Subject: Powder Coating of Aircraft Wing Struts

Purpose: This SAFO provides safety information to the public of potential problems associated with coating or modifying wing struts.

Background: A recent investigation of a repair station’s specifications for the application of powder coating revealed an airworthiness concern with a Piper airplane equipped with sealed wing struts. The airplane was in compliance with Airworthiness Directive (AD) 99-01-05, specifically the installation of sealed wing struts that terminates the 24-month repetitive inspection requirement.

Discussion: The powder coat application process requires heating the wing lift struts to 450 degrees Fahrenheit. At this temperature, a sealed wing lift strut would expand (swell up). To vent the heat and gas buildup of this process, a hole is drilled at the fork socket located at the bottom of the strut. This vent hole in the wing strut ultimately means that the struts are no longer sealed. Although drilling this hole is not specifically prohibited by AD 99-01-05 and installation of the lift strut assembly part number keeps the airplane within AD compliance, the intent of the AD was for the affected airplanes to be either 1) repetitively inspected every 24 months, or 2) equipped with sealed strut assemblies and those assemblies to remain sealed.

AD 99-01-05 requires these strut assemblies to be repetitively inspected if the seal is broken in any manner to include the drilling of a vent hole. The Federal Aviation Administration (FAA) is considering a revision to the AD to clarify this requirement and possibly add a requirement to determine if the struts are still sealed with appropriate follow-on action.

To further complicate the issue, the location of the vent hole in the socket of the fork boss does not make it apparent that the lift strut has been opened or modified. Thus, the part number of the sealed wing strut assembly shows compliance with the terminating action of AD 99-01-05. The drilling of a vent hole for the powder coating process brings into question the need for repetitive inspections. In addition, the vent hole creates a high stress concentration area, which affects the fatigue characteristics of the lift struts and degrades the structural performance of the strut. The high heat introduced by the powder coat process also causes the corrosion inhibitor inside the strut to boil, vaporize, and/or alter its protective properties. If unchecked and undetected, these conditions could lead to corrosion of the lift struts, which could result in a safety of flight issue such as wing separation.

AD 99-01-05 R1 which applied to certain aircraft equipped with wing lift struts was superseded by AD 2015-08-04 on June 3, 2015. AD 2015-08-04 retains all requirements of AD 99-01-05R1 and adds...
additional airplane models to the applicability section. The new AD applies to various aircraft and was
issued to detect and correct corrosion and cracking on the front and rear wing lift struts and forks, which
could cause the wing lift strut to fail. This failure could result in the wing separating from the airplane.

**Recommended Action:** Aircraft owners, operators, and maintenance personnel should:

- Initially inspect airplanes with sealed lift struts at the next annual inspection to determine if the
  struts have drill holes or other modifications that may have broken the seal.

- Annotate in the aircraft record if wing lift struts are no longer sealed.

- Identify externally if wing lift struts are no longer sealed so that maintenance personnel can
  readily ascertain this.

- Incorporate an FAA-approved procedure for reapplying the corrosion inhibitor and resealing the
  lift struts.

**Contact:** Questions or comments regarding this SAFO should be directed to the following:

- Piper Aircraft (All Models), Atlanta ACO 404-474-5500.

- Maule Aircraft (All Models), Atlanta ACO, 404-474-5500.

- Taylorcraft Aircraft (All Models) and Aircraft Parts and Development Corp (Models A-9, A-9A,
  and A-9B), Fort Worth ACO, ASW-150, 817-222-5100.

- Kansas City Aircraft Evaluation Group (AEG), 816-329-3233.