \times 1.3}{3} + 0.6$$

$$\frac{1}{2} W_p \approx 1.21 \text{ NM}$$

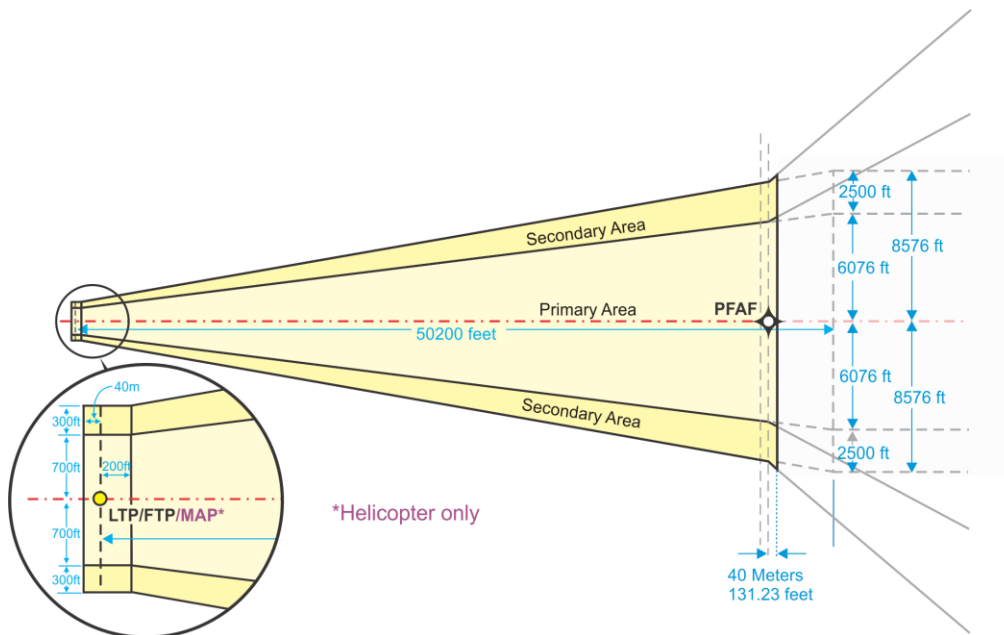
$$W_s = \frac{0.7 \times 1.3}{3} + 0.3$$

$$W_s \approx 0.60 \text{ NM}$$

Note: This formula may only be used when the intermediate and final legs are aligned.

b. LP.

Figure 3-2-2. LP Final OEA



(1) Length. The OEA begins 40 meters prior to the PFAF and ends 40 meters past the LTP/FTP (helipoint/FHP for helicopter procedures). Segment length is the distance from the PFAF location to the LTP/FTP (helipoint/FHP for helicopter procedures). The helipoint/FHP is a point on the FAC 2600 feet after the LP MAP for helicopter procedures. The maximum length is 10 NM.

(2) Width.

(a) Primary. The perpendicular distance from the course centerline to the outer boundary of the primary area is a constant 700 feet from a point 40 meters past (inside) the LTP/FTP (helipoint/FHP for helicopter procedures) to a point 200 feet prior to (outside) the LTP/FTP (helipoint/FHP for helicopter procedures). It expands from this point in a direction toward the PFAF. Calculate primary half width from the 200-foot point to a point 50200 feet from LTP (helipoint/FHP for helicopter procedures) using formula 3-2-2. The value of primary half width beyond the 50200-foot point is 6076 feet.

Formula 3-2-2. Primary Area Half Width

$$\frac{1}{2} w_p = 0.10752 \times d_{LTP} + 678.496$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

Note: $200 \text{ feet} \leq d_{LTP} \leq 50200 \text{ feet}$

Example:

$$\frac{1}{2} w_p = 0.10752 \times 30308.58 + 678.496$$

$$\frac{1}{2} w_p \approx 3937.27 \text{ feet}$$

(b) Secondary. The secondary width is a constant 300 feet from a point 40 meters past (inside) the LTP/FTP (helipoint/FHP for helicopter procedures) to a point 200 feet prior to (outside) the LTP/FTP (helipoint/FHP for helicopter procedures). It expands from this point in a direction toward the PFAF. Calculate secondary width from the 200-foot point to a point 50200 feet from LTP (helipoint for helicopter procedures) using formula 3-2-3. The value of secondary width beyond the 50200-foot point is 2500 feet.

Formula 3-2-3. Secondary Area Width

$$w_s = 0.044 \times d_{LTP} + 291.2$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

Note: $200 \text{ feet} \leq d_{LTP} \leq 50200 \text{ feet}$

Example:

$$w_s = 0.044 \times 30308.58 + 291.2$$

$$w_s \approx 1624.78 \text{ feet}$$

3-2-4. Obstacle Clearance. The minimum ROC in the primary area is 250 feet. The minimum ROC in the secondary area is 250 feet at the primary boundary, tapering uniformly to zero feet at the outer edge. Order 8260.3, chapter 3 precipitous terrain, remote altimeter, and excessive length of final adjustments apply. Calculate secondary ROC value in accordance with paragraph 1-3-1.b.

3-2-5. Minimum Descent Altitude. The MDA value is the sum of the controlling obstacle MSL plus the applicable ROC application from paragraph 3-2-4. Round the sum to the next higher 20-foot increment; e.g., 623 rounds to 640. The minimum HAT value is 250 feet.

Section 3-3. Lateral Navigation with Vertical Guidance (LNAV/VNAV) Final Segment

3-3-1. General. An LNAV/VNAV approach is a vertically-guided approach procedure using baro-VNAV or WAAS VNAV for the vertical guidance. Obstacle evaluation is based on the LNAV OEA dimensions, level surface OCS based on ROC, and a sloping baro-VNAV OCS. ROC adjustments for excessive length of final do not apply.

a. Vertical path restrictions. The true vertical path provided by baro-VNAV is influenced by temperature variations; i.e., during periods of cold temperature, the effective glidepath may be lower than published and during periods of hot weather, the effective glidepath may be higher than published (see paragraph 3-3-4.a(1) and Order 8260.19, paragraph 4-6-8).

b. Other restrictions. LNAV/VNAV approach procedures at airports where Remote Altimeter Setting Source (RASS) is in use, helicopter point-in-space (PinS) approaches, or where the final segment overlies precipitous terrain, must be annotated to indicate the approach is not authorized for baro-VNAV systems.

3-3-2. Alignment. Optimum final segment alignment is with the runway centerline extended through the LTP. TF-TF turns are not allowed in the FAS. SIAPs designed to support Simultaneous Offset Instrument Approaches (SOIA) operations are exempted from standard final segment alignment when compliant with Order 8260.3, paragraph 15-3-5. Where lowest minimums can only be achieved by offsetting the final course, it may be offset up to 15 degrees when the following conditions are met:

a. Offset ≤ 5 degrees. Align the course through LTP.

b. Offset > 5 degrees and ≤ 10 degrees. The course must cross the runway centerline extended at least 1500 feet (5200 feet maximum) prior to LTP.

c. Offset > 10 degrees and ≤ 15 degrees. The course must cross the runway centerline extended at least 3000 feet (5200 feet maximum) prior to LTP.

Note: For paragraphs 3-3-2.b and 3-3-2.c, determine the effective DA distance using formula 3-3-1. This distance can be used in formula 1-3-4 to determine the required altitude.

Formula 3-3-1. Offset Alignment Minimum Effective DA Distance

$$d_{DA\text{effective}} = \frac{V_{\text{KIAS}}^2 \times \tan(a_{\text{offset}} \times 0.5) \times 1852}{\tan(18) \times 68625.4 \times 0.3048} + d_{\text{LTP}}$$

Where:

V_{KIAS} = indicated final approach airspeed (knots) for applicable category

a_{offset} = degrees of offset between the final course and runway centerline extended

d_{LTP} = distance (feet) from LTP to the final course and runway centerline extended intersection

Example:

$$d_{DA\text{effective}} = \frac{165^2 \times \tan(12.5 \times 0.5) \times 1852}{\tan(18) \times 68625.4 \times 0.3048} + 4300.67$$

$$d_{DA\text{effective}} \approx 5113.16 \text{ feet}$$

3-3-3. Area. Apply paragraph 3-2-3.a.

3-3-4. Obstacle Clearance Surface (OCS). In the primary area, the elevation of the OCS at any point is the elevation of the OCS at the course centerline abeam it. The OCS in the secondary area is a 7:1 surface sloping upward from the edge of the primary area OCS perpendicular to the flight track (see formula 3-3-2 and figure 3-3-1). The controlling obstacle is the obstacle that having penetrated the sloping OCS requires the highest GPA or most adverse DA or an obstacle that penetrates the level OCS that causes the most adverse DA. See latest version of Order 8260.3, section 10-6 for obstacle assessment.

Formula 3-3-2. Secondary Area Adjusted Obstacle Elevation

$$O_{\text{adjusted}} = O_{\text{MSL}} - \frac{OBS_Y - \frac{1}{2} w_p}{7}$$

Where:

O_{MSL} = obstacle MSL elevation

OBS_Y = perpendicular distance (feet) from the final approach centerline to the obstacle

$\frac{1}{2} w_p$ = perpendicular distance (feet) from final approach centerline to primary area boundary

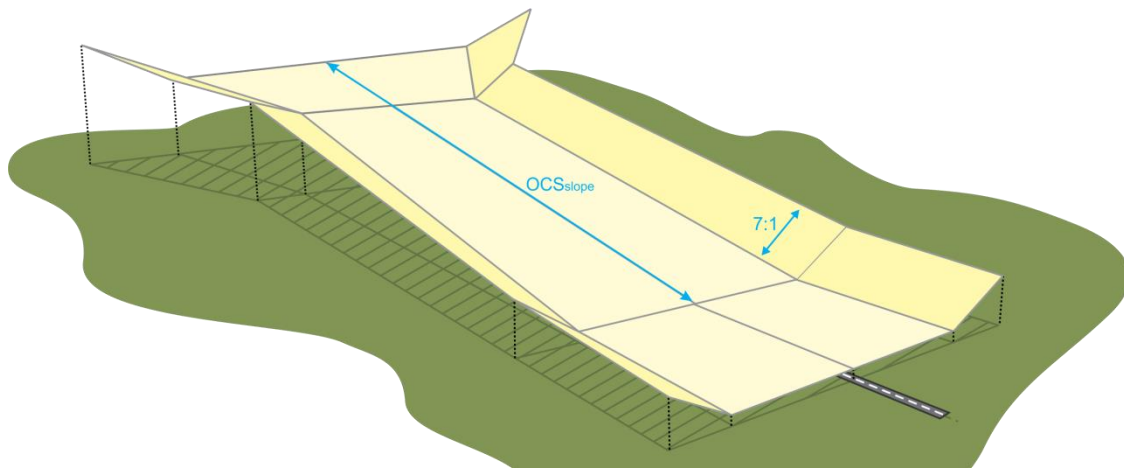
Example:

$$O_{\text{adjusted}} = O_{\text{MSL}} - \frac{OBS_Y - \frac{1}{2} w_p}{7}$$

$$O_{\text{adjusted}} = 3841.6 - \frac{4253.28 - 3645.67}{7}$$

$$O_{\text{adjusted}} \approx 3754.80 \text{ feet}$$

a. Sloping OCS (see Order 8260.3, paragraph 2-1-4). The primary area OCS slope varies based on critical temperatures and designed glidepath angle.

Figure 3-3-1. LNAV/VNAV Final Segment OCS

(1) Critical temperatures.

(a) Average cold temperature (ACT). In order to determine the critical low temperature, the ACT must first be established.

1. ACT based on historical temperature data.

a Source. The National Oceanic and Atmospheric Administration (NOAA), National Climatic Data Center (NCDC) is the official government source for historical temperature data.

b Reporting period. Reporting periods are established in calendar years (January through December) and must have a complete temperature record for the airport for the entire period. Ideally use the five full year period prior to the current year. If temperature data is incomplete, use the longest continuous period with complete data starting not more than six years prior to the current year. The minimum reporting period is not less than three full calendar years.

Example: The current date is in calendar year (CY) 2015. The ideal reporting period is January 1, 2010 to December 31, 2014; however, temperature data is only complete up to November 2014. Since the earliest allowable year for ACT determination is 2009, the specialist chooses a reporting period from January 1, 2009 to December 31, 2013.

c Calculate the ACT as follows:

1 Find the coldest month. For each year in the reporting period, find the month with the lowest average temperature. When two or more months have the same average temperature, choose the month with the lowest single day temperature.

Example: Jan 2009: 45.4°F; Dec 2010: 44.7°F; Dec 2011: 43.4°F; Feb 2012: 42.2°F; Jan 2013: 45.0°F.

2 Find the lowest temperature in the coldest month. Find the lowest reported temperature in each of the coldest months.

Example: Jan 2009: 37°F; Dec 2010: 35°F; Dec 2011: 35°F; Feb 2012: 29°F; Jan 2013: 35°F.

3 Average the lowest temperature. When Fahrenheit values are used, convert the result to Celsius. Round the Celsius value to the next warmer whole degree; e.g., -15.00 remains -15; -14.99 becomes -14. The resultant rounded value is the ACT.

Example:

$$ACT = \frac{37 + 35 + 35 + 29 + 35}{5}$$

$$ACT = 34.20^{\circ}\text{F}$$

$$ACT = (34.20 - 32)/1.8 \approx 1.22^{\circ}\text{C}$$

$$ACT = 2^{\circ}\text{C}$$

2. ACT based on standard Δ ISA. When historical temperature data is not available, determine ACT using a standard cold temperature deviation below airport ISA (Δ ISA).

a Calculate airport ISA. Calculate the airport ISA in degrees Celsius using formula 3-3-3:

Formula 3-3-3. Airport ISA

$$ISA_{\text{airport}} = 15 - 0.00198 \times \text{apt}_{\text{elev}}$$

Example:

$$ISA_{\text{airport}} = 15 - 0.00198 \times 5433.8$$

$$ISA_{\text{airport}} \approx 4.24^{\circ}\text{C}$$

b Determine standard Δ ISA. Select the standard Δ ISA for the airport using table 3-3-1:

Table 3-3-1. Standard Δ ISA Values

Airport Location	Value Below Airport ISA°C
Conus	-30°C
Alaska	-40°C
Hawaii and US territories	-20°C

c Determine the ACT. Add the standard Δ ISA deviation to the airport ISA value and round the calculated value to the next warmer whole degree; e.g., -15.00 remains -15; -15.01 through -15.99 becomes -15. The resultant rounded value is the ACT.

Example:

$$ACT = -30 + 4.24$$

$$ACT \approx -25.76^{\circ}\text{C}$$

$$ACT = -25^{\circ}\text{C}$$

(b) Determining the critical low and high temperatures. Normally, the critical low temperature limit is the calculated ACT. Where the ACT results in an effective glidepath of less than 2.5 degrees, the temperature limit is raised to achieve the required glidepath angle. The critical high temperature limit is the temp that yields an effective glidepath of 1.13 times the maximum allowed glidepath angle for the fastest designed category (not to exceed 54 degrees Celsius). Use formula 3-3-4, formula 3-3-5, and formula 3-3-6 to find both the critical low and critical high temperature.

Formula 3-3-4. Temperature Induced DA Deviation

$$\Delta DA_{\text{Temp}} = e^{\frac{d_{250} \times \tan(\theta_{\text{Temp}})}{r}} \times (r + \text{LTP}_{\text{elev}} + \text{TCH}) - (r + \text{TDZE} + 250)$$

Where:

d_{250} = distance (feet) from LTP to a point 250 feet above touchdown zone elevation (TDZE) using the designed GPA

θ_{Temp} = 2.5 for low temp deviations or $1.13 \times$ the maximum allowable GPA of the fastest CAT for high temp deviations.

Example:

$$\Delta DA_{\text{Temp}} = e^{\frac{3805.69 \times \tan(2.5)}{r}} \times (r + 5321.8 + 55) - (r + 5326.3 + 250)$$

$$\Delta DA_{\text{Temp}} \approx -33.26 \text{ feet}$$

Note: This calculation is accomplished twice. Once for low and once for high temperature deviations.

Formula 3-3-5. Δ ISA Adjusted

$$\Delta \text{ISA}_{\text{adjusted}} = \frac{\Delta DA_{\text{Temp}} \times [288 - 0.5 \times 0.00198 \times (\text{TDZE} + 250)]}{250 - \Delta DA_{\text{Temp}}}$$

Where:

ΔDA_{Temp} = calculated maximum DA deviation if flown at the respective critical temperatures

Example:

$$\Delta \text{ISA}_{\text{adjusted}} = \frac{-33.30 \times [288 - 0.5 \times 0.00198 \times (5326.3 + 250)]}{250 + 33.30}$$

$$\Delta \text{ISA}_{\text{adjusted}} \approx -33.20^{\circ}\text{C}$$

Note 1: This calculation is accomplished twice. Once for low and once for high temperature deviations.

Formula 3-3-6. Adjusted ACT

$$ACT_{\text{adjusted}} = ISA_{\text{airport}} + \Delta ISA_{\text{adjusted}}$$

Where:

ISA_{airport} = International Standard Atmosphere temperature °C for the airport

$\Delta ISA_{\text{adjusted}}$ = ISA adjustment (°C) required to achieve the respective low or high temp GPA

Example:

$$ACT_{\text{adjusted}} = 4.24 - 33.20$$

$$ACT_{\text{adjusted}} \approx -28.96^{\circ}$$

$$ACT_{\text{adjusted}} \approx -28^{\circ}$$

Note 1: $Critical_{\text{Low}}$ = warmer of ACT or ACT_{adjusted} rounded to the warmer whole degree.

Note 2: $Critical_{\text{High}}$ = colder of 54°C or ACT_{adjusted} rounded to the colder whole degree.

Note 3: $\Delta ISA_{\text{Low}} = Critical_{\text{Low}} - ISA_{\text{airport}}$

(c) High temperature descent rate. Use formula 3-3-7 to determine the maximum expected descent rate at standard and critical high temperatures (see Order 8260.19, paragraph 8-7-1).

Formula 3-3-7. Descent Rate

$$dr = \frac{(V_{KTAS} + 10) \times 1852}{0.3048} \times \frac{60 \times \tan(\theta)}{3600}$$

Where:

V_{KTAS} = KTAS using 250 feet above TDZE and final approach KIAS for fastest CAT

θ = either designed GPA or $1.13 \times$ maximum allowable GPA for the fastest CAT

Example:

$$dr = \frac{(184.10 + 10) \times 1852}{0.3048} \times \frac{60 \times \tan(3.50)}{3600}$$

$$dr \approx 1202.23 \text{ feet/minute}$$

(2) OCS slope. The OCS slope and origin are determined using formula 3-3-8, formula 3-3-9, and formula 3-3-10:

Formula 3-3-8. OCS Slope

$$OCS_{\text{slope}} = \frac{1}{\tan(\theta) \times [0.928 + 0.0038 \times (\text{Critical}_{\text{Low}} - \text{ISA}_{\text{airport}})]}$$

Where:

$\text{Critical}_{\text{Low}}$ = calculated temperature (°C) that ensures no less than a 2.5° GPA

$\text{ISA}_{\text{airport}}$ = International Standard Atmosphere temperature (°C) for the airport

Example:

$$OCS_{\text{slope}} = \frac{1}{\tan(3) \times [0.928 + 0.0038 \times (-28.96 - 4.24)]}$$

$$OCS_{\text{slope}} \approx 23.80$$

Formula 3-3-9. Sloping OCS Origin

$$d_{\text{origin}} = \frac{\text{TDZE} + 250 - (\text{LTP}_{\text{elev}} + \text{TCH})}{\tan(\theta)}$$

Example:

$$d_{\text{origin}} = \frac{5326.3 + 250 - (5321.8 + 55)}{\tan(3)}$$

$$d_{\text{origin}} \approx 3806.69 \text{ feet}$$

Formula 3-3-10. Distance from LTP that the Sloping OCS Application Begins

$$d_{\text{OCS}} = d_{\text{origin}} + r \times OCS_{\text{slope}} \times \ln\left(\frac{\text{TDZE} + 89 + r}{r + \text{LTP}_{\text{elev}}}\right)$$

Where:

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$d_{\text{OCS}} = 3806.69 + r \times 23.80 \times \ln\left(\frac{5326.3 + 89 + r}{r + 5321.8}\right)$$

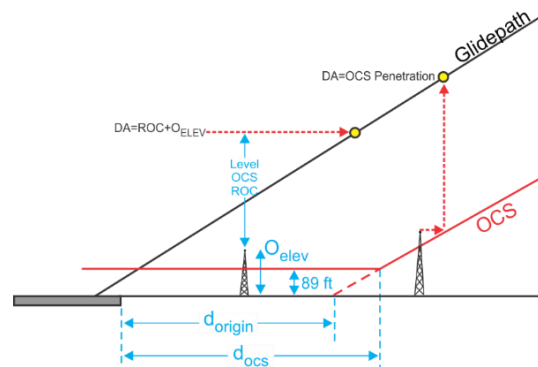
$$d_{\text{OCS}} \approx 6031.42 \text{ feet}$$

(3) Obstacle accuracies/ROC adjustments. If the controlling obstacle is within the sloping OCS area, apply the non-RNP procedure evaluation sequence based on the sloping surface evaluation method (see Order 8260.19, paragraphs 2-11-3 and 2-11-4). Do not apply Order 8260.3, chapter 3 precipitous terrain and/or RASS adjustments when determining preliminary DA.

(4) Sloping OCS DA adjustment (see figure 3-3-2). The MSL elevation of the sloping OCS at any distance from LTP (beyond the OCS origin) is determined using formula 3-3-11.

Where obstacles penetrate the OCS, determine the distance to the adjusted DA by entering the penetrating obstacle's actual or adjusted elevation (see formula 3-3-2) into formula 3-3-2.

Figure 3-3-2. LNAV/VNAV DA Adjustment



Formula 3-3-11. Sloping OCS Elevation

$$OCS_{elev} = e^{\left(\frac{d_{LTP} - d_{origin}}{r \times OCS_{slope}}\right)} \times (r + LTP_{elev}) - r$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$OCS_{elev} = e^{\left(\frac{19747.38 - 3911.63}{r \times 23.42}\right)} \times (r + 1125.4) - r$$

$$OCS_{elev} \approx 1801.61 \text{ feet}$$

Formula 3-3-12. Preliminary DA Distance Based on Sloping OCS Penetration

$$d_{DA} = (r + LTP_{elev}) \times OCS_{slope} \times \ln\left(\frac{r + O_{elev}}{r + LTP_{elev}}\right) + d_{origin}$$

Where:

O_{elev} = obstacle actual or $O_{adjusted}$ elevation (feet)

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$d_{DA} = (r + 1125.4) \times 23.42 \times \ln\left(\frac{r + 1263.7}{r + 1125.4}\right) + 3911.63$$

$$d_{DA} \approx 7150.61 \text{ feet}$$

b. Level OCS (see Order 8260.3, paragraph 2-1-3). A level OCS (89 feet above TDZE) is applied from 0.3 NM past the LTP/FTP to the point of intersection with the sloping OCS. If an

obstacle is in the secondary area (transitional surface), adjust the height of the obstacle using formula 3-3-2, then evaluate it at the adjusted height as if it is in the primary area.

(1) Obstacle clearance. Apply the applicable ROC from table 3-3-2 to the obstacle with the highest effective height to determine the preliminary DA.

(2) Obstacle accuracies/ROC adjustments. If the controlling obstacle is within the level OCS area, apply the non-RNP procedure evaluation sequence based on the level surface evaluation method (see Order 8260.19, paragraphs 2-11-3 and 2-11-4). Do not apply Order 8260.3, chapter 3 precipitous terrain and/or RASS adjustments when determining preliminary DA.

Table 3-3-2. Level OCS ROC by Approach Category

Aircraft Category	ROC
A	131
B	142
C	150
D/E	161

3-3-5. Decision Altitude (DA). The minimum HAT is 250 feet (see Order 8260.3, table 3-2-2). The DA based on the final segment evaluation is the highest of the following; the DA based on the minimum HAT, the DA based on the minimum offset DA distance (see paragraph 3-3-2), the DA based on the controlling obstacle. APV precipitous terrain and RASS adjustments apply. The DA based on the evaluation of the sloping OCS is not required to be higher than the MDA of an LNAV procedure established on the same chart. Order 8260.19, paragraph 8-7-1.b applies, except document the controlling obstacle and ROC based on whichever evaluation (level or sloping OCS) results in the highest DA.

3-3-6. Raising GPA or Adjusting TCH for Penetrating Obstructions. The GPA and/or TCH may be raised to eliminate or reduce final segment OCS penetrations.

Section 3-4. LPV/GLS Final

3-4-1. General. The OEA and associated OCSs are applicable to LPV and GLS final approach segments.

a. Exceptions. Paragraph 1-3-1 vertical path requirements apply except:

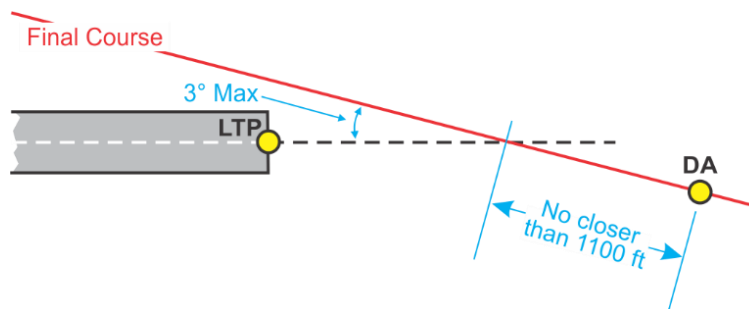
- (1) The maximum GPA is 3.5 degrees for procedures with a HAT less than 250.
- (2) The maximum TCH is 60 feet regardless of height group.
- (3) The maximum helipoint crossing height (HCH) is 20 feet for helicopter procedures to a helipad.
- (4) The maximum GPA is 6.4 degrees for helicopter PinS approach procedures. The minimum height above surface (HAS)/height above landing (HAL) is 250 feet for a GPA below 5.01 degrees. At 5.01 degrees and higher apply Order 8260.3, table 3-2-2, CAT A to determine the minimum HAS/HAL relating to the GPA.

b. Restrictions. For GLS procedures, final track intercept must occur within 20 NM of the airport.

3-4-2. Alignment.

a. The final course is normally aligned with the RCL (± 0.03 degrees) extended through the LTP (± 5 feet). Where an operations requirement necessitates an offset course, the FAC must intersect the RCL extended no closer than 1100 feet inside the DA point and the offset must not exceed three degrees measured from the point of intersection (see figure 3-4-1). The DA point for this evaluation is the point on the glideslope where the altitude is equal to the published DA less any RASS and precipitous terrain adjustments. Where the course is offset, the minimum HAT is 250. TF-TF turns are not allowed in the FAS. Helicopter procedures to heliports must be aligned per Order 8260.3, paragraph 12-2-4.a and are considered straight-in. SIAPs designed to support SOIA operations are exempted from standard final segment alignment when compliant with Order 8260.3, paragraph 15-3-5.

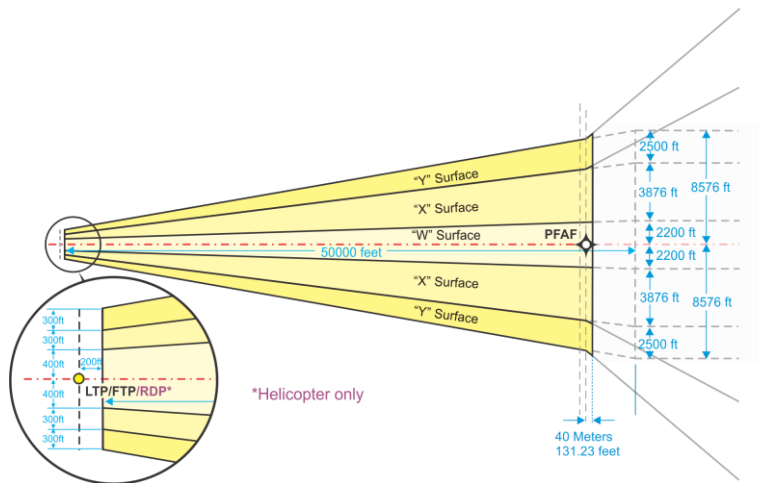
Figure 3-4-1. Offset Final Alignment



b. Helicopter PinS. PinS helicopter procedures align to a heliport/FHP located on the FAC 2600 feet after the LNAV MAP.

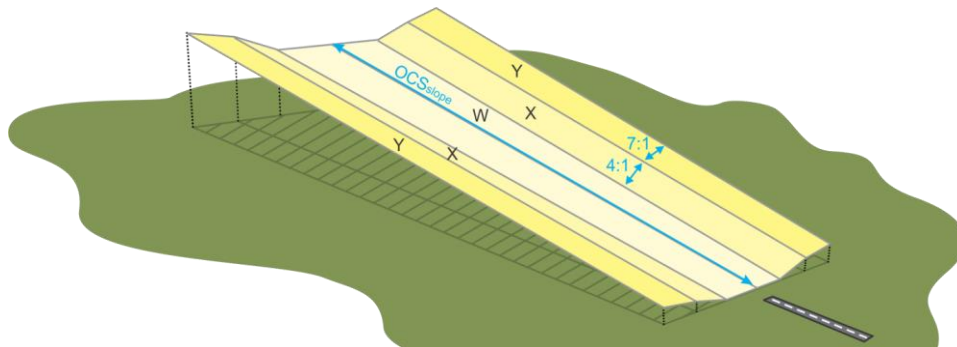
3-4-3. Area. The OEA originates 200 feet from LTP (heliport/FHP for helicopter procedures) and extends to a point 40 meters beyond the PFAF. It is centered on the final approach course and expands uniformly from its origin to a point 50000 feet from the origin. Where the PFAF must be located more than 50200 feet from LTP (heliport/FHP for helicopter procedures), the OEA continues linearly (boundaries parallel to course centerline) beyond that point (see figure 3-4-2).

Figure 3-4-2. LPV/GLS Final OEA



3-4-4. Obstacle Clearance Surface. The primary area OCS consists of the W and X surfaces. The Y surface is an early missed approach transitional surface. The W surface slopes longitudinally along the final approach track, and is level perpendicular to track. The X and Y surfaces slope upward from the edge of the W surface perpendicular to the final approach track (see figure 3-4-3). Obstacles located in the X and Y surfaces are adjusted in height to account for perpendicular surface rise and evaluated under the W surface. When computing OCS values for helicopter procedures, replace LTP with heliport/FHP. For procedures to a heliport that supports IFR procedures, replace TCH with HCH. For PinS procedures, the TCH is always zero (0).

Figure 3-4-3. LPV/GLS Final OCSs



a. OCS slope. Determine the OCS slope associated with a specific glidepath angle using formula 3-4-1.

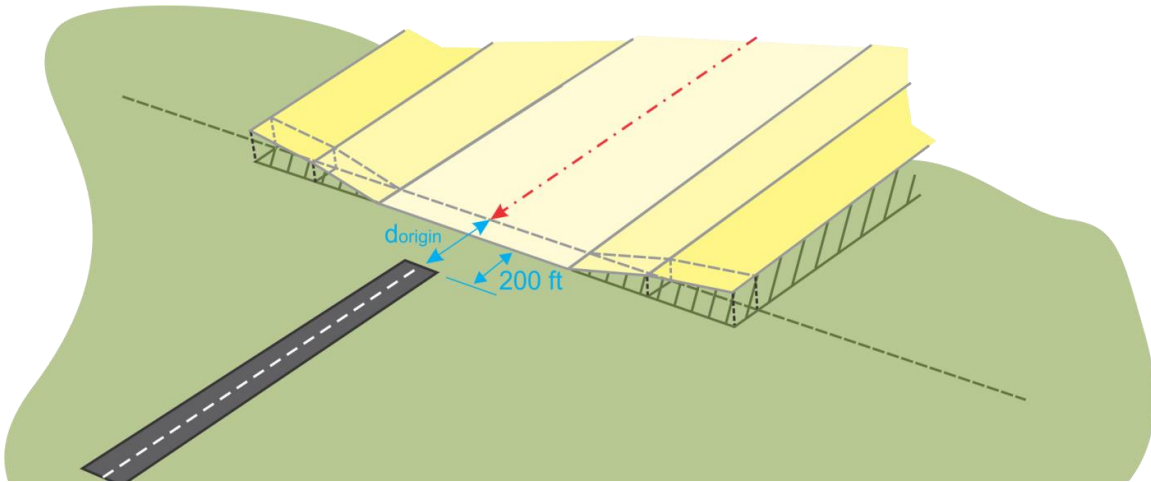
Formula 3-4-1. OCS Slope

$$\text{OCS}_{\text{slope}} = \frac{102}{\theta}$$

Example:

$$\begin{aligned}\text{OCS}_{\text{slope}} &= \frac{102}{3.1} \\ \text{OCS}_{\text{slope}} &\approx 32.90\end{aligned}$$

Figure 3-4-4. OCS Slope Origin



b. OCS origin. All OCS surfaces begin at LTP elevation at a point 200 feet from LTP (see figure 3-4-4) measured along course centerline and extends toward the PFAF. The longitudinal (along-track) rising W surface slope begins at a calculated distance from the LTP (see formula 3-4-2).

Formula 3-4-2. OCS Origin

$$d_{\text{origin}} = \text{greater of } 200 \text{ or } 1154 - \frac{\text{TCH}}{\tan(\theta)}$$

Example:

$$\begin{aligned}d_{\text{origin}} &= 1154 - \frac{55}{\tan(3.1)} \\ d_{\text{origin}} &\approx 138.45 \text{ feet} \\ d_{\text{origin}} &= 200 \text{ feet}\end{aligned}$$

c. Obstacle effective elevation (O_{EE}). Because the earth curves away from the OCS as distance from course centerline increases, the MSL elevation of an obstacle is reduced to account for this. Use formula 3-4-3 to calculate the O_{EE} .

Formula 3-4-3. Obstacle Effective Elevation

$$O_{EE} = O_{MSL} - \left[(r + LTP_{elev}) \times \left(\cos \left[\frac{OBS_Y \times 180}{r \times \pi} \right]^{-1} - 1 \right) + Q \right]$$

Where:

O_{MSL} = obstacle MSL elevation

OBS_Y = perpendicular distance (feet) from the course centerline to the obstacle

Q = obstacle adjustment (feet) for X or Y surface rise. Zero (0) if in the W surface

Example:

$$O_{EE} = 2768.9 - \left[(r + 1125.4) \times \left(\cos \left[\frac{1432.5 \times 180}{r \times \pi} \right]^{-1} - 1 \right) + 192.90 \right]$$

$$O_{EE} \approx 2575.95 \text{ feet}$$

d. W OCS.

(1) Width. The perpendicular distance from course centerline to the boundary is 400 feet at the origin, and expands uniformly to 2200 feet at a point 50200 feet from LTP/FTP. Calculate the OCS half-width using formula 3-4-4.

Formula 3-4-4. W OCS Half Width

$$W_{boundary} = 0.036 \times d_{LTP} + 392.8$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

Note: $200 \text{ feet} \leq d_{LTP} \leq 50200 \text{ feet}$

Example:

$$W_{boundary} = 0.036 \times 5462.03 + 392.8$$

$$W_{boundary} \approx 589.43 \text{ feet}$$

(2) OCS elevation. The LPV/GLS glidepath is considered to be a straight line in space extending from TCH. To protect the straight-line glidepath, the OCS is a flat plane (does not follow earth curvature). The elevation of the W OCS at any point is the elevation of the OCS at the course centerline abeam it. Calculate the W OCS angle using formula 3-4-5 and elevation at any distance beyond the OCS origin using formula 3-4-6.

Formula 3-4-5. OCS Angle

$$\text{OCS}_{\text{angle}} = \text{atan} \left(\frac{\theta}{102} \right)$$

Example:

$$\text{OCS}_{\text{angle}} = \text{atan} \left(\frac{3.1}{102} \right)$$

$$\text{OCS}_{\text{angle}} \approx 1.74^{\circ}$$

Formula 3-4-6. W OCS Elevation

$$\text{OCS}_{\text{elev}} = \frac{(r + \text{LTP}_{\text{elev}}) \times \cos(\text{OCS}_{\text{angle}})}{\cos \left[\frac{(\text{d}_{\text{LTP}} - \text{d}_{\text{origin}}) \times 180}{r \times \pi} + \text{OCS}_{\text{angle}} \right]} - r$$

Where:

$\text{OCS}_{\text{angle}}$ = calculated angle of the obstacle clearance surface

d_{LTP} = along track distance (feet) from LTP to point of interest

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$\text{OCS}_{\text{elev}} = \frac{(r + 1125.4) \times \cos(1.74)}{\cos \left[\frac{(5280 - 200) \times 180}{r \times \pi} + 1.74 \right]} - r$$

$$\text{OCS}_{\text{elev}} \approx 1280.35 \text{ feet}$$

(3) OCS evaluation. Compare the W OCS elevation abeam the obstacle location with the O_{EE} . Lowest minimums are achieved when the W surface is clear. See latest version of Order 8260.3, section 10-6 for obstacle assessment. To eliminate or avoid a penetration, take one or more of the following actions:

- (a) Remove or adjust the obstruction location and/or height.
- (b) Displace the runway threshold (NA for PinS procedures).
- (c) Raise GPA within the categorical limits.
- (d) Adjust DA.
- (e) Raise TCH (NA for PinS procedures).

e. X OCS.

(1) Width. The perpendicular distance from the course centerline to the outer boundary of the X OCS is 700 feet at the origin and expands uniformly to 6076 feet at a point 50200 feet

from LTP/FTP. Calculate the perpendicular distance from the course centerline to the X surface boundary using formula 3-4-7.

Formula 3-4-7. Perpendicular Distance to X Boundary

$$X_{\text{boundary}} = 0.10752 \times d_{\text{LTP}} + 678.496$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

Note: $200 \text{ feet} \leq d_{\text{LTP}} \leq 50200 \text{ feet}$

Example:

$$X_{\text{boundary}} = 0.10752 \times 5462.03 + 678.496$$

$$X_{\text{boundary}} \approx 1265.77 \text{ feet}$$

(2) Obstacle adjustment. The X OCS begins at the height of the W surface and rises at a slope of 4:1 in a direction perpendicular to the final approach course. The MSL elevation of an obstacle in the X surface is reduced by the amount of surface rise. Use formula 3-4-8 to determine the obstacle height adjustment (Q) for use in formula 3-4-3. Evaluate the obstacle in accordance with paragraph 3-4-4.d(3).

Formula 3-4-8. X OCS Obstacle Adjustment

$$Q = \frac{OBS_Y - W_{\text{boundary}}}{4}$$

Where:

OBS_Y = perpendicular distance (feet) from the course centerline to the obstacle

W_{boundary} = perpendicular distance (feet) from the course centerline to the W surface boundary

Example:

$$Q = \frac{1265.77 - 589.43}{4}$$

$$Q \approx 169.09 \text{ feet}$$

f. Y OCS.

(1) Width. The perpendicular distance from the course centerline to the outer boundary of the Y OCS is 1000 feet at the origin and expands uniformly to 8576 feet at a point 50200 feet from LTP/FTP. Calculate the perpendicular distance from the course centerline to the Y surface boundary using formula 3-4-9.

Formula 3-4-9. Perpendicular Distance to Y Boundary

$$Y_{\text{boundary}} = 0.15152 \times d_{\text{LTP}} + 969.696$$

Where:

d_{LTP} = along track distance (feet) from LTP to point of interest

Note: $200 \text{ feet} \leq d_{\text{LTP}} \leq 50200 \text{ feet}$

Example:

$$Y_{\text{boundary}} = 0.15152 \times 5462.03 + 969.696$$

$$Y_{\text{boundary}} \approx 1797.30 \text{ feet}$$

(2) Obstacle adjustment. The Y OCS begins at the height of the X surface and rises at a slope of 7:1 in a direction perpendicular to the final approach course. The MSL elevation of an obstacle in the Y surface is reduced by the amount of X and Y surface rise. Use formula 3-4-10 to determine the obstacle height adjustment (Q) for use in formula 3-4-3. Evaluate the obstacle in accordance with paragraph 3-4-4.d(3).

Formula 3-4-10. Y OCS Obstacle Adjustment

$$Q = \frac{X_{\text{boundary}} - W_{\text{boundary}}}{4} + \frac{\text{OBS}_Y - X_{\text{boundary}}}{7}$$

Where:

X_{boundary} = perpendicular distance (feet) from the course centerline to the X surface boundary

W_{boundary} = perpendicular distance (feet) from the course centerline to the W surface boundary

OBS_Y = perpendicular distance (feet) from the course centerline to the obstacle

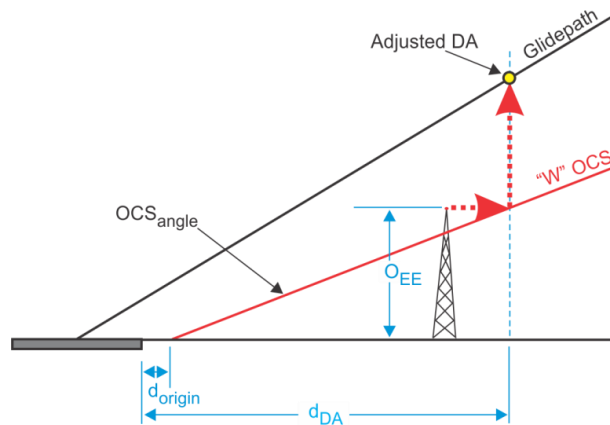
Example:

$$Q = \frac{1265.77 - 589.43}{4} + \frac{1432.5 - 1265.77}{7}$$

$$Q \approx 192.90 \text{ feet}$$

3-4-5. Decision Altitude (DA). When the final OCS is clear, the minimum HAT is 200 feet.

When the final OCS is penetrated or when the final course is offset, the minimum HAT is 250 feet. Minimum HAS/HAL for helicopter procedures is 250 feet. The DA is the highest of the following: the DA based on the applicable minimum HAT (HAS/HAL for helicopter procedures); the DA based on GPA (see Order 8260.3, table 3-2-2); and the DA based on evaluation of the final segment OCS. Calculate the adjusted DA distance for OCS penetrations using formula 3-4-11 for input into formula 3-4-12. Calculate the published DA distance from the LTP/FTP using formula 3-4-13. The published DA is not required to be higher than the MDA of an LNAV procedure established on the same chart.

Figure 3-4-5. LPV/GLS DA Adjustment**Formula 3-4-11. Adjusted DA Distance**

$$d_{DA} = \frac{r \times \pi}{180} \times \left(90 - OCS_{angle} - \text{asin} \left[\frac{\cos(OCS_{angle}) \times (r + LTP_{elev})}{r + O_{EE}} \right] \right) + d_{origin}$$

Where:

OCS_{angle} = calculated angle of the obstacle clearance surface

d_{origin} = distance (feet) from LTP to OCS origin

O_{EE} = calculated obstacle effective elevation (feet)

Example:

$$d_{DA} = \frac{r \times \pi}{180} \times \left(90 - 1.74 - \text{asin} \left[\frac{\cos(1.74) \times (r + 1125.4)}{r + 1271.5} \right] \right) + 200$$

$$d_{DA} \approx 4991.01 \text{ feet}$$

Formula 3-4-12. Altitude on LPV/GLS Glidepath

$$Z_{LPV} = \frac{(r + alt_b) \times \cos(\theta)}{\cos\left(\frac{d \times 180}{r \times \pi} + \theta\right)} - r$$

Where:

alt_b = beginning altitude (feet) (e.g., LTP elevation + TCH)

d = distance (feet) between the points of interest

Example:

$$Z_{LPV} = \frac{(r + (1125.4 + 55)) \times \cos(3.1)}{\cos\left(\frac{42041.91 \times 180}{r \times \pi} + 3.1\right)} - r$$

$$Z_{LPV} \approx 3500 \text{ feet}$$

Formula 3-4-13. Distance on LPV/GLS Glidepath

$$d_{LPV} = \frac{r \times \pi}{180} \times \left(90 - \theta - \text{asin}\left[\frac{\cos(\theta) \times (r + alt_b)}{r + alt_e}\right]\right)$$

Where:

alt_b = beginning altitude (MSL)

alt_e = ending altitude (MSL)

Example:

$$d_{LPV} = \frac{r \times \pi}{180} \times \left(90 - 3.1 - \text{asin}\left[\frac{\cos(3.1) \times (r + 1125.4 + 55)}{r + 3500}\right]\right)$$

$$d_{LPV} \approx 42041.91 \text{ feet}$$

3-4-6. Raising GPA for OCS Penetrations. Raising the GPA may eliminate OCS penetrations. To determine the revised minimum GPA, use formula 3-4-14.

Formula 3-4-14. GPA Adjustment

$$SRD = \sqrt{(r + O_{EE})^2 + (r + LTP_{elev})^2 - 2 \times (r + O_{EE}) \times (r + LTP_{elev}) \times \cos \left[\frac{(d_{LTP} - d_{origin}) \times 180}{r \times \pi} \right]}$$

$$\theta_{adjusted} = \tan \left[\arccos \left(\frac{SRD^2 + (r + LTP_{elev})^2 - (r + O_{EE})^2}{2 \times SRD \times (r + LTP_{elev})} \right) - 90 \right] \times 102$$

Where:

O_{EE} = calculated obstacle effective elevation (feet)

d_{LTP} = along track distance (feet) from LTP to penetrating obstacle

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$SRD = \sqrt{(r + 1274.5)^2 + (r + 1125.4)^2 - 2 \times (r + 1274.5) \times (r + 1125.4) \times \cos \left[\frac{(3992.7 - 200) \times 180}{r \times \pi} \right]}$$

$$SRD \approx 3795.85 \text{ feet}$$

$$\theta_{adjusted} = \tan \left[\arccos \left(\frac{3795.85^2 + (r + 1125.4)^2 - (r + 1274.5)^2}{2 \times 3795.85 \times (r + 1125.4)} \right) - 90 \right] \times 102$$

$$\theta_{adjusted} \approx 4.00^\circ$$

3-4-7. Adjusting TCH to Reduce/Eliminate OCS Penetrations. This paragraph is applicable only where the OCS origin is greater than 200 feet from the LTP. Adjusting TCH is the equivalent to relocating the glide slope antenna in ILS criteria. The goal is to move the OCS origin toward the LTP/FTP (no closer than 200 feet) to raise the OCS at the obstacle location. To determine the maximum W surface vertical relief that can be achieved by adjusting TCH, apply formula 3-4-15. If the result is greater than the penetration, you may determine the amount to increase TCH by applying formula 3-4-16. If this option is selected, re-evaluate the final segment using the revised TCH value.

Formula 3-4-15. Vertical Relief

$$Z_{TCH} = \frac{d_{\text{origin}} - 200}{OCS_{\text{slope}}}$$

Where:

d_{origin} = distance (feet) from LTP to OCS origin

Example:

$$Z_{TCH} = \frac{390.75 - 200}{34}$$

$$Z_{TCH} \approx 5.61 \text{ feet}$$

Formula 3-4-16. TCH Adjustment

$$\Delta TCH = \tan(\theta) \times OCS_{\text{slope}} \times p$$

Where:

p = penetration (feet) of the final OCS

Example:

$$\Delta TCH = \tan(3) \times 34 \times 4.04$$

$$\Delta TCH \approx 7.20 \text{ feet}$$

Section 3-5. Missed Approach General

3-5-1. General. The NavSpec is RNP APCH (missed flight phase with associated XTT of 1.00). *Exception: For LNAV and LP, apply ATT based on the applicable final flight phase at the beginning of section 1 (line C-D).* Optional NavSpec may be A-RNP (missed flight phase with associated XTT of 0.30), or RNP 0.3 for helicopters (missed flight phase with associated XTT of 0.30). Use an effective XTT of 2.00 where applicable per table 1-2-1 footnote 2. Secondary areas apply where specified, except for A-RNP. Line C-D location and altitude is defined using the base MDA/DA. The base MDA/DA is the lowest value resulting from subtracting any RASS and precipitous terrain adjustments from the lowest final segment MDA/DA (rounded to a publishable value).

3-5-2. Course-to-Altitude (CA) Leg. A CA leg is used as the first leg of an RNAV missed approach and must be followed by a DF leg. The CA leg must specify a course and altitude. The specified CA course must be an extension of the FAC. The specified CA altitude is determined as follows:

a. When a turning missed approach is based on a “climb-to” altitude before turning, the coded CA altitude is the charted “climb-to” altitude. The lowest permissible climb-to altitude is 400 feet above airport elevation (surface or heliport elevation as applicable for helicopters) (rounded to the nearest foot) plus any final precipitous terrain and primary RASS adjustments.

(1) A climb-to altitude is required for non-LPV/GLS procedures when the first fix after the coded MAP is not within one degree of the FAC extended and the lowest base MDA/DA is less than 400 feet above airport elevation (surface or heliport elevation as applicable for helicopters) (rounded to the nearest foot).

(2) A climb-to altitude is required for LPV/GLS procedures when the first fix after the coded MAP is not within one degree of the FAC extended. Additionally, the climb-to altitude must not be less than the highest Section 2 start altitude (see paragraph 3-7-1.b(2)).

b. When a charted climb-to altitude is not required by the preceding paragraph, the specified CA altitude is the DA, MDA, or 400 feet above the airport elevation (rounded to the nearest foot), whichever is lowest. For helicopter point-in-space (PinS) procedures, the specified CA altitude is the DA or MDA, whichever is lower.

3-5-3. MA Climb Gradient Termination. Where the default OCS slope is penetrated and the lowest HAT is required, a greater than standard CG may be required to clear the penetrating obstruction. Calculate the climb gradient termination altitude using formula 3-5-1 for input into the climb gradient formula 1-3-7. If a RASS adjustment is applicable, apply the CG associated with the lowest MDA/DA.

Formula 3-5-1. MA CG Termination Altitude

$$CG_{\text{term}} = 48 \times D_O + O_{\text{MSL}} - \frac{d_{\text{primary}}}{12} + AC_{\text{start}} - OCS_{\text{start}} + \text{RASS}$$

Where:

D_O = dist. (NM) from AC_{start} to primary obs. or point from which d_{primary} measured

O_{MSL} = obstacle MSL elevation

d_{primary} = dist. (feet) from edge of primary area to obs. Zero (0) if not in secondary

AC_{start} = starting altitude (feet) of the aircraft (e.g., SOC)

OCS_{start} = starting altitude (feet) of the OCS

RASS = remote altimeter setting source adjustment (feet) for final if applicable

Example:

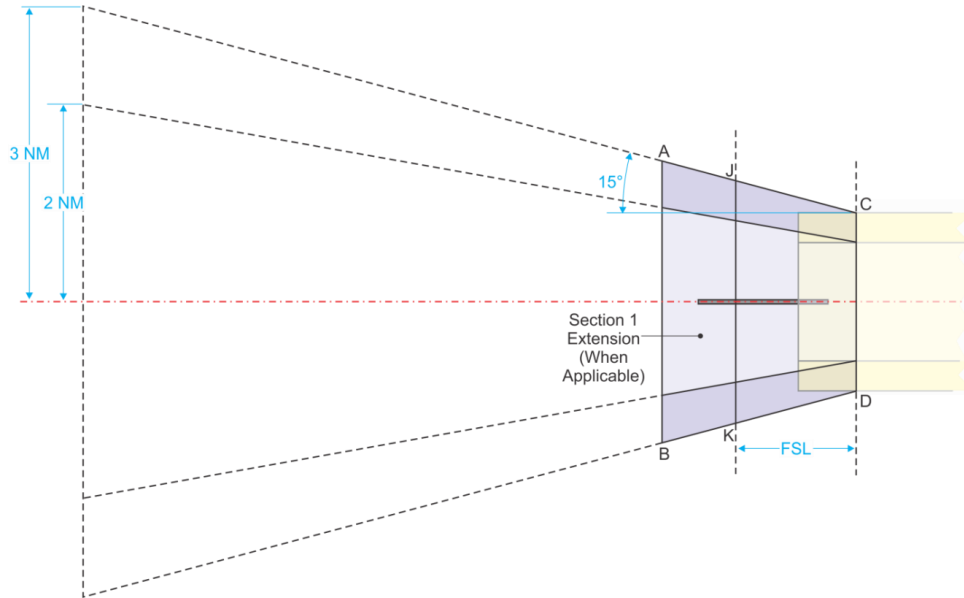
$$CG_{\text{term}} = 48 \times 3.95 + 2147.41 - \frac{537.41}{12} + 1442 - 1217.21$$

$$CG_{\text{term}} \approx 2517.02 \text{ feet}$$

Section 3-6. Missed Approach Section 1

3-6-1. Non-vertically Guided. Section 1 begins final segment ATT prior to the MAP (line C-D) along a continuation of the final course and extends to the SOC (line J-K) or the point where the aircraft is projected to cross 400 feet above airport/heliport elevation (rounded to the nearest foot) (line A-B); whichever is the greater distance from the MAP (see figure 3-6-1 and figure 3-6-2).

Figure 3-6-1. LNAV Missed Approach Section 1 Area



a. Length.

(1) Flat Surface Length (FSL). The section 1 flat surface begins either 0.30 NM for LNAV or 40 meters for LP prior to the MAP and extends the FSL distance to line J-K. Calculate the FSL using formula 3-6-1.

Formula 3-6-1. LNAV/LP Flat Surface Length

$$FSL = \frac{(V_{KTAS} + 10) \times 12}{3600} + 2 \times ATT_{NM}$$

Where:

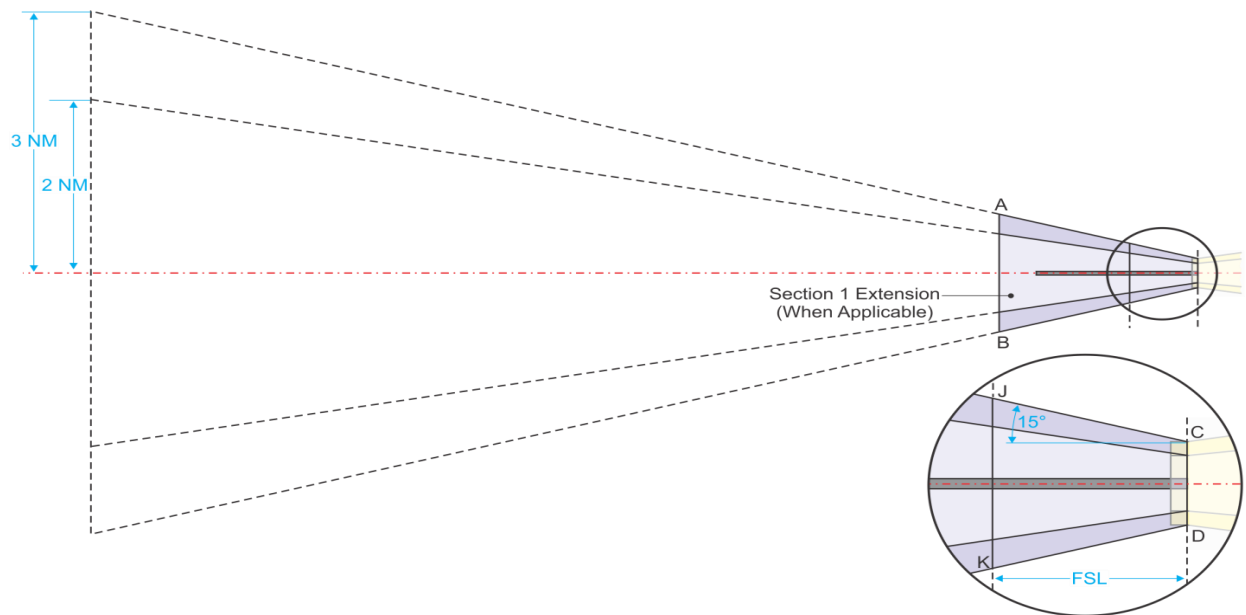
V_{KTAS} = calculated KTAS using MDA and final approach KIAS

ATT_{NM} = applicable final segment ATT in NM

Example:

$$FSL = \frac{(171.91 + 10) \times 12}{3600} + 2 \times 0.30$$

$$FSL \approx 1.21 \text{ NM}$$

Figure 3-6-2. LP Missed Approach Section 1 Area

(2) Section 1 extension.

(a) Base MDA \geq 400 feet above airport/heliport elevation (rounded to the nearest foot). No extension required. Line A-B is coincident with line J-K.

(b) Base MDA $<$ 400 feet above airport/heliport elevation (rounded to the nearest foot). Calculate the section 1 extension using formula 3-6-2 and locate line A-B at this distance beyond line J-K.

Formula 3-6-2. Section 1 Extension

$$S1_{\text{extension}} = \ln \left(\frac{r + \text{apt}_{\text{elev}} + 400}{r + \text{alt}_b} \right) \times \frac{r}{\text{CG}}$$

Where:

alt_b = either the Base MDA for LNAV/LP or Base DA for LNAV/VNAV

Example:

$$S1_{\text{extension}} = \ln \left(\frac{r + 2900 + 400}{r + 3205} \right) \times \frac{r}{315}$$

$$S1_{\text{extension}} \approx 0.30 \text{ NM}$$

b. Width. The secondary area outer boundary starts at the outer edge of the final segment secondary area at line C-D and splays at 15 degrees relative to the missed approach course until reaching 3 XTT width from centerline. The primary area outer boundary starts at the edge of the primary area at line C-D and splays at the angle required to reach a width of 2 XTT from centerline adjacent to the point where secondary outer boundary reaches 3 XTT width. Calculate the distance from course centerline to the primary and secondary boundary of the OEA at any

distance from line C-D using formula 3-6-3 or formula 3-6-4, depending on the final type. For LNAV there is no splay when an A-RNP RNP 0.30 is specified, the OEA remains linear.

Formula 3-6-3. LNAV and LNAV/VNAV MA Primary and Secondary Boundary

$$\begin{aligned} \text{MAS}_{Y_{\text{primary}}} &= \frac{D_{\text{splay}} \times \tan(15) \times 1.4}{2.1} + 0.6 \\ \text{MAS}_{Y_{\text{secondary}}} &= D_{\text{splay}} \times \tan(15) + 0.9 \end{aligned}$$

Where:

D_{splay} = along track distance (NM) from beginning of MA splay to point of interest

Example:

$$\begin{aligned} \text{MAS}_{Y_{\text{primary}}} &= \frac{6.04 \times \tan(15) \times 1.4}{2.1} + 0.6 \\ \text{MAS}_{Y_{\text{primary}}} &\approx 1.68 \text{ NM} \end{aligned}$$

$$\begin{aligned} \text{MAS}_{Y_{\text{secondary}}} &= 6.04 \times \tan(15) + 0.9 \\ \text{MAS}_{Y_{\text{secondary}}} &\approx 2.52 \text{ NM} \end{aligned}$$

Formula 3-6-4. LP MA Primary and Secondary Boundary

$$\begin{aligned} \text{MAS}_{Y_{\text{primary}}} &= \frac{D_{\text{splay}} \times \tan(15) \times (2 - \frac{1}{2} W_p)}{3 - W_s - \frac{1}{2} W_p} + \frac{1}{2} W_p \\ \text{MAS}_{Y_{\text{secondary}}} &= D_{\text{splay}} \times \tan(15) + W_s + \frac{1}{2} W_p \end{aligned}$$

Where:

D_{splay} = along track distance (NM) from beginning of MA splay to point of interest

$\frac{1}{2} W_p$ = perpendicular distance (NM) from FAS centerline to FAS primary boundary

W_s = final secondary area width (NM)

Example:

$$\begin{aligned} \text{MAS}_{Y_{\text{primary}}} &= \frac{6.04 \times \tan(15) \times (2 - 0.20)}{3 - 0.09 - 0.20} + 0.20 \\ \text{MAS}_{Y_{\text{primary}}} &\approx 1.27 \text{ NM} \end{aligned}$$

$$\begin{aligned} \text{MAS}_{Y_{\text{secondary}}} &= 6.04 \times \tan(15) + 0.09 + 0.20 \\ \text{MAS}_{Y_{\text{secondary}}} &\approx 1.91 \text{ NM} \end{aligned}$$

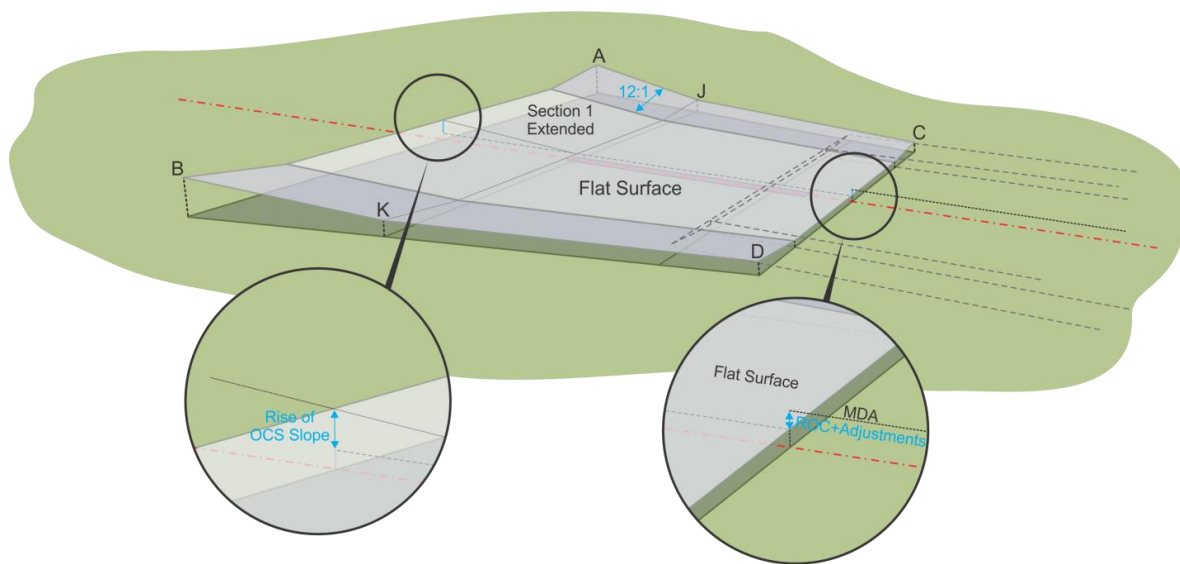
c. Obstacle clearance. Section 1 OCS consists of a flat surface and sloping surface extension (see figure 3-6-3). The MSL Height at the start of the Missed Approach Surface (HMAS) is equal to the base MDA less 100 feet, and excessive length of final adjustments. When obstacles are mitigated by one of the following, re-evaluate the missed approach segment.

(1) Flat surface. The missed approach primary surface remains flat from line C-D to line J-K. Obstacles must not penetrate the flat surface. Where obstacles penetrate the flat surface, raise the MDA by the amount of penetration.

(2) Section 1 extension. From line J-K to line A-B, a sloping OCS surface rises along the missed approach course centerline at a slope ratio commensurate with the obstructions (see paragraph 1-3-1.g(1)). Where obstacles penetrate this OCS, either adjust the MDA by the amount of penetration or apply a climb gradient to clear the obstacle (see paragraph 1-3-1.g(2)).

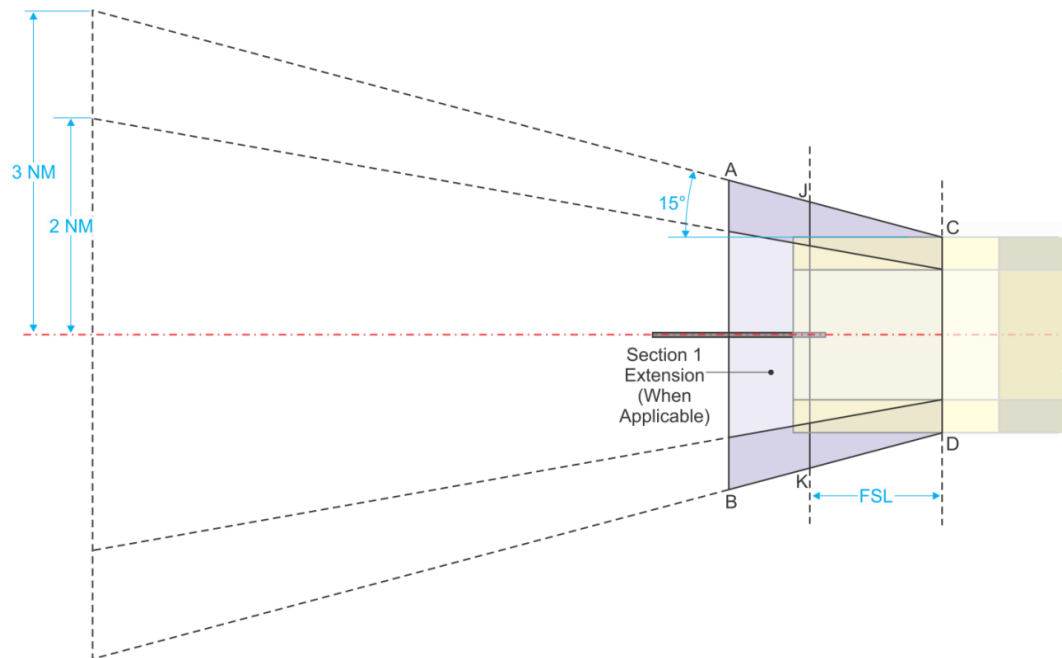
(3) Secondary areas apply from the edge of both the flat surface and section 1 extended with a 12:1 slope measured perpendicular to centerline.

Figure 3-6-3. LNAV Missed Approach Section 1 OCS



3-6-2. LNAV/VNAV. The same criteria specified for non-vertically guided procedures apply with the following exceptions (see figure 3-6-4).

- a. Line C-D is at the base DA point.

Figure 3-6-4. LNAV/VNAV Missed Approach Section 1 Area

b. Length. LNAV/VNAV FSL is based on 15 seconds at KTAS + 10 Kts tailwind without ATT. Calculate FSL using formula 3-6-5. Section 1 extension length is the same as non-vertically guided procedures (see formula 3-6-2).

Formula 3-6-5. LNAV/VNAV Flat Surface Length

$$FSL = \frac{(V_{KTAS} + 10) \times 15}{3600}$$

Where:

V_{KTAS} = calculated KTAS using DA and final approach KIAS

Example:

$$FSL = \frac{(173.02 + 10) \times 15}{3600}$$

$$FSL \approx 0.76 \text{ NM}$$

c. Width. The 15-degree OEA splay begins at the outer edge of the final segment secondary area at the base DA point (line C-D). Calculate the distance from course centerline to the primary and secondary boundary of the OEA at any distance from line C-D using formula 3-6-3. There is no splay when an A-RNP RNP 0.30 is specified, the OEA remains linear.

d. Obstacle clearance. The HMAS is determined by subtracting the applicable flat surface ROC for the CAT (see table 3-3-2) and applicable adjustments from the base DA. Where obstacles penetrate the flat surface, raise the DA by the amount of penetration. Where obstacles

penetrate the sloping OCS, either calculate the DA adjustment with formula 3-6-6 or apply a higher than standard climb gradient to clear the obstruction [see paragraph 1-3-1.g(2)]. In any case, obstacle mitigation requires the missed approach to be re-evaluated.

Formula 3-6-6. LNAV/VNAV DA Adjustment Value

$$\Delta DA = r \times e^{\left(\frac{p \times MA_{\text{slope}} \times \tan(\theta)}{r \times [1 + MA_{\text{slope}} \times \tan(\theta)]} \right)} - r$$

Where:

p = penetration (feet) of the MA OCS

MA_{slope} = missed approach OCS slope ratio

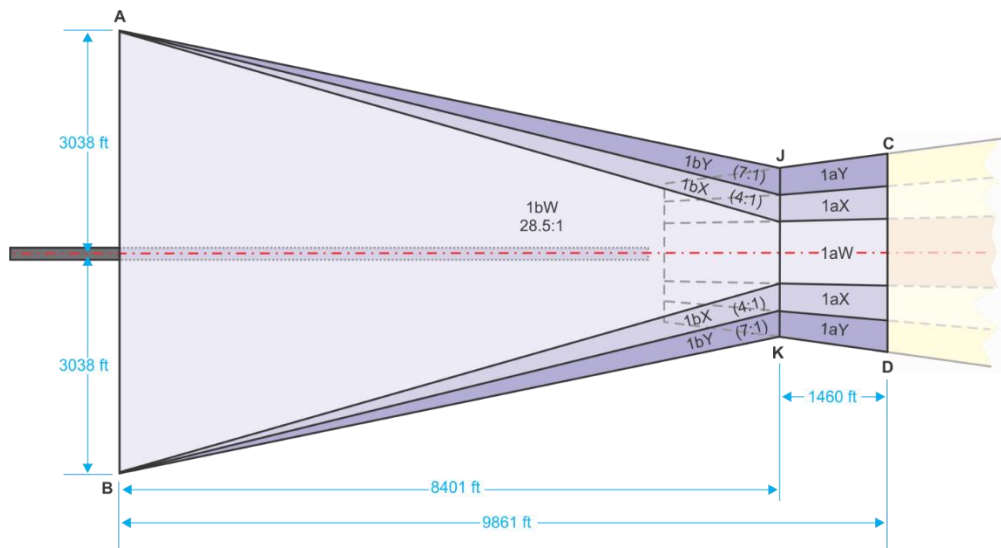
Example:

$$\Delta DA = r \times e^{\left(\frac{19.3 \times 40 \times \tan(3)}{r \times [1 + 40 \times \tan(3)]} \right)} - r$$

$$\Delta DA \approx 13.07 \text{ feet}$$

3-6-3. LPV/GLS. Section 1 begins at the base DA (line C-D) and ends at the SOC (line A-B). It accommodates height loss and establishment of missed approach climb gradient. Section 1 is centered on a continuation of the final approach track and is subdivided into sections 1a and 1b (see figure 3-6-5).

Figure 3-6-5. LPV/GLS Missed Approach Section 1 Area



a. Section 1a area/OCS elevation. Section 1a is a 1460 feet continuation of the FAS OCS beginning at the base DA point to accommodate height loss. The portion consisting of the continuation of the W surface is identified as section 1aW. The portions consisting of the continuation of the X or Y surfaces are respectively identified as section 1aX or 1aY. Calculate the width and elevation of the section 1aW, 1aX, and 1aY surfaces at any distance from LTP using the final segment formulas.

b. Section 1b areas. The section 1b area extends from line J-K at the end of section 1a for a distance of 8401 feet to line A-B. Section 1b is subdivided into sections 1bW, 1bX, and 1bY. Calculate the distance from course centerline to the boundary of these surfaces at any distance from the end of section 1a using formula 3-6-7.

(1) Section 1bW. Section 1bW extends from the end of section 1aW for a distance of 8401 feet. Its lateral boundaries splay from the width of the end of 1aW surface to a width of ± 3038 feet either side of the missed approach course at the 8401 foot-point.

(2) Section 1bX. Section 1bX extends from the end of section 1aX for a distance of 8401 feet. Its inner boundary is the outer boundary of the 1bW surface. Its outer boundary splays from the end of the 1aX surface to a width of ± 3038 feet either side of the missed approach course at the 8401-foot point.

(3) Section 1bY. Section 1bY extends from the end of section 1aY for a distance of 8401 feet. Its inner boundary is the outer boundary of the 1bX surface. Its outer boundary splays from the outer edge of the 1aY at the surface at the end of section 1a to a width of ± 3038 feet either side of the missed approach course at the 8401-foot point.

Formula 3-6-7. Section 1bW/X/Y Boundary

$$1b_{\text{boundary}} = \frac{d_{1a\text{End}} \times (3038 - 1a_{\text{boundary}})}{8401} + 1a_{\text{boundary}}$$

Where:

$d_{1a\text{End}}$ = along track distance (feet) from end of section 1a to the point of interest

$1a_{\text{boundary}}$ = perpendicular dist. (feet) from centerline to respective ending 1a outer boundary

Example:

$$1b_{\text{boundary}} = \frac{2591.8 \times (3038 - 481.06)}{8401} + 481.06$$

$$1b_{\text{boundary}} \approx 1269.90 \text{ feet}$$

c. Section 1b OCS elevations.

(1) Section 1bW. The surface rises from the elevation of the 1aW surface at the end of section 1a at a slope ratio of 28.5:1 (14.25:1 for helicopter procedures). Calculate the section 1bW surface elevation with formula 1-3-6 for input into formula 3-6-8.

(2) Section 1bX. The surface rises at a slope ration of 4:1 perpendicular to the missed approach course from the edge of the 1bW surface. Calculate the adjustment (Q) for X surface obstacles using formula 3-4-8.

(3) Section 1bY. The surface rises at a slope ratio of 7:1 perpendicular to the missed approach course from the edge of the 1bX surface. Calculate the adjustment (Q) for Y surface obstacles using formula 3-4-10.

Formula 3-6-8. Section 1b Surface Penetration

$$p = O_{\text{MSL}} - Q - 1bW_{\text{elev}}$$

Where:

O_{MSL} = obstacle MSL elevation

Q = obstacle adjustment (feet) for X or Y surface rise. Zero (0) if in the W surface

$1bW_{\text{elev}}$ = elevation (feet) of the 1bW surface abeam the obstacle

Example:

$$p = 1325.8 - 24.22 - 1282.70$$

$$p \approx 18.88 \text{ feet}$$

d. OCS penetration mitigation. To eliminate or avoid a penetration, take one or more of the following actions:

(1) Removing or adjusting the obstruction location and/or height.

(2) Raising GPA within categorical limits.

(3) Raising TCH (see paragraph 3-4-7).

(4) Adjusting DA. For a surface 1b penetration of p feet, the DA point must move Δd_{DA} (see formula 3-6-9) feet further from the LTP to raise the surface above the penetration.

Formula 3-6-9. Along-track DA Adjustment

$$\Delta d_{\text{DA}} = \frac{p \times 28.5 \times \text{FAS}_{\text{slope}}}{28.5 + \text{FAS}_{\text{slope}}}$$

Where:

p = penetration (feet) of the 1b OCS

$\text{FAS}_{\text{slope}}$ = final approach segment OCS slope ratio

Example:

$$\Delta d_{\text{DA}} = \frac{18.88 \times 28.5 \times 34}{28.5 + 34}$$

$$\Delta d_{\text{DA}} \approx 292.72 \text{ feet}$$

Section 3-7. Missed Approach Section 2

3-7-1. General.

a. Obstacle Evaluation Area (OEA). The section 2 OEA begins at the end of section 1, and splays at least 15 degrees relative to the nominal track to reach full width. Refer to individual chapters for MA section 1 information.

Figure 3-7-1. Reserved

b. Section 2 start altitude.

(1) Non-LPV/GLS.

(a) When line J-K and line A-B are coincident, the section 2 start altitude is MDA/DA as appropriate.

(b) When line J-K and line A-B are not coincident, the section 2 start altitude is airport elevation + 400 feet + adjustments (final precipitous terrain and RASS).

(2) LPV/GLS. Use formula 3-7-1 to calculate the section 2 start altitude.

Formula 3-7-1. LPV/GLS Section 2 Start Altitude

$$\text{Section 2 start altitude} = \text{DA} - \tan(\theta) \times 1460 + \frac{8401 \times 0.3048 \times 200}{1852}$$

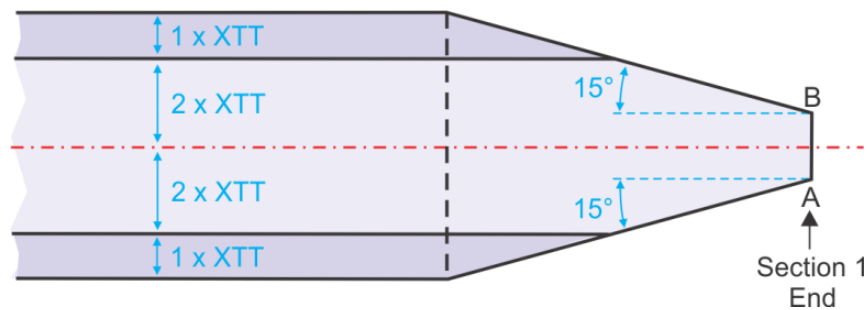
Example:

$$\text{Section 2 start altitude} = 1225 - \tan(3.1) \times 1460 + \frac{8401 \times 0.3048 \times 200}{1852}$$

Section 2 start altitude \approx 1422.45 feet

c. Obstacle Clearance Surface (OCS) slope and origin. The section 2 OCS slope (see formula 1-3-5) begins at line A-B. Where a higher than standard CG is required, apply the associated OCS from the SOC to the CG termination altitude, then revert to the default OCS.

3-7-2. Straight Missed Approach. The straight course is a continuation of the FAC. The straight MA section 2 OEA begins at the end of section 1 and splays at 15 degrees relative to the nominal track until reaching full primary and secondary width (see figure 3-7-2).

Figure 3-7-2. Missed Approach Section 2 Straight Area**3-7-3. Turning Missed Approach.****a. Early and inside turn construction.**

(1) First turn. The first MA turn may be specified at an altitude or be specified at a FB or FO fix.

(a) Turn-at-altitude or as soon as practicable. Order 8260.19, chapter 8 paragraph 8.6.6.d, “Missed approach instructions” applies except as modified by paragraph 3-5-2. A turn-at-altitude is followed by a DF leg usually ending with a DF-TF connection or holding pattern entry.

1. Turn initiation area (TIA). The TIA includes the applicable section 1 area starting at line C-D and ends where the specified turn altitude is reached (line L-L’). Construct the TIA from the end of section 1 to line L-L’ using straight MA criteria. Evaluate section 2 obstacles by applying the prescribed OCS slope along the shortest distance from the TIA boundary to the obstacle. The obstacle-based turn altitude is the sum of starting ROC and OCS elevation at the TIA boundary needed to clear section 2 obstacles. Because TIA length and the OCS elevation at the end of the TIA is dependent on aircraft altitude, establishing an obstacle-based turn altitude may be an iterative process. Calculate lengths from line C-D using formula 3-7-2, formula 3-7-3, and formula 3-7-5.

a Non-LPV/GLS TIA. Calculate the TIA length using the appropriate FSL value (see section 3-6). Where an increased CG terminates below the turn altitude, apply formula 3-7-2, otherwise apply formula 3-7-3.

Formula 3-7-2. TIA Length CG Termination Below Turn Altitude (Non LPV/GLS)

$$TIA_{\text{length}} = \frac{FSL \times r}{r + \text{alt}_b} + D_{\text{climb}_i} + D_{\text{climb}_{ii}}$$

Where:

FSL = length (NM) of the missed approach flat surface

alt_b = beginning altitude (feet) (i.e., the MDA for LNAV/LP or DA for LNAV/VNAV)

D_{climb_i} = climb distance (NM) required to reach the first CG_{term} from MDA or DA

$D_{\text{climb}_{ii}}$ = climb distance (NM) required to reach the turn altitude from the first CG_{term}

Example:

$$TIA_{\text{length}} = \frac{1.21 \times r}{r + 1900} + 4.44 + 3.26$$

$$TIA_{\text{length}} \approx 8.91 \text{ NM}$$

Formula 3-7-3. TIA Length Standard CG or CG Termination at or Above Turn Altitude (Non-LPV/GLS)

$$TIA_{\text{length}} = \frac{FSL \times r}{r + \text{alt}_b} + D_{\text{climb}}$$

Where:

FSL = length (NM) of the missed approach flat surface

alt_b = beginning altitude (feet) (i.e., either the MDA for LNAV/LP or DA for LNAV/VNAV)

D_{climb} = climb distance (NM) required to reach the turn altitude from MDA or DA

Example:

$$TIA_{\text{length}} = \frac{1.21 \times r}{r + 1900} + 6.67$$

$$TIA_{\text{length}} \approx 7.88 \text{ NM}$$

b LPV/GLS TIA. Where an increased CG terminates below the turn altitude, apply formula 3-7-4, otherwise apply formula 3-7-5.

Formula 3-7-4. TIA Length CG Termination Below Turn Altitude (LPV/GLS)

$$TIA_{\text{length}} = \frac{9861 \times 0.3048}{1852} + D_{\text{climb}_i} + D_{\text{climb}_{ii}}$$

Where:

D_{climb_i} = climb distance (NM) required to reach the first CG_{term} from SOC

$D_{\text{climb}_{ii}}$ = climb distance (NM) required to reach the turn altitude from the first CG_{term}

Example:

$$TIA_{\text{length}} = \frac{9861 \times 0.3048}{1852} + 3.79 + 3.26$$

$$TIA_{\text{length}} \approx 8.67 \text{ NM}$$

Formula 3-7-5. TIA Length Standard CG or CG Termination at or Above Turn Altitude (LPV/GLS)

$$TIA_{\text{length}} = \frac{9861 \times 0.3048}{1852} + D_{\text{climb}}$$

Where:

D_{climb} = climb distance (NM) required to reach the turn altitude from SOC

Example:

$$TIA_{\text{length}} = \frac{9861 \times 0.3048}{1852} + 6.01$$

$$TIA_{\text{length}} \approx 7.63 \text{ NM}$$

2. OEA construction after TIA (see table 3-7-1 and figure 3-7-3).

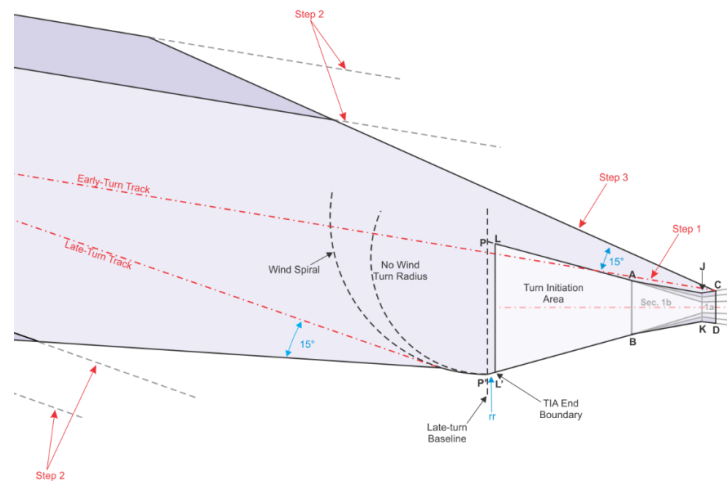
Table 3-7-1. Early/Inside Turn Tie-Back Points

Turn Angle (measured from FAC/line C-D intersection)	Tie-Back Point
≤ 75 degrees	Point C
> 75 degrees	Point D
> 165 degrees	Point P'

a *Step 1.* Construct a line (representing the early-turn flight track) from the tie-back point, to the fix.

b *Step 2.* Construct the inside turn primary and secondary OEA boundary lines parallel to this line.

c *Step 3.* From the tie-back point, construct a line slaying at 15 degrees to intersect the parallel boundary lines or leg end, whichever occurs earlier. Apply inside turn secondary areas only after the 15-degree splay line intersects the primary boundary line.

Figure 3-7-3. Early/Inside Turn Turn-at-Altitude Construction After TIA

(b) Turn-at-fix (FO or FB). TF legs are followed by a TF leg ending with holding pattern entry or TF-TF connection when the initial straight leg is less than full width. May be followed by an RF leg when the initial straight leg has reached full width, ending with an RF-TF or RF-RF connection. The recommended turn magnitude is 70 degrees; the maximum is 90 degrees. Use FB unless a FO is required for obstacle avoidance or where mandated by specific operational requirements. Use paragraph 1-2-5.d in cases where full primary and secondary width is achieved at a sufficient distance prior to the turn fix to permit construction. Otherwise, construct as follows:

1. Fix location. The first turn fix must be located on the final approach track extended. The distance to the turn fix must be sufficient to result in an early turn baseline at or beyond the end of section 1. Where the first fix must be located at the point the aircraft reaches or exceeds a specific altitude, calculate fix distance from line A-B with formula 1-3-9 using the standard or increased CG. If a fix is already established, calculate the aircraft altitude at the fix using formula 1-3-8. Calculate the aircraft SOC per paragraph 3-7-1.b.

2. Early turn baseline (line L-L'). The early turn baseline is established prior to the turn fix perpendicular to the inbound leg at the appropriate distance.

Table 3-7-2. Early turn Baseline Distance from Turn Fix

Fix Type	Distance
FB Fix	ATT + DTA*
FO Fix	ATT

*DTA = 0 for turns 10 degrees or less

3. Construction points. Inside turn construction uses points PI (primary intersection) and SI (secondary intersection). When defining these points, the leg prior to the turn fix is the inbound leg and the leg following the turn fix is the outbound leg.

a Point SI. This is the connection point for the secondary area boundary and is the intersection of line L-L' and the inbound leg secondary area boundary. When inbound leg has no secondary, points PI and SI are coincident.

b Point PI. This is the connection point for the primary area boundary and is the intersection of line L-L' and the inbound leg primary area boundary.

4. Inside turn construction. Primary/Secondary boundary expansion depends on the location of points PI and SI in relation to the outbound leg OEA boundary lines (extended when necessary) (see table 3-7-3 and figure 3-7-4, figure 3-7-5, and figure 3-7-6). Where standard expansion is suitable for one, but not both splays, find the outbound connection point for the non-standard splay abeam the standard connection point. If the table 3-7-3 expansion line results in smaller primary areas than those resulting from the 15-degree inbound leg splay, relocate point PI and/or increase the splay angle to maintain or increase the size of the inbound and outbound leg primary areas. Construct as follows:

Table 3-7-3. Inside Turn Expansion

Construction Point	External to Outbound leg OEA (both primary and secondary)	Internal to Outbound leg Secondary area	Internal to Outbound leg Primary area
Point PI	Greater of ½ Turn-angle or Inbound leg Splay	Greater of ½ Turn-angle or Inbound leg Splay	15 degrees relative outbound track
Point SI	Greater of ½ Turn-angle or Inbound leg Splay	15 degrees relative outbound track	15 degrees relative outbound track

a *Step 1.* Construct line L-L' at the appropriate table 3-7-2 distance.

b *Step 2.* Construct secondary area boundary.

1 Case 1. SI is external to the outbound leg OEA. Construct an expansion line from SI to the outbound leg secondary boundary.

2 Case 2. SI is internal to either the outbound leg secondary or primary area. In this case, an alternative construction point (point SI') may be required depending on turn magnitude. Point SI' is defined as the intersection of line L-L' and the outside-turn secondary area boundary. Construct the expansion line from either point SI or SI' to the outbound leg secondary boundary whichever results in the larger area. When the expansion line reaches the outbound secondary boundary before reaching the point of intersection of the inbound and outbound leg secondary boundaries, no expansion is required and the area is a simple connection of the inbound and outbound leg secondary boundaries.

c *Step 3.* Construct the primary area boundary. Construct an expansion line from point PI to the outbound leg primary boundary. When the *Step 2* expansion line reaches the outbound leg primary boundary before reaching point of intersection of the inbound and outbound leg primary boundaries, no expansion is required and the area is a simple connection of the inbound and outbound leg primary boundaries.

Figure 3-7-4. Early/Inside Turn, First Turn, Turn at Fix: PI and SI Collocated and External to Outbound Primary and Secondary

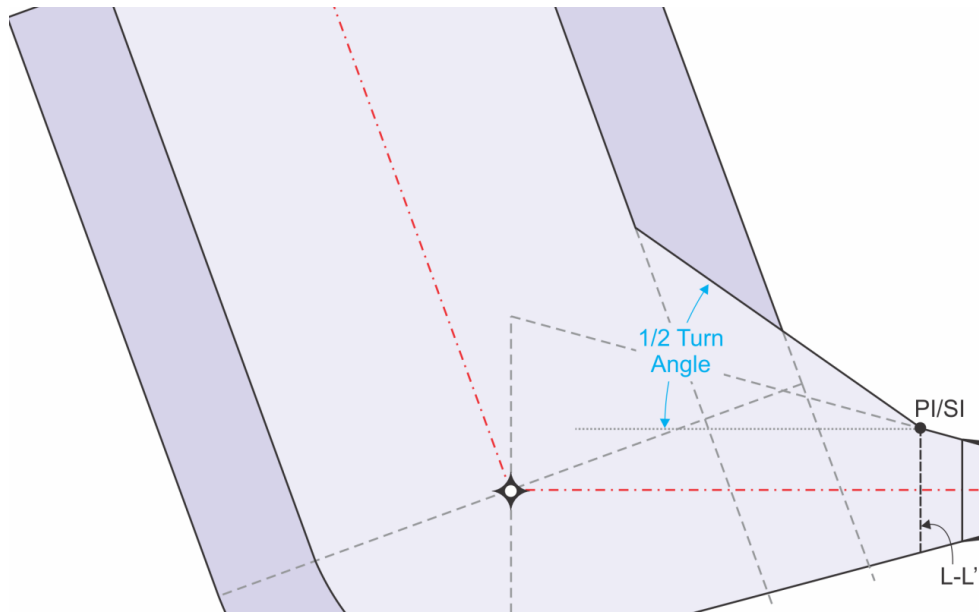
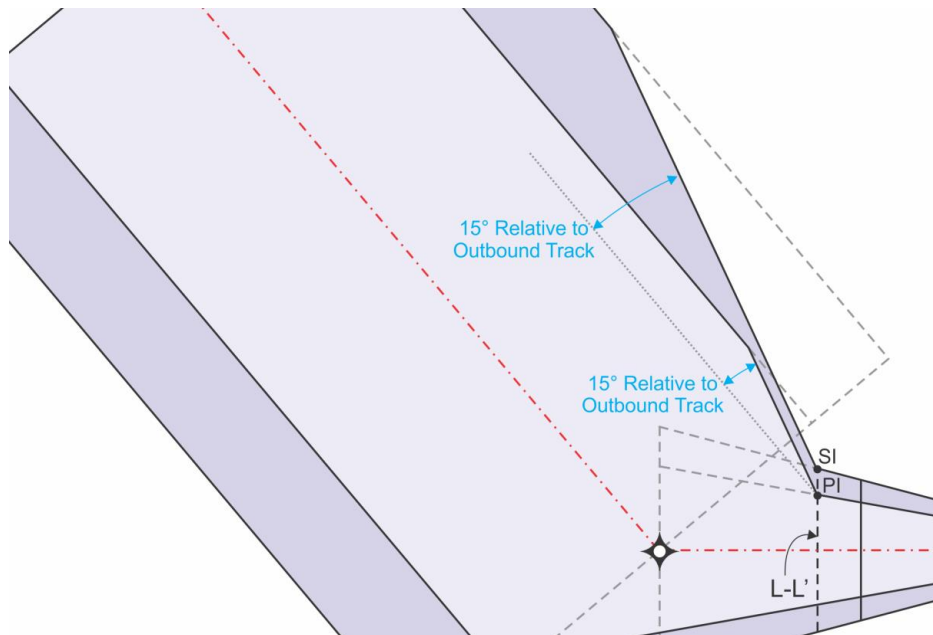
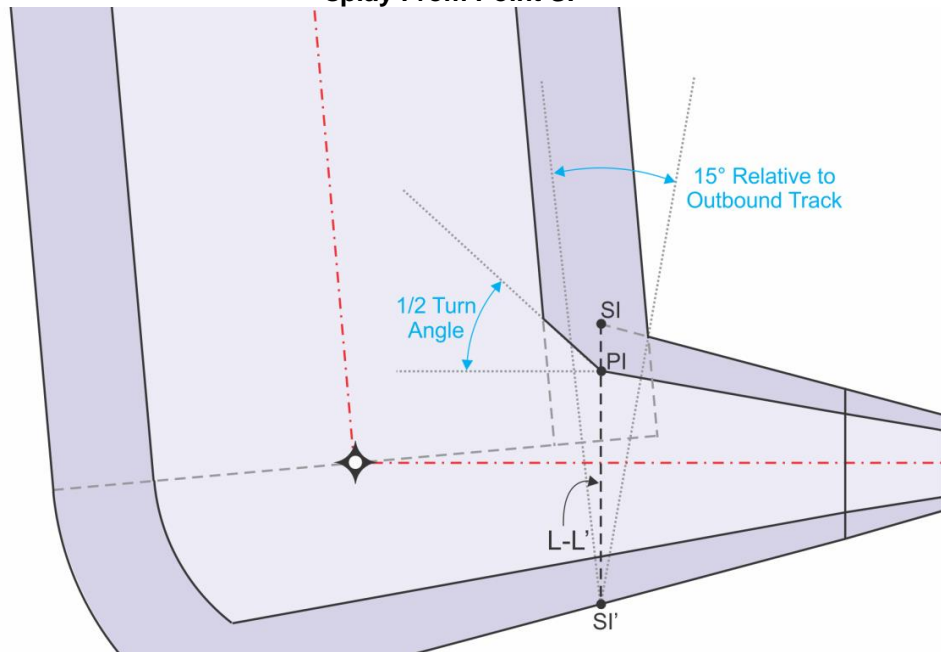


Figure 3-7-5. Early Inside Turn, First Turn, Turn at Fix: PI and SI internal to Outbound Primary



**Figure 3-7-6. Early/Inside Turn, First Turn, Turn at Fix:
PI and SI internal to Outbound Secondary,
splay From Point SI'**



(2) Second turn. The second MA turn is specified at either a FB or FO fix (see figure 3-7-7).

(a) DF-TF. This applies to a DF-TF connection following a turn-at-altitude.

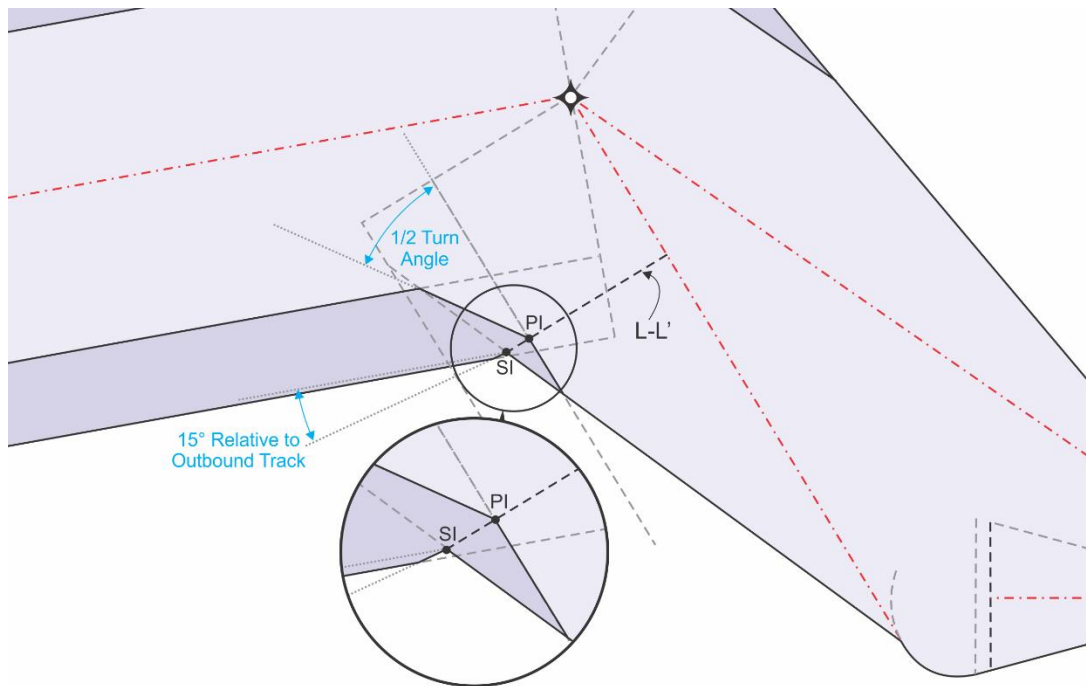
1. Construct in accordance with paragraph 3-7-3.a(1)(b) with the following exceptions:

a Paragraph 3-7-3.a(1)(b)1 does not apply.

b Line L-L' is established perpendicular either on the early- or late-turn track, whichever is on the inside of the turn.

c When full primary- and secondary-width is reached at the early-turn baseline (L-L') construct in accordance with paragraph 1-2-5.d.

(b) Other than DF-TF connections. Construct in accordance with paragraph 1-2-5.d.

Figure 3-7-7. Early/Inside Turn, Second Turn, Turn at FB Fix DF-TF Connection

b. Late and outside turn construction.

(1) Wind spiral application. Wind Spiral (WS) construction applies to late and outside turns for turn-at-altitude, turn-at-fix (FO) for the first turn, and DF-TF (FO) for the second turn. See paragraph 1-3-1.g for design parameters.

(a) WS number and baselines. Baseline locations for WS construction may be dependent on Reaction and Roll distance (see formula 1-2-12 and table 3-7-4).

Table 3-7-4. Wind Spiral Number and Distance to Late Turn Baseline

WS/Baseline Parameters	Turn-at-Fix (FO)	Turn-at-Altitude
WS Number	1 or 2	1, 2, or 3
WS1 & WS2 Distance Late-Turn Baseline from turn fix/point	$ATT + D_{rr}$	$TIA + D_{rr}$
WS3 Baseline	Not applicable	Parallel to line C-D

1. First turn baseline. For first turn construction, the late-turn baseline (line P-P') marks the construction line for wind spirals. Line P-P' is located at the table 3-7-4 distance with points P and P' placed at the continuation of the inbound leg's outer boundary (see paragraph 1-3-1.g(4)). The no-wind turn radius for the outside turn wind spiral (WS1) and inside turn wind spiral (WS2) will be measured perpendicular to the inbound track from their respective point P. If a third wind spiral is used (WS3), the baseline will be parallel to line C-D with the no-wind turn radius measured inward from point C.

2. Second turn baseline (see figure 3-7-15). To accommodate both inbound tracks, the second MA turn construction uses two WS baselines, line P'-P'' for WS1 and line P-P' for WS2. Each late turn baseline is oriented perpendicular to the early- and late-turn tracks at the table 3-7-4 distance. The baseline for the inbound track nearer the outside-turn boundary is designated line P'-P'', with point P'' placed on the extended outer boundary and point P' placed at the no-wind turn radius inward along the baseline from point P''. The baseline for the inbound track nearer the inside-turn boundary is designated line P-P', with point P' placed on inbound track and point P placed on the extended outer boundary. The no-wind turn radius for the inside turn wind spiral is measured inward along the baseline from point P.

(b) WS connection point and outside turn OEA boundary (see figure 3-7-8). Each WS has various connection options along its path, which predicate the outside turn OEA boundary. The chosen connection must provide the most reasonably conservative track and protection area. Where excessive splay is required to reach full-width protection, consider lengthening the leg, restricting the speed, category, etc. to avoid protection and/or construction difficulties. Consider full-width protection to exist at the fix where the splay line is tangent to a full-width-radius circle about the fix (see figure 3-7-11).

1. Turn-at-altitude connection point. For turn-at-altitude, the 15-degree or greater splay line that joins the outbound leg outer boundaries may originate from the WS/direct-to-fix tangent point (point 1), the WS to WS tangent line origin (point 2) or the WS to WS tangent line end (point 3). Where the turn angle is ≤ 105 degrees, or the divergence angle between the WS to WS tangent line and the direct-to-fix line is ≤ 15 degrees, apply the splay line from the WS to WS tangent line origin. DF secondary areas begin/exist only where full width primary exists.

2. Turn-at-fix (FO) Connection Point. For turns at a fix, the connection point and the OEA boundary is dependent on the WS boundary relative to the outbound leg OEA (extended). The connection point is the point of tangency on the WS in both cases. When the WS is not contained in the outbound leg OEA, the outside boundary is a 30-degree converging line relative to the outbound course (see figure 3-7-9). When the WS is contained in the outbound leg OEA, the outside boundary is a 15-degree or greater splaying line relative to the outbound course from the WS/outbound leg parallel point until reaching the outbound leg boundaries (see figure 3-7-10).

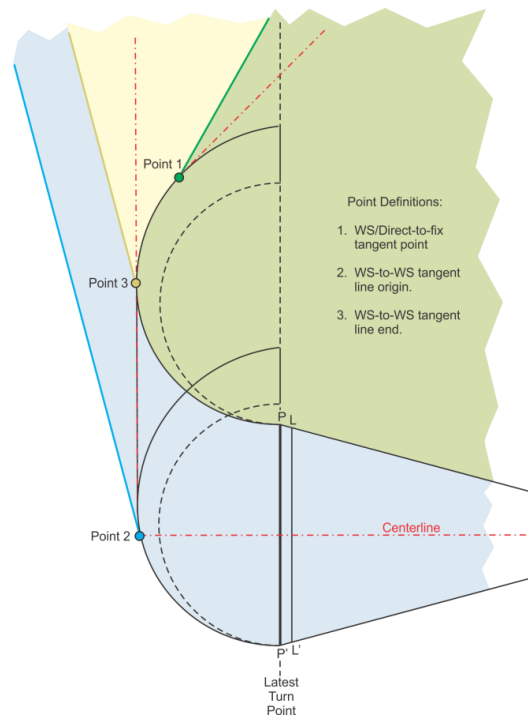
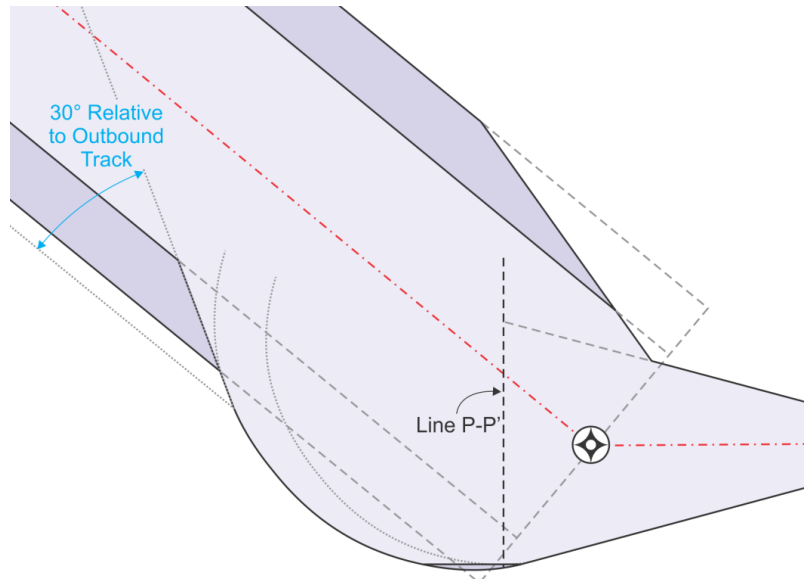
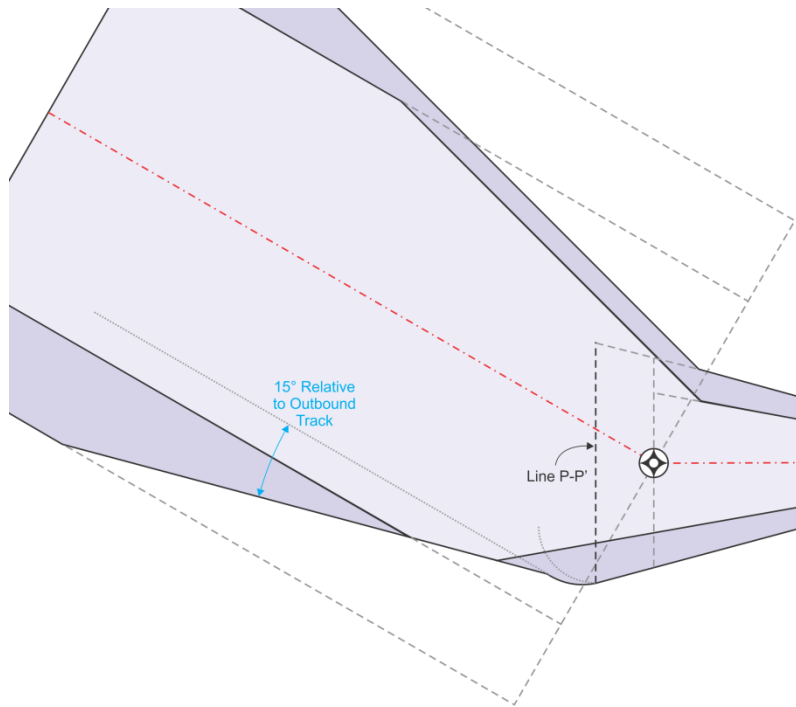
Figure 3-7-8. Late/Outside Turn, WS Connection Points and Outside Turn OEA Boundary**Figure 3-7-9. Late/Outside Turn, Wind Spiral, WS Not Contained in Outbound OEA**

Figure 3-7-10. Late/Outside Turn, Wind Spiral, WS Contained in Outbound OEA

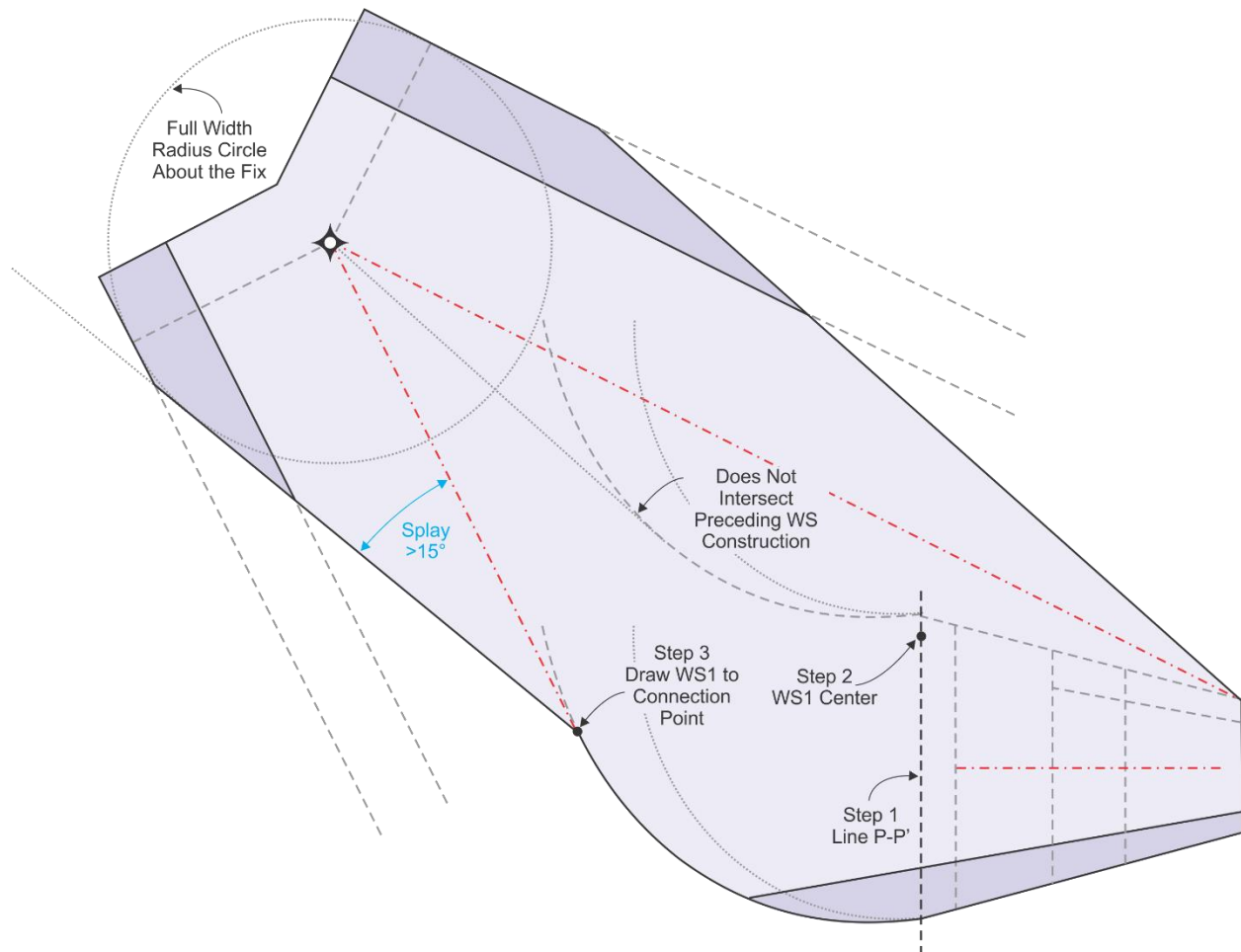
(c) Determining multiple WS necessity. To determine multiple wind spiral necessity, construct the additional WS in the direction of turn from its prescribed location to its connection point. Where the additional WS intersects the preceding WS construction (including the connecting and expansion lines), connect the wind spirals with a tangent line that is parallel to the WS center points. Otherwise, revert to the previous WS construction.

(2) First turn.

(a) Turn-at-altitude or turn-at-fix (FO).

1. *Step 1.* Construct the late-turn baseline (P-P') perpendicular to the straight missed approach track at the late-turn-point.

2. *Case 1.* Small turns using one WS (see figure 3-7-11).

Figure 3-7-11. Late/Outside Turn, First Turn, Case 1, Turn-at-Altitude or FO Fix with 1 WS

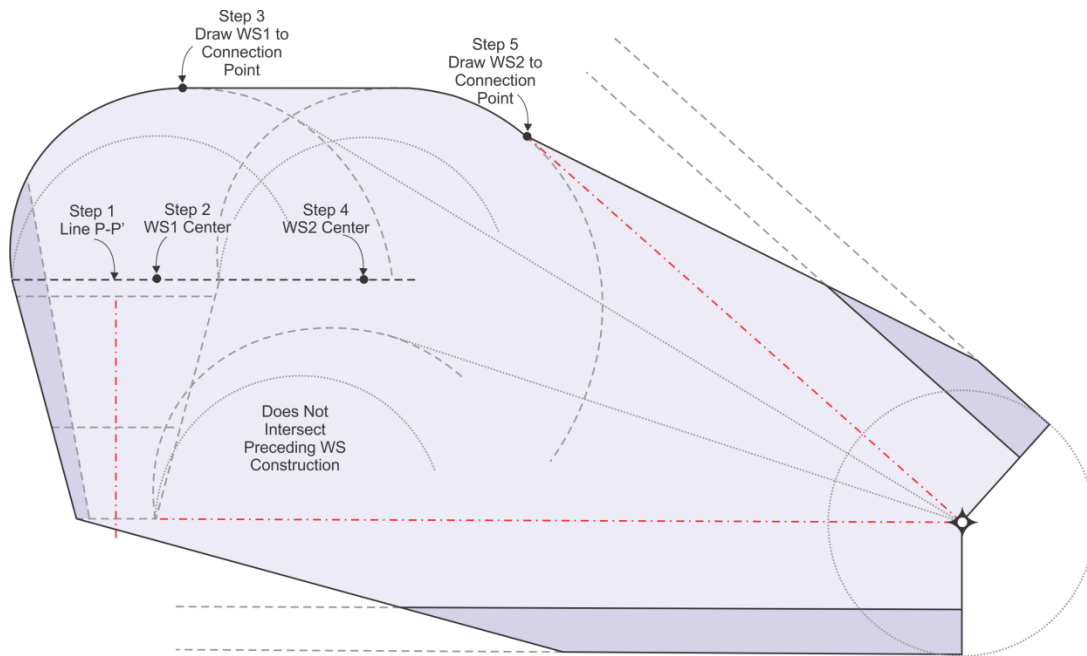
a Step 2. Locate the WS1 center on line P-P' at no-wind turn radius distance from point P'.

b Step 3. Construct WS1 from the outside-turn point in the direction of turn until reaching its connection point.

3. Case 2. Turns nearing or greater than 90 degrees using more than one WS (figure 3-7-12). Apply Steps 1-3 above, then:

a Step 4. Locate the WS2 center on P-P' at no-wind turn radius distance from point P.

b Step 5. Construct WS2 from the inside turn point in the direction of turn until reaching its connection point.

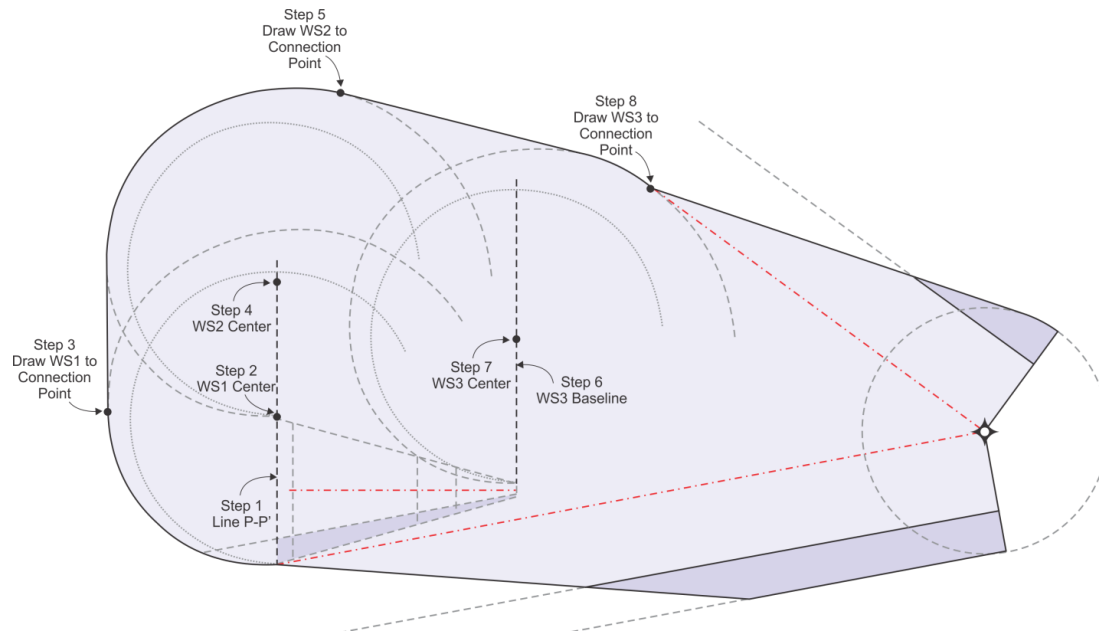
Figure 3-7-12. Late/Outside Turn, First Turn, Case 2, Turn-at-Altitude or FO Fix with 2 WS

4. *Case 3. Turns nearing or greater than 180 degrees using more than two WS (see figure 3-7-13). Apply Steps 1-5 above, then:*

a *Step 6. Construct the WS3 baseline perpendicular to the straight missed approach track along line C-D extended toward the turn side.*

b *Step 7. Locate the WS3 center on the baseline at no-wind turn radius distance from point C.*

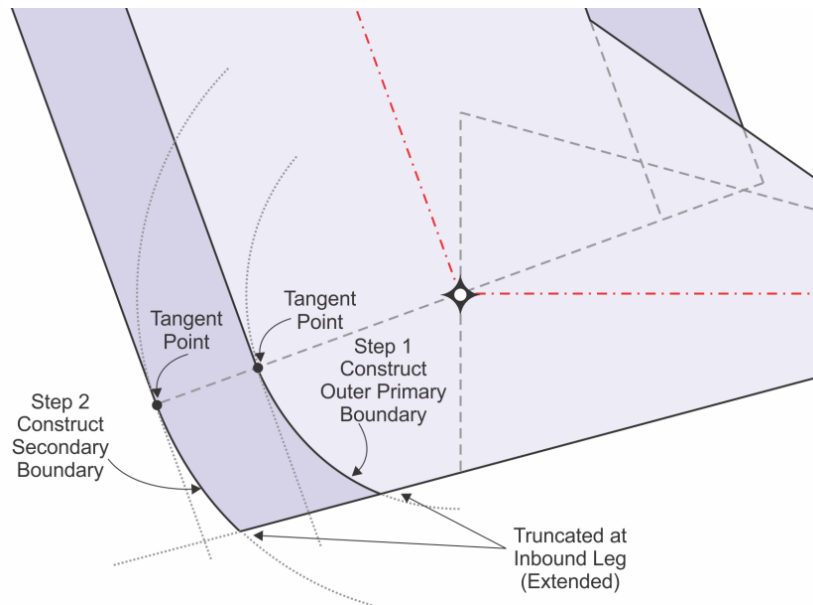
c *Step 8. Construct WS3 from this point in the direction of turn until reaching its connection point.*

Figure 3-7-13. Late/Outside Turn, First Turn, Case 3, Turn-at-Altitude or FO Fix with 3 WS

(b) Turn-at-fix (FB) (see figure 3-7-14).

1. *Step 1.* Construct the outer primary boundary using a radius equal to $2 \times$ segment XTT centered on the plotted fix position, truncated at the inbound leg extended outer boundary until tangent to the outbound leg primary boundary.

2. *Step 2.* Construct the secondary boundary using a radius equal to $3 \times$ segment XTT centered on the plotted fix position, truncated at the inbound leg extended outer boundary until tangent to the outbound leg secondary boundary.

Figure 3-7-14. Late/Outside Turn, First Turn, Turn-at- Fix, FB

(3) Second turn.

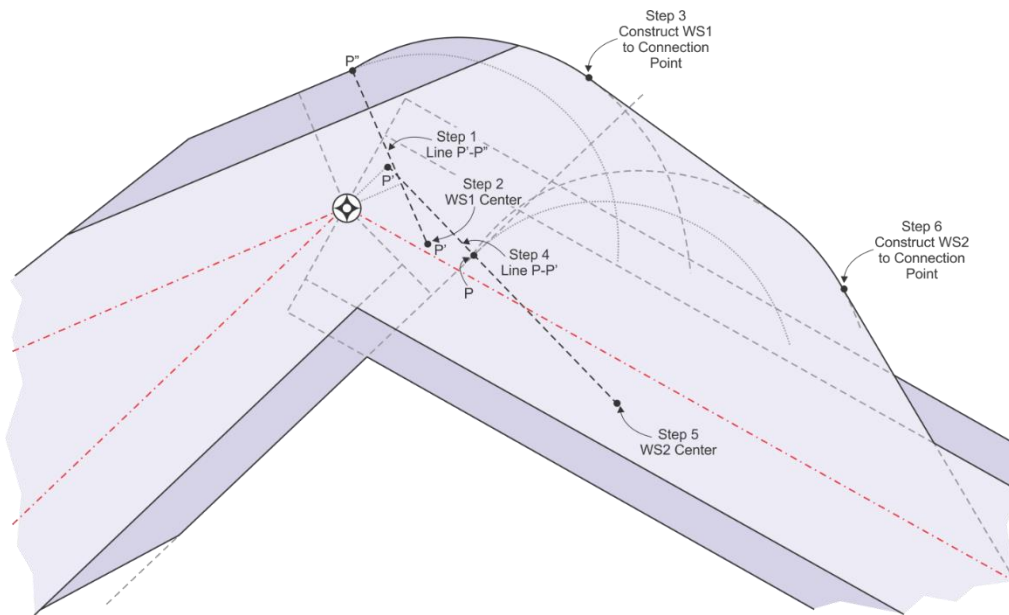
(a) Turn-at-fix (FO) (see figure 3-7-15).

1. *Step 1.* Construct the WS1 baseline, (line P'-P'') perpendicular to the DF track nearer the outside of the DF-TF turn, at the late turn point.

2. *Step 2.* Locate the WS1 center on line P'-P'' at no-wind turn radius distance from point P."

3. *Step 3.* Construct WS1 from the outside turn point in the direction of turn until reaching its connection point.

4. *Step 4.* Construct the WS2 baseline, (line P-P') perpendicular to the DF track nearer the inside of the DF-TF turn, at the late turn point.

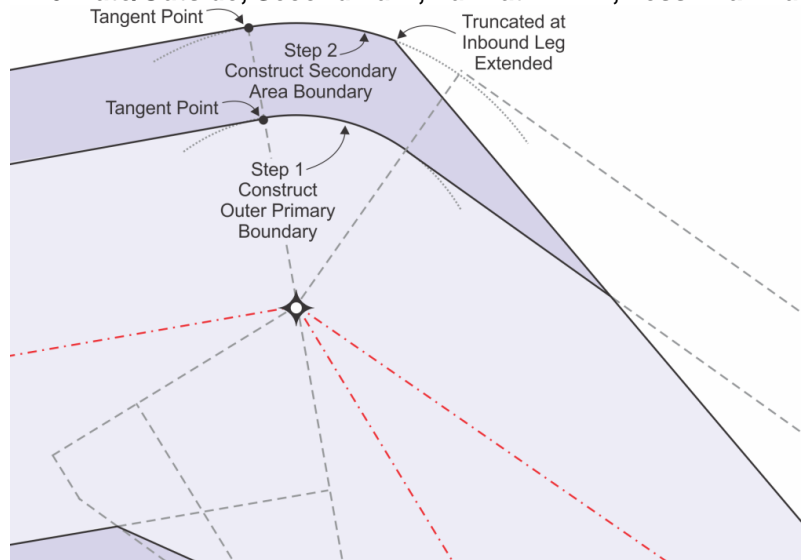
Figure 3-7-15. Late/Outside Turn, Second Turn, Turn-at Fix, FO

5. *Step 5.* Locate the WS2 center on line P-P' at no-wind turn radius distance from point P.

6. *Step 6.* Construct WS2 from the inside turn point in the direction of turn until reaching its connection point.

(b) Turn-at-fix (FB).

1. When the inbound leg outside boundary is less than full primary and secondary width reached perpendicular to the fix, construct in accordance with paragraph 3-7-3.b(2)(b) (see figure 3-7-16).

Figure 3-7-16. Late/Outside, Second Turn, Turn at FB Fix, Less Than Full OEA Width

2. When the inbound leg outside boundary is full primary and secondary width perpendicular to the fix, construct in accordance with paragraph 1-2-5.d.

3-7-4. Obstacle Clearance Surface. Apply a sloping OCS (see paragraph 1-3-1.g and paragraph 3-7-1.c). Calculate the aircraft SOC in accordance with paragraph 3-7-1.b. Where multiple measurements are required (e.g., a point of interest is equidistant from multiple primary boundaries, it lies along perpendiculars from multiple primary boundaries, etc.), apply the most adverse result from each of the combined primary/secondary measurements.

a. Primary area.

(1) Straight MA/TIA. Measure and apply the OCS slope from line A-B along the track distance to the point or abeam the point of interest (see figure 3-7-17).

(2) Turn-at-altitude (single and multiple legs). Apply the OCS slope along the shortest primary area distance from the TIA boundary to the point of interest (see figure 3-7-17 and figure 3-7-18, and figure 3-7-19).

(3) Turn-at-a-fix (single and multiple legs). Apply the OCS slope from line A-B (parallel to track) to L-L', then along the shortest primary distance to the point of interest (see figure 3-7-20). A-RNP obstacle distance is measured as the along-track distance from line A-B to a point abeam the obstruction.

Note: The shortest primary area distance is the length of the shortest line kept within the primary area that passes through the early-turn baseline of all preceding legs.

Figure 3-7-17. OCS, Primary Area, Turn-at-Altitude, Single leg, Small Turn

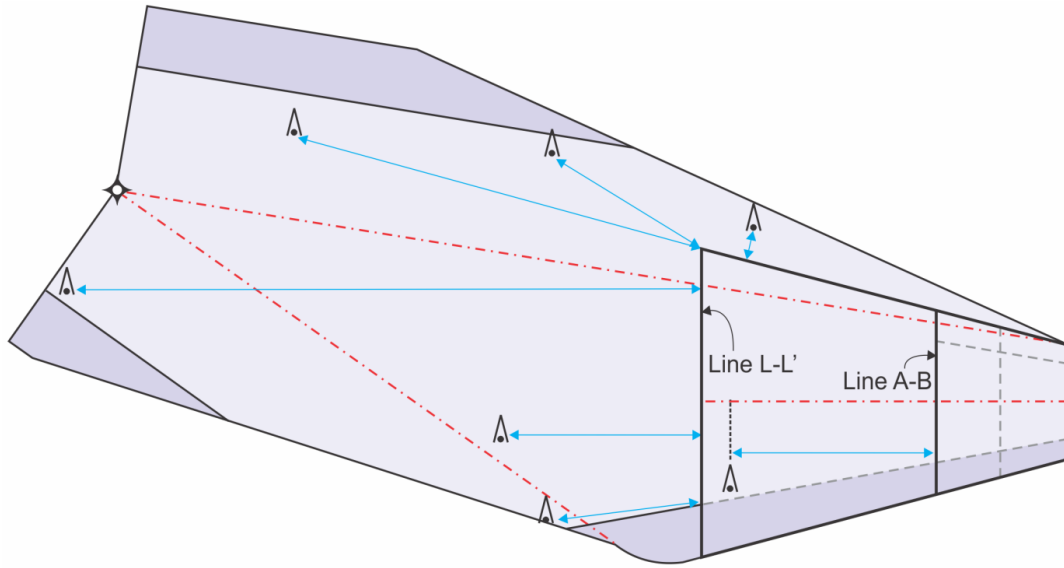


Figure 3-7-18. OCS Primary Area, Turn-at-Altitude, Single Leg, Large Turn

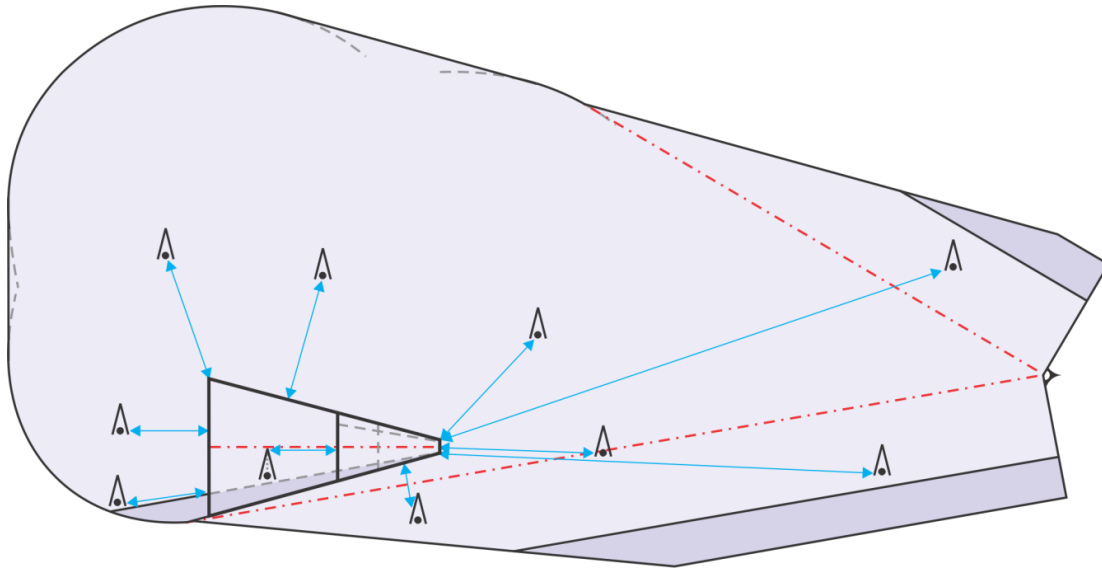


Figure 3-7-19. OCS, Primary Area, Turn-at-Altitude, Multiple Legs

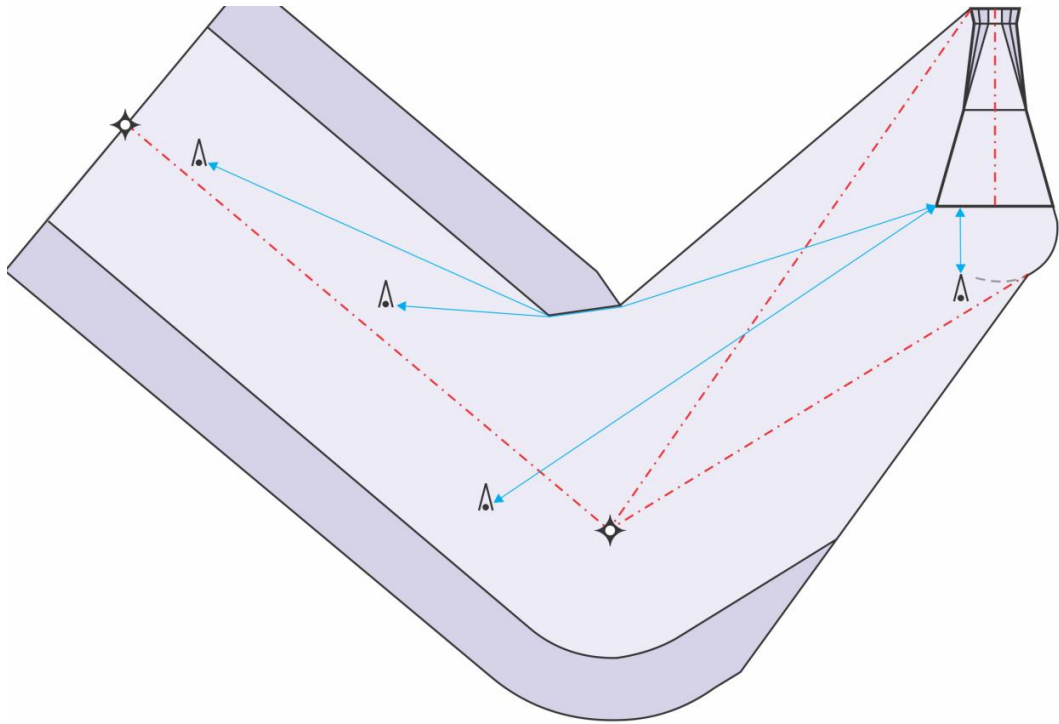
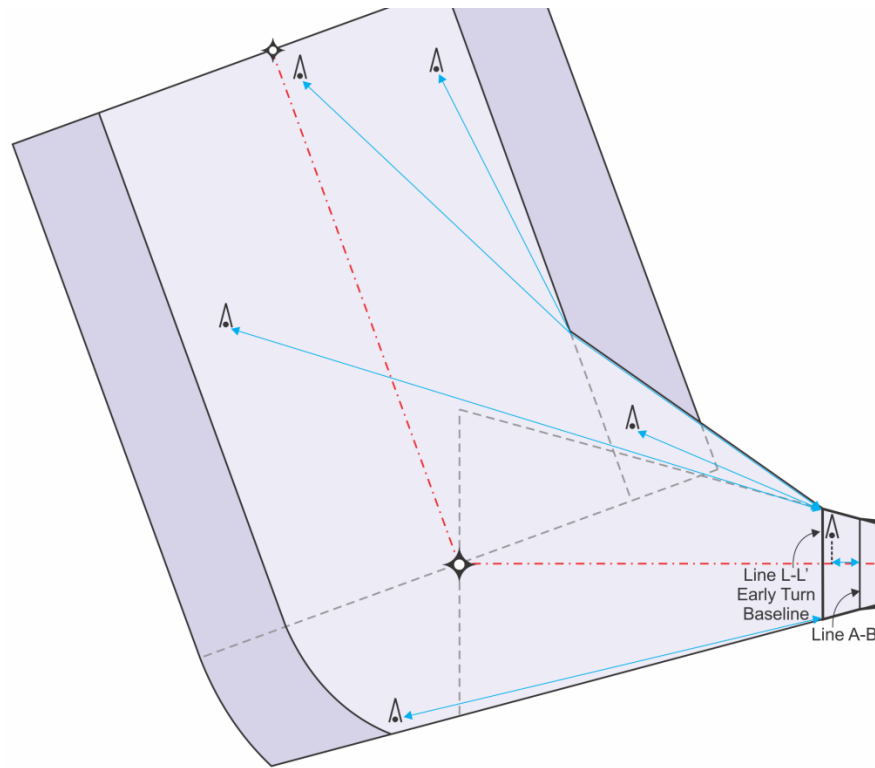
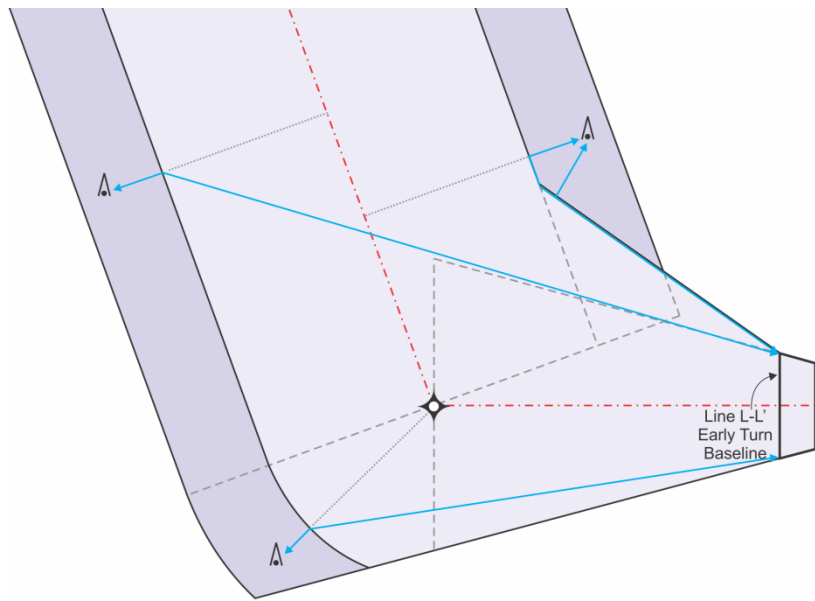


Figure 3-7-20. OCS, Primary Area, Turn-at-Fix

b. Secondary area. For secondary areas, calculate the primary OCS elevation as stated then apply a 12:1 OCS slope along the shortest secondary distance to the point of interest. In straight legs, this is normally perpendicular to the nominal track. In expansion areas, the slope rises in a direction perpendicular to the primary boundary (arc, diagonal corner-cutter, etc.) (see figure 3-7-21).

Figure 3-7-21. OCS, Secondary Area



Chapter 4. RNAV (RNP) Approach

Section 4-1. General Criteria

4-1-1. Concept and Design. This chapter applies to procedures covered by the RNP AR APCH NavSpec. RF legs are preferred for turns but TF legs may be used for procedure simplification or improved flyability (e.g., a small magnitude turn where the closeness of the waypoints leads to difficulty in charting). FO fixes that require turn construction are not authorized in RNP AR APCH segments.

a. OEA construction. Apply paragraph 1-2-5 with the following exceptions.

(1) Apply the largest RNP for the flight phase from table 1-2-1 unless a smaller value is required to achieve a desired ground track or is operationally required.

(2) RNP changes must occur at a fix and are linear (i.e., do not splay/taper 30 degrees relative to the course). RNP may be increased or decreased as needed prior to the FAS. RNP changes must not occur in the FAS. After crossing the LTP/FTP, RNP values may only increase. See figure 1-2-1, figure 1-2-2, and figure 1-2-3 for illustrations of RNP changes.

(3) The length of any leg between the initial approach fix and the missed approach point may be reduced to not less than 0.2 NM (regardless of RNP) when there are no more than three waypoints within 1 NM along-track distance and there is no leg shorter than 1 NM prior to a change in RF turn direction. No turns (> 0.03 degrees) allowed when applying this to TF-TF connections. Avoid establishing reduced length legs in sequence to avoid negative impact on chart/display readability.

b. Use of NAVAIDS. Do not incorporate a VOR/DME or VORTAC into an RNP AR procedure if the geodetic coordinates of the VOR and DME source are not identical to 0.01 seconds, regardless whether the facilities are considered collocated.

c. Lines of minima. No more than four lines may be established. Circling minimums are not established.

(1) Establish a default line. The default line should be based on RNP 0.30 in the FAS and either a default RNP AR APCH [see paragraph 4-3-1.a(1)] or RNAV 1 [see paragraph 4-3-1.a(2)] missed approach. When an RNP 0.30 line isn't possible, then the default line should be the largest value possible. See Order 8260.19, section 8-6.

(2) When the HAT value of the default line is ≥ 300 or no-lights visibility ≥ 1 SM, then:

(a) Additional lines may be established when at least a 50-foot reduction in HAT or $\frac{1}{4}$ SM reduction in visibility can be achieved using one or more of the following (as appropriate);

1. Reduced FAS RNP (values < 0.30 but ≥ 0.10).

2. Reduced RNP MAS [see paragraph 4-3-1.a(3)].

3. MAS CG.

(b) Additional lines based on reduced RNP may also be established to meet operational requirements, e.g., to achieve track-to-airspace or track-to-track separation when these minima reductions cannot be achieved.

(c) When applying paragraph 4-1-1.c(2)(a) or 4-1-1.c(2)(b), a default line must be still be published.

(3) A line based on reduced FAS RNP may be established to achieve minimum DA to final rollout point (FROP) distance. A default line is not required in this case.

4-1-2. Feeder Segment. Construct feeder segments as described in chapter 2 using one or more TF or RF legs. The NavSpec may be RNAV 1 (STAR/feeder/arrival flight phase with associated XTT of 1.00). When RF is used, the NavSpec is RNP 1 (STAR/feeder/arrival flight phase with associated XTT of 1.00). Optional NavSpec may be A-RNP (STAR/feeder/arrival flight phase with associated XTT of 2.00 or 1.00 as appropriate). Use an effective XTT of 2.00 where applicable per table 1-2-1, footnote 2. Secondary areas apply, except for A-RNP.

4-1-3. Initial Segment. Construct initial segments as described in chapter 1 using one or more TF or RF legs. The NavSpec is RNP AR APCH (initial flight phase with associated XTT of 1.00 - 0.10). Secondary areas do not apply.

4-1-4. Intermediate Segment. Construct intermediate segments as described in chapter 1 using one or more TF or RF legs. The NavSpec is RNP AR APCH (intermediate flight phase with associated XTT of 1.00 - 0.10). Secondary areas do not apply. FB turns at the PFAF are limited to a maximum of 15 degrees. See appendix D for transition to ILS/GLS/LPV.

Section 4-2. Final Approach Segment (FAS)

4-2-1. General. The NavSpec is RNP AR APCH (final flight phase with associated XTT of 0.30 - 0.10). Secondary areas do not apply. RNP AR APCH ops are 3D procedures in which the aircraft provides the pilot with lateral and vertical path guidance and deviation information throughout the procedure. During the FAS, the aircraft provides either baro-VNAV vertical guidance or SBAS LPV lateral and vertical guidance. The minimum HAT value is 250 feet.

a. Vertical path. Chapter 1 applies with the following additional guidance determining a PFAF on an RF leg.

(1) Use formula 4-2-1 and formula 4-2-2 to determine the PFAFs Cartesian coordinates relative to the LTP/FTP. The location of the PFAF may be calculated geodetically from these Cartesian values.

(a) *Step 1.* Determine the flight track distance from LTP/FTP to PFAF using formula 1-3-3.

(b) *Step 2.* Determine the distance from LTP/FTP to the FROP (see paragraph 4-3-5).

(c) *Step 3.* Subtract *Step 2* from *Step 1* to calculate the distance around the arc to the PFAF from the FROP. Use formula 1-2-4 to determine number of degrees of arc; conversely, use formula 1-2-3 to convert degrees of arc to length.

Formula 4-2-1. PFAF on an RF Leg, Cartesian Coordinate “X” Value

$$X = d_{\text{FROP}} + R \times \sin(\alpha)$$

Where:

d_{FROP} = distance from LTP to final rollout point

α = degrees of arc

R = arc radius

Note: This formula works with any unit so long as d_{FROP} and R share the same unit.

Example:

$$X = 9420.55 + 19079 \times \sin(98.9)$$

$$X \approx 28269.84$$

Formula 4-2-2. PFAF on an RF Leg, Cartesian Coordinate “Y” Value

$$Y = R - [R \times \cos (\alpha)]$$

Where:

α = degrees of arc

R = arc radius

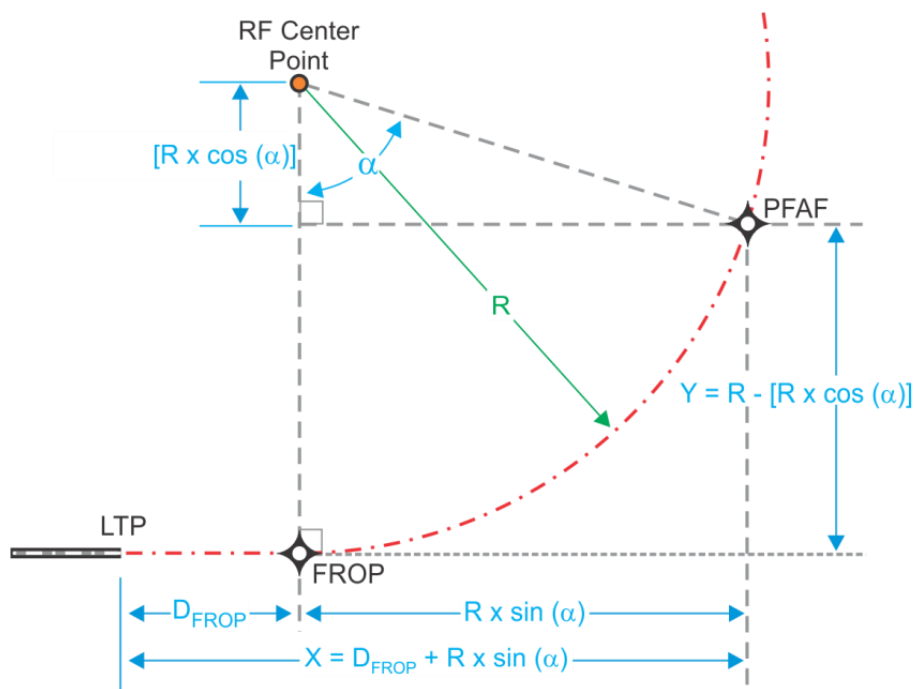
Note: This formula produces a value with the unit as R.

Example:

$$Y = 19079 - [19079 \times \cos (98.9)]$$

$$Y \approx 22030.72$$

Figure 4-2-1. Calculating Cartesian Coordinates, PFAF on an RF leg



b. Restrictions. The true vertical path provided by baro-VNAV is influenced by temperature variations; i.e., during periods of cold temperature, the effective glidepath may be lower than published and during periods of hot weather, the effective glidepath may be higher than published. Because of this phenomenon, minimum and maximum temperature limits (for aircraft that are not equipped with temperature compensating systems) are published on the approach chart. Additionally, RNP AR approach procedures are not authorized where a remote altimeter setting source (RASS) is in use. See paragraph 3-3-4.a(1) to determine the critical temperatures and ΔISA_{LOW} for vertical error budget (VEB) calculations.

c. Precipitous terrain. Do not apply the 10 percent HAT increase identified in Order 8260.3, paragraph 3-2-2.b to RNP AR procedures.

4-2-2. Alignment.

a. The optimum alignment is a TF leg straight in from PFAF to LTP on runway centerline extended. If necessary, the TF course may be offset by up to three degrees. Where the course is offset, it must cross runway centerline extended at least 1500 feet (5200 feet maximum) out from LTP.

b. Turns in the FAS and final rollout point (FROP). TF-TF turns are not allowed in the FAS. Where turns are necessary in the FAS an RF leg must be established. The RF leg must terminate and be followed by a TF leg meeting FAS alignment at or prior to reaching the minimum FROP distance. The *minimum* FROP distance is the greater of the point on the vertical path 500 feet above LTP/FTP elevation (rounded to the nearest foot) or the point in the FAS at distance equal to 0.5 NM prior to the DA point. If an RF leg is the last leg in the intermediate segment, the PFAF must meet the minimum FROP distance. Use formula 4-2-3 to determine the minimum FROP distance.

Formula 4-2-3. Minimum FROP Distance

$$d_{500} = \frac{500 - TCH}{\tan(\theta)}$$

$d_{FROP} = \text{greater of } d_{500} \text{ or } 0.5 \text{ NM} + \text{distance from LTP/FTP to DA}$

Example:

$$d_{500} = \frac{500 - 47}{\tan(3)}$$

$$d_{500} \approx 8643.75 \text{ feet}$$

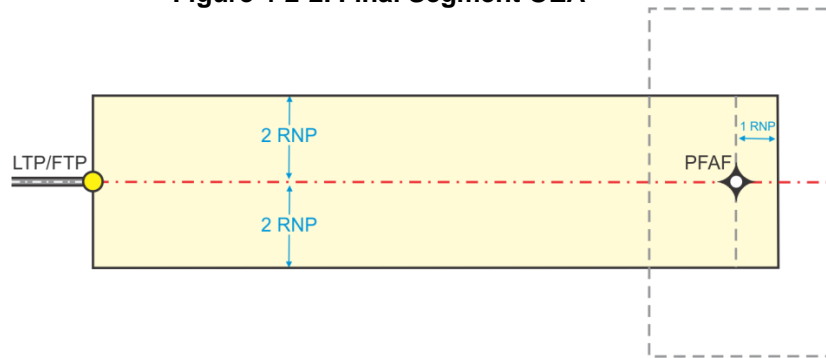
$$d_{FROP} \approx 8643.75 \text{ feet}$$

4-2-3. Area.

a. Length. The final segment OEA begins 1 x ATT prior to the PFAF and extends to the LTP/FTP.

b. Width. The final segment OEA width is 2 x XTT either side of designed path (see figure 4-2-2).

Figure 4-2-2. Final Segment OEA



4-2-4. Obstacle Clearance. An OCS based on the VEB is applied. The VEB origin varies depending on whether the procedure is designed for aircraft with wingspans ≤ 262 feet (wide body) or aircraft with wingspans ≤ 136 feet (narrow body). Design for wide body aircraft by default. When the DA can be reduced by at least 50 feet or visibility reduced by $\frac{1}{4}$ mile or where wide body operations are not anticipated (e.g., where the airport/runway infrastructure cannot support wide body aircraft) the approach may be designed for narrow body aircraft. In either case, the procedure must be properly annotated (see Order 8260.19, paragraph 4-6-10). The VEB origin also varies depending on whether evaluating a TF or RF leg (with corresponding changes to the OCS height when both leg types are in the FAS).

a. Calculating the VEB. Total VEB is calculated by adding bias errors to four thirds times the standard deviation variations combined via the root sum square method. This is completed for both the PFAF and 250 feet above TDZE to determine the OCS slope and origin.

(1) Bias errors are:

(a) Body geometry error accounts for the low point of the aircraft below the altimeter reference point. This may be the landing gear for aircraft on straight legs, but it could be a wing tip for aircraft on RF legs.

1. Narrow body (BG_{NB}). Wingspan less than or equal to 136 feet.

a Straight legs: fixed at 15 feet

b RF legs: greater of 15 feet or $\frac{136}{2} \times \sin(\phi)$

2. Wide body (BG_{WB}). Wingspan less than or equal to 262 feet.

a Straight legs: fixed at 25 feet

b RF legs: greater of 25 feet or $\frac{262}{2} \times \sin(\phi)$

(b) International Standard Atmosphere temperature Deviation (ISAD) is attributed to temperature induced altimeter error at the specified critical low temperature in relation to the airport ISA (ΔISA_{Low}) (see paragraph 4-2-1.b).

(c) Example bias error calculations.

$$BG_{NB} = \frac{136}{2} \times \sin(\phi)$$

$$BG_{NB} = \frac{136}{2} \times \sin(18)$$

$$BG_{NB} \approx 21.01 \text{ feet}$$

$$ISAD_{PFAF} = \frac{(PFAF_{alt} - LTP_{elev}) \times \Delta ISA_{Low}}{288 + \Delta ISA_{Low} - 0.5 \times 0.00198 \times PFAF_{alt} - (4500 - 1200) \times -20}$$

$$ISAD_{PFAF} = \frac{288 - 20 - 0.5 \times 0.00198 \times 4500}{(4500 - 1200) \times -20}$$

$$ISAD_{PFAF} \approx -250.43 \text{ feet}$$

$$ISAD_{250} = \frac{(TDZE + 250 - LTP_{elev}) \times \Delta ISA_{Low}}{288 + \Delta ISA_{Low} - 0.5 \times 0.00198 \times (TDZE + 250) - (1202 + 250 - 1200) \times -20}$$

$$ISAD_{250} = \frac{288 - 20 - 0.5 \times 0.00198 \times (1202 + 250)}{(1202 + 250 - 1200) \times -20}$$

$$ISAD_{250} \approx -18.91 \text{ feet}$$

(2) Standard deviation variations are:

(a) Actual Navigation Performance Error (ANPE). An aircraft is expected to be within a radius equal to the RNP for the leg 95% of the time. This additive accounts for the 5th percentile that may be operating outside of RNP.

(b) Waypoint Precision Error (WPR) contains the horizontal position error associated with the coordinate resolution stored in the database and Flight Management System (FMS) computational accuracy.

(c) Flight Technical Error (FTE), fixed at 75 feet, is a measure of the pilot and/or autopilot's ability to track the intended flight path.

(d) Altimetry System Error (ASE) is the difference between the pressure altitude displayed to the pilot when referencing ISA and the free stream pressure altitude.

(e) Vertical Angle Error (VAE) relates to the ability of the FMS to determine incremental height above the reference point using distance and intended flight path angle.

(f) Automatic Terminal Information System (ATIS), fixed at 20 feet, is a conservative assumption that the reported altimeter setting is within 0.02 inches of mercury of the actual pressure value.

(g) Example standard deviation calculations.

$$ANPE = 1.225 \times RNP \times \frac{1852}{0.3048} \times \tan(\theta)$$

$$ANPE = 1.225 \times 0.14 \times \frac{1852}{0.3048} \times \tan(3)$$

$$ANPE \approx 54.61 \text{ feet}$$

$$WPR = 60 \times \tan(\theta)$$

$$WPR = 60 \times \tan(3)$$

$$WPR \approx 3.14 \text{ feet}$$

$$FTE = 75 \text{ feet}$$

$$ASE_{PFAF} = -8.8 \times 10^{-8} \times PFAF_{alt}^2 + 6.5 \times 10^{-3} \times PFAF_{alt} + 50$$

$$ASE_{PFAF} = -8.8 \times 10^{-8} \times 4500^2 + 6.5 \times 10^{-3} \times 4500 + 50$$

$$ASE_{PFAF} \approx 77.47 \text{ feet}$$

$$ASE_{250} = -8.8 \times 10^{-8} \times (TDZE + 250)^2 + 6.5 \times 10^{-3} \times (TDZE + 250) + 50$$

$$ASE_{250} = -8.8 \times 10^{-8} \times (1202 + 250)^2 + 6.5 \times 10^{-3} \times (1202 + 250) + 50$$

$$ASE_{250} \approx 59.25 \text{ feet}$$

$$VAE_{PFAF} = \frac{PFAF_{alt} - LTP_{elev}}{\tan(\theta)} \times (\tan(\theta) - \tan(\theta - 0.01))$$

$$VAE_{PFAF} = \frac{4500 - 1200}{\tan(3)} \times (\tan(3) - \tan(3 - 0.01))$$

$$VAE_{PFAF} \approx 11.02 \text{ feet}$$

$$VAE_{250} = \frac{TDZE + 250 - LTP_{elev}}{\tan(\theta)} \times (\tan(\theta) - \tan(\theta - 0.01))$$

$$VAE_{250} = \frac{1202 + 250 - 1200}{\tan(3)} \times (\tan(3) - \tan(3 - 0.01))$$

$$VAE_{250} \approx 0.84 \text{ feet}$$

$$ATIS = 20 \text{ feet}$$

(3) Example VEB calculations.

$$VEB = BG - ISAD + \frac{4}{3} \sqrt{ANPE^2 + WPR^2 + FTE^2 + ASE^2 + VAE^2 + ATIS^2}$$

$$VEB_{PFAF} = 21.01 + 250.43 + \frac{4}{3} \sqrt{54.61^2 + 3.14^2 + 75^2 + 77.47^2 + 11.02^2 + 20^2}$$

$$VEB_{PFAF} \approx 435.50 \text{ feet}$$

$$VEB_{250} = 21.01 + 18.91 + \frac{4}{3} \sqrt{54.61^2 + 3.14^2 + 75^2 + 59.25^2 + 0.84^2 + 20^2}$$

$$VEB_{250} \approx 189.16 \text{ feet}$$

b. Calculating the OCS slope. The OCS slope is calculated using formula 4-2-4.

Formula 4-2-4. VEB OCS Slope

$$OCS_{\text{slope}} = \frac{(PFAF_{\text{alt}} - TDZE - 250) \times \tan(\theta)^{-1}}{PFAF_{\text{alt}} - TDZE - VEB_{PFAF} - 250 + VEB_{250}}$$

Where:

VEB_{PFAF} = amount of vertical error budget (feet) at the PFAF

VEB_{250} = amount of vertical error budget (feet) at a point 250 feet above TDZE

Example:

$$OCS_{\text{slope}} = \frac{(4500 - 1202 - 250) \times \tan(3)^{-1}}{4500 - 1202 - 435.50 - 250 + 189.16}$$

$$OCS_{\text{slope}} \approx 20.76$$

c. Calculating the OCS LTP/FTP to origin distance (d_{VEB}). The d_{VEB} is calculated with formula 4-2-5 by subtracting the distance required for the OCS to reach VEB_{250} below glidepath from the along glidepath distance projected from the threshold to a point 250 feet above TDZE.

Formula 4-2-5. VEB OCS Origin

$$d_{VEB} = \frac{TDZE + 250 - LTP_{\text{elev}} - TCH}{\tan(\theta)} - (TDZE + 250 - VEB_{250} - LTP_{\text{elev}}) \times OCS_{\text{slope}}$$

Where:

VEB_{250} = amount of vertical error budget (feet) at a point 250 feet above TDZE

Example:

$$d_{VEB} = \frac{1202 + 250 - 1200 - 55}{\tan(3)} - (1202 + 250 - 189.16 - 1200) \times 20.76$$

$$d_{VEB} \approx 2454.43 \text{ feet}$$

Formula 4-2-6. VEB OCS Elevation

$$\text{OCS}_{\text{elev}} = \text{LTP}_{\text{elev}} + \frac{d_{\text{LTP}} - d_{\text{VEB}}}{\text{OCS}_{\text{slope}}}$$

Where:

d_{LTP} = along-track distance (feet) from LTP to the point of interest

d_{VEB} = distance (feet) from LTP to VEB OCS origin as per leg type

Example:

$$\text{OCS}_{\text{elev}} = 1125.4 + \frac{19747.38 - 4063.28}{21.67}$$

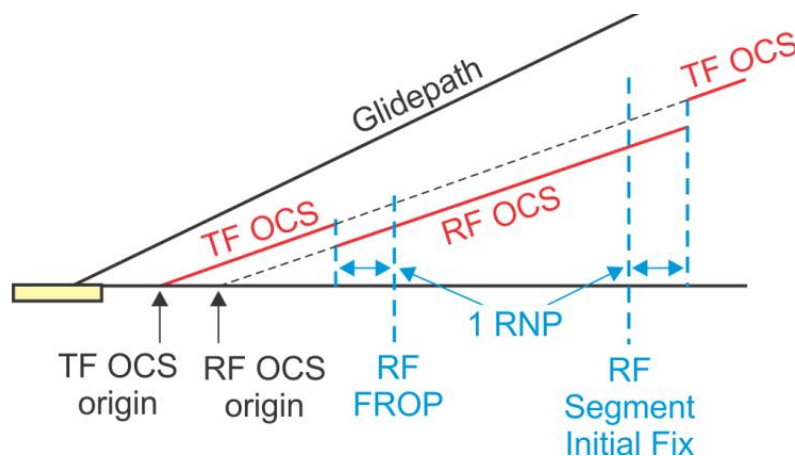
$$\text{OCS}_{\text{elev}} \approx 1849.17 \text{ feet}$$

d. Obstacle evaluation. Use formula 4-2-6 to determine the VEB OCS elevation. Apply the OCS applicable to the leg type to the entire leg (see paragraph 1-2-5.b(1)(b) and figure 4-2-3). See latest version of Order 8260.3, section 10-6 for obstacle assessment. Obstacles that penetrate an OCS may be mitigated by one of the following:

- (1) Reduce the RNP value for the leg or change the lateral path.
- (2) Raise GPA.
- (3) Raise TCH.
- (4) Adjust the DA (see formula 4-2-7).

Note: The VEB slope and origin is impacted by changes to RNP, GPA, and/or TCH. When paragraphs 4-2-4.d(1), 4-2-4.d(2), and/or 4-2-4.d(3) are applied, the VEB must be recalculated and the OCS re-evaluated.

Figure 4-2-3. Final Segment OCS Elevation



Formula 4-2-7. DA Distance Based on VEB OCS Penetration

$$d_{DA} = d_{LTP} + p \times OCS_{slope}$$

Where:

p = penetration (feet) of the OCS

d_{LTP} = along track distance (feet) from the LTP/FTP to obstacle

Example:

$$d_{DA} = 4514.5 + 8.3 \times 23.42$$

$$d_{DA} \approx 4708.89 \text{ feet}$$

e. Published decision altitude. If the FAS OCS is not penetrated, the minimum HAT value of 250 feet applies. Except when using an RNAV MAS [see paragraph 4-3-1.a(2)], the DA distance from LTP/FTP must also not be less than missed approach $d_{heightloss}$ (see formula 4-3-4) plus d_{VEB} .

Section 4-3. Missed Approach Segment (MAS)

4-3-1. General. The NavSpec is RNP AR APCH (missed flight phase with associated XTT of 1.00 - 0.10). Secondary areas do not apply. Optional NavSpec may be RNAV 1 or A-RNP (missed flight phase with associated XTT of 1.00 or 0.30) with secondary areas (except for A-RNP). Use of A-RNP missed approach XTT less than one requires Flight Standards approval.

a. MAS RNP. Use of RNP < 1.00 in the missed approach may limit availability of the procedure. In order to serve the largest number of users, specify a default or RNAV MAS where possible.

(1) Default MAS (RNP AR APCH with RNP 1.00). Use except where not operationally beneficial. The construction is a continuation of the FAC. The OEA expands from the FAS RNP to an RNP value of 1.00.

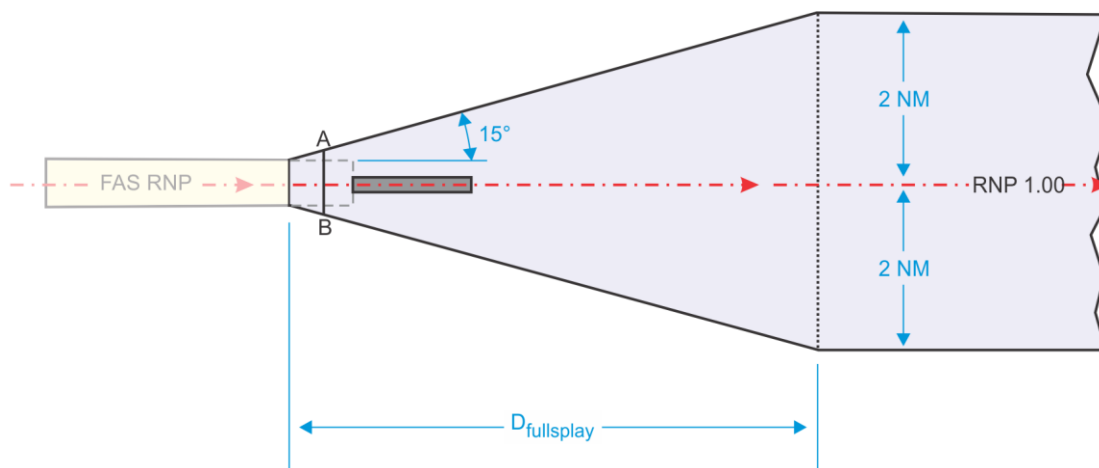
(2) RNAV MAS (RNAV 1 missed flight phase). Use where operationally beneficial (i.e., turn-at-altitude construction to allow a turn earlier than permitted by default MAS construction).

(3) Reduced RNP MAS (RNP AR APCH with RNP < 1.00). Use when previous options are not practical. Construct straight or turning (using RF legs) missed approach.

4-3-2. Default MAS OEA Construction. Establish the MAS using TF and/or RF legs.

a. Straight. The MAS leg expands from the FAS OEA DA at an angle of at least 15 degrees relative to course centerline until reaching a width of ± 2 NM (RNP 1.00) (see figure 4-3-1). RNP 1.00 width must be achieved at or prior to reaching the first missed approach fix or the clearance limit if no other fix exists beyond the DA. The along-track distance (NM) required to complete the splay ($D_{\text{fullsplay}}$) may be calculated using formula 4-3-1. When RNP 1.00 is not reached using a 15-degree splay, establish an RNAV MAS or splay at the angle required to reach ± 2 NM at the turn point/fix (S_{angle}).

Figure 4-3-1. Default MAS OEA, Straight



Formula 4-3-1. Along-Track Distance to Complete Splay

$$D_{\text{fullsplay}} = \frac{2 \times (1 - \text{RNP}_{\text{FAS}})}{\tan(S_{\text{angle}})}$$

Where:

RNP_{FAS} = RNP value (NM) of the final segment

S_{angle} = Missed approach OEA splay angle relative to the course centerline

Example:

$$D_{\text{fullsplay}} = \frac{2 \times (1 - 0.30)}{\tan(15)}$$

$$D_{\text{fullsplay}} \approx 5.22 \text{ NM}$$

b. Turning missed approach. Apply paragraph 4-3-2.a in conjunction with the following paragraphs.

(1) TF-TF connection. Use paragraph 1-2-5.d(1) in cases where RNP 1.00 width is achieved at a sufficient distance prior to the turn fix to permit construction. Otherwise, construct as follows:

(a) *Step 1.* Construct the leg following the turn fix (outbound leg) at full width.

(b) *Step 2.* Determine the DTA using formula 1-2-1. Establish point EB at the turn-side boundary of the inbound leg abeam the DTA. Establish point EE at the turn-side boundary of the outbound leg abeam the DTA.

(c) *Step 3.*

1. *Case 1* (see figure 4-3-2). Point EB is internal to the outbound leg OEA. In this case, no expansion is required. The area is a simple connection of the inbound and outbound leg boundaries.

2. *Case 2* (see figure 4-3-3). Point EB is external to the outbound leg OEA. In this case, expand the boundaries with a line connecting points EB and EE.

(2) TF-RF and RF-RF connections. Paragraph 1-2-5.d(3) applies.

Figure 4-3-2. Default MAS OEA Construction, Turning TF-TF Connection, Splay Greater than 15 degrees, Case 1

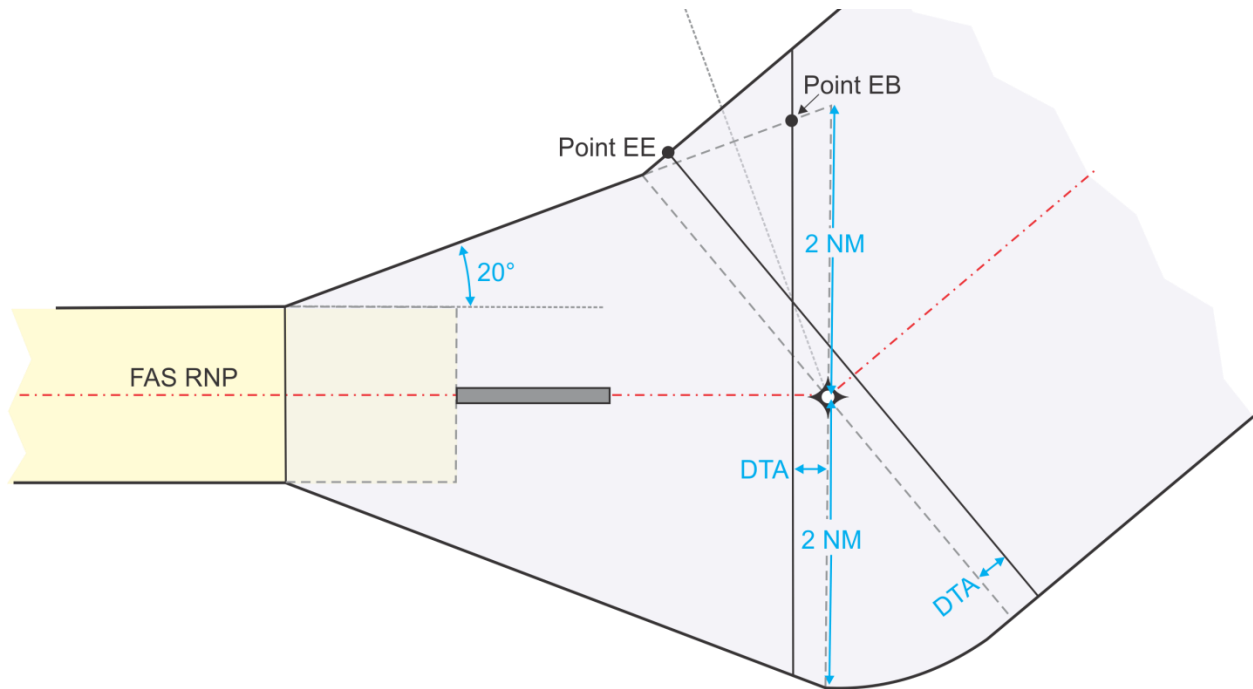
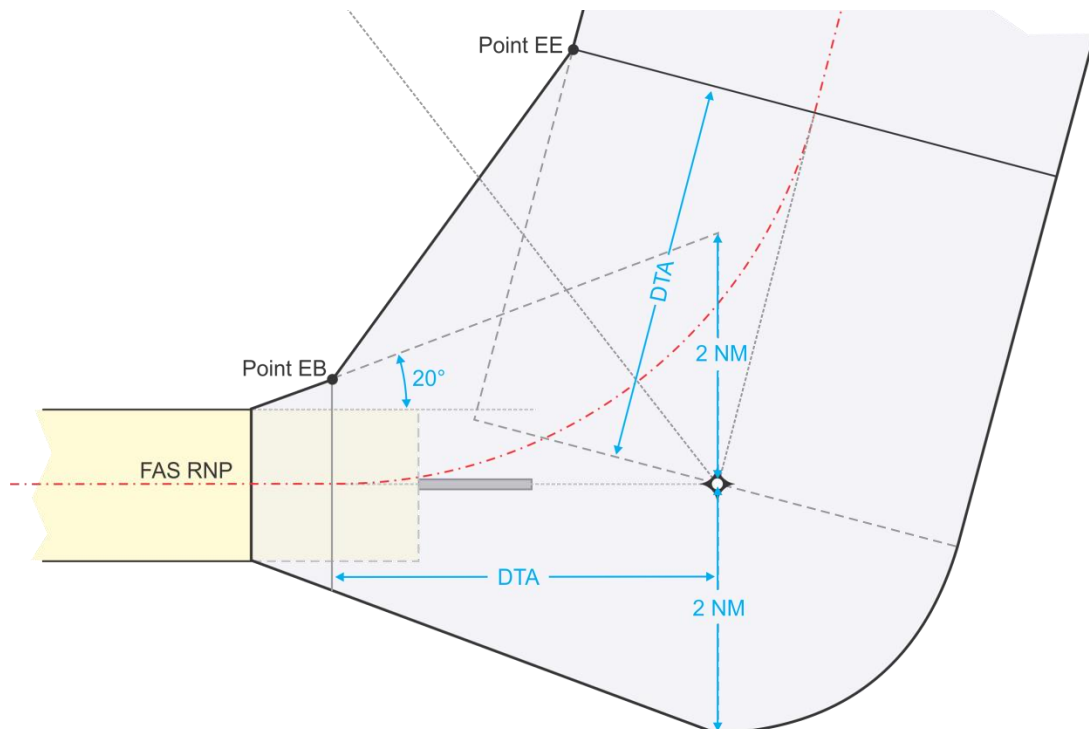


Figure 4-3-3. Default MAS OEA Construction, Turning TF-TF Connection, Splay Greater than 15 degrees, Case 2



4-3-3. RNAV MAS OEA Construction (see figure 4-3-4). Use LNAV/VNAV construction in accordance with section 3-5 and 3-7 with the following exceptions:

- a. DA must be no closer to LTP than the VEB OCS origin.
- b. RF legs do not require Flight Standards approval.
- c. Paragraph 3-6-2 applies except the 15-degree OEA splay begins at the outer edge of the final OEA at the DA point instead of the final secondary area (i.e., the secondary start width is zero). Calculate the distance from course centerline to the primary and secondary boundary of the OEA at any distance from line C-D using formula 4-3-2.

Formula 4-3-2. RNP AR RNAV MAS Primary and Secondary Boundary

$$MAS_{Y_{primary}} = \frac{D_{splay} \times \tan(15) \times (1 - RNP_{FAS})}{1.5 - RNP_{FAS}} + 2 \times RNP_{FAS}$$

$$MAS_{Y_{secondary}} = D_{splay} \times \tan(15) + 2 \times RNP_{FAS}$$

Where:

D_{splay} = along-track distance (NM) from beginning of MAS splay to point of interest

RNP_{FAS} = RNP value (NM) of the final segment

Example:

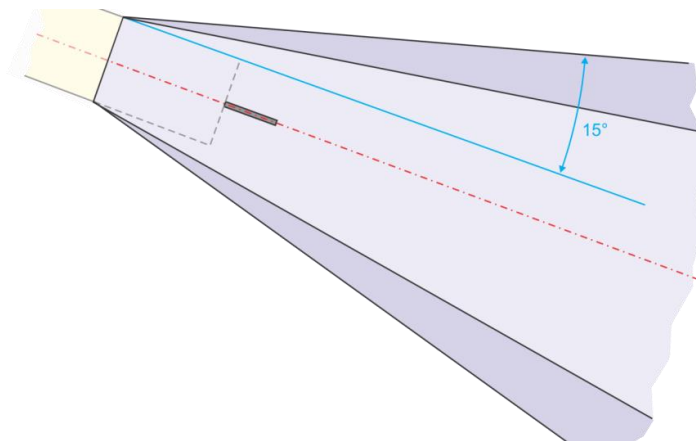
$$MAS_{Y_{primary}} = \frac{4.48 \times \tan(15) \times (1 - 0.30)}{1.5 - 0.30} + (2 \times 0.30)$$

$$MAS_{Y_{primary}} \approx 1.3 \text{ NM}$$

$$MAS_{Y_{secondary}} = 4.48 \times \tan(15) + 2 \times 0.30$$

$$MAS_{Y_{secondary}} \approx 1.8 \text{ NM}$$

Figure 4-3-4. Missed Approach Segment, RNAV MAS Area



4-3-4. Reduced RNP MAS. Establish using TF and/or RF legs. See paragraph 1-2-5.b for OEA construction. Where turns are necessary, the turn must occur after passing 500 feet above airport elevation (rounded to the nearest foot) and where possible, not prior to the DER. At no point may the turn be specified prior to $D_{MASturn}$ from DA (see formula 4-3-3).

Formula 4-3-3. Reduced RNP MAS, Minimum Distance DA to Turn

$$D_{MASturn} = \frac{(V_{KTAS} + 15) \times 10}{3600}$$

Where:

V_{KTAS} calculated KTAS using DA and final segment KIAS for the fastest CAT

Example:

$$D_{MASturn} = \frac{(170.38 + 15) \times 10}{3600}$$

$$D_{MASturn} \approx 0.51 \text{ NM}$$

4-3-5. OCS Evaluation. The following applies to default and reduced RNP missed approaches. For RNAV missed approaches see sections 3-5, 3-6, and 3-7.

a. General. The OCS is composed of sections 1a and 1b, which are separated by line A-B.

(1) Section 1a. The OCS extends from the DA point downward at the VEB OCS slope ratio for a distance of $d_{heightloss}$ using formula 4-3-4 measured along the final course track to line A-B.

Formula 4-3-4. Height Loss Distance

$$d_{heightloss} = \frac{50}{\tan(\theta)}$$

Example:

$$d_{heightloss} = \frac{50}{\tan(3)}$$

$$d_{heightloss} \approx 954.06 \text{ feet}$$

(2) Section 1b OCS. From line A-B, section 1b OCS rises at a 40:1 slope. Calculate the HMAS at line A-B using formula 4-3-5. Obstacle distance is measured as the along-track distance from line A-B to a point abeam the obstruction.

Formula 4-3-5. HMAS at Line AB

$$\text{HMAS} = \text{LTP}_{\text{elev}} + \frac{d_{\text{DA}} - d_{\text{VEB}} - d_{\text{heightloss}}}{\text{OCS}_{\text{slope}}}$$

Where:

d_{DA} = distance (feet) from LTP to DA

d_{VEB} = distance (feet) from LTP to VEB OCS origin

$d_{\text{heightloss}}$ = distance (feet) required to lose 50 feet on glidepath

Example:

$$\text{HMAS} = 1200 + \frac{3758.74 - 2454.55 - 954.06}{20.76}$$

$$\text{HMAS} \approx 1216.87 \text{ feet}$$

b. OCS penetrations. Obstacles must not penetrate the OCS. Where obstructions penetrate the OCS, mitigate using one or more of the following actions:

(1) Revise the MAS course.

(2) Specify a MAS climb gradient [see paragraph 1-3-1.g(2)]. Calculate the altitude above which the climb gradient is no longer required using formula 4-3-6. This altitude is used in formula 1-3-7 to calculate the required climb gradient.

Note: If the climb gradient exceeds 425 ft/NM, evaluate the MAS using the OCS slope appropriate for 425 ft/NM (18.82:1) and adjust DA for the remaining penetration per paragraph 4-3-5.b(3).

Formula 4-3-6. CG Termination Altitude

$$\text{CG}_{\text{term}} = \frac{(\text{O}_{\text{MSL}} - \text{OCS}_{\text{start}}) \times 8000 \times 0.3048}{1852} + \text{AC}_{\text{start}}$$

Where:

O_{MSL} = obstacle MSL elevation

$\text{OCS}_{\text{start}}$ = starting altitude (feet) of the OCS

AC_{start} = starting altitude (feet) of the aircraft (e.g., SOC)

Example:

$$\text{CG}_{\text{term}} = \frac{(2147.41 - 1217.37) \times 8000 \times 0.3048}{1852} + 1402$$

$$\text{CG}_{\text{term}} = 2626.52 \text{ feet}$$

(3) Adjust DA. To determine the DA required for an OCS penetration, calculate the HAT adjustment using formula 4-3-7 and add the result to the original DA. Use formula 1-3-3 to determine adjusted DA distance from the LTP/FTP.

Formula 4-3-7. HAT Adjustment for MAS OCS Penetration

$$\Delta DA = \frac{p \times \tan(\theta) \times MA_{\text{slope}} \times FAS_{\text{slope}}}{MA_{\text{slope}} + FAS_{\text{slope}}}$$

Where:

p = penetration (feet) of the MAS OCS

MA_{slope} = missed approach OCS slope ratio

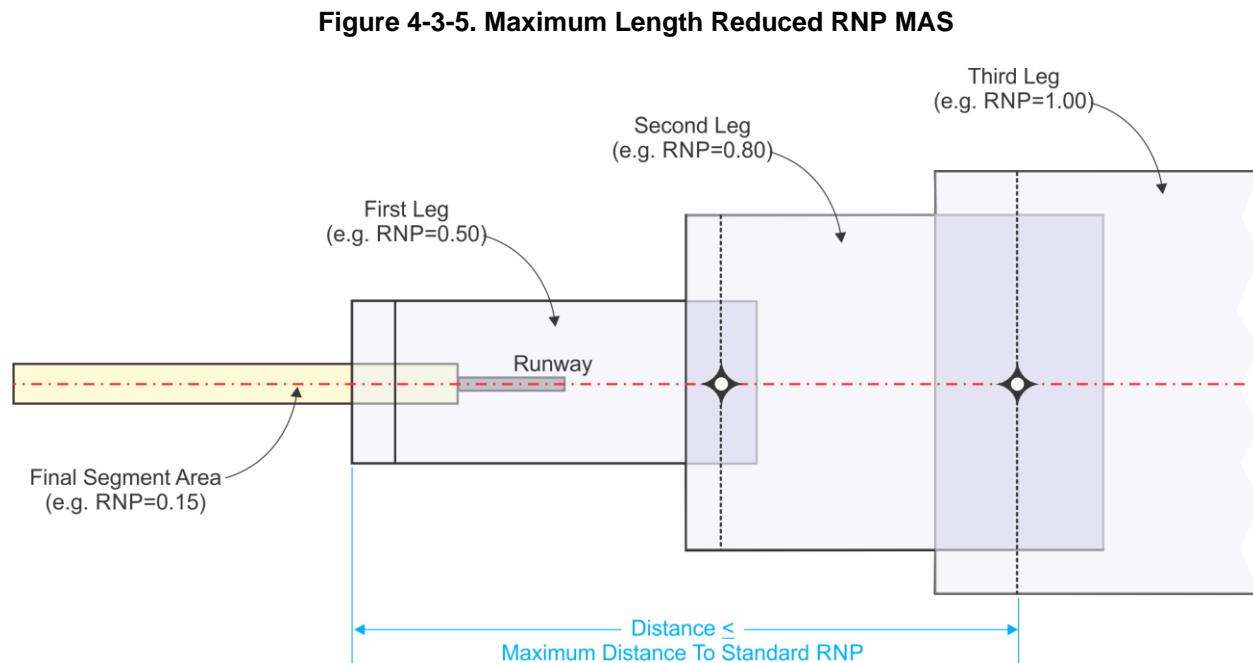
FAS_{slope} = final approach OCS slope ratio

Example:

$$\Delta DA = \frac{19.3 \times \tan(3) \times 40 \times 23.42}{40 + 23.42}$$

$$\Delta DA \approx 14.94 \text{ feet}$$

(4) Specify a reduced RNP MAS leg [see paragraph 4-3-1.a(3)]. Apply the reduced RNP MAS leg only as far as needed to clear the penetrating obstruction(s) and specify the largest RNP value possible. Calculate the maximum length of the reduced RNP leg using formula 4-3-8 (see figure 4-3-5). A reduced RNP MAS leg must be a TF or RF leg.



Formula 4-3-8. Max Length Reduced RNP MAS Leg

$$D_{\text{MASRNP}} = (\text{RNP}_{\text{MAS}} - 0.05) \times \frac{V_{\text{KTAS}} - 10}{8}$$

Where:

RNP_{MAS} = RNP value (NM) of the missed approach segment

V_{KTAS} = calculated KTAS using DA and final approach KIAS for the slowest CAT

Example:

$$D_{\text{MASRNP}} = (0.50 - 0.05) \times \frac{92.93 - 10}{8}$$

$$D_{\text{MASRNP}} \approx 4.66 \text{ NM}$$

Chapter 5. Departure Procedures

Section 5-1. General Criteria

5-1-1. General. The NavSpec is RNAV 1 (departure flight phase with associated XTT of 1.00). Optional NavSpecs may be RNP 1 (departure flight phase with associated XTT of 1.00), A-RNP (departure flight phase with associated XTT of 1.00 or 0.30), RNP AR DP (departure flight phase with associated XTT of 1.00 - 0.30), or RNP 0.3 (departure flight phase with associated XTT of 0.30) for helicopter. Use of A-RNP departure XTT less than 1.00 requires Flight Standards approval. RNP AR DPs are established using TF and/or RF legs. Secondary areas apply, except for RNP AR DP and A-RNP. For IFR departures from a heliport that supports IFR procedures, substitute all DER references with heliport departure reference point (HDRP). The HDRP is the point of intersection of the final approach and takeoff area (FATO) and departure course. For helicopter procedures, the departure reference line (DRL) is coincident with the HDRP, perpendicular to the departure course.

a. Initial climb area (ICA). Departure procedures begin with an ICA constructed in accordance with Order 8260.3, paragraph 13-1-2, except where modified by this order.

(1) The ICA must be long enough to allow LNAV engagement [500 feet above airport elevation (rounded to the nearest foot)].

(2) Use formula 1-3-9 to determine ICA length. Where a higher than standard climb gradient terminates prior to the minimum turning altitude, the ICA length will be the distance required to reach the CG termination altitude plus the distance between that altitude and the turning altitude at the reduced CG.

(3) An RNP AR departure with an RF turn at DER has no ICA. The OEA is constructed using a 15-degree splay relative to RF track originating from +/- 500 feet either side of DER (ICAB) (see figure 5-3-8).

b. Leg type limitations. See Order 8260.46, paragraph 3-1-6 for permissible leg types.

Note: The 'AER' in Order 8260.46, table 3-1-1 is for coding purposes. Bearings and distances used for OEA construction and leg length analysis are not measured from the AER.

(1) VA leg. VA legs are all-primary area and the initial departure course is aligned within 15 degrees of the extended runway centerline (see paragraph 5-2-1).

(2) VI leg. VI legs are all-primary area and the initial departure course is aligned with extended runway centerline (see paragraph 5-2-1). VI legs are associated with CF legs and due to possible Flight Management System route discontinuity, course changes of less than 10 degrees to intercept the CF leg are not authorized without approval from Flight Standards Service.

(3) DF leg. Secondary areas apply once the 15-degree splay from course line (early and late) have reached full primary width. DF-DF is only authorized when the first DF fix is within 15 degrees of the extended runway centerline.

(4) CF leg. Secondary areas apply once the 15-degree splay from course has reached full primary width or as defined in the turning departure section.

c. Leg length. Comply with paragraph 1-2-5.b(1) with the following exceptions.

(1) Do not develop legs exceeding 260 NM to ensure the geodesic path does not exceed the protected airspace for a great circle path.

(2) For LNAV engagement, the first leg must be designed to end at least 500 feet above airport elevation (rounded to the nearest foot). Only use a higher than standard CG for LNAV engagement when there is an operational need to bring the first turn closer to DER.

(3) The maximum allowable VA or VI leg length is 10 NM.

(4) For FO DF leg length feasibility check see appendix C.

Section 5-2. Straight Departure

5-2-1. Straight Departure.

a. VI is a straight departure leg and is aligned with runway centerline extended. VA, DF, CF, and TF are straight departure legs and are aligned within 15 degrees of the runway centerline. Evaluate straight departure legs in accordance with Order 8260.3, section 13-3 with the following provisions:

(1) The ICA is aligned along the runway centerline (departure course for helicopter departures from a heliport that supports IFR procedures) for the distance required for a climb to 500 feet above airport/heliport elevation (rounded to the nearest foot).

(2) Continue splay until reaching basic area width as defined in chapter 1 (see figure 5-2-1. If an A-RNP or RNP AR DP value is selected that results in an OEA narrower than the ending ICA width, taper inward at 30 degrees relative to course to join edge of primary area (see figure 5-2-2).

(3) The departure may not be offset from runway centerline when the first turn is an RF and the straight segment has not reached full OEA width construction (see paragraph 1-2-5.d).

(4) Straight route departure criteria apply only to the first course from DER. Any turn after the first departure course must be evaluated using turning criteria.

Figure 5-2-1. Straight Departure

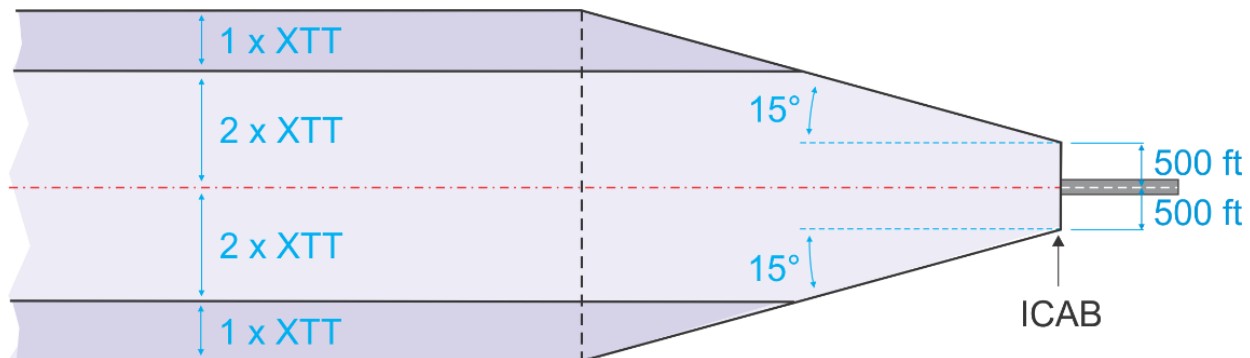
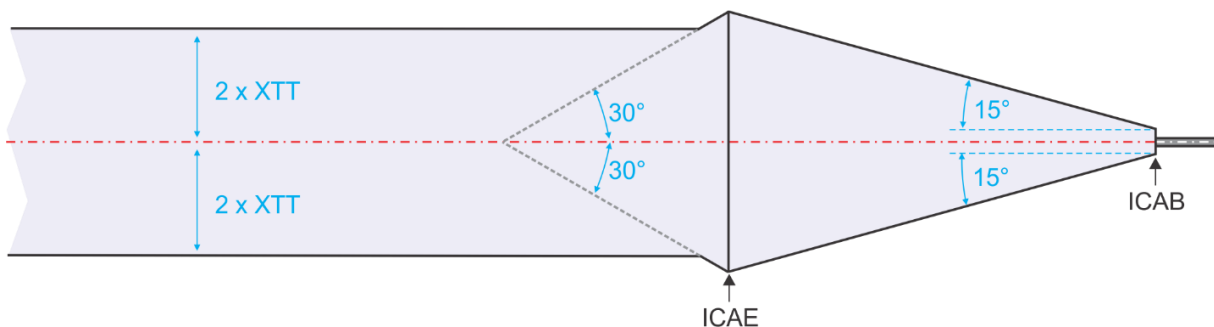


Figure 5-2-2. Straight Departure with 30-Degree Inward Taper



Section 5-3. Turning Departures

5-3-1. Turning Departure.

a. Early and inside turn construction.

(1) First turn. The first turn may be a FB or FO fix or may follow a VA or VI leg.

(a) Turn following VA leg (turn-at-altitude). The turn altitude will either be the minimum required altitude, operationally specified, or determined by obstacle evaluation. The specified turn altitude must equal or exceed the ICA end aircraft altitude.

1. The OEA consists of the ICA/straight segment, section 1, and section 2. Excluding the ICA, section 1 is defined as the OEA on the DER side of the DRL. Section 2 is the OEA on the start end of runway (SER) side of the DRL.

2. Construct the ICA/straight segment from the ICA baseline (ICAB) to line L-L'. Because straight segment length and the OCS elevation at L-L' are dependent on aircraft altitude, establishing an obstacle-based turn altitude may be an iterative process. If an A-RNP NavSpec RNP value is selected that results in an OEA narrower than the ending ICA/straight segment width, taper inward at 30 degrees relative to course.

3. OEA construction after ICA/straight segment (see table 5-3-1 and figure 5-3-1).

Table 5-3-1. Early-Turn Tie-Back Points

Turn Angle (measured from RWY C/L ICAE intersection)	Tie-Back Point
≤ 165 degrees	DRP
> 165 degrees	Point P'

a *Step 1.* Construct a line (representing the early-turn flight track) from the tie-back point, to the fix.

b *Step 2.* Construct the inner primary and secondary OEA boundary lines parallel to this line.

c *Step 3.* From the tie-back point, construct a line splaying at 15 degrees to intersect the parallel boundary lines or segment end, whichever occurs earlier. Apply secondary areas only after the 15-degree splay line intersects the primary boundary line. Where the inside turn boundary intersects the ICA inside 15-degree splay line the boundary must be a line connecting the departure reference point (DRP), point A, and full width point or original splay width abeam fix if full width has not been reached (see figure 5-3-2).

Figure 5-3-1. VA-DF Early/Inside Turn Construction After ICA

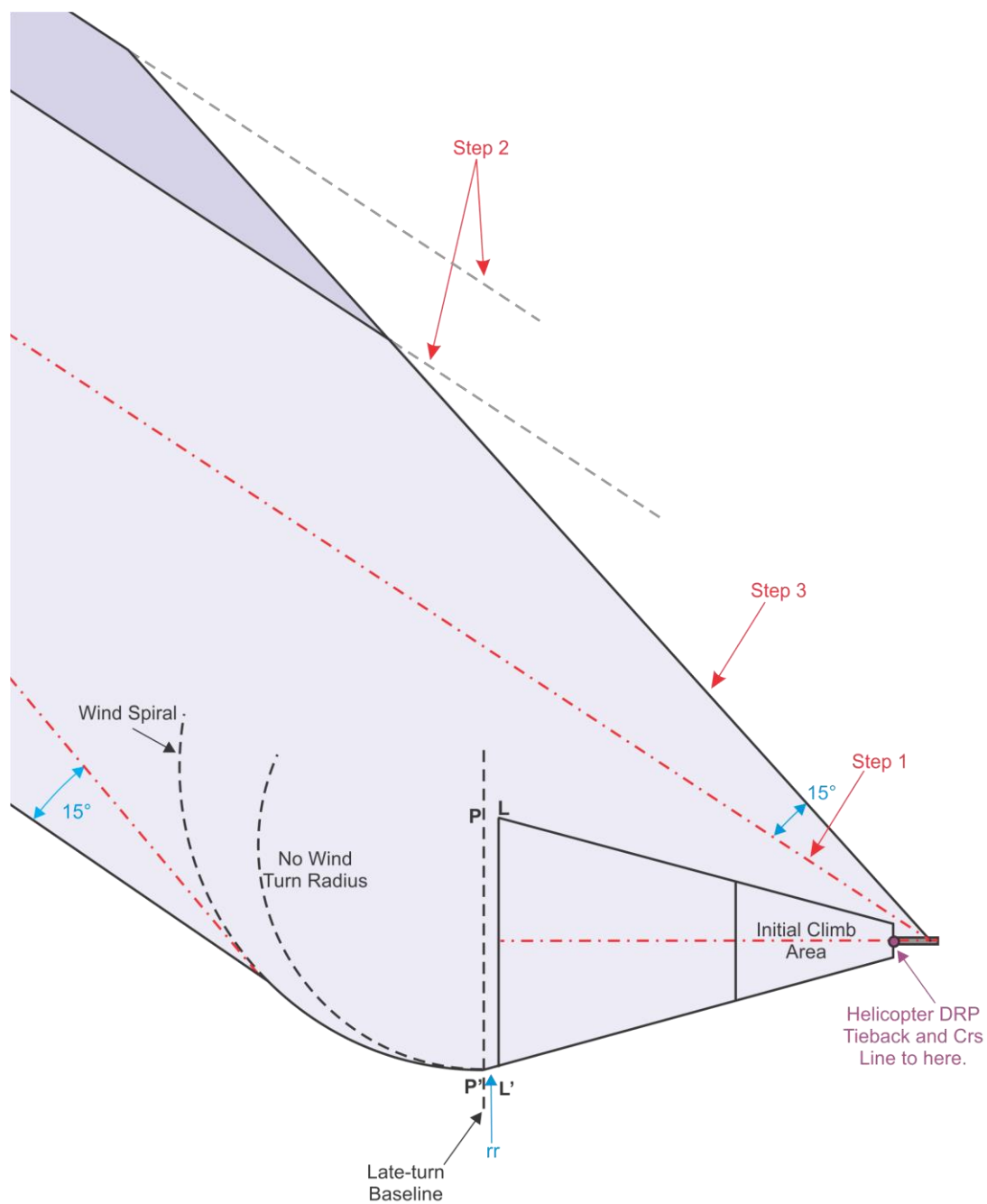
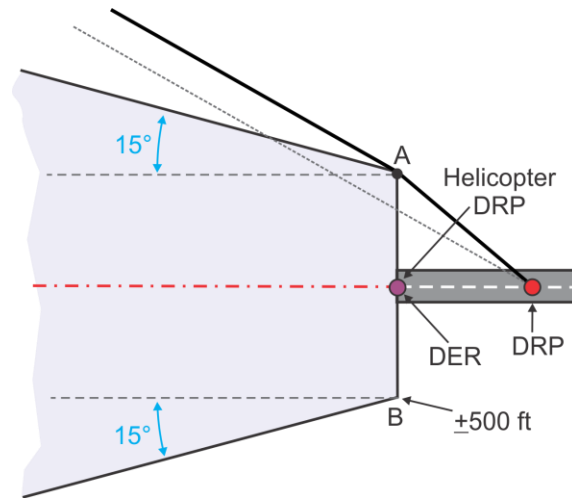


Figure 5-3-2. VA Leg Construction Alternate Inside Boundary



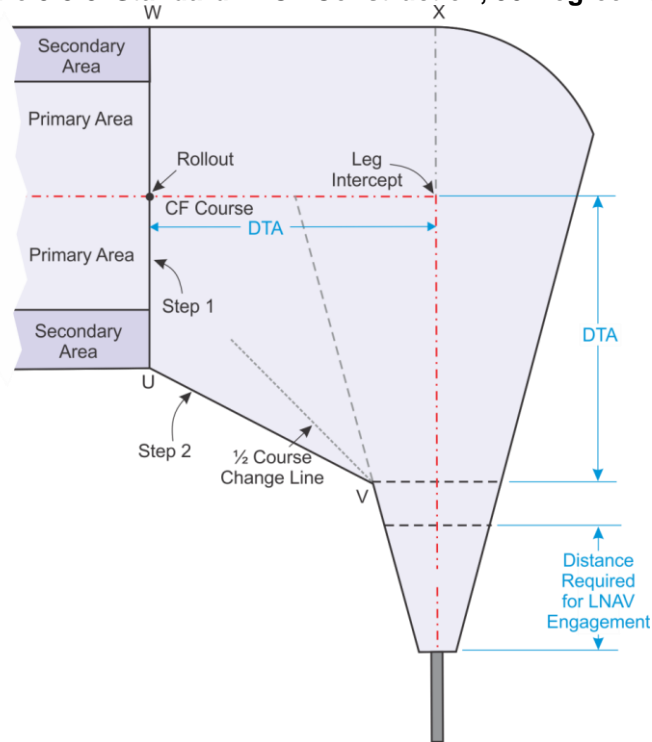
(b) Turn following VI leg. The VI leg contains an ICA constructed in accordance with paragraph 5-1-1.a terminating where LNAV engagement is reached (500 feet above airport elevation (rounded to the nearest foot)). The VI leg OEA is all primary area; secondary area may be applied in the CF leg at the rollout point.

1. OEA construction. Inside expansion starts where the VI DTA area begins (point V) with an angle drawn at one-half of the course change at leg intercept and ends where the angle converges with CF leg secondary boundary. Where the angle does not converge with secondary boundary, draw a line from point V to the inside secondary area boundary abeam the rollout (point U) for area completion (see figure 5-3-3).

a *Step 1.* Along the CF leg course, construct a line perpendicular to the CF leg at a distance from the intercept point equal to the calculated DTA based on the altitude at leg intercept. At the rollout point, the CF leg OEA is the full combined width of the primary and secondary areas.

b Step 2. Construct a line (representing the early-turn flight track) from the VI DTA (point V), to the CF leg full width abeam the rollout point.

Figure 5-3-3. Standard VI-CF Construction, 90-Degree Turn



2. Minimum VI-CF leg combination construction. Where a short initial departure leg must be developed, the VI leg length may be designed to the greater of 1 NM from DER/HDRP or the distance required to climb 500 feet above airport elevation (rounded to the nearest foot). For this early turn, the OEA is modified to be somewhat like a FB fix turn for inside expansion and additional protection is also provided for outside area expansion built similarly to a FO fix turn.

a For turns of 30 degrees or less, splay 15 degrees relative to the CF leg course from point ICAB and continue this line until intersecting the CF leg secondary area boundary (see figure 5-3-4). The secondary area begins where this line crosses the primary area boundary.

b For turns of more than 30 degrees, the inside boundary is from the DRP to the inside secondary area boundary abeam the rollout (point R). From point R, the secondary area tapers 30 degrees inward relative to the CF leg course until the CF leg standard primary area boundary (see figure 5-3-5 thru figure 5-3-7).

c CF leg construction. Along the CF leg course, a rollout point is established from leg intercept at a distance of the calculated DTA, based on the altitude at leg intercept. Establish a full primary and secondary width OEA at the rollout point; the area may or may not be fully utilized based upon the leg intercept turn.

Figure 5-3-4. Minimum VI-CF Construction, Turn 30 Degrees or Less

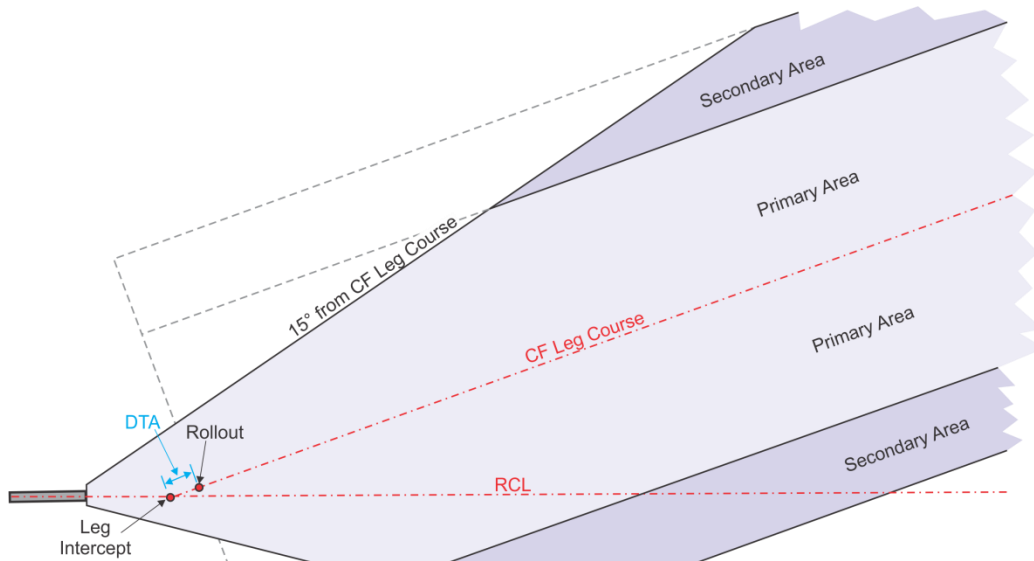


Figure 5-3-5. Minimum VI-CF Construction, Greater than 30-degree Turn

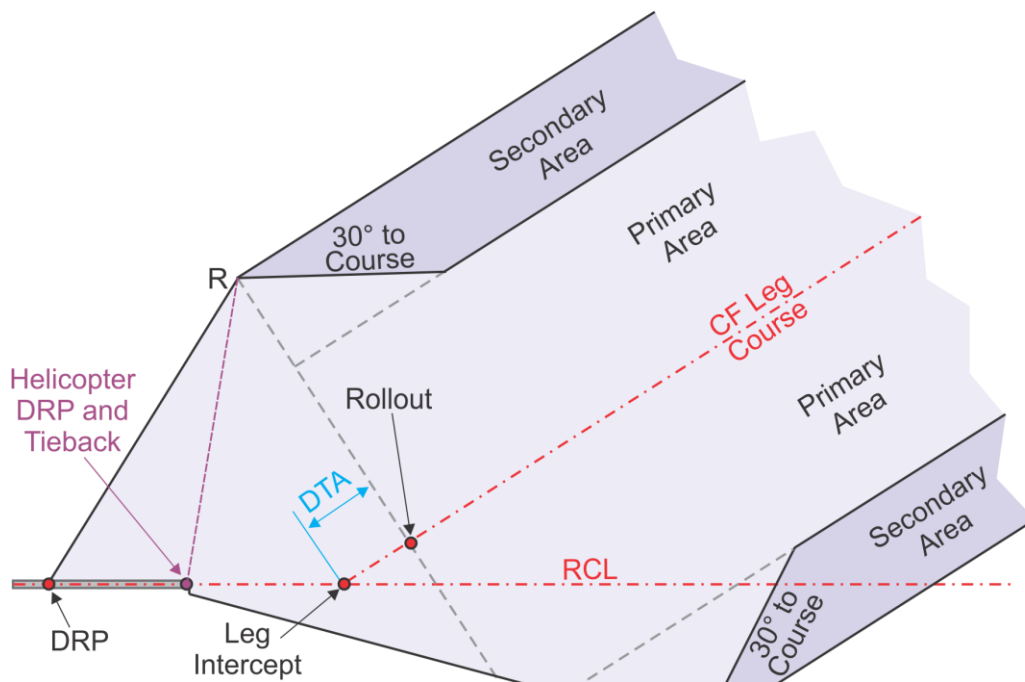


Figure 5-3-6. Minimum VI-CF Construction, 75-degree Turn

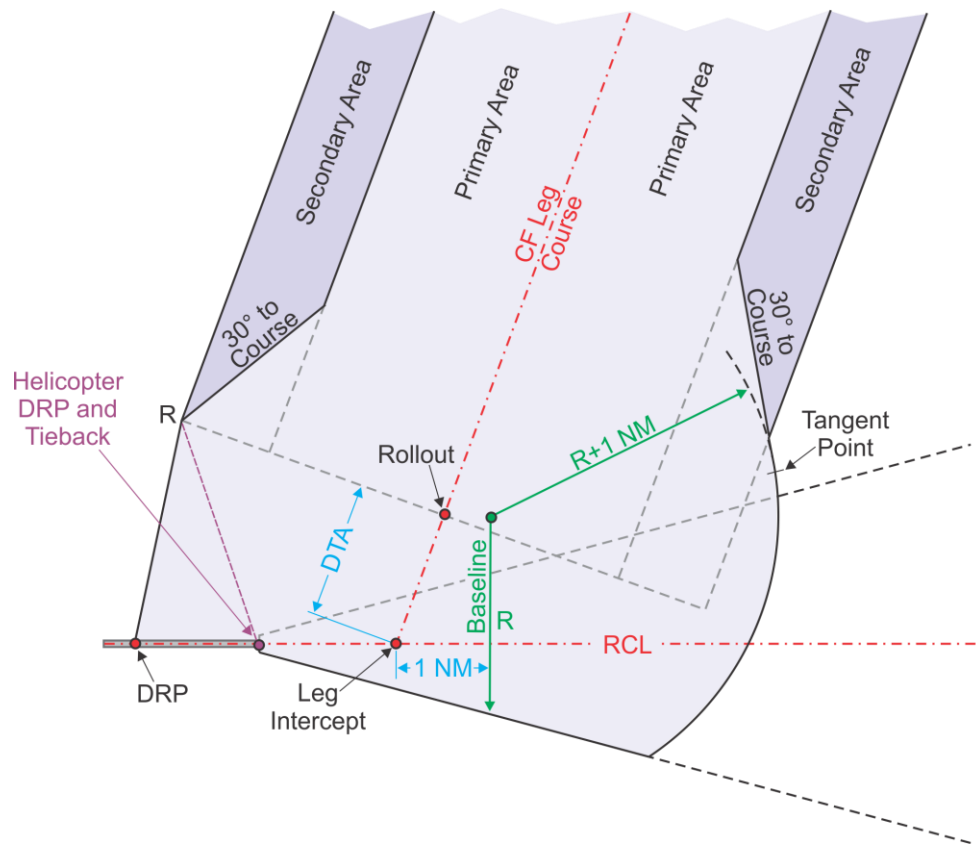
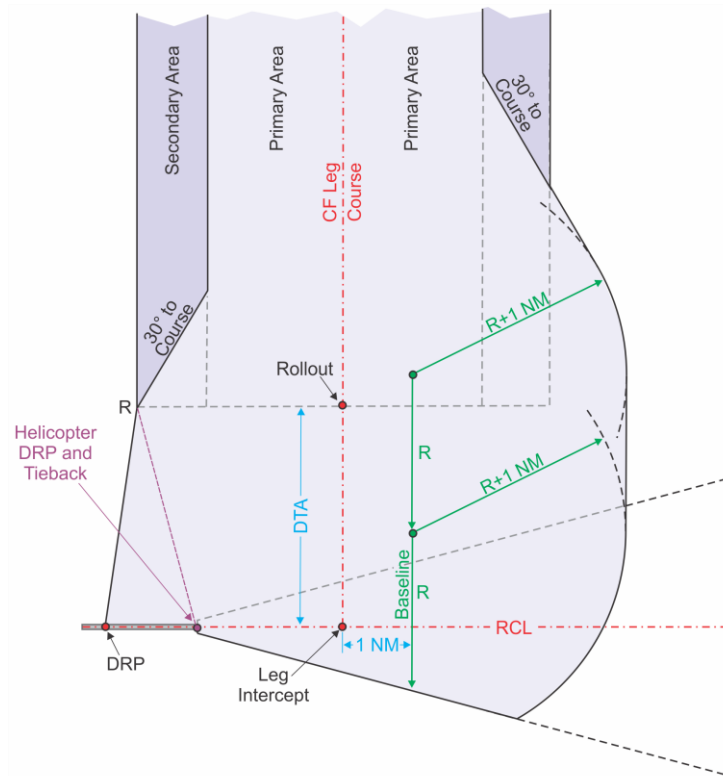
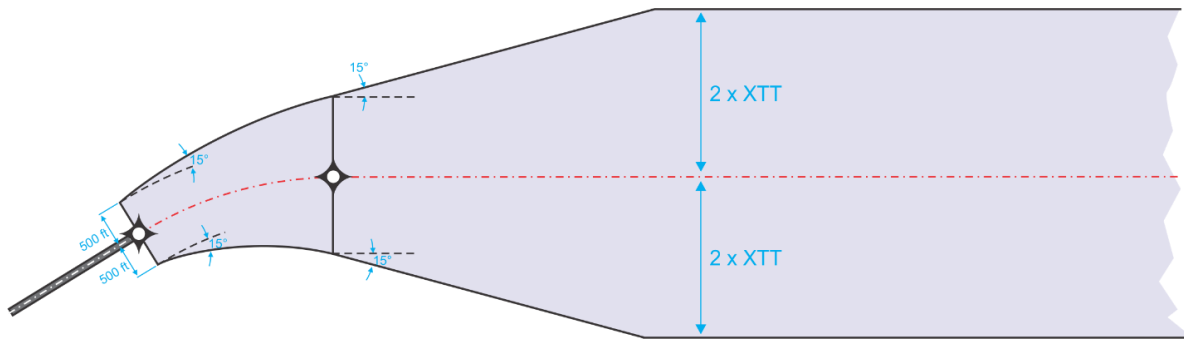


Figure 5-3-7. Minimum VI-CF Construction, 90-degree Turn

(c) Turn-at-fix (FO or FB). Use paragraph 1-2-5.d in cases where full primary and secondary width is achieved at a sufficient distance prior to the turn fix to permit construction. If an A-RNP or RNP AR DP NavSpec RNP value is selected that results in an OEA narrower at an RF start point than the ending ICA width, taper inward at 30 degrees relative to course to join edge of primary area. Otherwise, construct as follows:

1. Fix location. The first turn fix must be located on the runway centerline extended (or course line for paragraph 5-2-1.a construction). The distance to the turn fix must be sufficient to result in an early turn baseline at or beyond LNAV engagement (500 feet above airport elevation [rounded to the nearest foot]). Where the first fix must be located at the point the aircraft reaches or exceeds a specific altitude, calculate fix distance from DER with formula 1-3-9 using the standard or assigned CG. If a fix is already established, project the aircraft altitude at the fix in accordance with chapter 1. For RNP-1 or A-RNP departures, an RF start point may be located no closer to DER than the greater of 1 NM or LNAV engagement (500 feet above airport elevation [rounded to the nearest foot]). For RNP AR departure procedures, an RF start point may be located up to and including at the DER (see figure 5-3-8). If the RF start point is at DER or less than 1 NM from DER, chart restrictions to cross DER at or above 500 feet above airport elevation (rounded to the nearest foot) and a max airspeed of 210 KIAS at/until the RF start waypoint. If the RF start point is not at DER but is less than 1 NM from DER, publish a waypoint at DER (may be opposite end threshold if the coordinates are the same) this is to provide reference for the 500-foot crossing restriction. The RF must be tangent to the runway centerline (extended).

Figure 5-3-8. RF at DER

2. Early turn baseline (line L-L'). The early turn baseline is established perpendicular to the inbound leg at the appropriate table 5-3-2 distance prior to the fix. For RF turn construction, ATT need not be applied until sufficient distance from DER for OEA to reach full width.

Table 5-3-2. Early turn Baseline Distance Prior to Turn Fix

Fix Type	Distance
FB Fix	ATT + DTA*
FO Fix	ATT

*DTA = 0 for turns 10 degrees or less

3. Construction points. Inside turn construction uses points PI (primary intersection) and SI (secondary intersection). When defining these points, the leg prior to the turn fix is the inbound leg and the leg following the turn fix is the outbound leg.

a Point SI. This is the connection point for the secondary area boundary and is the intersection of line L-L' and the inbound leg secondary area boundary. When inbound leg has no secondary, points PI and SI are coincident.

b Point PI. This is the connection point for the primary area boundary and is the intersection of line L-L' and the inbound leg primary area boundary.

4. Inside turn expansion. Primary/Secondary boundary expansion depends on the location of points PI and SI in relation to the outbound leg OEA boundary lines (extended when necessary) (see table 5-3-3). Where standard expansion is suitable for one, but not both splays, find the outbound connection point for the non-standard splay abeam the standard connection point. If the primary expansion line would result in a smaller primary area, continue the inbound leg splay until full primary width is reached or until intersecting the angle bisector line whichever occurs first, then splay per table 5-3-3. For RF legs, splay 15 degrees relative to RF path beginning at points PI and SI on an early turn baseline L-L' abeam the RF start point (see figure 5-3-13). Where the splay doesn't reach full OEA width at a point that supports standard construction then connect points PI and SI on an early turn baseline L-L' abeam the RF

start point to the outbound leg OEA boundary lines perpendicular to the RF end point (see figure 5-3-12).

Table 5-3-3. Standard Inside Turn Expansion

Construction Point	External to Outbound leg OEA (both primary and secondary) (see figure 5-3-9)	Internal to Outbound leg Secondary area (see figure 5-3-11)	Internal to Outbound leg Primary area (see figure 5-3-10)
Point SI	Greater of $\frac{1}{2}$ Turn-angle or Inbound leg Splay	15-degrees relative outbound track	
Point PI	Greater of $\frac{1}{2}$ Turn-angle or Inbound leg Splay		15-degrees relative outbound track

5. Construction steps.

a *Step 1.* Construct line L-L' at the appropriate table 5-3-2 distance.

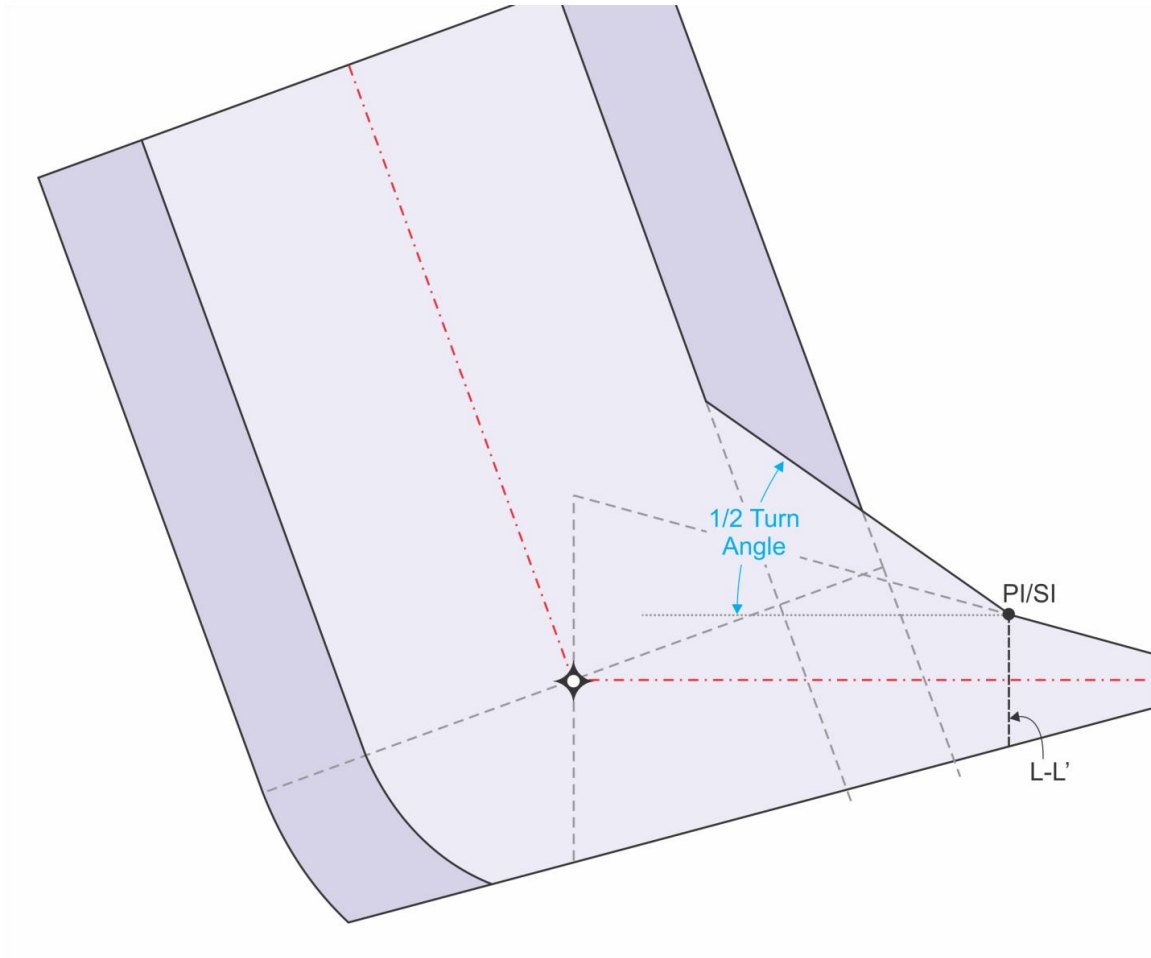
b *Step 2.* Construct secondary area boundary.

1 Case 1. SI is external to the outbound leg OEA. Construct an expansion line from SI to the outbound leg secondary boundary.

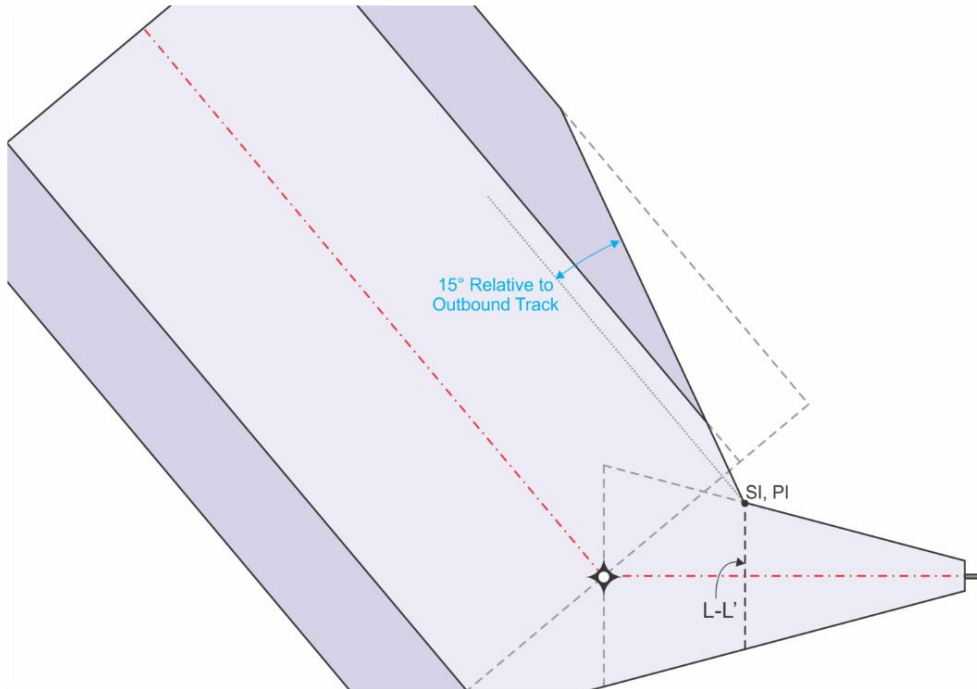
2 Case 2. SI is internal to either the outbound leg secondary or primary area. In this case, an alternative construction point (point SI') may be required depending on turn magnitude. Point SI' is the intersection of line L-L' and the outside-turn secondary area boundary. Construct the expansion line from either point SI or SI' to the outbound leg secondary boundary whichever results in the larger area. When the expansion line reaches the outbound secondary boundary before reaching the point of intersection of the inbound and outbound leg secondary boundaries, no expansion is required and the area is a simple connection of the inbound and outbound leg secondary boundaries.

c *Step 3.* Construct the primary area boundary. Construct an expansion line from point PI to the outbound leg primary boundary. When the *Step 2* expansion line reaches the outbound leg primary boundary before reaching point of intersection of the inbound and outbound leg primary boundaries, no expansion is required and the area is a simple connection of the inbound and outbound leg primary boundaries.

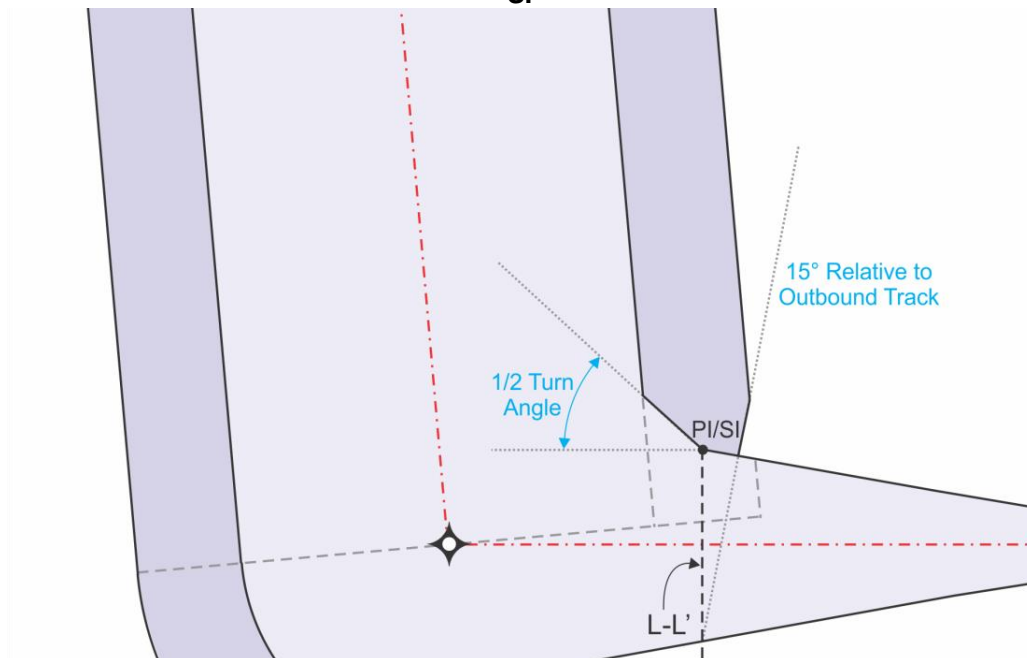
Figure 5-3-9. Early/Inside Turn, First Turn, Turn at Fix: PI and SI Collocated and External to Outbound Primary and Secondary



**Figure 5-3-10. Early Inside Turn, First Turn,
Turn at Fix: PI and SI internal to Outbound Primary**



**Figure 5-3-11. Early/Inside Turn, First Turn, Turn at Fix:
PI and SI internal to Outbound Secondary, Splay From Point
SI'**



**Figure 5-3-12. First Turn, Turn at Fix:
PI and SI Connecting to Abeam RF end Point**

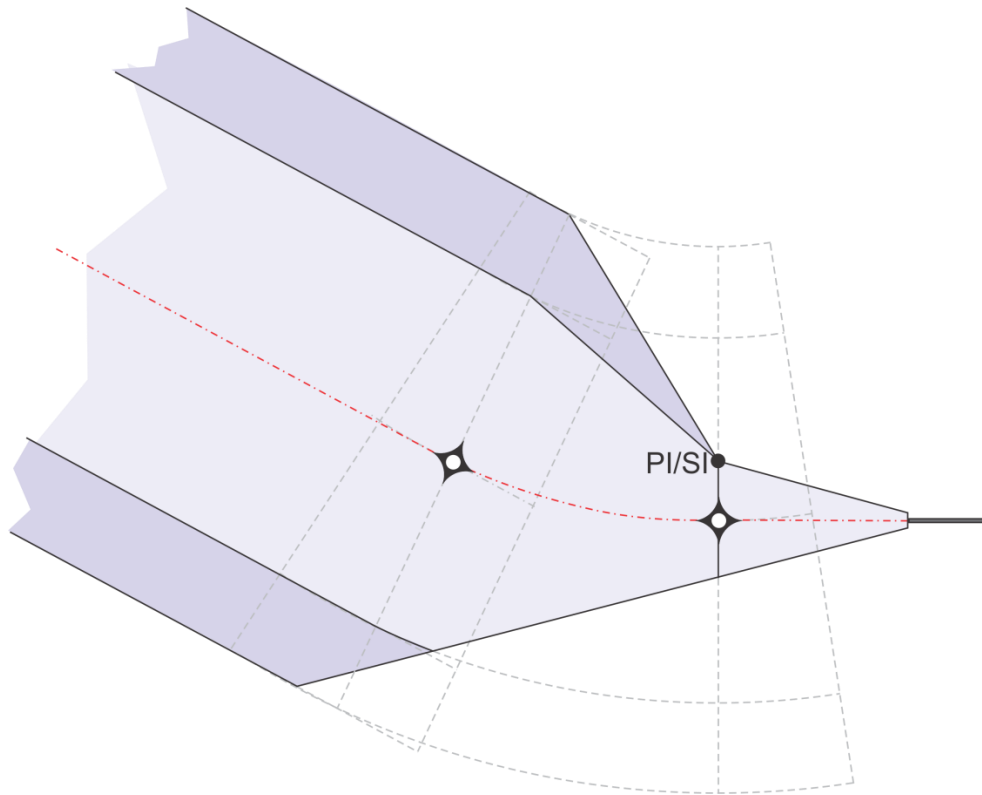
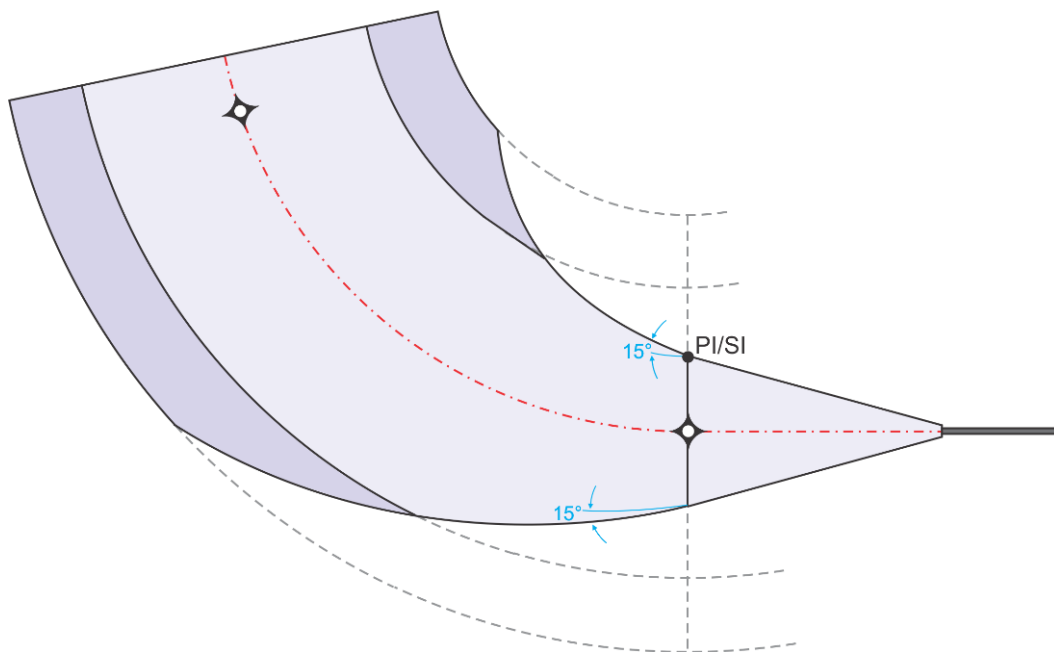


Figure 5-3-13. First Turn, Turn at Fix, RF



(2) Second turn. The second turn is specified at either a FB or FO fix.

(a) DF-TF or CF-TF.

1. Construct in accordance with paragraph 5-3-1.a(1)(c) with the following exceptions:

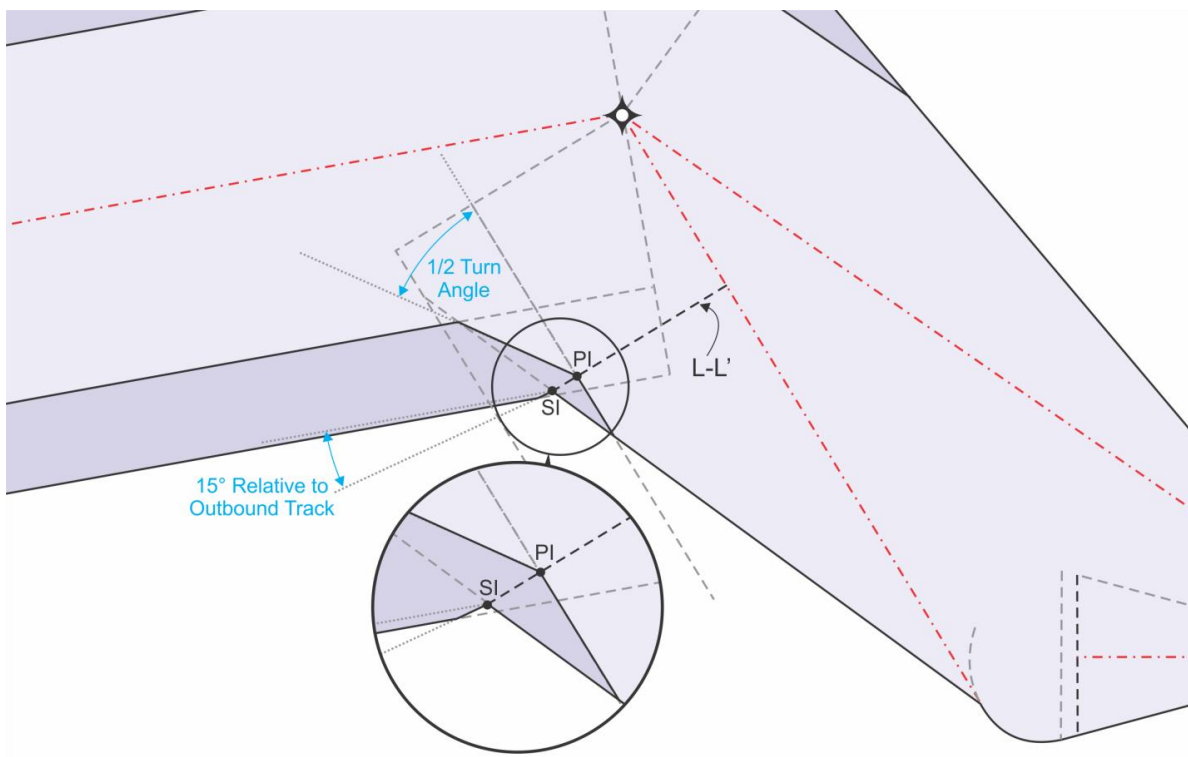
a Paragraph 5-3-1.a(1)(c)1 does not apply.

b For DF-TF, line L-L' is established perpendicular either on the early- or late-turn track, whichever is on the inside of the turn (see figure 5-3-14).

c When full primary and secondary width is reached at the early-turn baseline (L-L') construct in accordance with paragraph 1-2-5.

(b) Other connections. Construct in accordance with paragraph 1-2-5.

Figure 5-3-14. Early/Inside Turn, Second Turn, Turn at FB Fix DF-TF Connection



b. Late and outside turn construction.

(1) Wind spiral application. Wind Spiral (WS) construction applies to late and outside turns for turn-at-altitude, turn-at-fix (FO). See paragraph 1-3-1.g for design parameters.

(a) WS number and baselines. Baseline locations for WS construction may be dependent on Reaction and Roll distance (see formula 1-2-12 and table 5-3-4).

Table 5-3-4. Wind Spiral Number and Distance to Late Turn Baseline

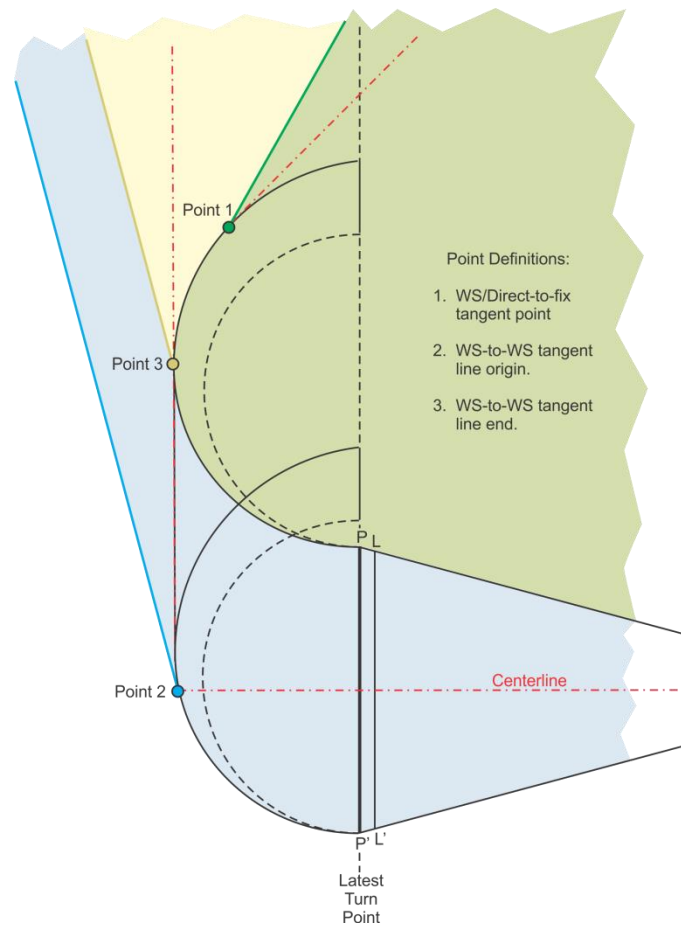
WS/Baseline Parameters	Turn-at-Fix (FO)	Turn-at-Altitude
WS Number	1 or 2	1, 2, or 3
WS1 & WS2 Distance Late-Turn Baseline (Line P-P') from turn fix/point	ATT + D _{rr}	ICA + D _{rr}
WS3 Baseline	ATT (early) for FO DF, otherwise not applicable	Parallel to DRL

1. First turn baseline. For first turn construction, the late-turn baseline (line P-P') marks the construction line for wind spirals. Line P-P' is located at the table 5-3-4 distance with points P and P' placed at the continuation of the inbound leg's outer boundary (see paragraph 1-3-1.g(4)). The no-wind turn radius for the outside turn wind spiral (WS1) and inside turn wind spiral (WS2) will be measured perpendicular to the inbound track from their respective point P. If a third wind spiral is used (WS3), the baseline will be parallel to DRL with the no-wind turn radius measured inward from DRP, for DF-DF baseline will be line L-L'.

2. Second turn baseline (see figure 5-3-22). To accommodate the two inbound tracks in the DF leg, the second turn construction uses two WS baselines, line P-P' for WS1 and line P'-P'' for WS2. Each late turn baseline is oriented perpendicular to the early- and late-turn tracks at the table 5-3-4 distance. The baseline for the inbound track nearer the outside-turn boundary is designated line P'-P'', with point P'' placed on the extended outer boundary and point P' placed at the no-wind turn radius inward along the baseline from point P. The baseline for the inbound track nearer the inside-turn boundary is designated line P-P', with point P' placed on inbound track and point P placed on the extended outer boundary. The no-wind turn radius for the inside turn wind spiral is measured perpendicular to the inside track inward along the baseline from point P.

(b) WS connection point and outside turn OEA boundary. Each WS has various connection options along its path, which predicate the outside turn OEA boundary. The chosen connection must provide the most reasonably conservative track and protection area. Where excessive splay is required to reach full-width protection, consider lengthening the leg, restricting the speed, category, etc. to avoid protection and/or construction difficulties. Consider full-width protection to exist at the fix where the splay line is tangent to a full-width-radius circle about the fix.

1. Turn-at-altitude (see figure 5-3-15). For turns at an altitude, the 15-degree or greater splay line that joins the outbound leg outer boundaries may originate from the WS/direct-to-fix tangent point (point 1), the WS to WS tangent line origin (point 2) or the WS to WS tangent line end (point 3). Where the turn angle is ≤ 105 degrees, or the divergence angle between the WS to WS tangent line and the direct-to-fix line is ≤ 15 degrees, apply the splay line from the WS to WS tangent line origin. DF secondary areas begin/exist only where full width primary exists.

Figure 5-3-15. WS Connection Points, Turn-at-Altitude

2. Turn-at-fix (FO). For turns at a fix, the connection point and the OEA boundary is dependent on the WS boundary relative to the outbound leg OEA (extended). The connection point is the point of tangency on the WS in both cases. When the WS is not contained in the outbound leg OEA, the outside boundary is a 30-degree converging line relative to the outbound course (see figure 5-3-16). When the WS is contained in the outbound leg OEA, the outside boundary is a 15-degree or greater splaying line relative to the outbound course from the WS/outbound leg parallel point until reaching the outbound leg boundaries (see figure 5-3-17).

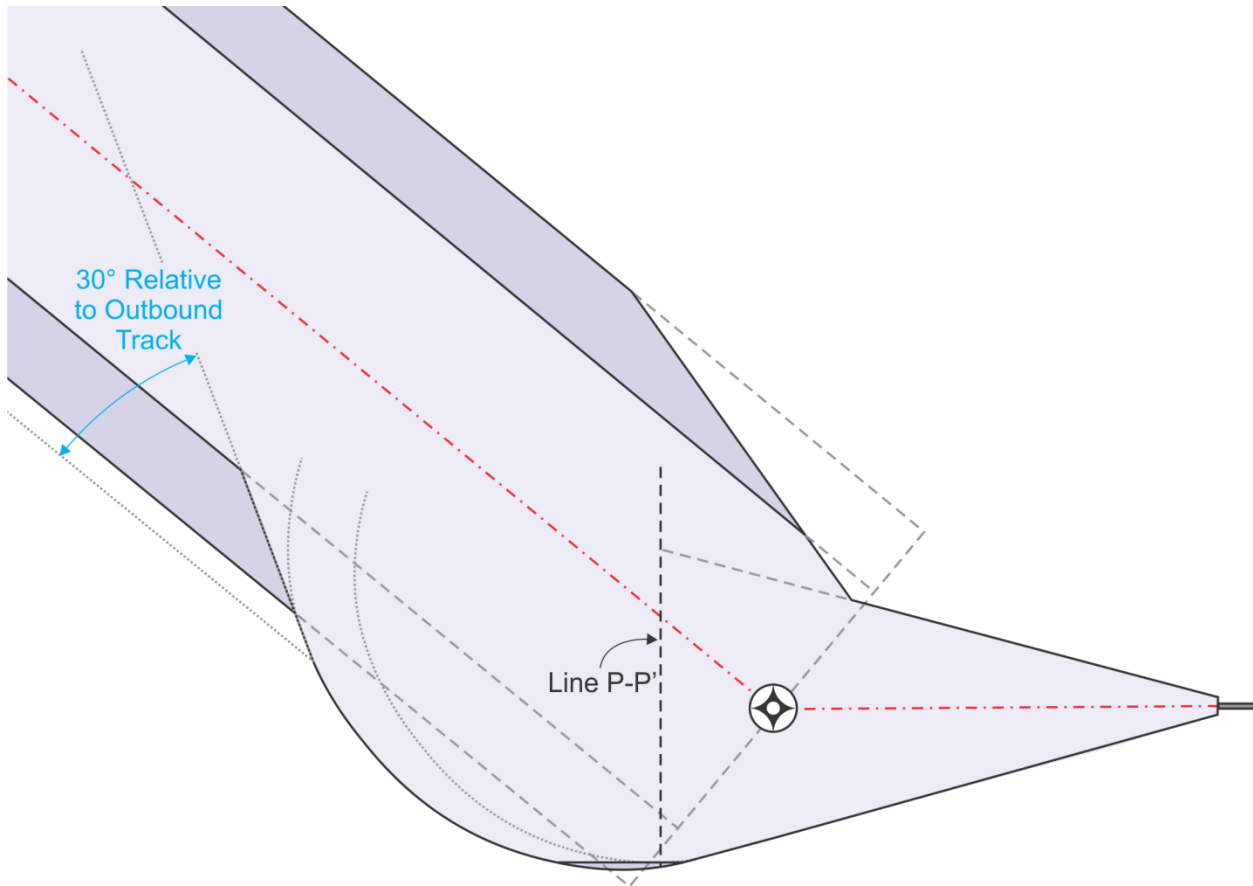
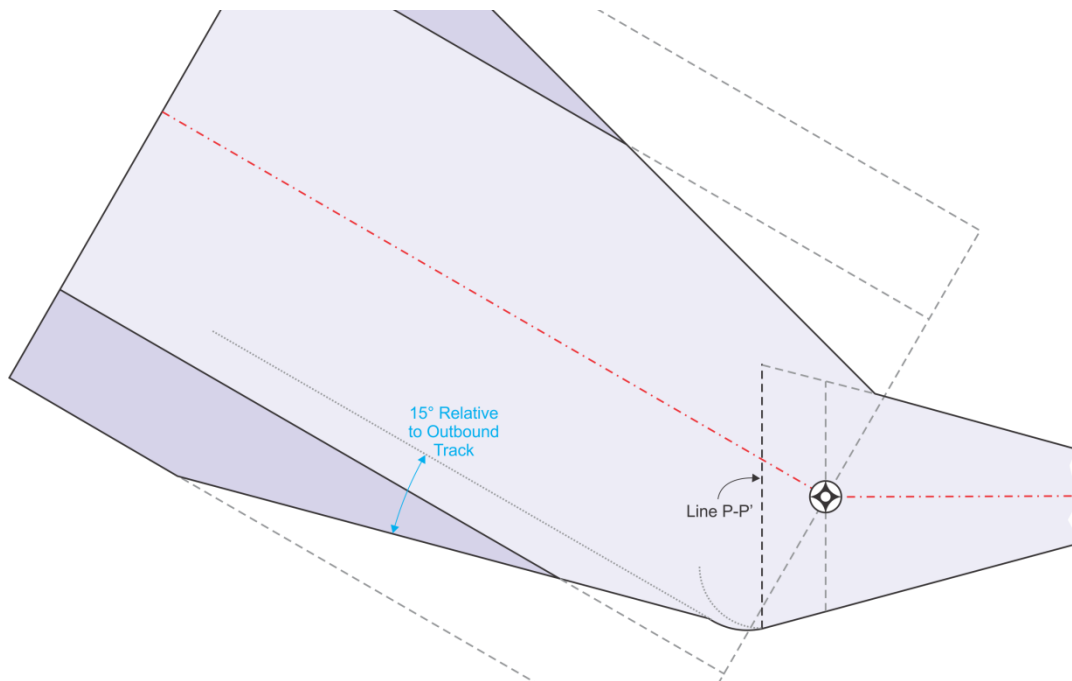
Figure 5-3-16. Late/Outside Turn, Wind Spiral, WS Not Contained in Outbound OEA

Figure 5-3-17. Late/Outside Turn, Wind Spiral, WS Contained in Outbound OEA

(c) Determining multiple WS necessity. To determine multiple wind spiral necessity, construct the additional WS in the direction of turn from its prescribed location to its connection point. Where the additional WS intersects the preceding WS construction (including the connecting and expansion lines), connect the wind spirals with a tangent line that is parallel to the WS center points. Otherwise, revert to the previous WS construction.

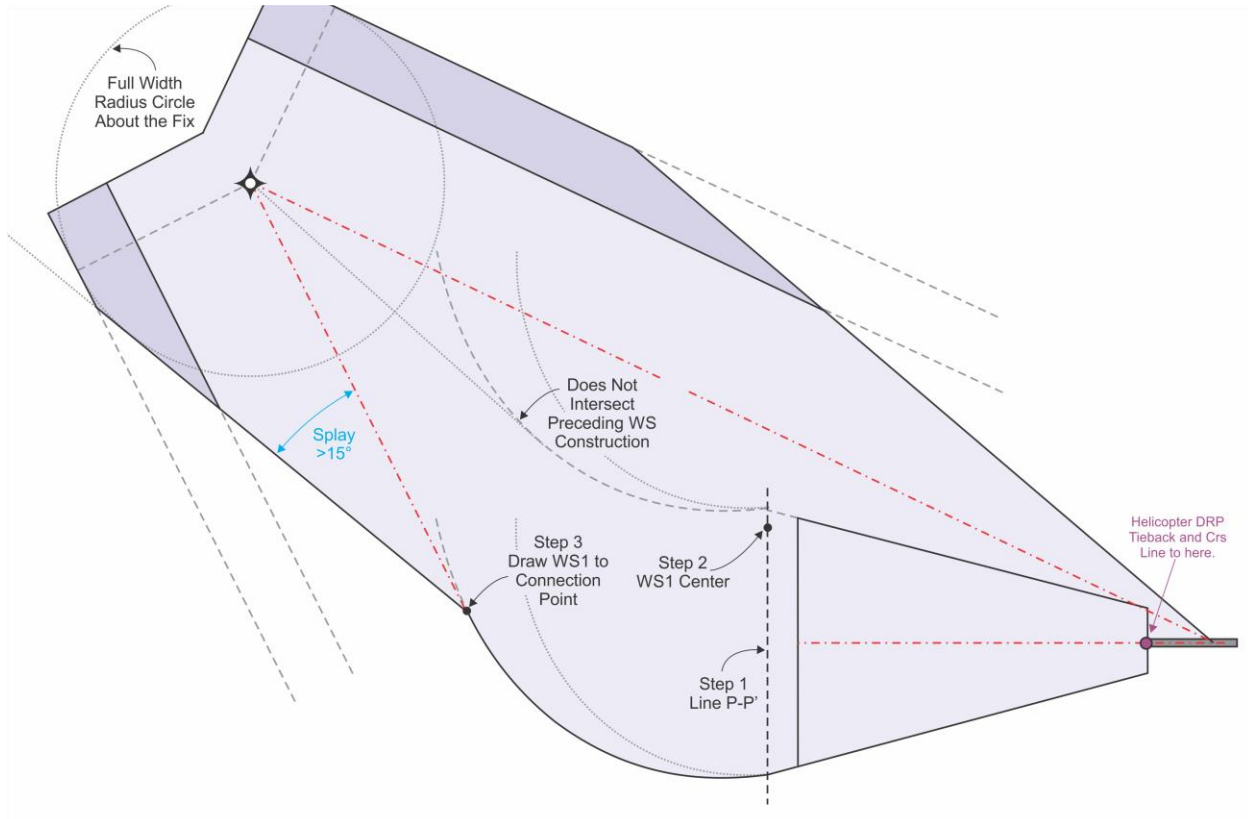
(2) First turn.

(a) Turn-at-altitude, turn-at-fix (FO), and minimum VI-CF construction.

1. *Step 1.* Construct the late-turn baseline (P-P') perpendicular to the runway centerline extended (or offset course if paragraph 5-2-1.a is used) at the late-turn-point. For minimum VI-CF construction late-turn point is 1 NM past the intercept point.

2. *Case 1* (see figure 5-3-18). Small turns using one WS.

Figure 5-3-18. Late/Outside Turn, First Turn, Case 1, Turn-at-Altitude or FO Fix with 1 WS

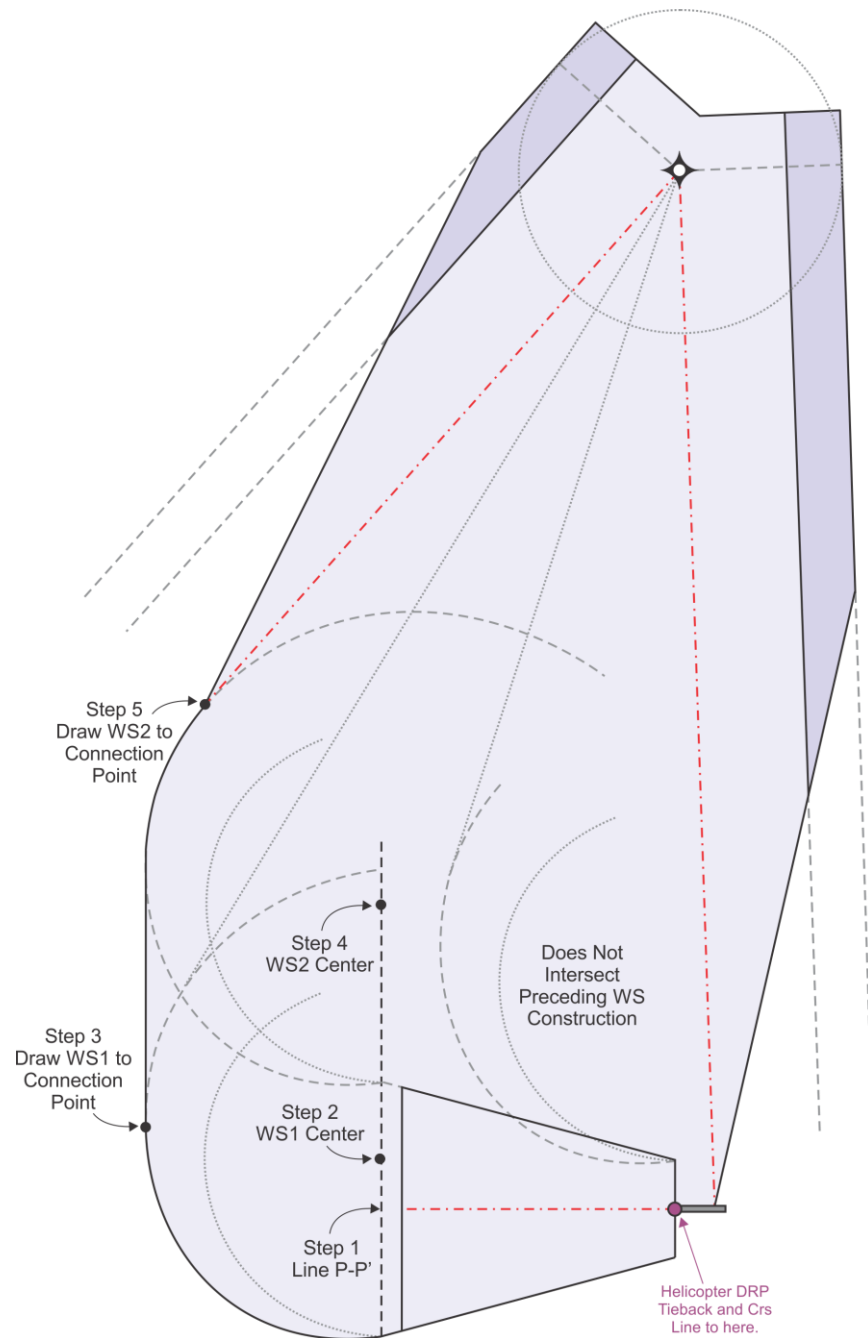


a *Step 2.* Locate the WS1 center on line P-P' at no-wind turn radius distance from point P'.

b *Step 3.* Construct WS1 from the outside turn point in the direction of turn until reaching its connection point. For minimum VI-CF construction use R (based on the highest projected turn altitude), place an arc center point at R distance from the outside turn point, construct an arc at distance R + 1 NM (this will result in a circle instead of a WS) (see figure 5-3-6).

3. *Case 2* (see figure 5-3-19). For turns nearing or greater than 90 degrees using more than one WS, complete *Steps 1-3* then:

**Figure 5-3-19. Late/Outside Turn, First Turn, Case 2,
Turn-at-Altitude or FO Fix with 2 WS**



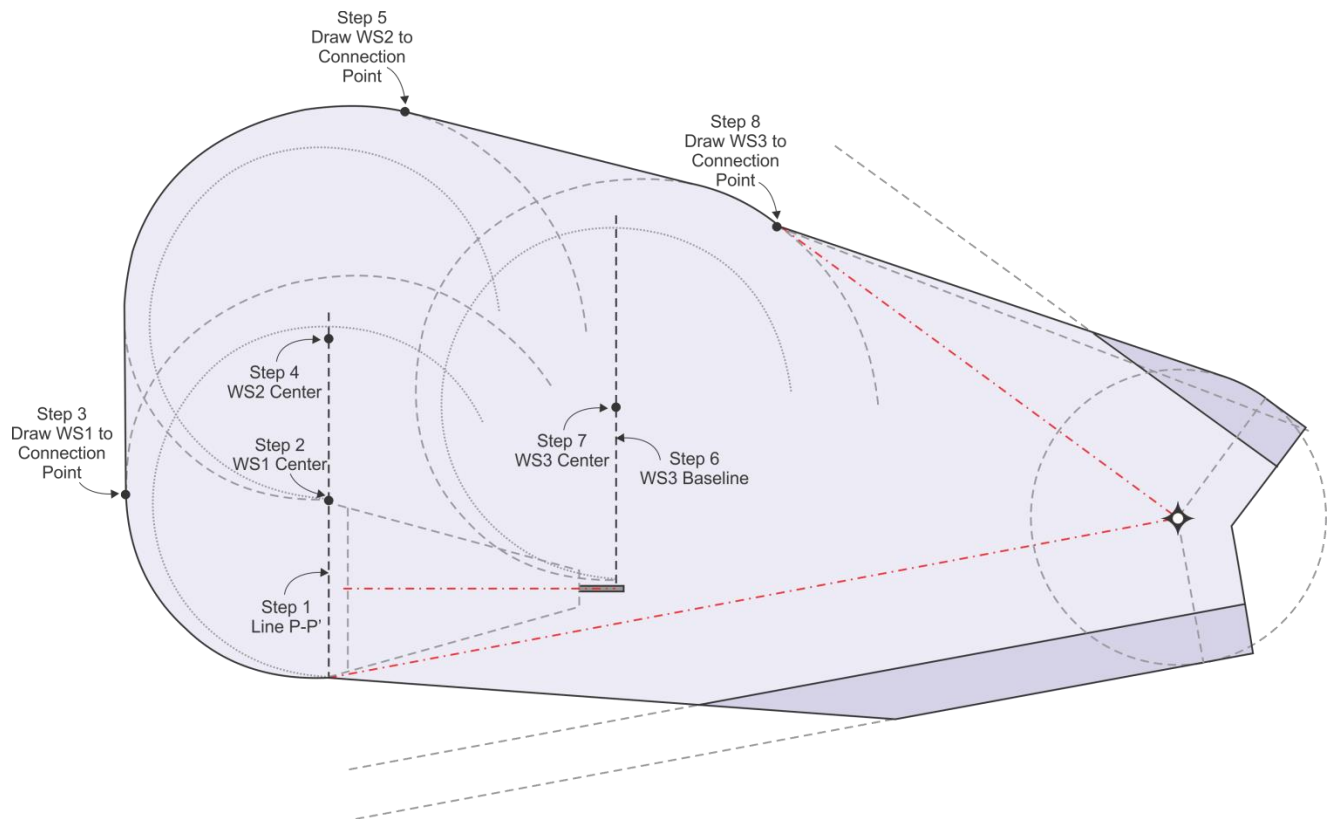
a *Step 4.* Locate the WS2 center on P-P' at no-wind turn radius distance from point P.

b *Step 5.* Construct WS2 from the inside turn point in the direction of turn until reaching its connection point. For minimum VI-CF construction use R (based on the highest

projected turn altitude), place an arc center point at $2 \times R$ distance from the outside turn point, construct an arc at distance $R + 1$ NM (this will result in a circle instead of a WS) (see figure 5-3-7).

4. *Case 3* (see figure 5-3-20). Turns nearing or greater than 180 degrees using more than two WS, complete *Steps 1-5* then:

Figure 5-3-20. Late/Outside Turn, First Turn, Case 3, Turn-at-Altitude with 3 WS



a *Step 6.* Construct the WS3 baseline perpendicular to the runway centerline along DRL extended toward the turn side. For FO DF the WS3 baseline is on line L-L' perpendicular to the runway centerline extended toward turn side.

b *Step 7.* Locate the WS3 center on the baseline at no-wind turn radius distance from DRP.

c *Step 8.* Construct WS3 from this point in the direction of turn until reaching its connection point.

(b) Turn-at-fix (FB) and standard VI-CF construction (see figure 5-3-21).

1. *Step 1.* Construct the outer primary boundary using a radius of $2 \times$ segment XTT centered on the plotted fix position, truncated at the inbound leg extended primary boundary until tangent to the outbound leg primary boundary (not applicable to VI-CF).

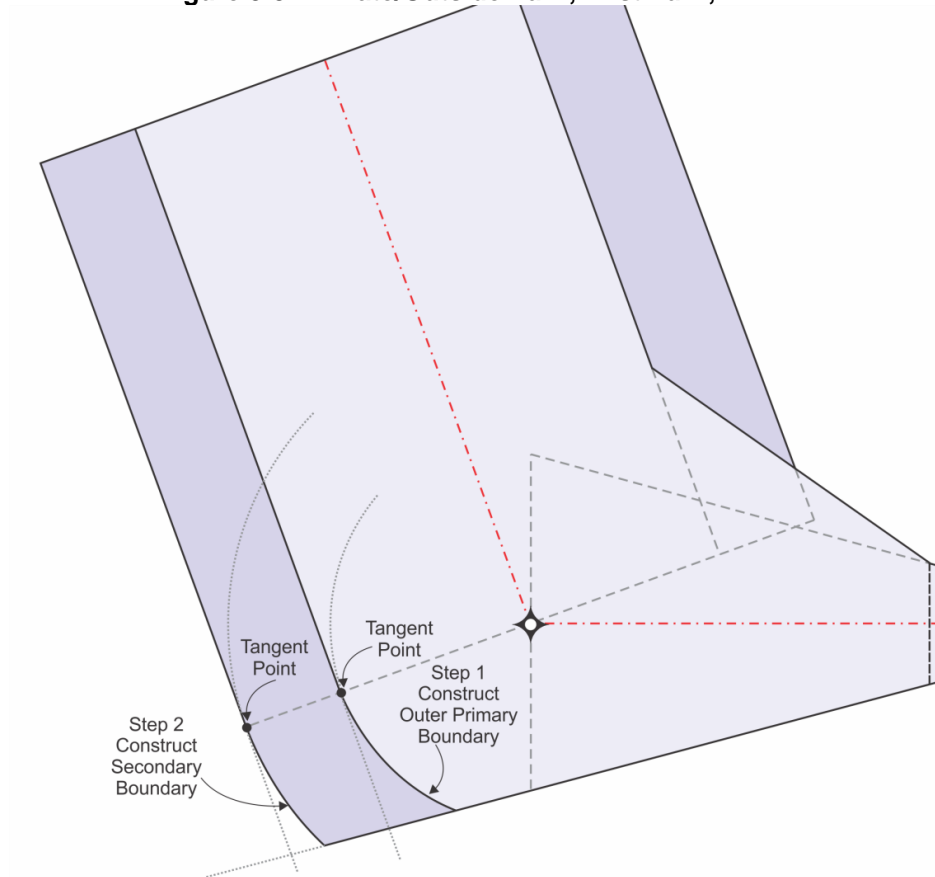
2. *Step 2.* Construct the secondary boundary using a radius of $3 \times$ segment XTT entered on the plotted fix position, truncated at the inbound leg extended outer boundary until tangent to the outbound leg outer boundary. For VI-CF this will be the outer primary boundary connecting to point X then parallel to CF course to point W (see figure 5-3-3).

(c) Turn-at-fix, RF construction.

1. *Step 1.* Construct the outer primary boundary by splaying at 15 degrees relative to RF path from point abeam RF start point. Where the splay doesn't reach full OEA width at a point that supports standard construction, then construct by truncating the RF full width primary area at the initial straight leg extended splay line (see figure 5-3-13).

2. *Step 2.* Construct the secondary boundary by splaying at 15 degrees relative to RF path from point abeam RF start point. Where the splay doesn't reach full OEA width at a point that supports standard construction, then construct by truncating the RF full width secondary area at the initial straight leg extended splay line (see figure 5-3-13).

3. Where the initial straight leg splay hasn't reached a width of $3 \times$ segment XTT by the end of the first RF and the succeeding legs is a TF or RF with same direction turn, the initial straight leg splay may continue into the next leg until full width is reached (see figure 5-3-12); otherwise, increase splay as necessary to reach full OEA width by the end point of the first RF.

Figure 5-3-21. Late/Outside Turn, First Turn, FB Fix

(3) Second turn.

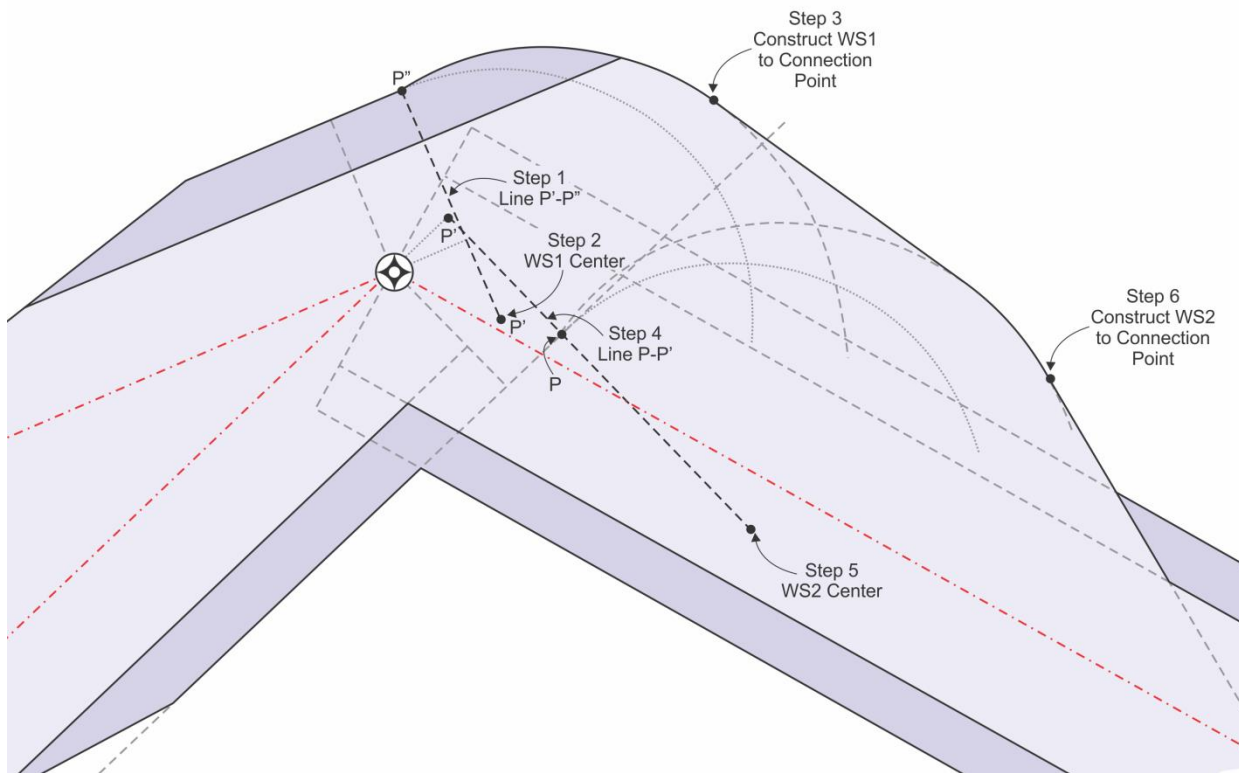
(a) Turn-at-fix (FO) (see figure 5-3-22).

1. *Step 1.* Construct the WS1 baseline, (line P'-P'') perpendicular to the DF track nearer the outside of the DF-TF turn at the late turn point.

2. *Step 2.* Locate the WS1 center on line P'-P'' at no-wind turn radius distance from point P."

3. *Step 3.* Construct WS1 the outside point in the direction of turn until reaching its connection point.

4. *Step 4.* Construct the WS2 baseline, (line P-P') perpendicular to the DF track nearer the inside the DF-TF at the late turn point.

Figure 5-3-22. Late/Outside Turn, Second Turn, FO Fix

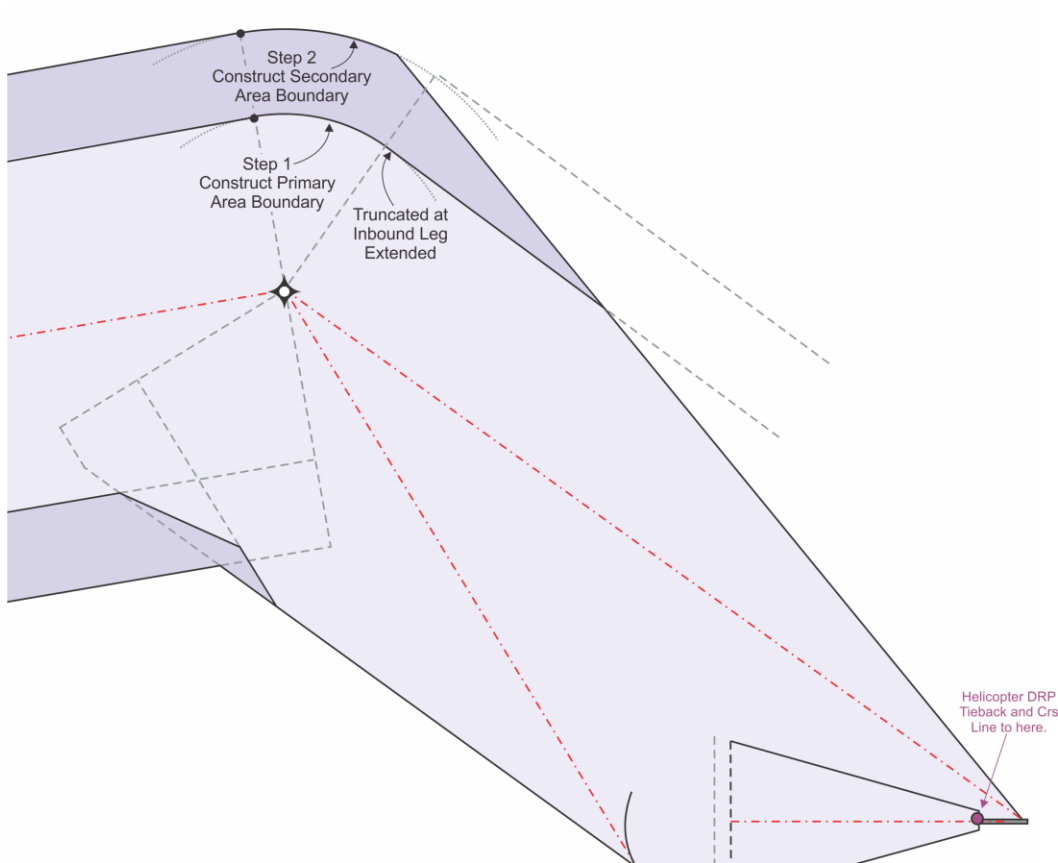
5. *Step 5.* Locate the WS2 center on line P-P' at no-wind turn radius distance from point P.

6. *Step 6.* Construct WS2 from the inside turn point in the direction of turn until reaching its connection point.

(b) Turn-at-Fix (FB).

1. When the inbound leg outside boundary is less than full primary and secondary width reached perpendicular to the fix, construct in accordance with paragraph 5-3-1.b(2)(b) (see figure 5-3-23).

**Figure 5-3-23. Late/Outside Turn, Second Turn, Turn at FB
Fix, Less Than Full OEA Width Perpendicular to Fix**



2. When the inbound leg outside boundary is full primary and secondary width perpendicular to the fix, construct in accordance with paragraph 1-2-5.

Section 5-4. Obstacle Evaluation

5-4-1. Obstacle Evaluations.

a. Utilize the general concepts from paragraph 1-3-1.g. Where an obstacle requires multiple measurements (an obstacle is equidistant from multiple primary boundary points, it lies along perpendiculars from multiple primary boundary points, etc.), apply the most adverse result from each of the combined primary/secondary measurements (see figure 5-4-1 through figure 5-4-7). For RNP AR DP apply horizontal and vertical accuracy adjustments to obstacles except for those excluded obstacles in Order 8260.19, paragraph 2-11-4.

b. Primary OCS.

(1) ICA. Measure distance to obstacles using the distance from ICAB along runway centerline extended. The ICA OCS begins at the MSL elevation of the ICAB (see formula 1-3-5 to determine OCS slope).

(2) Straight departure. Measure and apply the OCS slope along the shortest primary area distance from the ICA to a point at/abeam the obstacle. For RNP AR DP, obstacle distance is measured as the along track distance from the ICAB to a point abeam the obstruction.

(3) Turn-at-altitude (single and multiple legs). Apply the OCS slope along the shortest primary area distance from the ICA/straight segment boundary to a point at/abeam the obstacle.

(a) Section 1. For obstacles past the ICAB, measure from the closest point on the ICA/straight segment boundary. For obstacles between the DRL and ICAB, take the lesser of the distance from runway centerline and the closest point on the ICA boundary. The section 1 OCS begins at the MSL elevation of the OCS at ICAE.

(b) Section 2. Measure from DRP. The section 2 OCS begins at the specified turn altitude.

(4) Turn-at-a-fix (single and multiple legs). Apply the OCS slope along the shortest primary area distance from the ICA to a point at/abeam the obstacle. For RNP AR DP, obstacle distance is measured as the along track distance from the ICAB to a point abeam the obstruction. When the RNP AR DP has a DER crossing restriction of 500 feet above airport elevation (rounded to the nearest foot), the OCS begins 380 feet above DER.

(5) VI-CF.

(a) VI-CF standard construction and minimum VI-CF with turns of 30 degrees or less. Apply the OCS slope along the shortest primary area distance from the ICA boundary to a point at/abeam the obstacle.

(b) Minimum VI-CF with turns of more than 30 degrees. For obstacles past the DER, measure from the closest point on the ICA boundary. For obstacles between the DRP and DER, take the lesser of the distance from runway centerline and the closest point on the ICA boundary.

(6) Routes. Apply ICA, section 1 and section 2 methodology where applicable, then measure along the shortest primary area distance that passes through the early-turn baseline of all preceding legs.

Note: The shortest primary area distance is the length of the shortest line kept within primary area that passes through the early-turn baseline of all preceding legs.

c. Secondary OCS. For obstacles located in secondary areas, calculate the primary OCS elevation as stated, then apply a 12:1 OCS slope along the shortest secondary distance to the obstacle. In straight legs, this is normally perpendicular to the nominal track. In expansion areas, the slope rises in a direction perpendicular to the primary boundary (arc, diagonal corner-cutter, etc.).

d. Maximum Altitude Restrictions. Apply Order 8260.3, paragraph 13-6-1, for fix error use ATT. For RF legs, late ATT lies along the RF track and on the construction line anchored at the RF center point (see figure 5-4-8).

Figure 5-4-1. VA-DF Obstacle Distance Measurements, Primary and Secondary Area, Large Turn

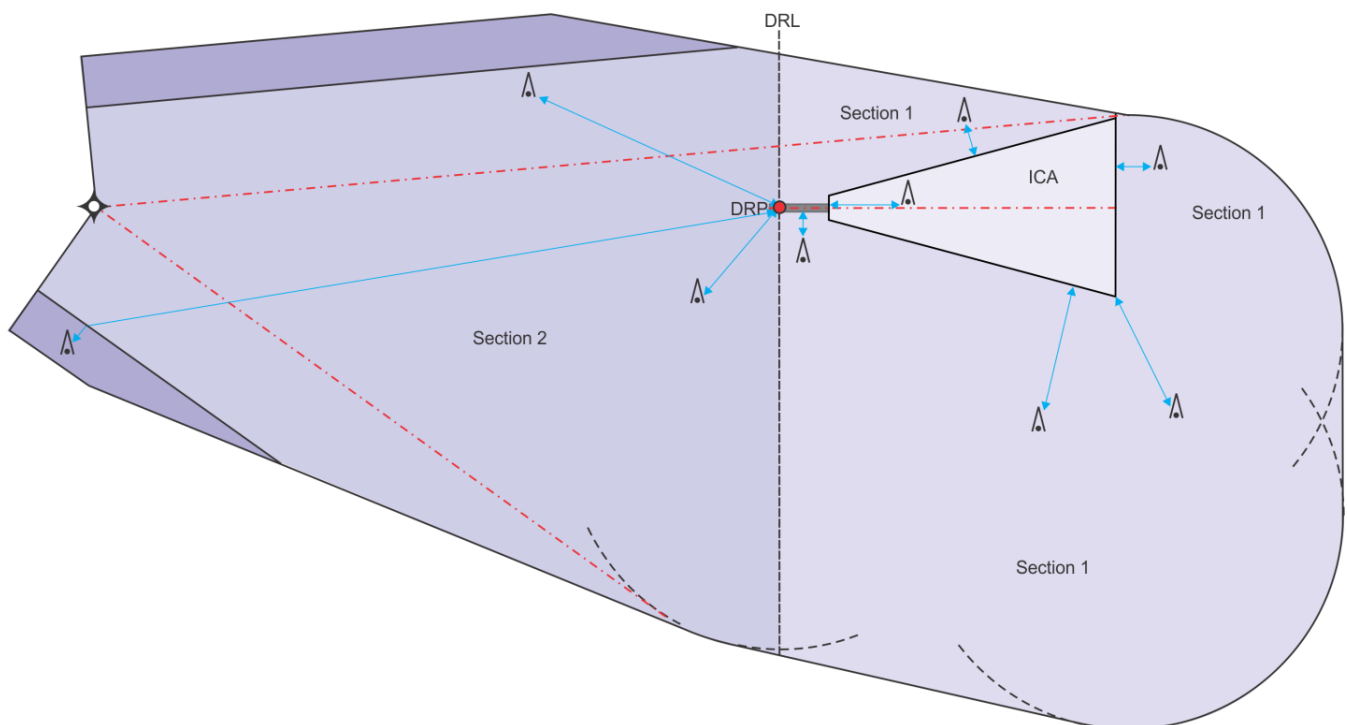


Figure 5-4-2. Obstacle Distance Measurements, Turn-at-Altitude, Small Turn

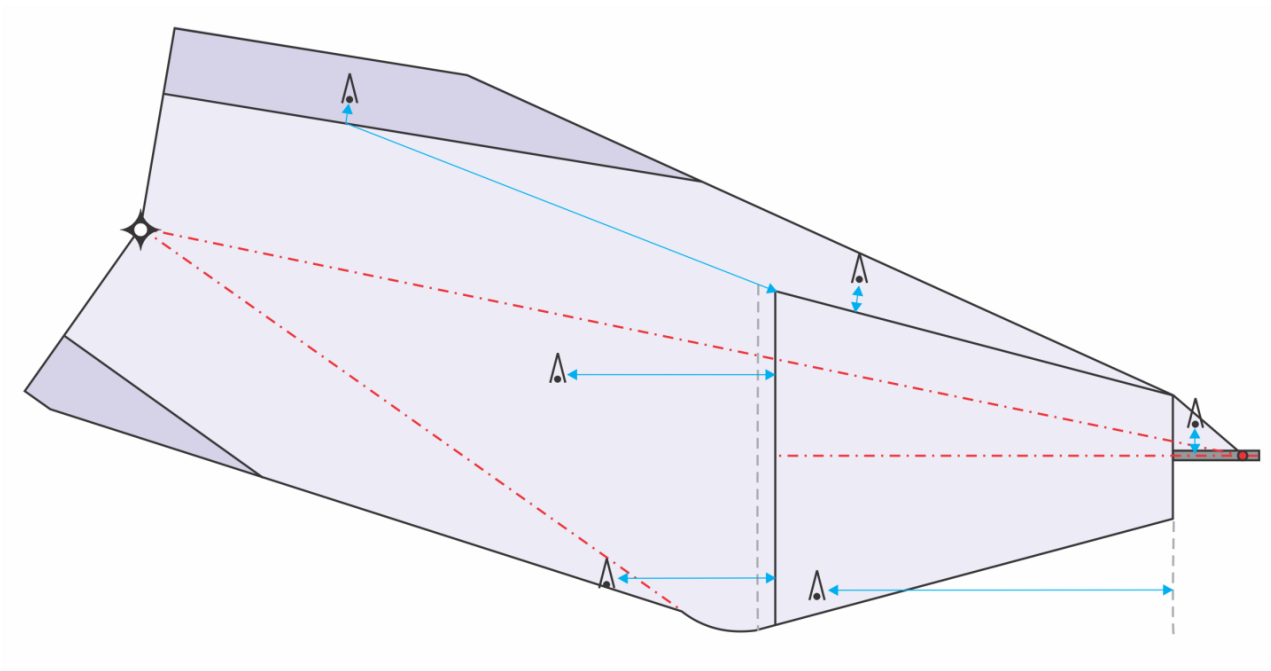


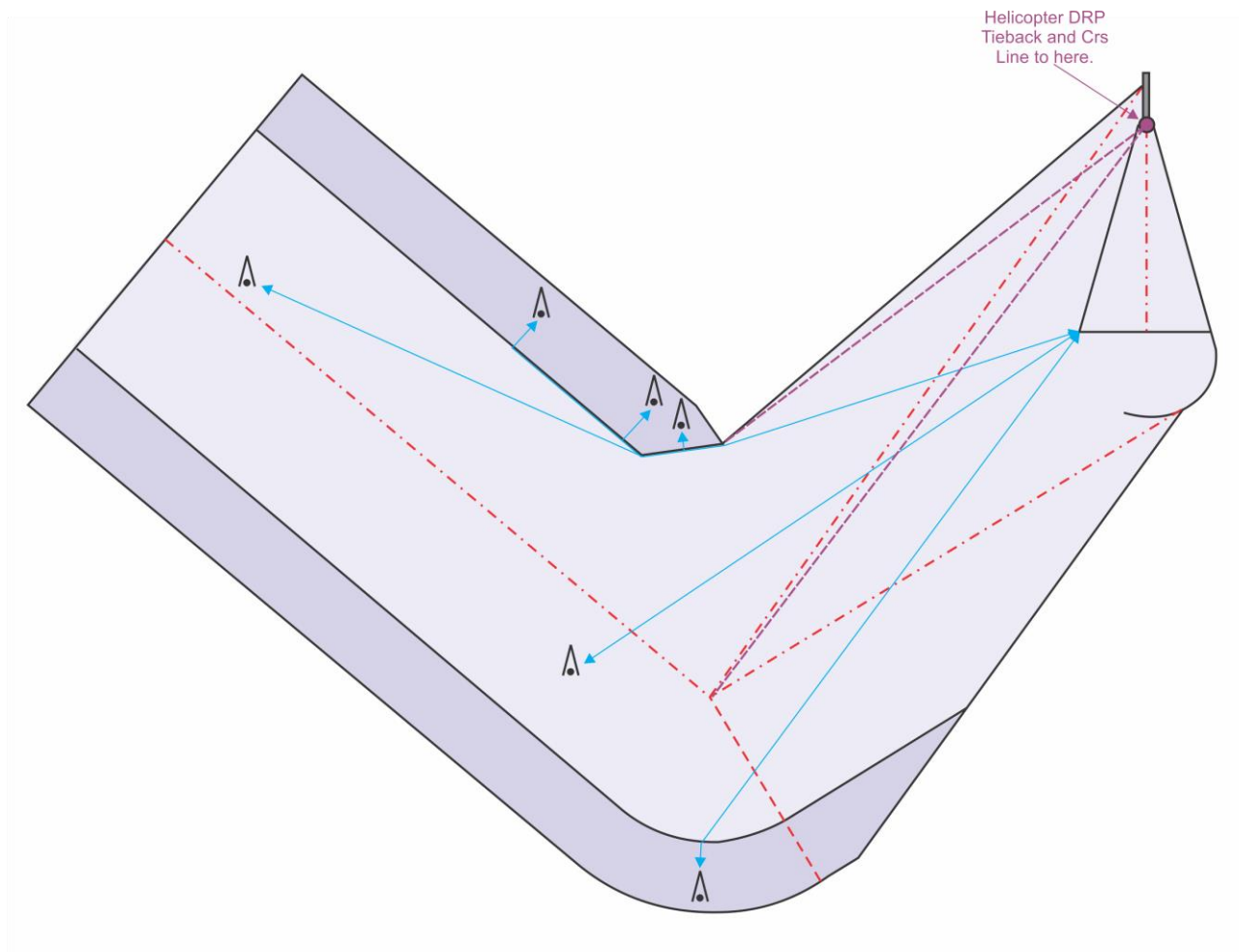
Figure 5-4-3. Obstacle Distance Measurements, Turn-at-Altitude, Multiple Legs

Figure 5-4-4. Obstacle Distance Measurements, Turn-at-FB Fix

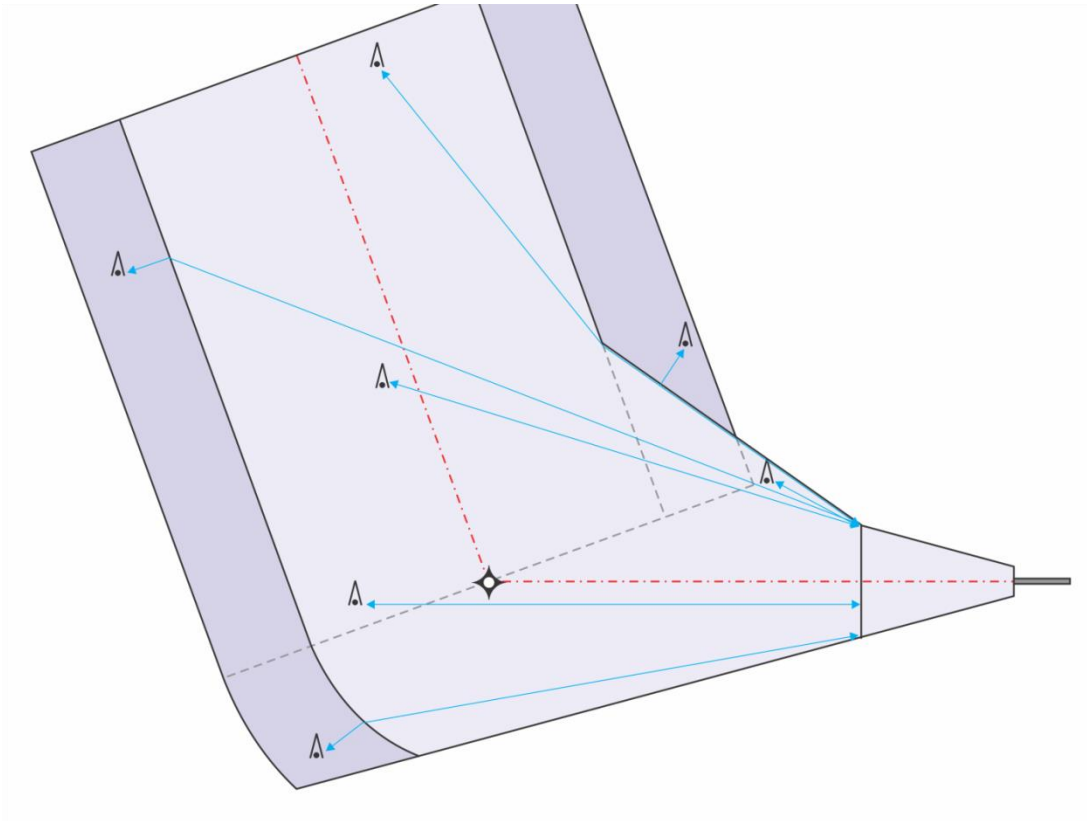


Figure 5-4-5. DF-DF Obstacle Distance Measurements, Turn-at-FO Fix

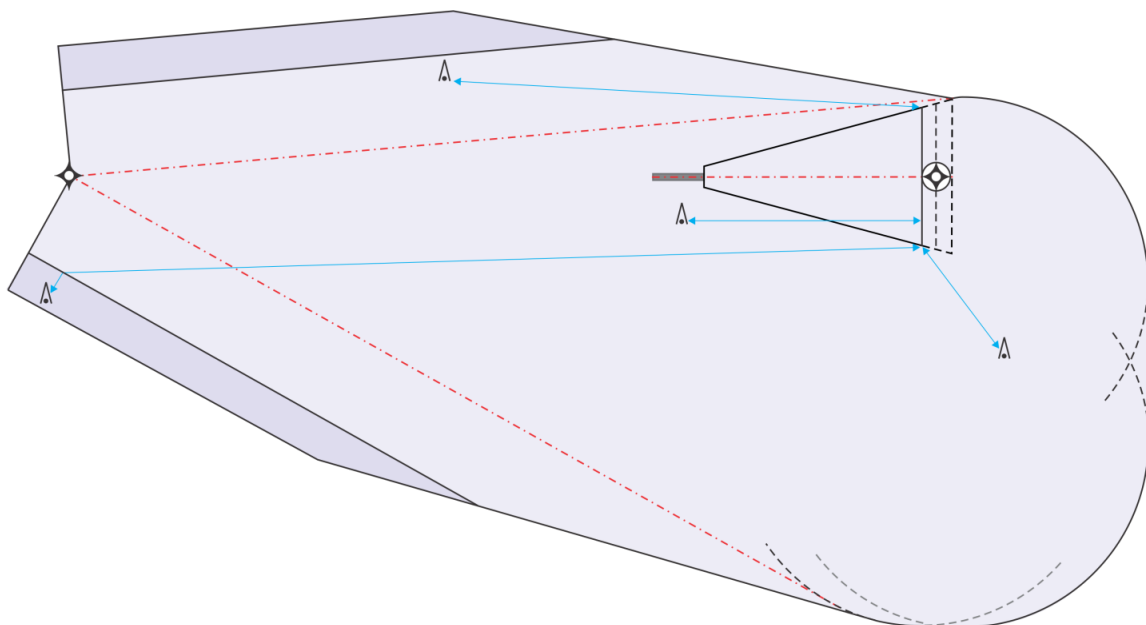


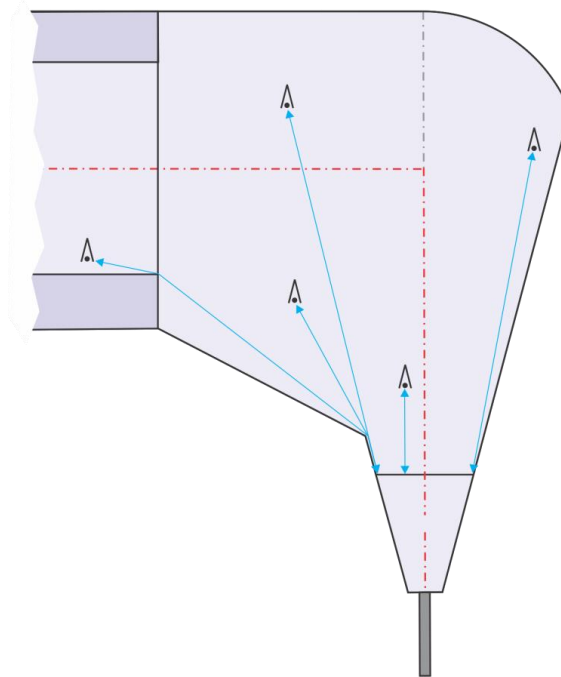
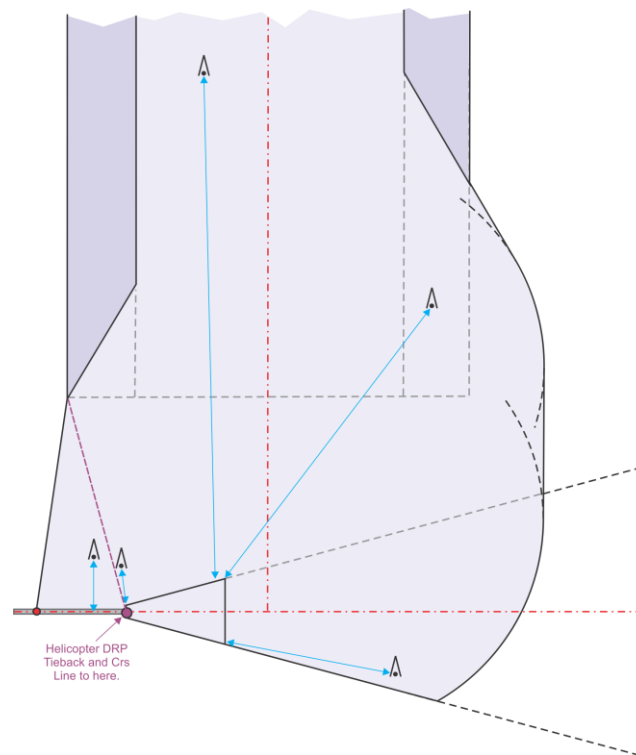
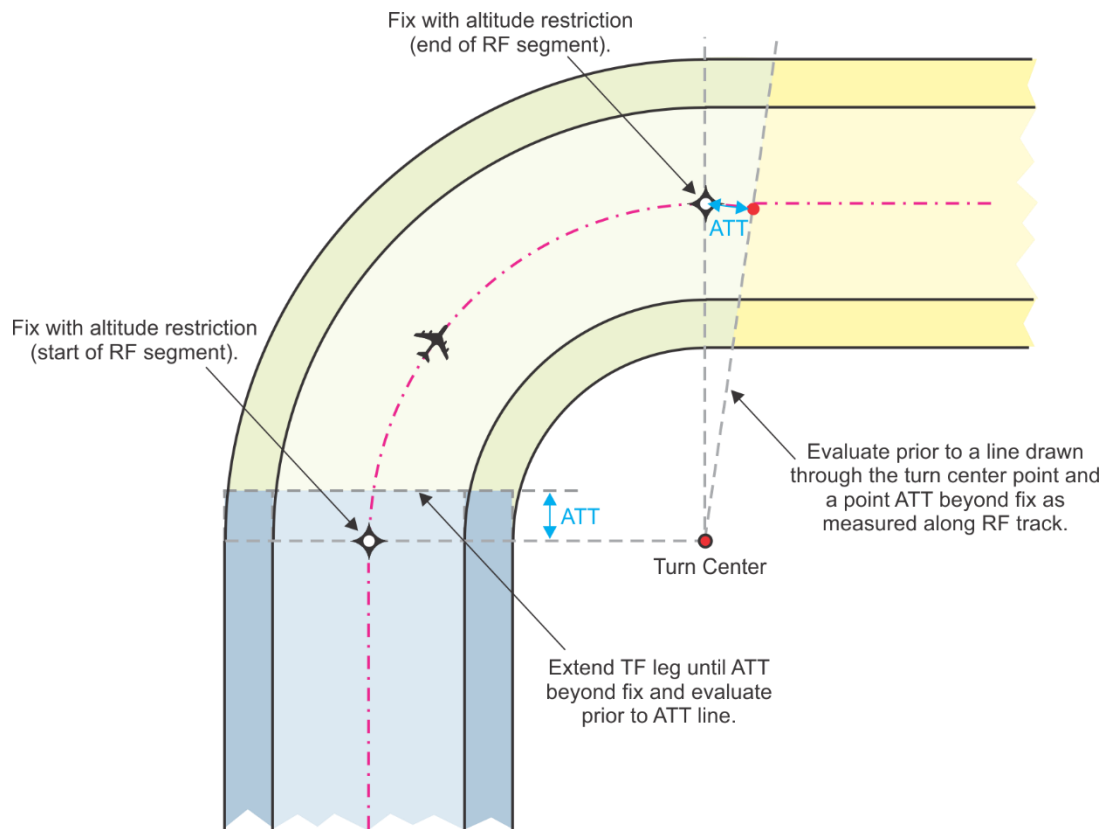
Figure 5-4-6. VI-CF Obstacle Distance Measurements**Figure 5-4-7. Minimum VI-CF with Turn Greater than 30 Degrees Obstacle Distance Measurements**

Figure 5-4-8. Maximum Altitude Restriction Evaluation

Section 5-5. Climb Gradient

5-5-1. Climb Gradient. Where the default OCS slope is penetrated, a greater than standard CG may be required to clear the penetrating obstruction. A CG for LNAV engagement when used along with a reduced higher than standard CG and/or a CG greater than 500 ft/NM (600 ft/NM for helicopters) requires Flight Standards approval.

a. When climb gradients are used for obstacle clearance, compute the climb gradient per paragraph 1-3-1.g and formula 1-3-7 using the CG termination altitude from formula 5-5-1.

Formula 5-5-1. CG Termination Altitude

$$CG_{\text{term}} = \frac{O_{\text{MSL}} - \frac{d_{\text{primary}}}{12} - DER_{\text{elev}}}{0.76} + DER_{\text{elev}}$$

Where:

DER_{elev} = departure end of runway elevation (feet)

d_{primary} = dist. (feet) from edge of primary to obstacle (zero if not in secondary)

Example:

$$CG_{\text{term}} = \frac{2147.41 - \frac{390.12}{12} - 1104.3}{0.76} + 1104.3$$

$$CG_{\text{term}} \approx 2434.04 \text{ feet}$$

b. Mitigating obstacle penetrations. The preferred method of obstacle mitigation is to use a less onerous route. The next choice is a climb gradient, using paragraph 1-3-1.g(2). Also see Order 8260.3, chapter 13 and Order 8260.46.

c. Climb in a holding pattern. Where required, apply climb-in-hold criteria contained in Order 8260.3, section 16-7.

Section 5-6. Vector SID

5-6-1. General. When operationally advantageous, a SID may be developed starting with or including a vector leg followed by an RNAV route. (See appendix E for implementation considerations.) A-RNP and RNP AR DP NavSpecs are not authorized.

5-6-2. SID Starting with Vector Leg. When designing an RNAV route following vectors, establish the first waypoint as an Initial Fix leg (IF). No OEA construction or DME/DME screening is required prior to the IF of the RNAV route being joined.

a. The IF beginning an RNAV route must fall within an area and be at an altitude protected by an MVA/MIA. The IF altitude may not be less than 500 feet above airport elevation (rounded to the nearest foot). ATC must be consulted and consideration must be given to the time/distance required for radar identification and normal vectoring when designing the IF.

b. A single RNAV course must be defined from the IF.

c. The IF must be a FB fix. A TF leg must follow the IF. General OEA construction applies (see paragraph 1-2-5).

d. The length of the first leg must be sufficient to accommodate a 90-degree turn at the IF. Use standard turn parameters. Where a shorter leg is needed, reduce airspeed in increments of not less than five KIAS until the desired length is achieved (see table 1-2-2). An alternative method to allow design of a shorter first leg is to limit the amount of intercept to 45 degrees, this must be approved and applied by ATC.

e. Each SID is limited to one common route and must start with a single IF.

f. OCS slope and origin. A flat surface evaluation is conducted from the IF early ATT to late ATT. A sloping OCS (see formula 1-3-5) originates at IF late ATT. The OCS starting elevation is the MVA/MIA altitude minus ROC. Where multiple MVA/MIA sectors apply, the most demanding surface must be used. Climb gradients are not authorized as mitigation for obstacles that penetrate the flat or sloping OCS, the surface must be raised by increasing fix altitudes or redesigning the route.

5-6-3. SID Including Vector Legs. A SID may start with an RNAV route that has a manual termination followed by vectors to join an RNAV route (Open SID). Construct and evaluate the route off the runway using the applicable straight or turning criteria. No OEA construction or DME/DME screening is required between the fix beginning a manual termination leg and the IF of the RNAV route being joined.

a. The RNAV route off the runway must terminate with either an FM or VM.

b. An FM beginning with a FB fix is preferred. This combination encompasses the most operators.

c. An FM that results in a turn must start with a FB fix.

d. An FM that doesn't require a turn may be either a FB or FO fix.

e. A VM must start with a FO.

f. The fix beginning a manual termination leg must fall within an area and at an altitude protected by either an MVA/MIA, free vector area, ATC prominent obstacle display, or diverse vector area (DVA).

(1) When a sectorized DVA is used, the heading/course of the manual termination leg must comply with the heading limitations of that DVA sector.

(2) If the manual termination leg starts with a FO fix and a sectorized DVA is used, the turn to the heading/course must be contained within that DVA sector.

g. For an RNAV route following a vector segment, comply with paragraphs 5-6-2.a thru 5-6-2.f.

Section 5-7. Helicopter Departures

5-7-1. General. For departures from IFR runways or heliports that support IFR procedures, apply criteria in accordance with sections 5-1 through 5-6. ICAB, HDRP, and DRL for departures from a heliport that supports IFR procedures are located at the edge of the FATO departure side.

a. PinS departures may be conducted from VFR heliports, unmarked landing areas, and VFR runways that do not have an ODP. PinS departures are designed to allow a pilot to navigate to a point where IFR flight may commence called the initial departure fix (IDF). The departures are categorized as Proceed VFR or Proceed Visually.

(1) Proceed VFR does not provide obstruction evaluation until reaching the IDF flat surface area where IFR obstruction clearance begins.

(2) Proceed Visually does provide obstruction evaluation from the departure point to the IDF.

(3) Positive course guidance (PCG) and obstruction clearance must be provided from the IDF to the latest ATT at the departure termination fix WP for Proceed VFR PinS departures.

b. For Proceed Visually departures an IDF is established between 0.55 NM and 5.0 NM from the departure point. Consideration should be given to requirements for air traffic, mission support, obstacles, and airspace when determining the location of the IDF.

(1) While proceeding outbound on the approach inbound course may be an option, it may not serve other requirements.

(2) An outbound heading/course and obstruction clearance must be provided from the helipad to the nominal position of the IDF. If needed, a non-standard climb gradient as applicable is provided from the HCH to cross the IDF.

c. The departure procedure consists of an IDF flat surface area, which is a level surface area to initiate the departure procedure. Section 1 begins at the edge of the IDF flat surface area and extends in the direction of flight at full width utilizing primary and secondary areas (see figure 5-7-1). Section 1 ends at the first waypoint after the IDF. The standard ATT/XTT for the RNP-1 DP is based on table 1-2-1. Section 2 begins at the end of Section 1 and continues until the departure procedure is terminated or transition segments begin.

(1) An ATT/XTT of 0.30 may be used for Section 1 on an RNP-1 DP provided the following conditions are met:

- (a) The procedure is issued as a special,
- (b) Section 1 has a maximum length of 10 NM, and
- (c) Section 1 ends at a flyby waypoint.

(d) Section 1 OEA outside turn boundary is a 3 x XTT (2 x XTT primary) arc around the turn point to the point of tangency parallel to the outbound course then splays at 15 degrees until reaching full width for the subsequent leg, the OEA inside turn boundary is constructed consistent with paragraph 5-3-1.a(1)(c) to the point of tangency parallel to the outbound course then splays at 15 degrees until reaching full width for the subsequent leg.

d. The departure procedure must join the en route structure at an altitude that permits en route flight, to include airspace and obstacle clearance. If the departure procedure does not terminate at a fix/NAVAID on an ATS route, the termination altitude must allow for continued level flight in all directions. If unable, raise the termination altitude or place restrictions on the procedure.

5-7-2. Procedure Design.

a. Standards. Utilize the following standards for procedure design.

(1) Optimum and minimum leg length is 3 NM. Minimum leg length for turns equal to or greater than 70 degrees is 3.5 NM. Maximum leg length is 10 NM and maximum turn is 90 degrees.

(2) Utilize standard climb airspeed of 80 KIAS and bank angle of 13 degrees until reaching the desired target altitude. After reaching the desired target altitude, evaluate at an airspeed of 140 KIAS and bank angle of 15 degrees.

(3) Apply table 1-2-1 for all segments except the IDF flat surface area to achieve desired ATT/XTT. For standard departure criteria apply 1 NM ATT/XTT, for RNP 0.3 departures or for RNP 1 departures with the first segment of RNP 0.3 [per paragraph 5-7-1.c(1)] apply 0.30 ATT/XTT in the IDF flat surface area.

(4) Standard ROC of 250 plus adjustments (altimeter, precipitous terrain) is applied in level surface areas.

(5) Altimeter setting adjustment is applied for altimeter sources more than 5 NM from the IDF to the altimeter source in accordance with Order 8260.3, chapter 3.

(6) A 20:1 slope for the primary area surface and a 6:1 slope for secondary area surface are applied after the flat surface area.

(7) The first departure turn-at-a-fix may be a fly-by or fly-over fix. Use fly-by unless a fly-over is required for obstacle avoidance or where mandated by specific operational requirements.

(8) For a Proceed Visually departure, the maximum turn angle at the IDF is 30 degrees.

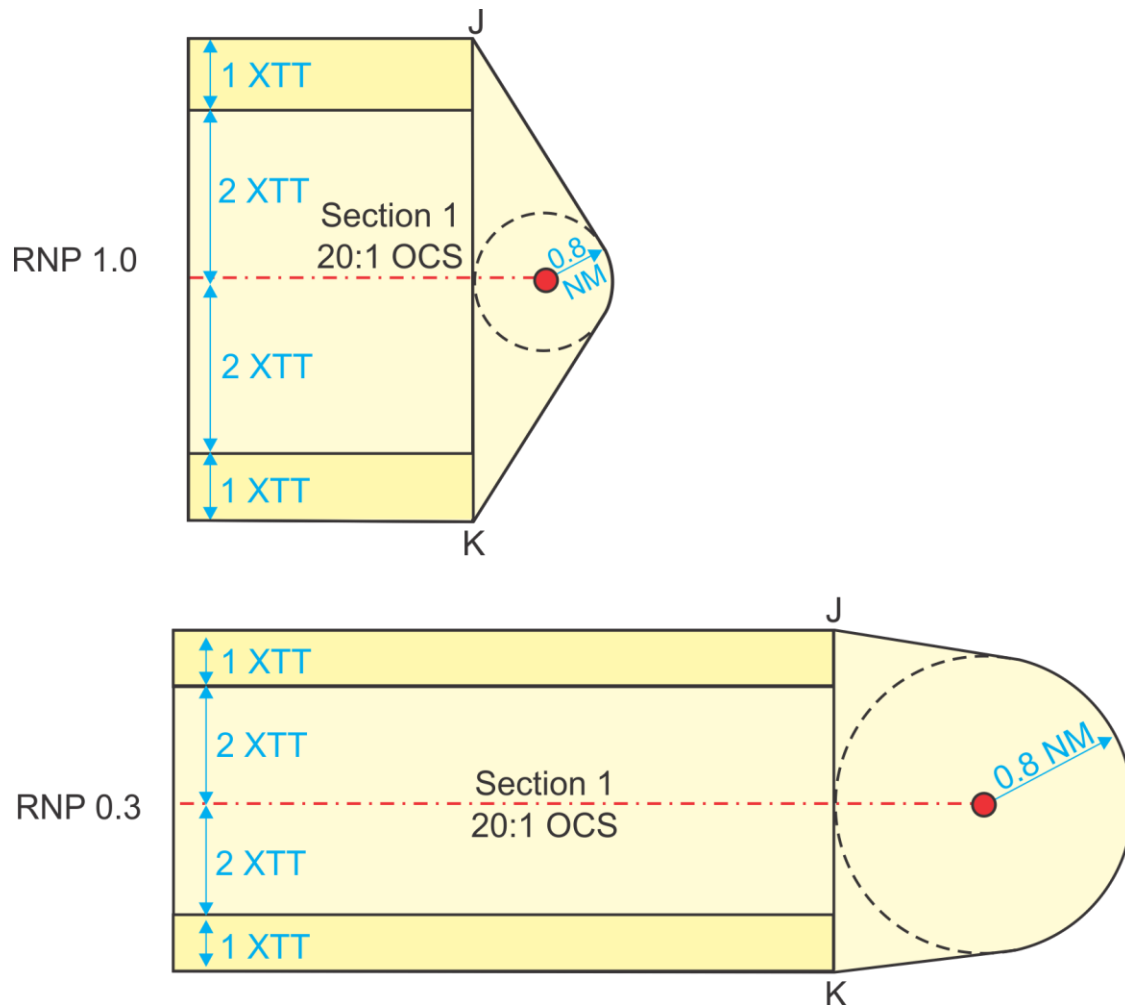
b. Obstacle Evaluation Area (OEA).

(1) The Proceed VFR OEA consists of a flat surface area around the IDF followed by a 20:1 sloping surface from departure side of the flat surface area to the departure procedure termination point. Obstacle evaluation is not conducted from the helipad to the IDF.

(2) The Proceed Visually OEA consists of a sloping surface from the helipad to the IDF point of earliest reception in the flat surface area (see paragraph 5-7-4), the flat surface area around the IDF, and a 20:1 sloping surface from the flat surface area to the departure procedure termination point.

5-7-3. Flat Surface Area. A level surface is evaluated using a 0.8 NM radius around the IDF with expansion lines tangent to the radius extending to the secondary boundary points at 3 RNP on the J-K line either side of the departure course (see figure 5-7-1). The minimum IDF crossing altitude is determined by the elevation of the highest obstacle within the IDF flat surface area plus 250 ROC and adjustments (including precipitous terrain within the IDF flat surface area), rounded up to the next 20-foot increment.

Figure 5-7-1. IDF Surface Area



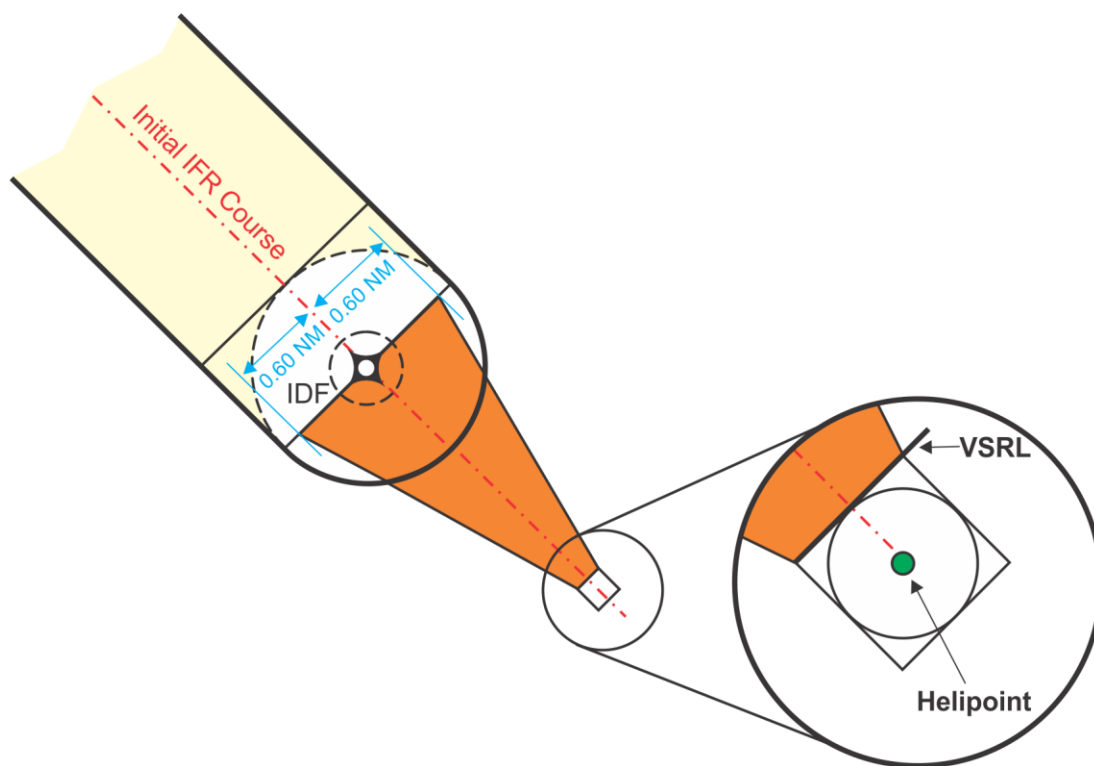
5-7-4. Visual Segment for Proceed Visually Departures. The visual segment begins at the helipad and continues to the IDF point of earliest reception (see figure 5-7-2).

a. Area.

(1) Length. The visual segment OEA begins at the Visual Surface Reference Line (VSRL) and ends at the IDF point of earliest reception. The VSRL is located at the edge of the departure side of the FATO.

(2) Width. The visual segment splay begins at the VSRL and splays from the VSRL endpoints to 0.6 NM either side of the IDF, perpendicular to the visual segment course.

Figure 5-7-2. Proceed Visually Departure Visual Surface Area

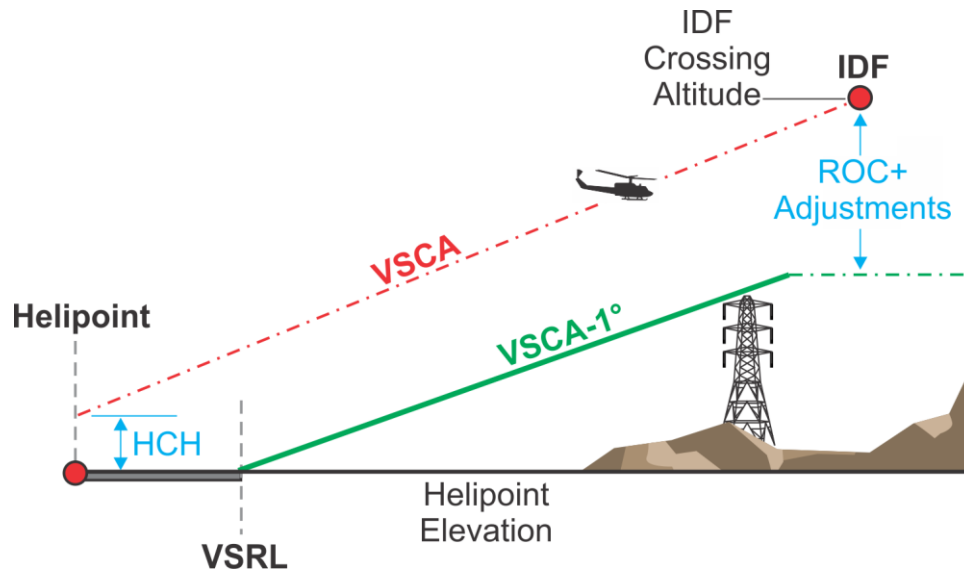


b. Visual Segment Climb Angle (VSCA). The VSCA begins at the HCH above the helipoint, and ends at the IDF at the IDF crossing altitude (see figure 5-7-3). The optimum angle will allow an aircraft to climb at 400 ft/NM.

c. Visual Segment OIS. The OIS begins at the VSRL at the helipoint elevation, and extends upward toward the IDF at an angle one degree less than the VSCA. The OIS rises to the point it reaches an altitude equal to the IDF crossing altitude minus ROC and adjustments (including precipitous terrain within the IDF area) after which it becomes a level surface to the end of the IDF area (see figure 5-7-3). Measure obstacles using the shortest distance to the VSRL. Obstacles should not penetrate the OIS, if any obstacles penetrate the OIS on initial evaluation, take one of the following actions, listed in preferential order:

- (1) Remove or adjust obstacle location and/or height to eliminate the penetration, or
- (2) Raise the VSCA (Maximum 8.13 degrees) to achieve an OIS angle that clears the obstacle (verify that the helicopter meets the new climb performance), or
- (3) Raise the HCH to ≤ 20 feet. Consult with the operator to determine ability of the helicopter fleet to hover at the adjusted HCH. When this procedure is applied, raise the OIS origin above the heliport elevation by the amount that the HCH is increased.

Figure 5-7-3. Proceed Visually Surface Climb Area Profile

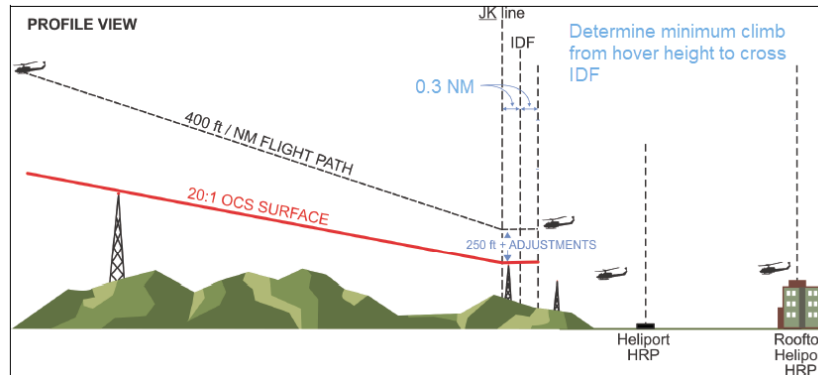


5-7-5. Procedure Climb Area. The procedure climb area begins at the IDF flat surface J-K line and continues to a point 0.3 NM beyond the departure procedure termination fix.

a. Sloping OCS. Section 1 begins at the IDF flat surface area J-K line (see figure 5-7-1) and continues to the termination fix (for a straight departure) or until the first turn (see figure 5-7-4). Section 2 begins after the first turn and continues until the end of the departure procedure. Apply an OCS slope of 20:1 for the primary area, with a secondary area OCS slope of 6:1 measured along the shortest secondary distance to the point of interest. If the OCS is penetrated, raise the IDF crossing altitude, utilize a higher-than-standard climb gradient, or redesign the departure procedure to clear the obstacle. Raising the IDF crossing altitude may cause an increased ceiling and visual requirement for Proceed Visually procedures.

b. Level surface. The sloping OCS is applied during the climb until en route ROC is attained. En route flight may commence once en route ROC is attained.

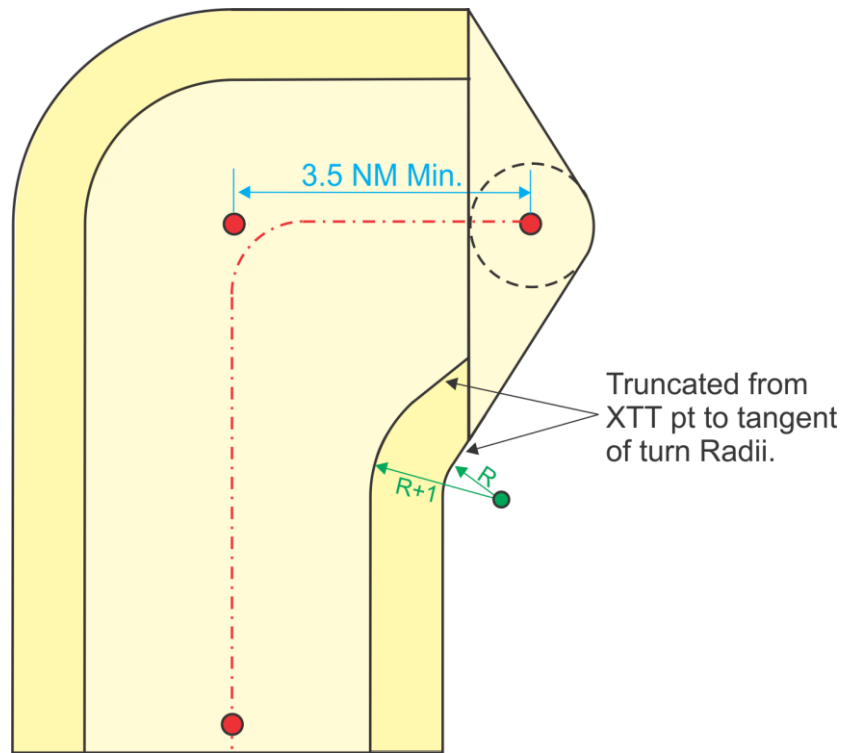
Figure 5-7-4. IDF Flat Surface Area



c. **OEA Configuration.** The OEA is constructed based on the RNP requirement for the procedure or procedure segment. It is designed utilizing procedure design standards listed in paragraph 5-7-2.a. Turns are dependent on the course change, airspeed, bank angle, turn altitude, and heliport elevation.

(1) *Step 1.* From the J-K line, construct Section 1 with the appropriate primary and secondary areas in the direction of flight. Primary area width will be $2 \times \text{XTT}$ either side of course, and secondary area width will be $1 \times \text{XTT}$ either side of the primary area. Continue Section 1 to the next waypoint plus ATT. If the desired departure procedure altitude is achieved, terminate the segment 0.3 NM beyond the WP.

(2) *Step 2.* If a turn is desired, construct Section 2 continuing the primary and secondary widths from Step 1 (see figure 5-7-5). The outside turn will maintain the radius connecting Section 1 and Section 2. Apply paragraph 1-2-5.d(1) to determine the inside radius, substituting heliport elevation for airport elevation in formula 1-2-8. If the inside turn connection begins prior to the J-K line, truncate from the J-K line to the tangent of the turn radius. Once the desired departure procedure altitude is reached, terminate the segment using ATT of 0.30 after the WP. If another turn is necessary, repeat *Step 2*.

Figure 5-7-5. Turning Obstacle Evaluation Area

- d. For obstacle evaluation measurements, apply section 5-4 (Obstacle Evaluation).

Appendix A. Administrative Information

- 1. Distribution.** This order is distributed electronically only.
- 2. Terms.** See Order 8260.3 for definitions. The following words have the meanings shown:
 - a. May. Action is permissible.
 - b. Must. Action is mandatory.
 - c. Should. Action is desirable.
 - d. Will. Indicates a presumption that action is to be taken.
- 3. Acronyms.** Users of this order can refer to table A-1 for an alphabetical listing of frequently used acronyms and abbreviations.

Table A-1. Acronyms and Abbreviations

ACT	average cold temperature
ANP	actual navigation performance
ANPE	actual navigation performance error
AP	autopilot
ARP	airport reference point
ASE	altimetry system error
ATD	along-track distance
ATIS	automated terminal information system
ATS	air traffic service
ATT	along track tolerance
CA	course to altitude
CAT	category
CFR	Code of Federal Regulations
CY	calendar year
DA	decision altitude
DER	departure end of runway
DF	direct-to-fix
DG	descent gradient
DME	distance measuring equipment
DP	departure procedure
DRL	departure reference line
DRP	departure reference point
DTA	distance of turn anticipation
DVA	diverse vector area
FAC	final approach course
FAF	final approach fix
FAS	final approach segment
FATO	final approach and takeoff area
FB	fly-by
FD	flight director
FEP	final end point

FHP	fictitious heliport
FL	flight level
FM	fix to a manual termination
FO	fly-over
FPAF	flight path alignment point
FROP	final rollout point
FSL	flat surface length
FTE	flight technical error
FTP	fictitious threshold point
GARP	global navigation satellite system azimuth reference point
GBAS	ground based augmentation
GLS	ground based augmentation landing system
GNSS	global navigation satellite system
GPA	glidepath angle
HAL	height above landing
HAS	height above surface
HAT	height above touchdown
HCH	heliport crossing height
HDRP	heliport departure reference point
HIL	hold-in-lieu-of-procedure turn
HMAS	height of missed approach surface
IAF	initial approach fix
IAP	instrument approach procedure
ICA	initial climb area
ICAB	initial climb area baseline
IDF	initial departure fix
IF	intermediate fix

IFP	instrument flight procedure
IFR	instrument flight rules
ILS	instrument landing system
ISA	International Standard Atmosphere
KIAS	knots indicated airspeed
KTAS	knots true airspeed
LGSA	lateral guidance sector angle
LNAV	lateral navigation
LNAV/VNAV	lateral navigation with vertical guidance
LP	localizer performance
LPV	localizer performance with vertical guidance
LTP	landing threshold point
MA	missed approach
MAHWP	missed approach holding waypoint
MAP	missed approach point
MAS	missed approach segment
MCA	minimum crossing altitude
MDA	minimum descent altitude
MEA	minimum en route IFR altitude
MIA	minimum instrument altitude
MSL	mean sea level
MVA	minimum vectoring altitude
NA	not authorized
NAVAID	navigational aid
NavSpec	navigation specification
NCDC	National Climatic Data Center
NM	nautical mile
NOAA	National Oceanic and Atmospheric Administration
OCS	obstacle clearance surface
OEA	obstacle evaluation area
OIS	obstacle identification surface
PBN	performance based navigation
PCG	positive course guidance
PFAF	precise final approach fix
PI	primary intersection
PinS	point-in-space
RASS	remote altimeter setting source
RCL	runway centerline
RF	radius-to-fix
RNAV	area navigation

RNP	required navigation performance
RNP AR	required navigation performance with authorization required
ROC	required obstacle clearance
SBAS	satellite based augmentation services
SER	start end of runway
SI	secondary intersection
SIAP	standard instrument approach procedure
SID	standard instrument departure
SOC	start-of-climb
SOIA	simultaneous offset instrument approach
STAR	standard terminal arrival route
TAA	terminal arrival area
TCH	threshold crossing height
TDZE	touchdown zone elevation
TF	track-to-fix
TIA	turn initiation area
TR	turn rate
VAE	vertical angle error
VDA	vertical descent angle
VDP	visual descent point
VEB	vertical error budget
VFR	visual flight rules
VM	manual termination
VNAV	vertical navigation
VOR	very high frequency omnidirectional radio
VORTAC	very high frequency omnidirectional radio collocated with tactical air navigational aid
VSCA	visual segment climb angle
VSRL	visual surface reference line
WAAS	Wide Area Augmentation System
WCH	wheel crossing height
WP	waypoint
WPR	waypoint precision error
WS	wind spiral
XTT	cross track tolerance

4. Suggestions for Improvements. Please forward all comments on deficiencies, clarifications, or improvements regarding the contents of this order to:

- a. The Directives Management Officer at 9-AWA-AFB-120-Directives@faa.gov or
- b. Flight Technologies and Procedures Division at 9-AWA-AFS400-COORD@faa.gov.

Your suggestions are welcome. [FAA Form 1320-19](#), *Directives Feedback Information*, is available at the link provided as well as located at the end of this order for your convenience.

Appendix B. Use of Historical Wind Studies for Determining Tailwind

1. Purpose. This appendix addresses the use of five-year historical wind analysis to determine historical tailwind in lieu of calculated tailwind.

2. Acceptable Source. Historical winds must be obtained or derived from the NOAA/NCEP hourly operational weather prediction system numerical forecast models (i.e., Rapid Refresh (RAP)).

a. Use wind data from the last five-years of “zero-hour” data (best estimate of current conditions taken from aircraft measurements, weather stations, radar winds, and satellite imagery system).

b. Project the wind data on a Lambertian Conformal Hybrid-Isentropic grid at a resolution of not more than 20 km.

c. Derive the 99th percentile wind speeds by altitude using linear interpolation from the nearest grid point(s) to the applicable turn waypoint. For FB or FO turns, use the fix location for wind. For RF use the turn center location for wind.

d. 99th percentile winds speeds may be further sub-divided by direction.

3. Use of Directional Wind Speeds. Historical winds sub-divided by direction may be used to support TF/FB, TF/FO, and RF turn construction only. Use the greatest 99th percentile wind speed from any direction that results in a tailwind during the turn (i.e., the “tailwind direction sector”).

a. For TF legs, the turn start and end point is the FB or FO waypoint. For RF legs, the turn start point is the RF leg start waypoint and turn end point is the RF leg end waypoint.

b. For TF/FB, TF/FO, and RF legs, the start bearing is the reciprocal of inbound leg at the turn start point.

c. For TF/FB and RF turns, the end bearing is the reciprocal of the outbound leg at turn end point.

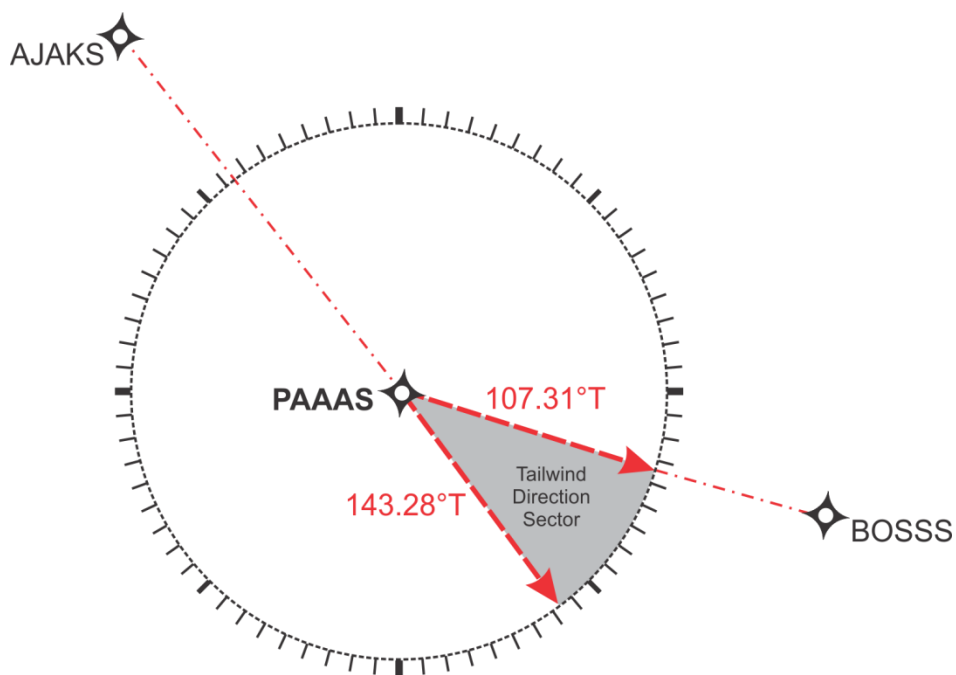
d. For TF/FO turns, the end bearing 30 degrees from the reciprocal of the outbound leg at turn end point.

e. Examples:

(1) TF/FB turn: KDEN ILS or LOC Rwy 34L, tailwind direction sector for turn at PAAAS (see figure B-1).

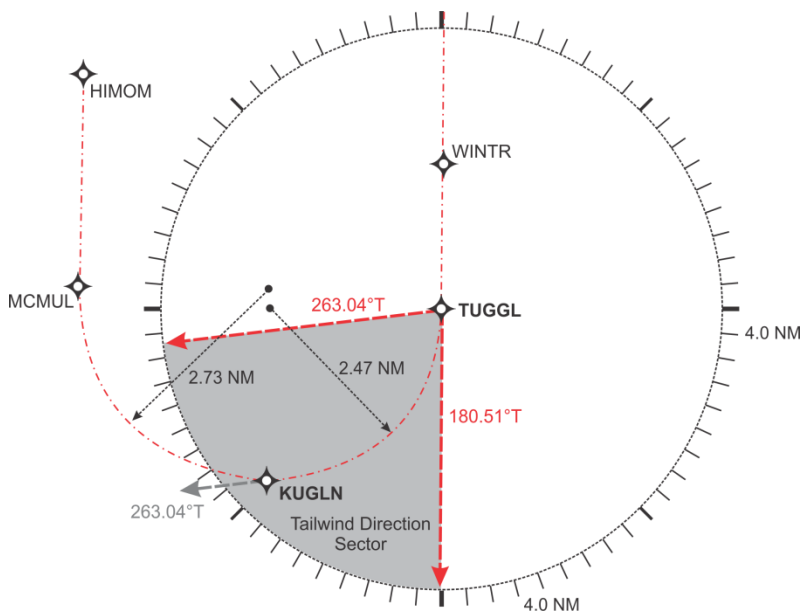
(a) Start bearing; reciprocal of inbound leg at PAAAS = 107.31 (true).

(b) End bearing; reciprocal of outbound leg at PAAAS= 143.28 (true).

Figure B-1. TF/FB Turn Tailwind Direction Sector

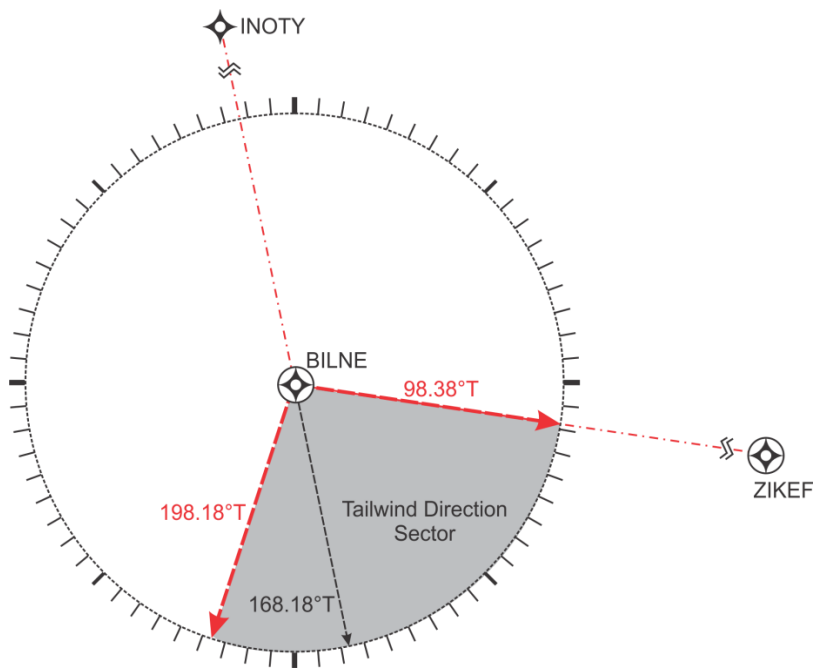
(2) RF leg: KDEN RNAV (RNP) Z RWY 34L, tailwind direction sector for KUGLN - TUGGL (see figure B-2).

- (a) Start bearing; reciprocal of inbound leg at KUGLN = 263.04 (true).
- (b) End bearing; reciprocal of outbound leg at TUGGL = 180.51 (true).

Figure B-2. RF Leg Tailwind Direction Sector

(3) TF/FO turn: PAAK INOTY ONE (RNAV) Departure, tailwind direction sector for turn at BILNE (see figure B-3).

Figure B-3. TF/FO Turn Tailwind Direction Sector



Appendix C. DF Leg Feasibility Analysis

1. Purpose. This analysis checks the DF leg for feasibility and is separate from the OEA evaluation. It is only intended to be a preliminary screening for flyability and does not ensure OEA construction.

2. Applicability. This analysis applies to DF legs used in departures and missed approach.

a. DF following turn-at-altitude (see figure C-1).

(1) Determine the earliest turn point (TP).

(a) Departure. Measure from the DRP at airport elevation with 1100 ft/NM climb gradient until reaching the earliest climb-to altitude or the DER, whichever occurs first. If the climb-to altitude is not reached by the DER, continue the climb determination starting at the DER using a climb gradient at 500 ft/NM until 10000 feet, then 350 ft/NM until 18000 feet, then 200 ft/NM above 18000 feet.

(b) Missed approach. The earliest TP is line C-D.

(2) Determine the latest TP.

(a) Departure. Commencing at the DER at DER elevation, the latest TP is where an aircraft reaches the climb-to altitude at a climb gradient of 200 ft/NM or the minimum climb gradient required for obstacle clearance whichever is higher.

(b) Missed approach. Commencing at the aircraft (see paragraph 3-7-1.b), the latest TP is where an aircraft reaches the climb-to altitude at a climb gradient of 200 ft/NM or the minimum climb gradient required for obstacle clearance whichever is higher.

(3) Determine the turn radius. Use the highest altitude in the turn (see paragraph 1-2-5.c).

b. DF following FO fix [(applicable to departure only, see figure C-2)].

(1) Determine the turn points. The early TP is ATT prior to the FO fix. The late TP is ATT + reaction and roll distance (see formula 1-2-12) after the FO fix.

(2) Determine the turn radius. Use the assumed altitude at the fix (see paragraph 1-2-5.c).

3. Analysis. Given the location of the DF leg termination fix and the outbound track from this fix, verify each of the following. When all three conditions are met, the DF leg is feasible.

a. DF fix location. The fix is on or outside all paths from the earliest TP until the latest TP based on the specified turn radius.

b. Turn following fix. The turn at the fix is 90 degrees or less.

- c. DTA. Where the fix is a FB, the required DTA is available.

Figure C-1. DF Following Turn-at-Altitude

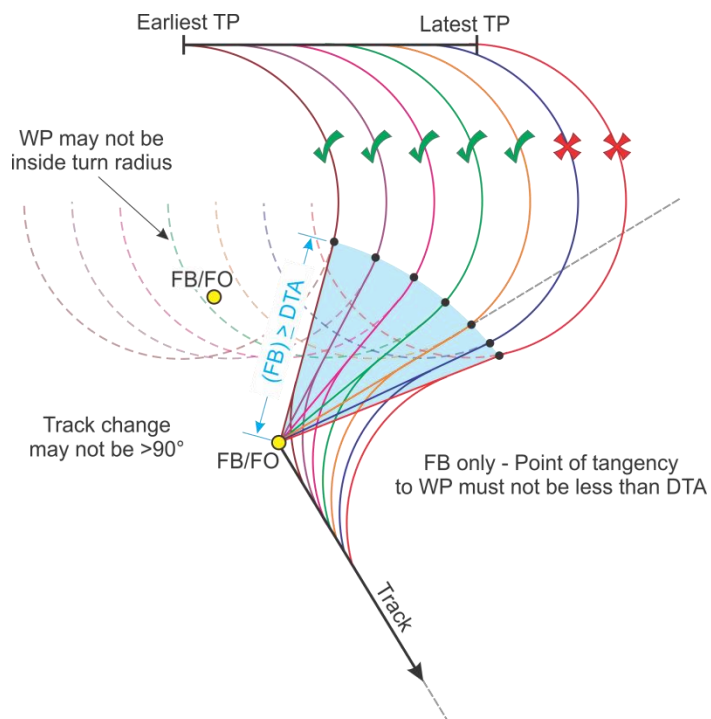
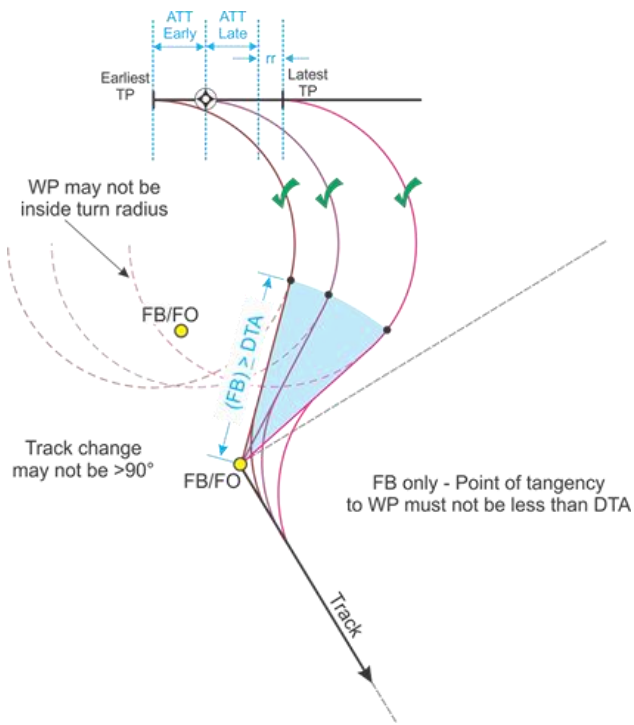


Figure C-2. DF Following FO fix



Appendix D. PBN Transition to ILS/GLS/LPV Final

1. Purpose. When establishing an intermediate segment that transitions to an ILS/GLS/LPV final, the design must account for high temperature conditions that may cause higher than indicated true altitudes during the glide slope capture.

2. Transition to ILS/GLS/LPV Final. Design steps.

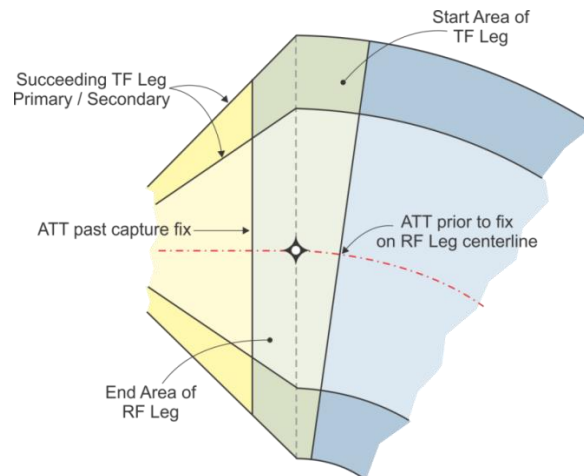
a. Step 1. Evaluate the final, intermediate, and initial segments in accordance with the applicable paragraphs, except:

(1) Establish a capture fix. Construct a TF leg aligned with the FAC that is common to all intermediate segments. The start fix of this leg is designated the capture fix. Alternatively where operationally necessary, the capture fix may be placed on or at the start of an RF leg. The preliminary location of the capture fix may be less than 2 NM but no closer than 1 NM prior to the PFAF.

(2) Intermediate OEA.

(a) RNAV (GPS) construction (see chapter 3). The segment remains at full width until reaching 2 NM prior to the PFAF or the capture fix, whichever is closer to the PFAF. The primary and secondary boundaries abeam this point taper uniformly (relative to path) to the X and Y OCS outer boundaries abeam the PFAF. The preceding leg construction follows the tapering boundaries until reaching the appropriate ATT beyond the fix (see figure D-1 for RF example).

Figure D-1. Intermediate OEA, RNAV Construction



(b) A-RNP (see chapter 3) construction. Use the same RNP value throughout the entire intermediate construction and verify that the final segment supports the selected RNP with formula D-1. Where the selected RNP is less than the minimum, increase the RNP or decrease the length of final. The capture leg ends at the plotted position of the PFAF.

Formula D-1. Minimum RNP

$$\text{Min}_{\text{RNP}} = \frac{0.3048 \times (\text{FAS}_{\text{Length}} \times 0.10752 + 678.496)}{1852 \times 2}$$

Where:

$\text{FAS}_{\text{Length}}$ = length (feet) of the final approach segment

Example:

$$\text{Min}_{\text{RNP}} = \frac{0.3048 \times (50201.07 \times 0.10752 + 678.496)}{1852 \times 2}$$

$$\text{Min}_{\text{RNP}} = 0.50$$

b. Step 2. Specify a mandatory altitude at the capture fix which results in as close to a two-degree descent angle as operationally possible (see formula D-2). Where the descent angle is greater than two degrees, lengthen the segment and/or alter the preceding segment to provide a lower fix altitude.

Formula D-2. Capture Leg Descent Angle

$$\text{Capture}_{\theta} = \text{atan} \left(\frac{\text{Capture}_{\text{alt}} - \text{PFAF}_{\text{alt}}}{\text{Capture}_{\text{Length}}} \right)$$

Where:

$\text{Capture}_{\text{alt}}$ = MSL altitude (feet) of the capture fix

$\text{Capture}_{\text{Length}}$ = distance (feet) between the capture fix and PFAF

Example:

$$\text{Capture}_{\theta} = \text{atan} \left(\frac{2300 - 2000}{8754.28} \right)$$

$$\text{Capture}_{\theta} \approx 1.96^{\circ}$$

c. Step 3. Calculate the actual aircraft altitude at the capture fix using the default 40°C Δ ISA in formula D-3. Alternatively, determine the average hot temperature using a process similar to paragraph 3-3-4.a(1)(a) and calculate its deviation from ISA using formula 3-3-3.

Formula D-3. High Temperature Capture Fix Altitude

$$\text{High}_{\text{alt}} = \frac{(\text{Capture}_{\text{alt}} - \text{LTP}_{\text{elev}}) \times \Delta\text{ISA}_{\text{High}}}{288 + \Delta\text{ISA}_{\text{High}} - 0.5 \times 0.00198 \times \text{Capture}_{\text{alt}}} + \text{Capture}_{\text{alt}}$$

Where:

$\text{Capture}_{\text{alt}}$ = MSL altitude (feet) of the capture fix

$\Delta\text{ISA}_{\text{High}}$ = 40°C or historical high temperature deviation from ISA

Example:

$$\text{High}_{\text{alt}} = \frac{(2300 - 1000) \times 40}{288 + 40 - 0.5 \times 0.00198 \times 2300} + 2300$$

$$\text{High}_{\text{alt}} \approx 2459.64 \text{ feet}$$

d. Step 4. Calculate the minimum capture leg length using formula D-4. If the designed length is greater than or equal to the minimum, the design is satisfactory. Otherwise, lengthen the leg and reevaluate if fix altitudes change.

Formula D-4. Minimum Capture Leg Length

$$\text{Min}_{\text{Length}} = \frac{r \times \pi}{180} \times \left(90 - \theta - \text{asin} \left[\frac{\cos(\theta) \times (r + \text{LTP}_{\text{elev}} + \text{TCH})}{r + \text{High}_{\text{alt}}} \right] \right) - \text{FAS}_{\text{Length}}$$

Where:

High_{alt} = calculated high temperature capture fix altitude

$\text{FAS}_{\text{Length}}$ = length (feet) of the final approach segment

Example:

$$\text{Min}_{\text{Length}} = \frac{r \times \pi}{180} \times \left(90 - 3 - \text{asin} \left[\frac{\cos(3) \times (r + 1000 + 45)}{r + 2459.64} \right] \right) - 18221.16$$

$$\text{Min}_{\text{Length}} \approx 8443.89 \text{ feet}$$

Appendix E. Vector SID Implementation

1. Purpose. When establishing a vector SID certain prerequisites should be considered, the design should account for ATC capability and operational need.

2. Vector SID Authorization. The use a vector SID is not authorized when:

- a. Surveillance is not available.
- b. 4-D trajectory management is required (i.e., future NextGen implementation).

3. Vector SID Application. A vector SID that includes a vector leg between routes (Open SID) should only be used when:

- a. The initial leg(s) off the runway requires a specific ground track,
- b. The intermediate leg(s) is (are) anticipated to require frequent ATC vectoring, and
- c. The final (PBN route) leg(s) require(s) a specific ground track.

Note: See table E-1 for specific requirements to consider an Open SID.

Table E-1. Cases for Application of Open SID

	Initial (climb) leg(s) A	Intermediate (vector) leg B	Final (PBN route) leg(s) C
Case 1	Terrain or obstacles near the initial climb out path	Requires dispersion of ground tracks to distribute noise footprint	Requires a path to avoid terrain or obstacles terrain
Case 2	Noise sensitive areas or congested airspace near the initial climb out path	Frequently requires ATC intervention to separate departing traffic e.g. from hub/satellite airfields	Joins a PBN route (where containment is assumed at the entry point)
Case 3	Special Use Airspace (e.g., Prohibited or restricted area) near the initial climb out path	Requires attaining a minimum altitude that enables a large angle turn to be accomplished (to reduce DP departure track miles)	Requires a path around airspace

Note: If at least one condition from each of the three vertical columns is present, an Open SID may be considered. For example, where 1A, 1B, and 1C are present or if 1A, 2B, and 3C apply, an Open SID is a viable design option.

Directive Feedback Information

Please submit any written comments or recommendations for improving this directive or suggest new items or subjects to be added to it. Also, if you find an error, please tell us about it.

Subject: FAA Order _____

To: Directive Management Officer, _____

(Please mark all appropriate line items)

An error (procedural or typographical) has been noted in paragraph _____ on page _____ .

Recommend paragraph _____ on page _____ be changed as follows:
(attached separate sheet if necessary)

In a future change to this order, please include coverage on the following subject *(briefly describe what you want added)*:

Other comments:

I would like to discuss the above. Please contact me.

Submitted by : _____ Date: _____

Telephone Number: _____ Routing Symbol: _____