DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

Aviation Rulemaking Advisory Committee; Transport Airplane and Engine Issue Area—Phase 2 of Low Speed Alerting Task

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of new task assignment for the Aviation Rulemaking Advisory Committee (ARAC).

SUMMARY: The FAA assigned the Aviation Rulemaking Advisory Committee (ARAC) a new task to identify and develop recommendations on additional requirements for low speed alerting. Phase 1 of the task addresses new standards for transport category airplanes. Phase 2 of the task addresses possible retrofit standards for existing transport category airplanes. This notice is to inform the public that the ARAC working group has completed activity for Phase 1 of the task and will begin activity for Phase 2.

FOR FURTHER INFORMATION CONTACT: Joe Jacobsen, Airplane & Flight Crew Interface Branch, ANM–111, Transport Airplane Directorate, Federal Aviation Administration, 1601 Lind Ave, SW., Renton, Washington 98057; telephone (425) 227–2011, facsimile (425) 227– 1149; e-mail *joe.jacobsen@faa.gov.*

SUPPLEMENTARY INFORMATION:

Background

The FAA established ARAC to provide advice and recommendations to the FAA Administrator on the FAA's rulemaking activities with respect to aviation-related issues. With respect to low speed alerting, the FAA previously revised regulations in the area of flight guidance (autopilot) and performance and handling qualities in icing conditions to improve transport airplane standards for low speed protection (in the case of icing, stall warning standards were enhanced). However, as a result of several recent loss-of-control accidents and incidents, the FAA has identified a need for additional low speed safeguards, in addition to the regulatory actions that have already been taken. The committee addressed the Phase 1 task—new part 25 standards under the existing Avionics System Harmonization Working Group within the Transport Airplane and Engine Issues Group. (The FAA published a notice of Phase 1 task assignment in the Federal Register (75 FR 16902) on April 2, 2010.) The committee will also address the Phase 2 task-parts 25/121/ 129 retrofit standards under the existing

Avionics Systems Harmonization Working Group within the Transport Airplane and Engine Issues Group.

The Task

ARAC was initially tasked with providing information that will be used to develop standards and guidance material for low speed alerting systems. This information may result in standards that complement existing stall warning requirements. The working group provided a report that addressed several low speed alerting technical questions, relative to new aircraft designs (Phase 1 task—new part 25 standards), and provided the rationale for their responses.

Since the Phase 1 task is complete, ARAC is now tasked with providing information that will be used to develop possible retrofit standards and guidance material for low speed alerting systems. This information may result in standards that complement existing stall warning requirements. The working group will also be expected to provide a report that addresses the following low speed alerting technical questions, relative to existing aircraft designs (Phase 2 task-part 25/121/129 retrofit standards), and provide the rationale for their responses. If the recommendation for retrofit is the same as for new designs, the working group should state the rationale and not repeat the information previously reported. If there is disagreement within the working group, those items should be documented, including the rationale from each party and the reasons for the disagreement.

• How timely is the airplane in alerting the crew of flight below the intended operating speed?

• How timely relative to stall warning?

• Is alerting instantly recognizable, clear, and unambiguous to the flightcrew?

• How are nuisance alerts minimized?

• Does the alerting operate under all operating conditions, configurations, and phases of flight, including icing conditions?

• Does the alerting operate during manual and autoflight?

• After reviewing airworthiness, safety, cost, benefit, and other relevant factors, including recent certification and fleet experience, are there any additional considerations that should be taken into account?

• Is coordination necessary with other harmonization working groups (e.g., Human Factors, Flight Test)? (If yes, coordinate and report on that coordination.)

 If improvements are needed for low speed alerting in the existing fleet, should the FAA adopt a design approval holder (part 26) requirement to mandate development of design changes, or would an operational rule be sufficient? In responding, the working group should address the factors set forth in "FAA Policy Statement: Safety—A Shared Responsibility—New Direction for Addressing Airworthiness Issues for Transport Airplanes" (70 FR 40166, July 12, 2005). The ARAC working group should provide information that could lead to standards for low speed alerting that can be satisfied with practical design approaches.

Schedule

The required completion date for Phase 2 of the task is 15 months after the FAA publishes this notice in the **Federal Register**.

ARAC Acceptance of Task

ARAC accepted the task and assigned it to the existing Avionics Systems Harmonization Working Group in the Transport Airplane and Engine Issue Area. The working group serves as support to ARAC and assists in the analysis of assigned tasks. ARAC must review and approve the working group's recommendations. If ARAC accepts the working group's recommendations, it will forward them to the FAA.

Working Group Activity

The Avionics Systems Harmonization Working Group must comply with the procedures adopted by ARAC. As part of the procedures, the working group must:

1. Prepare a work plan on how to complete the task, including the rationale for this plan. Present the plan for consideration to the Transport Airplane and Engine Issues Group following publication of this notice.

2. Give a detailed conceptual presentation of the proposed recommendations prior to proceeding with the work stated in item 3 below.

3. Draft the appropriate documents and required analyses and/or any other related materials or documents.

4. Provide a status report at each meeting of the ARAC held to consider Transport Airplane and Engine Issues.

Participation in the Working Group

The Avionics Systems Harmonization Working Group is composed of technical experts having an interest in the assigned task. We recommend the existing working group be expanded to include individuals involved in current fleet operations so there is appropriate representation for the Phase 2 task. A working group member need not be a representative or a member of the full committee.

If you have expertise in the subject matter and wish to become a member of the working group, write to the person listed under the caption FOR FURTHER **INFORMATION CONTACT** expressing that desire. Describe your interest in the task and state the expertise you would bring to the working group. We must receive all requests by March 17, 2011 for the meeting scheduled to start from March 15 to 17, 2011, located at the Cessna Conference Center, 6711 West 31st Street South, Wichita, Kansas 67215. The assistant chair, the assistant executive director, and the working group co-chairs will review the requests and advise you whether or not your request is approved.

If you are chosen for membership on the working group, you must represent your aviation community segment and actively participate in the working group by attending all meetings and providing written comments when requested to do so. You must devote the resources necessary to support the working group in meeting any assigned deadlines. You must keep your management chain and those you may represent advised of working group activities and decisions to ensure that the proposed technical solutions do not conflict with your sponsoring organization's position when the subject being negotiated is presented to ARAC for approval. Once the working group has begun deliberations, members will not be added or substituted without the approval of the assistant chair, the assistant executive director, and the working group co-chairs.

The Secretary of Transportation determined that the formation and use of the ARAC is necessary and in the public interest in connection with the performance of duties imposed on the FAA by law. Meetings of the ARAC are open to the public. Meetings of the Avionics Systems Harmonization Working Group will not be open to the public, except to the extent individuals with an interest and expertise are selected to participate. The FAA will make no public announcement of working group meetings.

Issued in Washington, DC, on February 28, 2011.

Pamela Hamilton-Powell,

Executive Director, Aviation Rulemaking Advisory Committee.

[FR Doc. 2011–4761 Filed 3–2–11; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

[Docket No. FAA-2011-0146]

Notice of Intent To Review Structure of the Aviation Rulemaking Advisory Committee

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice and request for public comment.

SUMMARY: The FAA is considering restructuring the Aviation Rulemaking Advisory Committee (ARAC). This notice is to inform the public of FAA's intent and invites the public to provide any ideas or thoughts it may have on this matter.

DATES: Send your comments on or before April 4, 2011.

ADDRESSES: You may send comments identified by Docket Number FAA–2011–0146 using any of the following methods:

• *Government-wide rulemaking Web site:* Go to *http://www.regulations.gov* and follow the instructions for sending your comments electronically.

• *Mail:* Send comments to the Docket Management Facility; U.S. Department of Transportation, 1200 New Jersey Avenue, SE., West Building Ground Floor, Room W12–140, Washington, DC 20590.

• *Fax:* Fax comments to the Docket Management Facility at 202–493–2251.

• *Hand Delivery:* Bring comments to the Docket Management Facility in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Privacy: We will post all comments we receive, without change, to *http:// www.regulations.gov*, including any personal information you provide. Using the search function of our docket Web site, anyone can find and read the comments received into any of our dockets, including the name of the individual sending the comment (or signing the comment for an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477–78).

Docket: To read background documents or comments received, go to http://www.regulations.gov at any time or to the Docket Management Facility in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. SUPPLEMENTARY INFORMATION:

Background

The Aviation Rulemaking Advisory Committee (ARAC) was established in February 1991 to provide FAA's Administrator with industry and public input in the form of information, advice, and recommendations to be considered in the full range of FAA rulemaking activities. These factors are consistent with the dictates of the Administrative Procedures Act (APA). The exchange of ideas that occurs through the ARAC process affords the FAA additional opportunities to obtain firsthand information and insight from those parties who are most affected by existing and proposed regulations.

ARAC consists of approximately 55 member organizations selected by the FAA as most representative of the various viewpoints of those impacted by FAA regulations. The organizations provide a membership fairly balanced in terms of points of view of those represented and the functions to be performed by the committee. The committee is composed of organizations representing air carriers, airports, flight attendants, manufacturers, pilots, public interest and advocacy groups, repair stations, and consumer groups. Members serve in a representative capacity. In addition, an Executive Committee (ExCom) was formed to provide overall administrative oversight for committee activities. The ExCom consists of the ARAC Chair and Vice Chair, who serve as chairperson and vice chairperson, respectively for ExCom; assistant chairpersons representing aeronautical technical subject areas (presently, air carrier operations, maintenance, occupant safety, general aviation certification and operations, noise, aircraft certification, airport certification, transport airplane and engine, rotorcraft, and training and qualifications) with active projects only in transport airplane and engine, and air carrier operations.

The goal of ARAC is to assemble the strongest expertise possible to address particular issues facing the aviation industry and traveling public. The committee conducts its business in open deliberations in the form of public meetings (working groups are exempted). As an advisory body, ARAC has consistently exercised its independence and freedom to provide the FAA recommendations that are not influenced or predetermined by the government. Since 1998, ARAC has submitted more than 110 documented recommendations or products to the March 11, 2013

Federal Aviation Administration 800 Independence Avenue, SW Washington, D.C. 20591

Attention:	Lirio Liu, Director, Office of Rulemaking
Subject:	ARAC Recommendation, Avionics System Harmonization Working Group (ASHWG)
Reference:	Tasking Notice 77 FR 11844 (March 3, 2011)

Dear Lirio,

On behalf of the Aviation Rulemaking Advisory Committee, I am pleased to submit the attached report and presentations as an ARAC recommendation. This report addresses the Phase 2 – Low Airspeed Alerting (Retrofit Applications) and presents the following key findings and recommendations:

- (1) A detailed examination of low airspeed/low energy events (accidents and incidents over a 20 year period), including all contributing factors, not just the lack of a low airspeed alert.
- (2) Examination of operational and safety data to help determine whether a low airspeed alert would have operational benefit.
- (3) Sufficient quantification of the effectiveness of a low airspeed alert (future JSIT report pending).
- (4) A cost-benefit analysis to determine how the system can be "practically" implemented for existing aircraft.

The ASHWG formally requests that when this data is gathered, the FAA task the ARAC to reconvene and review that data for further recommendations.

If a low airspeed alert is required in the future, the ASHWG recommends that the information from this report be used to develop the rules and associated guidance. There must be an integrated approach that incorporates design changes with flight crew procedures and pilot training. There may be multiple mitigations to improve low airspeed awareness.

To facilitate FAA/EASA harmonization and implementation, any associated rulemaking should be reviewed by the ASHWG to ensure that it is aligned with the findings in this report.

The ARAC approved the report for transmittal to the FAA during its March 5th, 2013 meeting. I want to thank all the members of the RPWG for their hard work on both phases of this report.

Sincerely,

Dan Elwell ARAC Chairman

Copy: Renee Butner – FAA Office of Rulemaking ARAC members Loran Haworth – FAA Representative Joe Jacobsen – FAA, Transport Airplane Directorate

Contents

1	Ва	ckground2
	1.1	The First Task2
	1.2	The Second Task3
	1.3	Definitions, Acronyms, and Abbreviations3
2	Pr	ocess Followed5
3	Sc	ope6
4	Ke	y Findings and Recommendations7
5	Te	chnical Questions
	5.1	How timely is the airplane in alerting the crew of flight below the intended operating speed?.8
	5.2	How timely relative to stall warning (alphafloor)?8
	5.3	Is alerting instantly recognizable, clear, and unambiguous to the flightcrew?9
	5.4	How are nuisance alerts minimized?9
	5.5 inclu	Does the alerting operate under all operating conditions, configurations, and phases of flight, ding icing conditions?9
	5.6	Does the alerting operate during manual and autoflight?10
		After reviewing airworthiness, safety, cost, benefit, and other relevant factors, including nt certification and fleet experience, are there any additional considerations that should be taken account?
	5.8 Fligh	Is coordination necessary with other harmonization working groups (e.g. Human Factors, t Test)? (If yes, coordinate and report on that coordination)
	-	If improvements are needed for low speed alerting in the existing fleet, should the FAA/EASA at a design approval holder (part 26) requirement to mandate development of design changes, or d an operational rule be sufficient?11
	State Tran infor	In responding, the working group should address the factors set forth in "FAA Policy ement: Safety—A Shared Responsibility—New Direction for Addressing Airworthiness Issues for sport Airplanes" (70 FR 40166, July 12, 2005). The ARAC working grup should provide mation that could lead to standards for low speed alerting that can be satisfied with practical gn approaches
A	ppend	lix A – Aircraft Survey
A	ppend	dix B – Survey Results
A	ppend	dix C– Example Implementations

1 Background

At the agency's request, the Aviation Rulemaking and Advisory Committee (ARAC) provide advice and recommendations to the Federal Aviation Administration (FAA) on aviation-related rulemaking activities.

As a result of several loss-of-control accidents and incidents, the FAA and EASA identified a need for additional low airspeed safeguards, therefore, the agencies issued Amendment 14 CFR Part 25-121, issued Oct 9, 2007 and CS 25 Amendment 3, effective Sept 19, 2007, *Performance and Handling Qualities in Icing Conditions* to address handling and low speed protection requirements in icing conditions. In addition:

- (1) In June 2007 the FAA revised Advisory Circular AC 25-11A and EASA introduced CS-25 amendment 11, AMC 25-11, which includes guidance for low airspeed *awareness*.
- (2) In November 2010 the FAA revised 14 CFR § 25.1322 and EASA issued CS-25 amendment 11 for flightcrew *alerting*.
- (3) Information from AC 25.1329-1B (and the associated AMC) provides information which may be helpful in determining how to address low airspeed conditions. These were released in 2006.

To augment the regulatory actions taken, the Transport Airplane and Engine Issues Group (TAEIG) assigned the Avionics Systems Harmonization Working Group (ASHWG) to provide information for developing standards and guidance on low airspeed alerting systems (LAS), which could complement existing stall warning requirements. The ASHWG activity was broken into two tasks.

1.1 The First Task

The ASHWG addressed the following ten (10) technical questions relative to new aircraft designs in its first report provided to the ARAC and FAA in April 2011:

- (1) How much time is needed to alert the crew in order to avoid stall warning or excessive deviation below the intended operating speed?
- (2) What would make the alerting instantly recognizable, clear, and unambiguous to the flight crew?
- (3) How could nuisance alerts be minimized?
- (4) Could the alerting operate under all operating conditions, configurations, and phases of flight, including icing conditions?
- (5) Could the alerting operate during manual and auto flight?
- (6) Could the system reliability be made consistent with existing regulations and guidance for stall warning systems?
- (7) Are there any regulations or guidance material that might conflict with new standards?
- (8) What recommended guidance material is needed?
- (9) After reviewing airworthiness, safety, cost, benefit, and other relevant factors, including recent certification and fleet experience, are there any additional considerations that should be taken into account?
- (10) Is coordination necessary with other harmonization working groups (e.g., Human Factors, Flight Test)? (If yes, coordinate and report on that coordination.)

1.2 The Second Task

This report provides answers to the following low speed alerting technical questions relative to *existing* aircraft designs), including a recommendation as to whether retrofit requirements should be the same as new designs.

- (1) How timely is the airplane in alerting the crew of flight below the intended operating speed?
- (2) How timely relative to stall warning?
- (3) Is alerting instantly recognizable, clear, and unambiguous to the flightcrew?
- (4) How are nuisance alerts minimized?
- (5) Does the alerting operate under all operating conditions, configurations, and phases of flight, including icing conditions?
- (6) Does the alerting operate during manual and autoflight?
- (7) After reviewing airworthiness, safety, cost, benefit, and other relevant factors, including recent certification and fleet experience, are there any additional considerations that should be taken into account?
- (8) Is coordination necessary with other harmonization working groups (e.g., Human Factors, Flight Test)? (If yes, coordinate and report on that coordination.)
- (9) If improvements are needed for low speed alerting in the existing fleet, should the FAA and EASA adopt a design approval holder (part 26) requirement to mandate development of design changes, or would an operational rule be sufficient?

Note that the terms "low airspeed" and "low energy" are both used in this report. A low airspeed alert is intended to provide awareness to the flight crew that the aircraft's airspeed is reaching a point where the energy level of the aircraft is being compromised.

1.3 Definitions, Acronyms, and Abbreviations

The following are key definitions for this report:

- Alphafloor The point automated low energy protection will engage.
- Low Airspeed Alert Provides visual, aural or tactile awareness that the aircraft's airspeed is reaching a point where the decrease exceeds a pre-determined threshold.
- Low Energy Alert Provides awareness that the Angle of Attack exceeds a pre-determined low energy threshold.

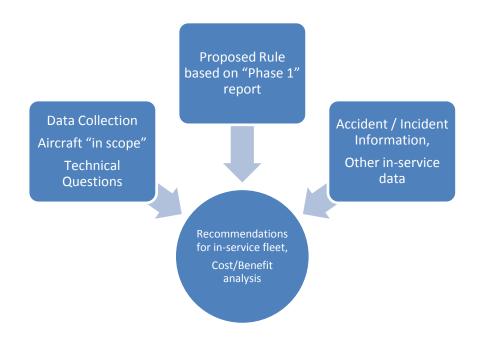
The following are acronyms and abbreviations used in this report:

- AC/AMC Advisory Circular/Acceptable Means of Compliance
- AoA Angle of Attack
- ARAC Aviation Rulemaking Advisory Committee
- ASHWG Avionics Systems Harmonization Working Group
- EASA European Aviation Safety Agency
- EICAS Engine Instrument and Crew Alerting System
- FAA Federal Aviation Administration

- FBW Fly-by-Wire
- JSIT Joint Safety Implementation Team
- LAS Low Airspeed Alerting System
- PFD Primary Flight Display
- TAEIG Transport Airplane and Engine Issues Group
- TAWS Terrain Awareness and Warning System

2 Process Followed

In order to perform this task, the ASHWG prepared a work plan, and presented a summary of that plan to the TAEIG.



Following approval of the work plan, information was collected from aircraft manufacturers to identify existing designs' capability to provide low airspeed awareness and alert functionality.

The ASHWG were provided with a review of relevant accident information to understand whether a low airspeed alerting function could have played a role in reducing loss of control. Specifically, a briefing was provided of a summary of six events occurring from 1999 – 2009, where failure to maintain proper airspeed resulted in a loss of control.

Three key factors were discussed in creating the findings for this report:

- (1) Distractions in the flight deck.
- (2) The effectiveness of the alerting in aircraft; and
- (3) Lack of flight crew system knowledge resulting from current training.

As a result of this data collection and group discussion, the ASHWG were able to generate this report.

3 Scope

Aircraft certificated under 14CFR/CS 25 and operated under 14CFR/CS 121, 129 and 135 (and the international equivalent) manufactured in the Americas and Europe. Representing the global air transport fleet of approximately 24,000; approximately 85% of the total current air transport fleet of 28,000.

All of the aircraft in this analysis were certificated prior the update 14CFR/CS 25.1322, 25.1329, and their associated advisory material. The flight deck capability of the aircraft was categorized into major groups:

- (1) Representing 21% of the fleet studied, aircraft with no low airspeed alert before stick shaker, no PFD (i.e., glass display), and minimal alerting (i.e., no crew alerting "system" or EICAS.
- (2) Aircraft with no low airspeed alert before stick shaker, but with PFD and centralized alert capability (but no low airspeed alert); approximately 15% of aircraft studied.
- (3) Aircraft with PFD and visual low airspeed alert; approximately 10% of the aircraft studied.
- (4) Aircraft with PFD and both visual and aural low airspeed alert. Boeing and Airbus aircraft representing approximately 45% of the fleet.

NOTE: Approximately 8-9 % of the 24,000 aircraft covered by this report did not have manufacturer data available, so are excluded from the analysis.

Appendix A and B provide the survey used to help generate this information and the survey results.

4 Key Findings and Recommendations

Imposing a rule on existing aircraft to incorporate a <u>practical</u> method for implementing low airspeed alerts can only be substantiated by:

- (1) A detailed examination of low airspeed/low energy events (accidents and incidents over a 20 year period), including all contributing factors, not just the lack of a low airspeed alert.
- (2) Examination of operational and safety data to help determine whether a low airspeed alert would have operational benefit.
- (3) Sufficient quantification of the effectiveness of a low airspeed alert (future JSIT report pending).
- (4) A cost-benefit analysis to determine how the system can be "practically" implemented for existing aircraft.

The ASHWG formally requests that when this data is gathered, the FAA task the ARAC to reconvene and review that data for further recommendations.

If a low airspeed alert is required in the future, the ASHWG recommends that the information from this report be used to develop the rules and associated guidance. There must be an integrated approach that incorporates design changes with flight crew procedures and pilot training. There may be multiple mitigations to improve low airspeed awareness.

To facilitate FAA/EASA harmonization and implementation, any associated rulemaking should be reviewed by the ASHWG to ensure that it is aligned with the findings in this report.

5 Technical Questions

The answers to technical questions 5.1 through 5.6 are limited to aircraft designs that incorporate a low airspeed alert system; i.e., approximately 45% of the in-service aircraft.

The terms "low airspeed" and "low energy" are used in this report; both are intended to provide awareness that airspeed is reaching a point where the aircraft may be compromised.

- (1) Example of Low Airspeed Alert: The alert is primarily a function of airspeed, configuration and minimum maneuver speed. It is set when the airspeed decreases 30 percent into the lower amber band. It is reset when airspeed increases above the amber band.
- (2) Example of Low Energy Alert: The alert is a function of configuration, deceleration rate and flight path angle; it is presented when the AoA exceeds an alpha low energy threshold; The AoA corresponds to when it is impossible to recover a long term positive flight path by only increasing lift. The crew's attention is drawn to the speed scale and indicates the need to adjust thrust.

5.1 How timely is the airplane in alerting the crew of flight below the intended operating speed?

The aircraft alerts the crew to airspeed that may result in negative operational situations. Manufacturers provide different protections based upon other alerting capabilities. Both design approval holders that have low airspeed alerting systems complied with the standard established by the authorities.

Existing designs with low airspeed alerting systems are dependent on stall warning information or to other implemented protections.

The designs are not intended to alert for a deviation <u>from</u> the intended operating speed. Rather there are alerts that a stall warning is approaching or of a low energy situation. Both systems are timely in that they indicate a potentially adverse airspeed situation.

5.2 How timely relative to stall warning (alphafloor)?

Design approval holders ensured no spurious activation of the alert occurred over a variety of flight conditions and airplane configurations. Evaluations were also performed to ensure that the alert reset properly. These testing and analyses established compliance with the standards established by the authorities.

The alert was tested in wings level and in turn at different slat/flap configurations, different acceleration rates, and with & without airbrakes. Operational scenarios (for example approach, cruise, and climb) were evaluated to validate minimal nuisance alerts and subjectively validate acceptable alerting prior to stick shaker. One of the worst-case scenarios for evaluating nuisance alerting was during go-around, with one engine inoperative.

The setting of the low energy alert aims at providing enough time to the pilot to manually recover an adequate level of energy through thrust adjustment, before engagement of any protection mechanism if applicable, for low deceleration rates. The approach cases were considered the most significant, so a one second response time was considered to evaluate the effectiveness of a timely thrust increase before stick shaker was activated.

For nominal deceleration rates (1-2 kts per second), the low airspeed alert is intended to provide the pilot sufficient time to increase thrust and minimize the possibility of decelerating to stick shaker activation.

5.3 Is alerting instantly recognizable, clear, and unambiguous to the flightcrew?

The low airspeed alert systems which provide two senses of attention-getting characteristics are considered to be instantly recognizable by the FAA/EASA, however exceptions have been previously approved (for example, an aural "SPEED SPEED SPEED" voice with a visual indication on the display, coincident with an amber/red band directly on the speed tape, with no written messages. In each case the specific content of the alert makes it clear and unambiguous.

Alerts which provide a visual only sense may not be instantly recognizable under all operating conditions; these represent a small subset of the population.

5.4 How are nuisance alerts minimized?

Existing designs input filtering and large margins from normal operating speeds as techniques to minimize nuisance alerts. Some designs filter airspeed inputs while other designs filter Angle of Attack (AoA). Designs also reduce the likelihood where there are large and sudden fluctuations in airspeed or AoA (e.g., in turbulence).

Nuisance alerts have also been minimized by other conditions such as a fixed number excursion below a pre-determined low airspeed value or accounting for failure of a suitable speed protection mechanism (e.g. autopilot/autothrottle).

A reset of the low airspeed alert (from on to off) typically occurs after the aircraft has recovered to a point when the actual airspeed rises by a fixed value (e.g. 5 kts) above the top of the amber low speed band.

5.5 Does the alerting operate under all operating conditions, configurations, and phases of flight, including icing conditions?

Low airspeed alerting operates in most (but not all) operating conditions; there are phases of flight where low airspeed alerting may not be warranted; for example, during take-off, prior to flap retraction, or above 2500 feet. While the alert is helpful in approach and landing conditions when low energy situation is more likely to occur; there may be other protections from airspeed deviations for take-off, climb, in cruise and en-route. There are also certain abnormal system conditions (e.g., air data failure, alpha data failure) where the alert will be inoperative. The designs include operations during icing conditions.

With respect to configuration, in certain FBW aircraft load factor is used in setting the alert parameter, whereas all alert systems account for the effects of normal weight and center of gravity variations.

5.6 Does the alerting operate during manual and autoflight?

Yes.

5.7 After reviewing airworthiness, safety, cost, benefit, and other relevant factors, including recent certification and fleet experience, are there any additional considerations that should be taken into account?

- (1) Aircraft with a history of low airspeed awareness issues be clearly identified.
 - (a) Any relevant accident/incident information and the specific reasons why low airspeed contributed to those events.
 - (b) Detailed information on the effectiveness of any installed low airspeed alert must be clearly identified.
 - (c) Precursor information from operational databases by aircraft type for low airspeed conditions that did not result in a reportable incident/accident. from:
 - Line Operations Safety Audit (LOSA)
 - Flight Operations Quality Assurance (FOQA)
 - Aviation Safety Action Program (ASAP)
 - Aviation Safety Reporting System (ASRS)
- (2) A comprehensive solution to address the need for low-airspeed alerting in existing fleets should be considered, based on the recommendations contained within section 5.10 of this report.

The range of necessary changes to implement low airspeed alert system is variable; the cost and benefit are also variable. This report provides representative examples of a potential integrated and functional solution. One or more technical solutions may be the most practical to incorporate in existing aircraft types. Several example "functional solutions" are described in Appendix C.

Any change will require updated flight crew procedures and pilot training to ensure proper management of the aircraft energy state when presented with a low airspeed alert.

(3) If a low airspeed alert system retrofit is mandated, a cost-benefit analysis must be performed to establish whether the implementation would be economically feasible.

The analysis must be done on a certification-basis. Aircraft variability is vital to understanding the feasibility of implementing the alert system vs. its expected effectiveness. Focus on those aircraft with known concerns of low-airspeed control, as opposed to a broad analysis, to support a specific aircraft safety finding. Expected longevity of the aircraft type being analyzed must be taken into account. For example, there may be plans to make a specific aircraft type obsolete (retire) as a result of future airspace requirements.

- (a) Costs should consider:
 - The proposed rule and advisory information for existing aircraft
 - The proposed technical solution
 - An estimate to develop and certificate the proposed technical solution (non-recurring per aircraft type)

- An estimate to retrofit the proposed technical solution (recurring per aircraft type)
- Operational manual changes
- Training
- (b) Benefits should consider the following:
 - Current accident/incident rate by aircraft type that would be prevented by implementation of a low airspeed alert
 - Average cost per accident/incident

5.8 Is coordination necessary with other harmonization working groups (e.g. Human Factors, Flight Test)? (If yes, coordinate and report on that coordination)

Yes, coordination with other harmonization working groups has already occurred.

An advisory circular developed by the FAA-Industry Stall/Stickpusher Working Group in 2010 was published by the FAA in August 2012 (reference: AC 120-109, Stall and Stick Pusher Training) and provided training procedures for stall and stick pusher recovery. The preventions in this AC should be updated to include low airspeed alerting awareness and recovery procedures.

An aviation rulemaking committee (208ARC) addressing stall and loss of control avoidance and recovery training used the information from this AC and additional inputs from industry on Loss of Control to address the pilot training aspects in their final report submitted to the FAA in December 2012. AFS-210 should be made aware of the contents of our report to ensure that low airspeed alerting awareness and recovery procedures are included in the subsequent Upset Recovery and Loss of Control NPRM that should be published for comment in late Spring, 2013.

Coordination is required with the Airplane State Awareness Joint Safety Implementation Team (JSIT), who will be generating a cost-benefit analysis. Reports from JSIT will provide the detailed data required to substantiate the effectiveness of low-airspeed alerting into existing aircraft. The report is to provide other means of mitigation that could help reduce loss of control, for those aircraft which were examined (those with specifically known loss of control accidents/incidents).

5.9 If improvements are needed for low speed alerting in the existing fleet, should the FAA/EASA adopt a design approval holder (part 26) requirement to mandate development of design changes, or would an operational rule be sufficient?

A Part 26 requirement is not necessary. If a broad requirement is deemed necessary an operational rule (e.g., part 121) would be sufficient. The operational rule would have to be specific to low airspeed alerting, regardless of the existing aircraft systems on board, and would be based on the information provided in this report.

5.10 In responding, the working group should address the factors set forth in "FAA Policy Statement: Safety—A Shared Responsibility—New Direction for Addressing Airworthiness Issues for Transport Airplanes" (70 FR 40166, July 12, 2005). The

ARAC working group should provide information that could lead to standards for low speed alerting that can be satisfied with practical design approaches.

If the FAA can justify that a rule is deemed necessary, the ASHWG recommends rulemaking and guidance that existing fleets may be able to utilize based on the information already contained within AC 25.1329-1B and AC 25.1322-1. This does not directly reference AC 25.1329-1B (and EASA Amendment CS-25/4) but instead incorporates the appropriate wording from this AC. AC 25.1329-1B was written for flight guidance systems for forward fit applications.

The following material from the referenced guidance information may be helpful in understanding the aspects of existing material relevant to low airspeed alerting:

Low Airspeed Alerting should be developed in accordance with AC 25.1322-1. A low airspeed alert should be considered as a caution level alert which precedes a warning condition (such as a stall warning), to provide immediate flight crew awareness and subsequent flight crew response.

Caution alerts should be developed in accordance with AC/AMC 25.1322-1, Paragraph 6.d:

d. Caution Alerts.

(1) The alert elements used for caution are typically identical to those used for warnings, as both require immediate flightcrew awareness.

(2) Some caution alerts are related to conditions that are precursors to potential time-critical warning conditions. In these cases, the alerting system elements associated with the caution should be consistent with the elements for related time-critical warnings (described in paragraph 6b of this AC). For example, reactive windshear warnings, ground-proximity warnings, and caution alerts can develop into time-critical warning alerts.

Two senses for attention getting should be provided. The low airspeed alert should be sufficiently specific to direct the attention of the flight crew as to the energy state of the airplane.

Under conditions where multiple alerts are occurring, or during certain failure conditions, the flight crew's workload may be significantly challenged, and any one specific alert may be missed.

Certain failure conditions may reduce the confidence of the flight crew to believe that one or more alerts are valid. For example, if the airspeed information presented to the flight crew were unreliable, the crew may not believe that the logic to set the low airspeed alert is working correctly.

Note that these considerations are not necessarily specific to low airspeed alerting - that is, alerts from legacy aircraft designs which are not in compliance with the recently updated 14 CFR/CS §25.1322, and specifically those where a suitable attention-getting means is necessary, may exhibit similar behaviors.

It is also important to note that none of the aircraft for which low airspeed "incidents" were evaluated by the JSAT had a low airspeed alerting function which would be in compliance. Prioritization of low airspeed alerts should be developed in accordance with AC/AMC 25.1322-1, paragraph 8.a:

a. Rules and General Guidelines.

(1) All flight deck alerts must be prioritized into warning, caution, and advisory categories (§ 25.1322(b)).

(2) To meet their intended function(s), alerts must be prioritized based upon urgency of flightcrew awareness and urgency of flightcrew response (§ 25.1301(a)). Normally, this means time-critical warnings are first, other warnings are second, cautions are third, and advisories are last (§ 25.1322(b)).

(3) Depending on the phase of flight, there may be a need to re-categorize certain alerts from a lower urgency level to a higher urgency level. Furthermore, prioritization within alert categories may be necessary. For example, when near threatening terrain, time-critical aural warnings must be prioritized before other warnings within the warning-alert category

(25.1322(c)(1)). AC 25-23, Airworthiness Criteria for the Installation Approval of a Terrain Awareness and Warning System (TAWS) for Part 25 Airplanes, also identifies situations where prioritization within alert categories is necessary.

(4) The prioritization scheme within each alert category, as well as the rationale, should be documented and evaluated, by following the guidance in paragraph 13, *Showing Compliance for Approval of a Flightcrew-Alerting System*, of this AC.

(5) Documentation should include the results of analyses and tests that show that any delayed or inhibited alerts do not adversely impact safety.

The intended function of the low airspeed alert should be documented, and the alert design should be incorporated according to its intended function.

A low airspeed alert may still be needed for systems that provide a speed protection function. Factors which should be considered include the reliability of the speed protection, the availability of the speed protection function in other than normal flight control laws and in particular flight phases, and speed protection failure conditions where a low airspeed alert may still be needed. Alternatively, aircraft fitted with a high incidence protection system that can demonstrate the loss of AOA protection is improbable (remote) may constitute an Equivalent Level of Safety (Ref 14 CFR §25.1309(b)(2); CS 25.1309(b)(3)).

Standard stall warning and high-speed alerts are not always timely enough for the flight crew to intervene to prevent unacceptable speed excursions. Low Airspeed Alerting should be shown to be appropriate and timely to ensure flightcrew awareness and enable the pilot to keep the airplane within an acceptable margin from the low speed range of the normal flight envelope.

Data regarding crew recognition and response from the Human Engineering Compendium by Boff/Lincoln may be helpful to develop a more "complete" timeline, from condition to expected recovery.

For practical reasons, on existing airplanes where integration of new alerts into the flight deck would be very challenging, incorporating low airspeed alerts into existing designs should consider the guidance contained in AC/AMC 25.1322-1, paragraph 14:

14. Integrating Flightcrew-Alerting System Elements into the Existing Fleet.

a. General.

(1) This material provides recommendations to applicants on how to retrofit existing airplanes so they comply with § 25.1322 without major modifications to the current flightcrew alerting system.

(2) System upgrades to existing airplanes should be compatible with the original airplane's flightcrew-alerting philosophy. The existing alerting system might not be able to facilitate the integration of additional systems and associated alerts due to limitations in the system inputs, incompatible technologies between the airplane and the system being added, or economic considerations.

(a) We discourage incorporating a new additional master visual function into the flightcrewalerting system. If it is not feasible to include additional systems and associated alerts in the existing master visual function, an additional master visual function may be installed, provided that it does not delay the flightcrew's response time for recognizing and responding to an alert.

(b) Where possible, new alerts should be integrated into the existing flightcrew alerting system. If these alerts cannot be integrated, individual annunciators or an additional alerting display system may be added.

(c) Not all alerts associated with failure flags need to be integrated into the central alerting system. However, for those alerts requiring immediate flightcrew awareness, the alert needs to meet the attention-getting requirements of § 25.1322(c)(2) as well as the other requirements in § 25.1322. Thus, a master visual or master aural alert may not be initiated, but an attention-getting aural or tactile indication must still accompany an attention-getting visual failure flag to meet the attention-getting requirement of § 25.1322(a)(1), which requires attention-getting cues through at least two different senses for warning and caution alerts.

b. Visual Alerts. Following the guidance in paragraphs 5 and 6 of this AC, determine whether or not the added system features will require activation of an airplane master visual alert.

c. Aural Alerts.

(1) Using the guidance in this AC, determine if an added system will require activating an aural alert.

(2) The new aural alert should be integrated into the existing aural alerting system and functions. If this is not possible, a separate aural alerting system may be installed, provided that a prioritization scheme between existing aural alerts and the new aural alerts is developed so that each alert is recognized and can be acted upon in the time frame appropriate for the alerting situation. This may require a demonstration of any likely combination of simultaneous alerts.

After the new and existing alerts have been merged, follow the guidance in this AC for determining how to prioritize the alerts.

d. Tactile Alerts.

(1) Using the guidance in this AC, determine if an added system will require activating a tactile alert.

(2) If possible, incorporate the new tactile alert into the existing aural alerting system. If this is not possible, a separate tactile alerting system may be installed, provided that the following elements are included:

(a) A prioritization scheme between existing tactile alerts and the new tactile alerts should be developed so that each alert is recognized and can be acted upon in the time frame appropriate for the alerting situation. After the new and existing alerts have been merged, follow the guidance in this AC for determining how to prioritize the alerts.

(b) A means to ensure that an individual alert can be understood and acted upon. This may require a demonstration of any likely combination of simultaneous alerts.

In addition to design, appropriate flight crew procedures and training for proper reaction in response to the alert must be provided.

Flight crew procedures to facilitate corrective action from the low airspeed condition:

- (1) The need to continue flying the airplane
- (2) The recognition of the low airspeed condition
- (3) An assessment of the aircraft's energy state, and other conditions which may be a factor in determining appropriate corrective action
- (4) Roles and responsibilities between flight crew members
- (5) The corrective action necessary to avoid a stall condition, and recover to safe flight

Pilots need to be trained in crew procedures, CRM measures, stall recovery and airspeed management to provide the knowledge and skills to avoid negative aircraft situations that result from low airspeed, and to respond correctly and consistently to the alerts.

Appendix A – Aircraft Survey

The following survey was administered to aircraft manufacturers, and the attached data in Appendix B was collected through the survey to identify where and how low airspeed alerting has been implemented in existing fleets.

Survey - Low Airspeed Indications, Alerting & Protection/Limiting

As a result of several recent accidents and incidents, the FAA has identified a possible need for additional low airspeed safeguards and tasked the Aviation Rulemaking Advisory Committee (ARAC) to answer technical questions on this subject. The ARAC assigned this task to the Avionics Systems Harmonization Working Group (ASHWG). To accomplish this task, the ASHWG is collecting information on the low airspeed indications, alerting and protection/limit functions available on current commercial airplanes. The ASHWG will provide information to help develop recommendations on whether there should be regulatory requirements and guidance material for retrofit of low airspeed alerting on existing aircraft.

Any rulemaking that the FAA might undertake based on the ARAC recommendations would be subject to a cost-benefit analysis. Detailed information for the FAA tasking to ARAC can be found at 76 FR 11844. The survey requests information on low airspeed flight deck indications, alerting and protection/limiting functions as well as technical information on input parameters to these functions. The following are brief definitions of terms to help in understanding the survey.

- "Indications" for low airspeed conditions information presented full time on a display or indicator.
- "Alerting" for low airspeed conditions additional information presented to the flight crew (visual and/or aural) only under specific predefined conditions.
- "Protection/limiting" for low airspeed or approach to stall conditions functions that automatically provide assistance to the flight crew (e.g., throttle advance, increase in stick forces), but only under specific predefined conditions.

The ASHWG strictly adheres to ethical standards, public law, and federal policies for safeguarding the confidentiality of all participants in this survey. Completion of this survey is voluntary and all responses to the survey that are released will not contain survey participant information.

The survey should take approximately 20-30 minutes to complete per airplane model. Please complete the survey within 30 days of receipt. It is recommended that you review the attached survey file and gather all the necessary information before completing the online survey.

Thank you for participating in this survey.

1) Select your airplane model:

• □Airbus A300-600 or A310 All

- · DAirbus A318/319/320/321/330/340/380 All
- · □ATR ATR42 All
- · □ATR ATR72 All
- □BaE J31 All
- ■Boeing 717 All
- □Boeing 727 All
- · □Boeing 737 -300, -400, -500 Conv
- · □Boeing 737 -300, -400, -500 EADI F/S
- · □Boeing 737 -300, -400, -500 EADI Spd Tape
- □Boeing 737 -600, -700, -800, -900
- □Boeing 747 -200
- · □Boeing 747 -400
- · □Boeing 757 -200 EADI F/S
- · □Boeing 757 -200, -300 EADI Spd Tape
- □Boeing 767 -200, -300 EADI F/S
- · □Boeing 767 -200, -300 EADI Spd Tape
- □Boeing 767 -400
- Boeing 777 All
- · □Boeing DC9 All
- □Boeing MD 80 All
- □Boeing MD 90 All
- □Boeing MD10 All
- □Boeing MD11 All
- Bombardier CRJ -100, -200, -400, -440
- · □Bombardier CRJ -700, -701, -702
- · □Bombardier CRJ -705, -900
- □Bombardier DHC8 -100, -200, -300
- · □Bombardier DHC8 -400
- · □Embraer 120 All
- · □Embraer 135 All
- · □Embraer 140 All
- · □Embraer 145 All
- · □Embraer 170 All
- · □Embraer 175 All
- · □Embraer 190 All
- · □Saab 340 All

Section 1 - General System Capabilities

2) 1-1. What general system capabilities does the airplane have to support new flight deck indications and alerting? (Check all that apply)

- · □Primary Flight Display (with speed tape)
- · □Alert message system (visual message list)
- · □Master caution/warning light
- · □Aural tone and/or voice capability
- · □Enhanced Ground Proximity Warning System
- □Angle of attack data
- □Flap data
- · □Anti-ice active data
- · □Other (please specify)

If you selected other, please specify

Section 2 - INDICATIONS for Low Airspeed Awareness

3) 2-1.What low airspeed awareness indications or cues (other than alerts) are presented on the airspeed indicator or airspeed tape? (Check all that apply)

[Reference AC 25-11A Appendix 1, Paragraph 2.3 provides information for low airspeed awareness]

- · \Box Colored bands
- · \Box Trend vectors
- · \Box Speed bugs
- · □Other (please specify)

If you selected other, please specify

4) 2-2. What other indications exist that support low airspeed awareness, although it may not be the primary function? (Check all that apply)

- · □Pitch limit indicator
- · \Box Angle of attack indicator
- · □Other (please specify)
- If you selected other, please specify

Section 3 - ALERTING Functionality for Low Airspeed Conditions (prior to

stall warning)

5) 3-1. What additional visual indications are presented to the flight crew for a low airspeed alert, prior to stall warning? (Check all that apply)

[CFR 14 Part 25.1322, Paragraph (c) (2) provides requirements for alerting indications]

- · □Discrete indicator (lamp)
- · \Box Master caution light
- · □Indicator on Crew Alerting display
- · □Indicator on Primary Flight Display
- Change in display of current airspeed (i.e., flash, color change, etc)
- Change in display of angle of attack or angle of attack threshold (i.e., flash, color change, etc)
- · □Other (please specify)

If you selected other, please specify

6) 3-2. What aural indications are presented to the flight crew for a low airspeed alert, prior to stall warning? (Check all that apply, and specify in Comments)

[CFR 14 Part 25.1322, Paragraph (c) (2) provides requirements for alerting indications]

- · □Voice (please specify)
- □Tone (please specify)
- · □Other (please specify)

Additional comments

7) 3-3. What input parameters are used in the logic for the low airspeed alert? (Check all that apply)

- · □Airspeed rate of change
- □Angle of attack
- $\cdot \ \square$ Barometric altitude
- · □Radio altitude
- · □Minimum maneuver speed
- · □Stick shaker speed
- · □Manual or automatic flight state
- □Thrust/power parameters
- □Time
- · □Other (please specify)

If you selected other, please specify

8) 3-4. Is the low airspeed alert adjusted for the following conditions/configurations? (Check all that apply)

- □Flaps setting
- · □Speedbrake extension
- ·□CG

- · □Load factor/g-loading
- □ Icing conditions
- · □Other (please specify)

If you selected other, please specify

9) 3-5. What trip point is used to activate the low airspeed alert? (Check all that apply)

- $\cdot \ \ \Box X$ kts or X% in the low speed amber band
- · $\Box X\%$ above stall speed
- $\cdot \ \Box X$ degrees angle of attack
- · □Low airspeed alert is same as stall warning
- · □Other (please specify)

If you selected other, please specify

10) 3-6. How do you minimize nuisance alerts? (Check all that apply)

- · □Hysteresis (e.g. delay in reset)
- □Filtering
- □Large margins from normal operating speed
- · \Box Special combinations of input parameters
- ■Manual inhibit
- · □Automatic inhibit
- · □Other (please specify)

If you selected other, please specify

11) 3-7. What circumstances or conditions are used to inhibit the low airspeed alert? (Check all that apply)

- · □Baro Altitude
- $\cdot \ \square$ Radio Altitude
- · \Box Priorities with other alerts
- □Phase of flight (e.g., takeoff, approach)
- · □Non-normal configurations
- □Other (please specify)

If you selected other, please specify

12) 3-8. Is there a design requirement or goal for a minimum time margin between the low airspeed alert activation and stall warning activation? (assuming these are two independent points)

- · □Yes (please specify below)
- · □No

Additional comments

13) 3-9. Can you provide a description or illustration or logic diagram or equation that describes how the low airspeed alert is activated?

- _Yes (If so, email to the point of contact identified in the introduction)
- · □No

14) 3-10. How did you determine that the Low Airspeed Alert is timely (i.e., provides the pilot sufficient time to avoid stall warning, or some other identified point)? (Check all that apply)

- ■Analysis
- □In-service history
- □Flight test
- · □Flight simulator or lab testing
- · □Other (please specify)

If you selected other, please specify

15) 3-11. Is the alerting functionality you have described above implemented on all airplanes or only some through a customer option, STC or later add-on?

- · □All
- · \Box Some through option, STC or later add-on

16) 3-12. If you selected some through option, STC or later add-on, please specify the number of airplanes modified versus the number in the fleet:

Number of airplanes modified _

Number airplanes in the fleet _____

Section 4 - PROTECTION/LIMITING functionality (automated assistance)

for low airspeed or approach to stall conditions?

17) 4-1. What protection/limiting functionality is available to automatically assist the pilot for low airspeed conditions, prior to stall warning? (Check all that apply)

- · □Autothrottle "wakeup"/automatic thrust activation
- · \Box Stick pusher
- · \Box Automatic pitch control
- · □Increased column/stick forces
- · □Angle of attack protection
- · \Box Auto-slat extension
- · \Box Angle of attack limit
- · □Other (please specify)

If you selected other, please specify

18) 4-2. What protection/limiting functionality is available to automatically assist the pilot for approach to stall conditions, at/after stall warning? (Check all that apply)

- $\cdot \ \Box$ Stick pusher
- · □Automatic pitch control
- · □Increased column/stick forces
- · □Angle of attack protection
- · \Box Auto-slat extension
- □Angle of attack limit
- · □Other (please specify)

If you selected other, please specify

Appendix B – Survey Results

Appendix B contains the data which was collected through the survey questions in Appendix A, to identify where and how low airspeed alerting has been implemented in existing fleets.

		1-1. What ge	neral system ca	pabilites does	the airplane i	have to supp	ort new flight o	leck indicati	ons and alertin	g?	
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Primary Flight Display (with speedtape)	Alert message system (visual message list)	Master caution/war ning light	Aural tone and/or voice capability	Enhanced Ground Proximity Warning System	Angle of attack data	Flap data	Anti-ice active data	Other	OtherText
Round dial	Boeing DC9 All (433)	Na	Να	Yes	Yes	Yes	Yes	Yes	No	No	EGPWS is available as an
No SW cptr	Boeing 727 All (826)	No	No	No	Yes	Yes	Yes	Yes	No	No	SIC
	Boeing 747-200 (202)	Na	No	No	Yes	Yes	Yes	Yes	No	No	0.0
Round dial	Embraer 120 All (126)	No	No	Yes	Yes	No	Yes	Yes	Yes	No	
Have SW cptr to	Boeing 737 -300, -400, -500 Conv (1550)	No	No	Yes	Yes	Yes	Yes	Yes	No	No	EGPWS is available as an
upport	Boeing MD 80 All (1916)	No	Yes	Yes	Yes	Yes	Yes	Yes	No	No	STC
	Boeing MD 90 All (108)	Na	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
	Saab 340 All (197)	No	No	Yes	Yes	Yes	No	No	No	No .	
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	Yes	Yes	Yes	Yes	Yes	No	No	
	Boeing 757 200 EADI F/S (971)	No	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
	Boeing 767-200, -300 EADI F/S (880)	No	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	Yes	No	Yes	Yes	Yes	Yes	Yes	No	No	
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Ice Detector detected ice
	Bombardier CRJ -700, -701, -702 (215)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	
	Bombardier CRJ -705, -900 (105 -900)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	
	Embraer 135 All (137)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
	Embraer 140 All (74)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
	Embraer 145 All (503)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
	Embraer 170 All (76) Embraer 175 All (54)	Yes Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	- 71
PFD with visual	Embraer 199 All (51)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
low airspeed alert	Boeing 717 All (155)	Yes	Yes	Yes	Yes	Ves	Yes	Yes	No	No	
indication only. (3-1)	Boeing MD10/11 All (257)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	No	
<i>C</i> -11	Airbus A300.600 or A310 All (A300.158; A310.70)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
	Dealer 737 CBR 700 000 (3008)		12							12	
10	Boeing 737 -600, -700, -800, -900 (3908) Boeing 747-400 (675)	Yes	No Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
Visual & aural low airspeed alert	Boeing 767-400 (38)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	No	2
indication. (3.1, 3-2)	Boeing 777 All (981)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	
0001300 3	Airbus A318/319/320/321/330/340/380 All (767) (318-321724; 330-38043)	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	No	

		2-1.What low airspeed	awareness indications or c	ues (other than alerts) are p	resented on the airspeed i	ndicator or airspeed tape?
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Colored bands	Trend vectors	Speed bugs	Other	OtherText
Round dial	Boeing DC9 All (433)	No	No	No	No	
No SW cptr	Boeing 727 All (826)	No	No	Nio	No	
no swiepu	Boeing 747-200 (202)	140	No	Nio	No	
Round dial	Embraer 120 All (126)	No	No	Ma	Yes	Fast/Slow indicator
Have SW cptr to		No	No	Nao	No	
support	Boeing MD 80 All (1016)	No.	No	Na	No	
	Boeing MD 99 All (198)	No	No	No	No	
		No	No	Yes	No	
		A Decimination of the second	No	ALC: NOT THE OWNER OF THE OWNER OWNER OF THE OWNER O		
	Boeing 737 -300, -400, -500 EADI F/S (190) Boeing 757-200 EADI F/S (971)	No No	No	No	Yes	Fast/Slow indicator
	Boeing 757-200 EAD(F/S (971) Boeing 767-200, -300 EAD(F/S (880)	No	Nó	No	Yes	1 Baccon announ
		Yes	Yes	Yes	No	
Have EFIS & alert capability.		Yes	Yes	Yes	No	
(1-1)		30 C	1837	0.44		
	Boeing 767 .200, .300 EADI Spd Tape (46)	Yes	Yes	Yes	No	
	Bombardier CRJ-100, 200, 400, 440 (718)	Yes	Yes	Yes	No	
	Bombardier CRJ -700, -701, -702 (215)	Yes	Yes	Yes	Na	
	Bombardier CRJ -705, -900 (105 -900)	Yes	Yes	Yes	No	
	Embraer 135 All (137)	Yes	Yes	Ne	No	
	Embraer 140 All (74)	Yes	Yes	NIO	No	When airspeed trend vector
	Embraer 145 All (503)	Yes	Yes	tila	Na	touches the red band.
2	Embraer 170 All (76)	Yes	Yen	Ne	No	airspeed miling digits becom
PFD with visual	Embraer 175 All (54)	Yes	Yes	Nα	No	amber.
low airspeed alert	Embraer 190 All (51)	Yes	Yes	Na	No	
indication only.	Boeing 717 All (155) Boeing MD10/11 All (257)	Yes	Yes Yes	Yes	Na	
(3-1)	Airbus A300-600 or A310 All (A300-158; A310-70)	Yes	Yes	Yes	No	
	Boeing 737 -500, -700, -800, -900 (3908)	Yes	Yes	Yes	Na	
Visual & aural low	Boeing 747-400 (675)	Yes	Yes	Yes	No	
airspeed alert	Boeing 767-400 (38)	Yes	Yes	Yes	No	
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	Yes	Y'es	No	
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	Yes	Yes	Yes	No	

		2-2. What other	indications exist that	support low air	speed awareness, although it may not be the primary function?
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Pitch limit indicator	Angle of attack indicator	Other	OtherText
Round dial	Boeing DC9 All (433)	No	No	No	
No SW cptr	Boeing 727 All (826)	No	No	No	
no strepu	Boeing 747-200 (202)	No	No	No	
	Embraer 120 All (126)	No	No	No	
Round dial	Boeing 737 -300, 400, -500 Conv (1550)	No	No	No	
Have SW cptr to support	Boeing MD 80 All (1016)	Yes	No	No	
apport	Boeing MD 90 All (108)	Yes	No	No	
	Saab 340 All (197)	No	No	No	
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	No	
	Boeing 757-200 EADI F/S (971)	Yes	No	No	
	Boeing 767-200, -300 EADI F/S (880)	Yes	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	Yes	Nio	No	
capability. (1-1)	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	Yes	No	No	
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	Yes	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	No	
	Bombardier CRJ -700, -701, -702 (215)	No	No	No	
	Bombardier CRJ -705, -900 (105 -900)	No	No	No	
	Embraer 135 All (137)	Yes	No	No	
	Embraer 140 All (74)	Yes	No	No	
	Embraer 145 All (503)	Yes	No	No	
	Embraer 170 All (76)	Yes	No	No	
PFD with visual	Embraer 175 All (54)	Yes Yes	No No	No	
low airspeed alert	Embraer 190 All (51) Boeing 717 All (155)	Yes	No	No No	
indication only.	Boeing MD10/11 AB (257)	Yes	No	No	
(3-1)		No	No	Na	
	Boeing 737 -600, -700, -800, -900 (3908)	Yes	Yes	Yes	The Angle of attack indicator is an available option, the "other" is the pilot- selectable flight path vector
Visual & aural low	Boeing 747-400 (675)	Yes	No	Yes	There is a pilot-selectable flight path vector
airspeed alert	Boeing 767-400 (38)	Yes	Yes	Yes	Pilot-selectable fight path vector
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	Yes	Yes	The Angle of attack indicator is an available option; the "other" is the pilot- selectable flight path vector
	Airbus A318/319/328/321/330/340/380 All (767) (318-321724; 330-38043)	Yes	No	No	

		3.1. What ad	iditional visual indi	cations are pres	ented to the fligt	is craw for a low	aimpeed alert, j	rior to stall	warning?	
Model Capabilities (Reference survey question is parantheses)	Airplane model (number in fleet at the time of data collection)	Discrete Indicator (Iamp)	Master caution fight	Indicator on Crew Alerting display	Indicator on Primary Flight Display	Change in display of current airspeed (Le. Sash, color change, etc)	Change it display of AOA, or AOA threshold i.e., flash, color change, etc)	Other	Other Text	
Round dial	Boeing DC9 All (433)	140	No	No	No	No		No		
No SW cptr	Boeing 727 All (826)	140	No	No	No	No		No		
	Boeing 747-200 (202)	No	No	No	No	No	No	No.		
Round dist	Embraer 120 All (126)	140	No	Np	No	No		Yes	Dedicated equipment to indicate fast/slow airspeed	
ANUCCESSION CONTRACTOR	Boeing 737 300, 400, 500 Conv (1550) Boeing MD 80 All (1016)	flo	No	740	140	No. No		Na:		
apport	Boeing MD 80 All (1016)	140	740.	No	140	140	140	filo		
interest i	Boeing MD 90 All (108)	Nu	Ne	No	Na	No		No		
	Snab 340 All (197)	140	No	No	No	No	No	No		
	Boeing 737 -300, -400, -500 EAD(F/S (190)	140	No	No	Yes	No	No	No		
	Boeing 757-200 EADI F/S (971)	140	No	No	Yes	No		No	10 kts below target speed	
	Boeing 767-200, -300 EADI F/S (880)	No	No	No	Yes	No	No	Na		
	Boeing 737 -300, 400, -500 EADt Spd Tape (58)	No	No	No	No	No	No	No		
capability.	Boeing 757-200, 300 EADI Spd Tape (37 -300 only)	140	No.	No	No	No	140	No		
(1.1)	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	No	No	No	No	No		
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	No	No	No	No	Yes		
	Bombardier CRJ 700, 701, 707 (215)	No	No	No	No	No	No	Yes	Engine Cont Ignition light and message	
	Bombardier CRJ -705, -900 (105 -900)	No	No	No	No	No	No	Yes		
	Embrane 135 All (137)	No.	Hu	No	Yes	Yes		No		
	Embraes 140 All (74)	Fin	Nu	No	Yes	Yes	No	Net		
	Embraer 145 All (503)	No	No	140	Yes	Yes	No	No		
	Embraer 170 All (76)	No	Net	110	Ves	Yes	No	No		
PFD with visual	Embraer 175 All (54)	140	No	140	Yes	Yes		No.		
ow airspeed stert	Embraer 190 All (51)	140	No	No	Yes			No		
ndication only.	Boeing 717 All (155)	140	No	No	Yes	Yes		No.		
3.1)	Boeing MD10/11 All (257)	140	140	No	Yes	Yes	No	Nas.	and a second	
	Airbus A300.600 or A310 All (A300.158; A310-70)	No	Yes	No	Yes	No	No	Yes	3 alerts considered 11 dynamic red speedtape on PFD. 3] AP doconnect at VLS-10 associated to master caution flashing 3] alphafoor autoaut of TOGA Thrust by autoconnection of ATHE "THRUST LATCH" and on FMA(PFD)	
									1.00 00	
	Boeing 737 600, 700, 800, 900 (3908)	No	No	No	Yes	Yes	No	No.		
a charter of a second contra	Boeing 747.400 (675)	No	Yes	Yes	Yes	Yes		No		
airspeed alort	Boeing 767-400 (38)	No	Yes	Yes	Yes	Yes	No	No		
indication.	CAR WILLIAM 1	Sal	2211	NAL:	100	1000	6 55	2011		
(3-1, 3-2)	Boeing 777 All (981)	No	Yes	Yes	Yes	Yes	No	No		
2.1 CA14	Airbox A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	Yes	Yes	Yes	No	Yes	No	three alerts considered - 1) dynamic amber & red speedtape on PFD, 2) alfa floor amber flashing on ECAM & PFD, 3) AP disconnect at alphaprot + 1*	

		3-2. What sural indicat	tions are presented to th	ne flight crew for a low i	sirspeed alert, prior to stall warning?
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Voice (please specify)	Tone (please specify)	Other (please specify)	CommentText
Round dial	Boeing DC9 All (433)	Na	No	No	
No SW cptr	Boeing 727 All (826)	No	No	No	
in art spu	Boeing 747-200 (202)	No	No	No	
lound dial	Embraer 120 All (125)	No	No	Yes	The sound produced by stick shaker motor is loud enough to be perceived by pilots
Have SW cptr to	Boeing 737 -300, -400, -500 Conv (1550) Boeing MD 80 All (1016)	No	No	No	· · · · · · · · · · · · · · · · · · ·
upport	Looning and to real (1010)	140.	mo	140	
	Boeing MD 90 All (108)	No	No	No	-
	Saab 340 All (197)	No	No	No	
	Boeing 737 -300, 400, -500 EADI F/S (190)	No	No	No	
	Boeing 757-200 EADI F/S (971)	No	No	No	
	Boeing 767-200, .300 EADI F/S (880)	No	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	No	No	
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	No	
	Bombardier CRJ -700, -701, -702 (215)	No	No	No	
	Bombardier CRJ -705, -900 (105 -900)	No	No	No	
	Embraer 135 All (137)	No	No	Yes	
	Embraer 140 All (74)	No	No	Yes	
	Embraer 145 All (503)	No	No	Yes	The sound produced by stick shaker motor is loud enough to b
	Embraer 170 All (76)	No	No	Yes	perceived by pilots
PFD with visual	Embraer 175 All (54)	No	No	Yes	
ow airspeed alert	Embraer 190 All (51)	No	No	Yes	V
ndication only.	Boeing 717 All (155)	No No	No	No	
3-1)	Boeing MD10/11 All (257)			No	"cardo, charas" adap AD decrement 8 stocksholar
	Airbus A300-600 or A310 All (A300-158; A310-70)	No	No	Yes	"cavalry charge" when AP disconnet & stickshaker
	Boeing 737 -600, -700, -800, -900 (3908)	Yes	No	No	Service Bulletin available to enable voice - "AIRSPEED LOW, AIRSPEED LOW"
Visual & aural low	Boeing 747-400 (675)	No	Yes	No	
airspeed alert	Boeing 767-400 (38)	No	Yes	No	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
ndication. 3-1, 3-2)	Boeing 777 All (981)	No	Yes	No	Standard EICAS caution tone
	Airbus A318/319/320/321/330/340/380 All (767) (318-321724; 330-38043)	Yes	No	No	"Speed-Speed-Speed" below 2500ft

		3-3. What	input param	eters are use	ed in the logic	for the lo	w airspeed al	ert?					
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Airspeed	Airspeed rate of change	Angle of attack	Barometric altitude	Radio eltitude	Minimum maneuver speed	Stick shaker speed	Manual or automatic flight state	Thrust or power parameters	Time	Other	Other Text
Round dial	Boeing DC9 All (433)	No	No	No	Nio	No	No	No	No	No	No	No	
No SW cptr	Boeing 727 All (826)	No	No	No	No	No	No	No	No	No	No	No	-
	Boeing 747-200 (202)	No	No	No	No	No	No	No	No	No	No	No	
	Embraer 120 All (126)	No	No	Yes	No	No	No	Yes	No	No	No	Na	
Round dial	Boeing 737 -300, 400, -500 Conv (1550)	No	No	No	No	No	No.	No	No	No	No	No	
lave SW cptr to support	Boeing MD 80 AH (1016)	No	No	No	Ne	No	No	840	No	No	No	No	
	Boeing MD 90 All (108)	No	No	No	No	No	No	No	No	No	Na	No	
	Saab 340 All (197)	No.	No	No	No	No	No	êlo.	No	No	No	No	2
	Boeing 737 -300, -400, -500 EADI F/S (190)	Yes	Na	No	No	No	No	No	No	No	Na	No	
	Boeing 757-200 EADI F/S (971)	Yes	140	No	No	No	No	No	No	No	No	No	
	Boeing 767-200, -300 EADI F/S (880)	Yes	No	No	No	No	No	No	No	No	No	No	
Have EFIS & alert capability. (1-1)	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	No	No	No	No	No	No	No	No	
	Boeing 757-200, 300 EADI Spd Tape (37 -300 only)	No	No	No	No	No	No	No	No	No	No	No	
	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	No	No	No	No	No	No	No	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	Yes	No	No	No	No	No	No	No	Yes	
	Bombardier CRJ -700, -701, -702 (215)	No	No	Yes	No	No	No	No	No	No	No	Yes	Macti & AoA rate
	Bombardier CRJ -705, -900 (105 -900)	No	No	Yes	No	140	No	No	No	No	No	Yes	
	Embraer 135 All (137)	No	No	Yes	No	No	Ne	Yes	No	Na	No	No	- 21 P
	Embraer 140 All (74)	No	No	Yes	No	No	No.	Yes	No	140	No	No	
	Embraer 145 All (503) Embraer 170 All (76)	No. Yes	No Yes	Yes	No	No	No	Yes	No	No	No	No	-
PED with visual	Embraer 175 All (54)	Yes	Yes	Yes	Fig	feo	No	Yes	tio	Wes	No	No	
ow aimpeed alert	Embraer 190 All (51)	Yes	Yes	Yes	No	No	No	Yes.	No	Yes	No	No	
indication only.	Boeing 717 All (155)	Yes	No	No	No	No	Yes	Yes	No	No	No	No	
(3-1)	Boeing MD10/11 All (257)	Yes	Na	No	No	No	Yes	Yes	140	No	110	No	
	Airbus A306 600 or A310 All (A300 158: A310 70)	Yes	Yes	Yes	No	rio	No	No	740	Na	110	No	
				2			2						
-	Boeing 737 -600, -700, -800, -900 (3908) Boeing 747 -400 (675)	Yes	Na	No	No	No	Yes	Yes	No	No	No	No	-
Visual & aural low	Boeing 767-400 (675) Boeing 767-400 (38)	Yes	No	Na	No	No	Yes	Yes	No	No	No	No	-
airspeed alert indication.	Personal Visit Links	1.42	1.14				1.00				110	110	
(3-1, 3-2)	Boeing 777 All (981)	Yes	No	No	No	No	Yes	Yes	tio .	No	No	No	
and a	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	Yes	Yes	Yes	No	Yes	No	No	No	No	No	Yes	Mach (for alphaprot alphamax)

		3-4. Is the low	sinspeed alert adj	iusted for the fo	llowing condition	ons/configurations?			
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Flaps setting	Speedbrake extension	Weight	CG	Load factor/g- loading	Icing conditions	Other	OtherText
Round dial	Boeing DC9 All (433)	No	No	No	No	No	No	No	
No SW cptr		No	No	No	No	No	No	No	
no sir cpu	Boeing 747-200 (202)	No	No	No	No	No	No	No	
Round dial	Embraer 120 All (126)	Yes	No	No	No	Na	Yes	No	
Have SW cptr to	Boeing 737 -300, 400, -500 Conv (1550)	Nn	No	No	No	No	No	No	
support	Boeing MD 80 All (1916)	No	No	No	No	No	No	14e	
		No	No	Na	No	No	No	160	
	Saab 340 All (197)	No	No	No	No	No	No	No	
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	Na	No	Na	No	file	
	Boeing 757-200 EADI F/S (971)	No	No	No	No	No	No	No	
	Boeing 767-200, -300 EADI F/S (880)	No	No	No	No	No	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	No	No	No	No	
capability.	Boeing 757-200, 300 EADI Spd Tape (37 -300 only)	No	No	No	No	No	No	No	
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	140	No	No	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	Yes	Na	No	No	No	No	Yes	1
	Bombardier CRJ 700, 701, 702 (215)	Yes	No	No	No	No	No	Yes	Mach
	Bombardier CRJ -705, -900 (105 -900)	Yes	No	No	No	No	No	Yes	
	Embraer 135 All (137)	Yes	No	No	\$30	No.	Yes	No	
	Embraor 140 All (74)	Yes	No.	140	No	No	Yes	Na	
	Embraer 145 All (503)	YKS	No	No	ê lo	No	Yes	No	
	Embraor 170 All (76)	Yes	No	No	No	No	Yes	No	
PFD with visual	Embraer 175 All (54)	Yes	No	140	No	Να	Yes	No	
low airspeed alert.	Embraer 190 All (51) Boeing 717 All (155)	Yes Yes	No	Yes	No Yes	No Yes	Yes No	No No	
indication only.	Boeing MD10/11 All (257)	Yes	No	Yes	Yes	Yes	No	No	
(3-1)	Airbus A386-500 or A310 All (A300-158: A310-70)	Yes	Yes	Yes	No	Yes	110	No	
	Boeing 737 -800, -700, -800, -900 (3908)	Yes	No	Yes	Yes	Yes	Yes	No	The adjustments are made to the
Visual & aural low	Boeing 747-400 (675)	Yes	No	Yes	Yes	Yes	Yes	No	amber band/barber pole, which
airspeed alert	Boeing 767-400 (38)	Yes	No	Yes	Yes	Yes	Yes	No	directly affects alert trip point
indication. (3-1, 3-2)	Boeing 777 All (981)	Yea	Ne	Yes	Yes	Yes	Yes	No	100 MAR
	Airbus A318/319/320/321/330/340/380 All (767) (318-321724; 330-380-43)	Yes	Yes	Yes	No	Yes	No	No	

	s Airplane model (number in fleet at the time of data collection)	3.5. What trip point is used to activate the low airspeed alert?								
Model Capabilities (Reference survey question in parentheses)		X kts or X% in the low speed amber band	X% above stall speed	X degrees angle of attack	Low airspeed alert is same as stall warning	Other	Other Text			
Round dial	Boeing DC9 All (433)	No	No	No	No	No				
No SW cptr	Boeing 727 All (826)	No	No	No	No	No				
NO SW CDB	Boeing 747-200 (202)	No	No	No	No	No				
Round dial	Embraer 120 All (126)	No	No	Yes	Yes	No				
Have SW cptr to		No	No	No	No	No No				
support.	Boeing MD 80 All (1916)	No	No	1740	IND .	NED .				
	Bosing MD 90 All (108)	No	No	No	No	No				
	Seeb 340 All (197)	No	No	No	740	No				
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	No	No	Yes	Difference from target airspeed (10 kts.)			
	Boeing 757-200 EADI F/S (971)	No	No	No	Na	Yes	Difference from target airspeed			
	Boeing 767-200, 300 EADI F/S (880)	No	No	No	No	Yes	Difference from target arspeed			
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	No	No				
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	No	No	No	No				
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	No	No	No				
		No	No	Yes	No	No				
	Bombardier CRJ .700, .701, .702 (215)	No	No	Yes	Yes	No				
	Bombardier CRJ -705, -900 (105 -900)	No	No	Yes	No	No				
9	Embraer 135 All (137)	No	No	Yes	Yes	No				
	Embraer 140 All (74)	No	No	Yes	Yes	tio :				
	Embraer 145 All (503)	No	No	Yes	Yes	No				
	Embraer 170 All (76)	Yes	No	Yes	Yes	No				
PFD with visual	Embraer 175 All (54) Embraer 190 All (51)	Yes Yes	No	Yes Yes	Yes	No No				
low airspeed alert	(Empraer 190 All (51) (Boeing 717 All (155)	Yes	No	No	Pio	No				
indication only.	Boeing MD10/11 All (257)	Yez	No	No	No	No				
(3-1)	Airbus A300.600 or A310 All (A300.158; A310.70)	Yes	Yes	Yes	No	No				
	Boeing 737 600, 700, 800, 900 (3908)	Yes	No	No	No	No				
Visual & aural low	Boeing 747.400 (675)	Yes	No	No	No	No.	30%			
airspeed alert	Boeing 767-400 (38)	Yes	Na	No	No	No				
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	No	No	No	No				
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	No	Yes	No	No				

		3-6. How do you	minimize nuisa	nce alorts?					
Model Capabilities (Reference survey question to parentheses)	Airplane model (number in flent at the time of data collection)	Hystoresis (e.g. delay in reset)	Filtering	Large morgins from normal operating speed	Special combinations of input parameters	Manual inhibit	Automatic inhibit	Other	Other Text
Round dial		No	No	No	No	No	No	No	
No SW cptr		No	No	No	No	No	No	No	
No Stv cpu	Boeing 747-200 (202)	No	No	No	No	No	No	No	
	Embraer 120 All (126)	20.	100	in the second	No	122		No	
Round dial		No. No	No	Yes	Nin	No.	No	No	
Have SW cptr to		tiko:	No	No	No	into-	TND IND	No	
support	and and an an and the full	174		100	1.00				
	NAME AND A DESCRIPTION OF A DESCRIPTION OF A DESCRIPTION OF A DESCRIPTIONO	No	No	No	No	No	No	No	
	Saab 340 All (197)	No	No	No	No	No	No	No	
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	No	No	No	No	No	
	Boeing 757 200 EADI F/S (971)	No:	No	No	No	No.	No	No	
	Boeing 767-200, -300 EADI F/S (800)	No.	No	No	No	No	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	No	No.	No	No	
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	No	No	No	No	No	No	
(1.1)	Boeing 767 -200, -300 EADI Spd Tape (45)	No.	No	No	No	No	No	No	
		No	Yes	Yes	No	No	No	No	
		No	Vea	Yes	No	No	No	No	
		No	Yes	Yes.	No	No	No	No	
-		No	Yes	Yes	No	No	No	No	
		No	Yes	Yen	No	No	No	No	
		No	Ves	Yes	No	ING.	No	No	
		Yes	Yes	Yes	No	No	No	No	
PFD with visual		Ves	Yes	Yes	No	Na	No	No	
low airspeed alert		Yes	Yes	Yes	No	No	No	No	
indication only.		No	No	140	No	No	No	140	
(2-1)	a second second	No.	Nb .	No Yes	No	No	No Yes	No	
	Annous Autoropy, or Autor an paudo-108; Auto-709	rea	1.89	149	140		1.83	1000	The alert is triggered when current airspeed
	Boeing 737 -600, -700, -800, -900 (3908)	Yes	Yes	Yes	No	No	No	No	has decreased so as to "use up" 30% of the
Visual & aural low	Boeing 747-400 (675)	Yes	Yes	Yes	No	No	No	No	minumum maneuver speed margin to
airspeed alert		Yea	Yes	Yes	No	No	No	No	stickshaker. The alert is removed when
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	Yes	Yes	No	No	No	No	airspeed is greater than minimum maneuver speed
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	Yes	Yes	No	No	Yes	No	

		3-7. What circum	stances or condition	s are used to inhibit	the low airspeed ale	n?		
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Baro Altitude	Radio Altitude	Priorities with other alerts	Phase of flight (e.g., takeoff, approach)	Non-normal configurations	Other	Other Text
Round dial	Boeing DC9 All (433)	No	No	No.	No	No	No	
No SW cptr	Boeing 727 All (826)	No	No	No	No	No	No	
no su cho	Boeing 747-200 (202)	No	No	No.	No	No	No	
	Embraer 120 All (126)	No	No	No	No	No	Yes	Weight On Wheels equal to ground inhibits LAA.
Round dial	Boeing 737 -300, -400, -500 Conv (1550)	No	No.	No	No	No	No	
Have SW cptr to	Boeing MD 80 All (1016)	No	No	No	No	No	No	
apport	Boeing MD 90 All (108)	No	No	No	No	No	No	
	Saab 340 All (197)	No	No	No	No	No	No	
	The Instance of the United States and the States of the St	No	No	No	No	No	No	
	Boeing 737 -300, 400, 500 EADI F/S (190) Boeing 757 200 EADI F/S (971)	No	No	No	No	No	No	
	Boeing 767-200, 300 EADI F/S (880)	No	No	No	No	No	No	
Have EFIS & alert	Boeing 737 -300, 400, -500 EADI Spd Tape (58)	No	No	No	No	filo	No	
capability.	Boeing 757-200, 300 EADI Spd Tape (37-300 only)	No	No	No	No	No	No	
(1-1)		No	No	No	No	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	No	No	No	No	
	Bombardier CRJ -700, -201, -702 (215)	No	No	No	No	No	No	
		1.000	and the second se	1 Mar 10		1000		
	Bombardier CRJ -705, -900 (105 -900) Embraer 135 All (137)	No	No	No	No	No	No	
	Embraer 135 All (137) Embraer 140 All (74)	No	No	No	No	No	Yes	
	Embraer 145 All (503)	No	No	No	No	No	Yes	Weight On Wheels equal to
	Embraer 170 All (76)	No	No	No	No	No	Yes	ground inhibits LAA
PFD with visual	Embraer 175 All (54)	No	No	No	No	No	Yes	
low airspeed alert	Embraer 190 All (51)	No	No	No	No	No	Yes	
indication only.	Boeing 717 All (155)	No	No	No	No.	No	No	
(3-1)	Boeing MD10/11 All (257)	No	No	No	Na	No	No	
	Airbus A300 600 or A310 All (A300-158; A310-70)	No	Yes	No	No	No	No	
	Boeing 737 -500, -700, -800, -900 (3908)	Na	No	No	Yes	No	Yes	The low speed alert is inhibited until first flap
Visual & aural low	Boeing 747-400 (675)	No	No	No	Yes	No	Yes	retraction after takeoff. The
airspeed alert	Boeing 767-400 (38)	No	No	No	Yes	No	Yes	logic detects that the flaps
indication. (3-1, 3-2)	Boeing 777 All (981)	No	No	No	Yes	No	Yes	have changed after takeoff
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	Yes	No	No	No	Yes	slat/flap conf for "Speed- Speed-Speed" alert

			uirement or goal for a minimum time margin between tivation and stall warning activation?	3.9. Can you provide a description or illustration or logic diagram or equation that describes how the low airspeed alert is activated?	
Model Capabilities (Reference survey question in parentheses)	Airplane model (number in fleet at the time of data collection)	Yes or No	CommentText	Yes or No	
Round dial No SW cptr	Boeing DC9 All (433) Boeing 727 All (826) Boeing 747-200 (202)				
Round dial Have SW cptr to support	Embraer 120 All (126) Boeing 737 -300, 400, 500 Conv (1550) Boeing MD 80 All (1016)	No		Yes	
mpport	Boeing MD 90 All (108) Saab 340 All (197)				
	Boeing 737 300, 400, 500 EADI F/S (190) Boeing 757-200 EADI F/S (971) Boeing 767-200, 300 EADI F/S (880)	No No No		No No No	
Have EFIS & alert capability. (1-1)	Boeing 737 -300, -400, -500 EADI Spd Tape (58) Boeing 757-200, -300 EADI Spd Tape (37 -300 only)				
	Boeing 767 -200, -300 EADI Spd Tape (46) Bombardier CRJ-100, -200, -400, -440 (718) Bombardier CRJ -700, -701, -702 (215)	No No		Yes Yes	
	Bombardier CRJ -705, -900 (105 -900) Embraer 135 All (137) Embraer 140 All (74)	No No No	Low Speed alert is the same as Stall	Yes Yes Yes	
PFD with visual	Embraer 145 All (503) Embraer 170 All (76) Embraer 175 All (54)	No No No	Low Speed alert is the same as Stall Low Speed alert is the same as Stall Low Speed alert is the same as Stall	Yes Yes Yes	
low airspeed alert indication only. (3-1)	Embraer 190 All (51) Boeing 717 All (155) Boeing MD10/11 All (257)	No No No	Low Speed alert is the same as Stall	Yes No No	
	Airbus A300-600 or A310 All (A300-158; A310-70)	No	Single time margin value cannot address all dynamic conditions	Yes	
Visual & aural low airspeed alert	Boeing 737 -600, -700, -800, -900 (3908)	No		Yes	
	Boeing 717-400 (575) Boeing 767-400 (38)	No No		Yes Yes	
indication. (3-1, 3-2)	Boeing 777 All (981)	No		Yes	
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	single time margin value cannot address all dynamic conditions	Yes	

	s .	3-10. How did you determine that the Low Airspeed Alert is timely (i.e., provides the pilot sufficient time to avoid stall warning, some other identified point)?							
Model Capabilities (Reference survey question in parentheses)		Analysis	In service history	Flight test	Flight simulator or lab testing	Other	Other Text		
Round dial	Boeing DC9 All (433)	No	No	No	No	No			
No SW cptr	Boeing 727 All (826)	No	No	No	No	No			
na svi cpu	Boeing 747-200 (202)	No	No	No	No	No			
Round dial	Embraer 120 All (125)	No	No	Yes	No	Yes	The values used for stick shaker activation follows the certification requirements (25.207) and are confirmed via flight test		
Have SW cptr to	Boeing 737 300, 400, 500 Conv (1550)	No	No	No	No	No	The second		
support	Boeing MD 89 All (1015)	No	No	No	No	No			
Contract.	Boeing MD 90 All (106)	Nα	No	No	Ne	No			
	Saab 340 All (197)	No	No	No	No	No			
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	No	No	No	No			
	Boeing 757-200 EADI F/S (971)	No	No	No	No	140			
	Boeing 767-200, -300 EADI F/S (880)	No.	No	No	No	No			
Have EFIS & alert	Boeing 737 .300, .400, .500 EADI Spd Tape (58)	No	No	No	No	No			
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	No	No	No	No			
(1-1)	Boeing 767 .200, .300 EADI Spd Tape (46)	No	No	No	No	No			
	Bombardier CRJ-100, -200, -400, -440 (718)	No	No	No	No	Yes	AND COMPANY AND		
	Bombardier CRJ -700, -701, -702 (215)	No	No	No	No	Yes	Has not been verified		
	Bombardier CRJ -705, -900 (105 -900)	No	No	No	No	Yes			
27 E 1	Embraer 135 All (137)	No	No	Yes	No	Yes			
	Embraer 140 All (74)	No	No	Yes	No	Yes	The values used for stick shaker activation		
	Embraer 145 All (503)	140	No	Yes	No	Yes	follows the certification requirements (25 207)		
	Embraer 170 All (76)	No	No	Yes	No	Yes	and are confirmed via fight test.		
PFD with visual	Embraer 175 All (54) Embraer 190 All (51)	No	No	Yes	No	Yes			
low airspeed alert	Boeing 717 All (155)	No	No	No	No	No			
indication only.	Boeing MD10/11 All (257)	No	No	No	No	No			
(3-1)	Airbus A309-509 or A310 All (A309-158; A319-70)	Yes	Yes	Yes	Yes	No			
Visual & aural low airspeed alert indication. (3-1, 3-2)	Boeing 737 -600, -700, -800, -900 (3908)	Yes	Yes	Yes	Yes	No			
	Boeing 747-400 (675)	Yes	Yes	Yes	Yes	No			
	Boeing 767-400 (38)	Yes	Yes	Yes	Yes	No			
	Boeing 777 All (981)	Yes	Yes	Yes	Yes	No			
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	Yes	Yes	Yes	Yes	No			

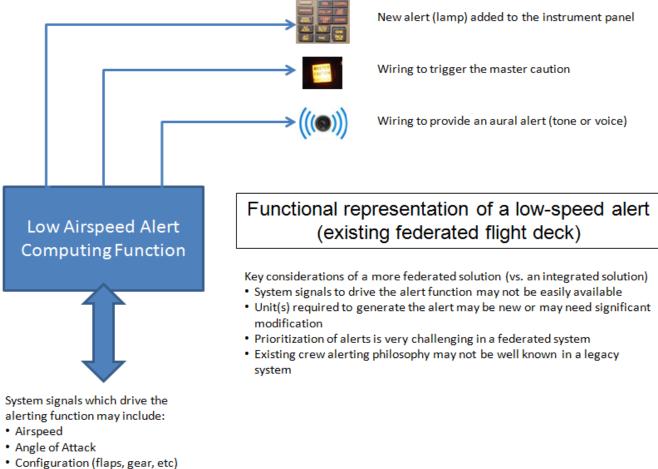
Model Capabilties (Reference survey question in parentheses)		implemented on	ng functionality you have described above all airplanes or only some through a STC or later add-on?	3-12. If you selected some through option, STC or later add-on, please specify the number of airplanes modified versus the number in the fleet:			
	Airplane model (number in fleet at the time of data collection)	AII	Some through option, STC or later add-on	Number of airplanes modified	Number airplanes in the fleet		
Round dial	Boeing DC9 All (433)	No	No				
No SW cptr	Boeing 727 All (826)	No	No				
NO SW CPU	Boeing 747-200 (202)	No	No				
Round dial	Embraer 120 All (126)	Yes	No				
Have SW cptr to	Boeing 737 -300, -400, -500 Conv (1550)	No	No				
support	Boeing MD 80 All (1016)	No	No				
	Boeing MD 90 All (108)	No	No				
	Saab 340 All (197)	No	No				
	Boeing 737 -300, -400, -500 EADI F/S (190)	Yes	No				
	Boeing 757-200 EADI F/S (971)	Yes	No				
	Boeing 767-200, -300 EADI F/S (880)	Yes	No				
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No				
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	No				
(1-1)	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No				
	Bombardier CRJ-100, -200, -400, -440 (718)	Yes	No				
	Bombardier CRJ -700, -701, -702 (215)	Yes	No				
	Bombardier CRJ -705, -900 (105 -900)	Yes	No				
	Embraer 135 All (137)	Yes	No		4		
	Embraer 140 All (74)	Yes	No	S			
	Embraer 145 All (503)	Yes	No	3			
	Embraer 170 All (76)	Yes	No		2		
PFD with visual	Embraer 175 All (54)	Yes	No	2	<u>z</u> ,		
low airspeed alert	Embraer 190 All (51)	Yes	No		21 C		
indication only.	Boeing 717 All (155)	Yes	No	*	1		
(3-1)	Boeing MD10/11 All (257)	Yes	No	×	50 C		
	Airbus A300-600 or A310 All (A300-158; A310-70)	Yes	No				
Visual & aural low airspeed alert	Boeing 737 -600, -700, -800, -900 (3908)	No	Yes		400 3700		
	Boeing 747-400 (675)	No	Yes		350 680		
	Boeing 767-400 (38)	Yes	No				
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	No				
	Airbus A318/319/320/321/330/340/380 All (767) (318-321724; 330-38043)	Yes	No				

Model Capabilities (Reference survey question in parantheses)	Airplane model (number in fleet at the time of data collection)	4.1. What protection/limiting functionality is available to automatically assist the pilot for low airspeed conditions, prior to stall warning?								
		Autothrottle "wakeup"/aut omatic thrust activation	Stick pusher	Automatic pitch control	Increased column/stick forces	Angle of attack protection	Auto-slar extension	Angle of attack limit	Other	Other Text
Round dial	Boeing DC9 All (433)	Na	No	No	No.	No	No	No	No	
No SW cptr	Boeing 727 All (826)	No	No	No	No	No	No	No	No	
no sw cpo	Boeing 747-200 (202)	Na	Na	No	No	Na	No	No	No	
Decent all all	Embraer 120 All (126)	No	No	No	No	140	No	No	Yes	Stick Shaker
Round dial	Boeing 737 -300, -400, -500 Conv (1550)	No	No	No	No	No	Yes	No	No	
Have SW optr to support	Boeing MD 80 All (1916)	No	No	No	No	No	Yes	No	Ne	
mappear	Boeing MD 90 All (108)	No	No	No	No	No	Yes	No	No	
	Saab 340 All (197)	No	No	No	No	No	140	No	No	
	Boeing 737 .300, .400, .500 EADI F/S (190)	No	Na	No	No	No	Yes	No	No	
	Boeing 757-200 EADI F/S (971)	No	No	No	No	Nio	Yes	No	No	
	Boeing 767-200, 300 EADI F/S (880)	No	No	No	No	No	Yes	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	Na	No	No	No	No	Yes	No	No	
capability.	Boeing 757-200, -300 EADI Spd Tape (37 -300 only)	No	Na	No	No	No	Yes	140	No	
(1.1)	Boeing 767 -200, -300 EADI Spd Tape (46)	140	No	No	No	No	Yes	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	No	Na	No	No	No	No	No	No	
	Bombardler CRJ -700, -701, -702 (215)	No	No	No	No	No	No	No	No	
	Bombardier CRJ -705, -900 (105 -900)	No	No	No	No	No	No	No	No	
	Embraer 135 All (137)	No	No	No	No	No	No	No	Yes	
	Embraer 140 All (74)	filo	No	No	No	No	140	No	Yes	
	Embraer 145 All (503)	No	No	No	No	No	No	No	Ves	Slick Shaker
120100000000000000000000000000000000000	Embraer 170 All (76)	No	140	No	No	Na	Na	Yes	No	
PFD with visual	Embraer 175 All (54)	No	No	No	No	No.	No No	Yes	No	
low airspeed alert	Embraer 190 All (51) Boeing 717 All (155)	Na. Yes	No	No	No	No	No	Yes No	No	
indication only.	Boeing MD10/11 All (257)	Yes	No	No	1540	Na	Yes	this	No	
(3-1)	Airbus A300-600 or A310 All (A300-158; A310-70)	Yas	Na	Yes	Yes	teo	No	No	Yes	alphalock function stat retraction inhibition (confl to confl) with "blue" discrete tamp on frontdesk
Visual & aural low	Boeing 737 -600, -700, 800, -900 (3908) Boeing 747 -400 (675)	Yes Yeo	No.	No No	No	No No	Yes	No No	No No	
airspeed alort	Boeing 767-400 (38)	No	No	No	No	No	Yes	No	No	
indication. (3-1, 3-2)	Boeing 777 All (981)	Yes	Na	No	Yes	Na	Yes	No	No	
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 338-380-43)	Yes	Να	Yes	No	Yes	Yes	tio.	Yes	alphalock function slat retraction inhibition to go from conf1 to confi clean

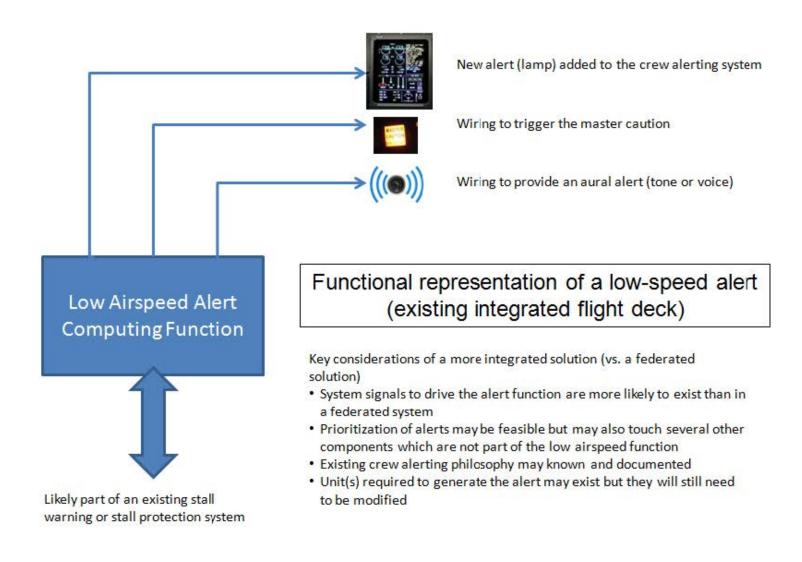
Model Capabilities (Reference survey question in parantheses)		4-2. What protection/limiting functionality is available to automatically assist the pilot for approach to stall conditions, at/after stall warning?							
		Stick pusher	Automatic pitch control	Increased column/stick forces	Angle of attack protection	Auto-siat extension	Angle of attack limit	Other	Other Text
Round dial	Boeing DC9 All (433)	No	No	No	No	No	Na	No	
No SW cptr	Boeing 727 All (826)	No	No	No	No	No	No	No	
но этера	Boeing 747-200 (202)	Na	No	No	No	No	No	Na	
Round dial	Embraer 120 All (126)	Yes	No	No	140	No	140	No	
Have SW optr to		No		No	No	No	140	No	
support	Boeing MD 80 All (1915)	Yes	No	No	No:	110	No	No	The second secon
mapport	Boeing MD 90 All (108)	Yes	No	No	No	No	No	Yes	Pylon tap provides additional nose down capability at full column forward
	Saab 340 All (197)	Yes	No	No	No	No	No	Na	
	Boeing 737 -300, -400, -500 EADI F/S (190)	No	the second s	No	No	No	No	Na	
		No	No	Yes	No	No	No	No	
	Boeing 767-200, 300 EADI F/S (880)	No	No	Yes	No	No	No	No	
Have EFIS & alert	Boeing 737 -300, -400, -500 EADI Spd Tape (58)	No	No	No	No	No	No	Na	
capability.	provide the second strategy and so the second strategy and the second strategy and s	No	No	V	No	No	No	No	
(1.1)		102		Yes	100 C	1.00	1 1111	100	
in march	Boeing 767 -200, -300 EADI Spd Tape (46)	No	No	Yes	No	No	No	No	
	Bombardier CRJ-100, -200, -400, -440 (718)	Yes	No	No	No	No	No	Yes	The second s
	Bombardier CRJ -700, -701, -702 (215)	Yes		No	No	No	No	Yes	AP disconnect at shaker/stall warning
	Bombardier CRJ -705, -900 (105 -900)	Yes		No	No	No	No	Yes	
		Yes		No	No	No	140	fig	
	Embraer 140 All (74)	Yes	No No	No	No	No No	140	140	
	Embraer 145 All (503) Embraer 170 All (76)	Yes	No	No	No	No	No. Yes	Na	
PED with visual	Embraer 175 All (54)	No	No	No	No	No	Yes	No	Pitch trim up inhibition
low airspeed alert		tig	No	No	No	No	Yes	No	Price one of everynees
indication only.	Boeing 717 All (155)	Yes	No	No	No	No	No	No	
(3-5)	Boeing MD10/11 All (257)	No	No	No	No	No	No	tia .	
(2-3)	Airbus A300-600 or A310 All (A300-158: A310-70)	Na	No	No	No	Na	No	Na	
Visual & aural low airspeed alert indication. (3-1, 3-2)	Boeing 737-600, -700, -800, -900 (3908)	No	No	Yes	No. Y	No	No	No	
	Boeing 737-600, -700, 400, 300 (1908) Boeing 747-400 (675)	No	No	No	No	No	No	No	
	Boeing 767-400 (38)	No	No	Yes	No	No	No	No	
	Boeing 777 All (981)	No	No	Yes	No	No	No	No	
	Airbus A318/319/320/321/330/340/380 All (767) (318-321-724; 330-380-43)	No	No	No	No	Na	Νσ	No	

Appendix C- Example Implementations

This appendix illustrates two possible functional implementations – one for a federated configuration and one for an integrated configuration. These are representative functional examples and will vary between aircraft types.



Icing on/off





Mr. Dan Elwell
Chair, Aviation Rulemaking
Advisory Committee
Airlines for America
1301 Pennsylvania Ave, NW, Suite 1100
Washington, DC 20004

800 Independence Ave., SW. Washington, DC 20591

AUG 1 1 2014

Dear Mr. Elwell:

The FAA received the Low Airspeed Alerting Phase 2 Task Report from ARAC in March 2013. This report was developed by the Avionics Systems Harmonization Working Group (ASHWG) in response to a 2011 tasking from the FAA, which asked for industry information to support a potential requirement of low airspeed alerting in all airplanes operating under 14 CFR parts 121 and 129. The Phase II Final Report does not contain an assessment of the potential cost and benefits of implementation of such systems, as the ASHWG was awaiting completion of other studies on this subject. Those studies are now complete and their results should be available to the ASHWG soon.

Since receipt of the report, the FAA has internally evaluated several potential options for proposed alerting systems, based on our estimates of their cost and predicted effectiveness at preventing future loss-of-control accidents resulting from unobserved airspeed loss and stall. Per the FAA Office of Rulemaking Committee Manual, Part 3, section 2.4, the FAA may seek additional clarification from an ARAC working group on work related to a completed tasking through the ARAC Chair.

The FAA is requesting ARAC to reconvene the ASHWG to review the FAA's evaluation of the systems and the additional information now available from other concluded studies on this subject. Specifically, we propose to meet with available members of the ASHWG, present the options we have evaluated, and gather additional information as to their suitability for addressing the hazard of unobserved airspeed decay. The ASHWG should provide recommendations based on its review and plan to discuss the recommendations during the September 2014 ARAC meeting. The FAA would like to note that it has not initiated rulemaking on this matter; therefore, this discussion should be considered a continuation of the FAA's previous tasking on this topic.

Sincerely,

Designated Federal Officer

1103 1 1 AVA