

# Leading Edge Protective Coating Against Fluid and Particulate Erosion for Turbofan Blades

**Presented to:** FAA Office of Environment and Energy

**By:** Delta TechOps (DTO)  
MDS Coating Technologies (MCT)  
America's Phenix, Inc. (AP)

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# LE Protective Coating Against Fluid and Particulate Erosion for Turbofan Blades



## Benefits:

Based on 1% fuel savings for Mainline and Regional commercial carriers:

- Fuel savings between 80M to 100M gal per year
- 750M to 1.0B kg CO<sub>2</sub> / year
- 700M to 1.0B g NO<sub>x</sub> / year

## Risk

- Potential fatigue debit impact of coating Ti blades
- Insufficient coating durability

## Mitigation

- Adjust coating process parameters
- Test & compare to eroded blades in operation

## Objectives:

- Quantify performance degradation
- Optimize coating protection via component tests
- Demonstrate coating protection on operational a/c

## Work Statement:

- Conduct engine tests on degraded & O/H<sup>1</sup> blades
- Conduct CFD analysis on degraded & O/H<sup>1</sup> blades
- Conduct fluid erosion tests at AFRL<sup>2</sup> SuRE<sup>3</sup> rig
- Flight certify optimal coating candidate
- Conduct flight service evaluation

<sup>1</sup> Overhaul <sup>2</sup> Air Force Research Lab <sup>3</sup> Supersonic Rain Erosion

## Accomplishments / Milestones

- Engine test completed with used and new blades
- CFD<sup>4</sup> model completed on used blade
- SuRE test successfully completed
- Flight certification tests completed

## Schedule:

- Blade Condition / Operational Analysis – **COMPLETE**
- JT8D Fluid Erosion Component Test – **COMPLETE**
- Other Engine Types Fluid Erosion Test – **COMPLETE**
- Flight Certification – **COMPLETE**
- Flight Service Evaluation – **April 2018 to Dec 2020**

<sup>4</sup> Computational Fluid Dynamics

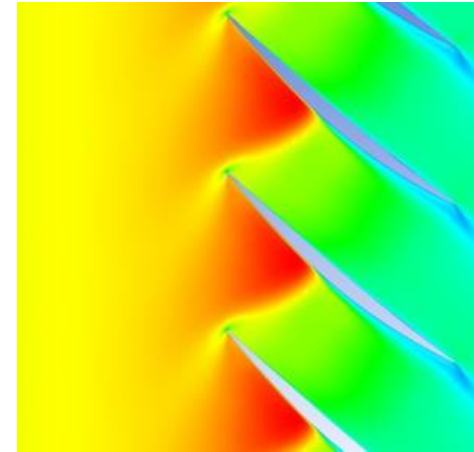
# Phase I – Data, Test & Simulate



Blade Condition Analysis



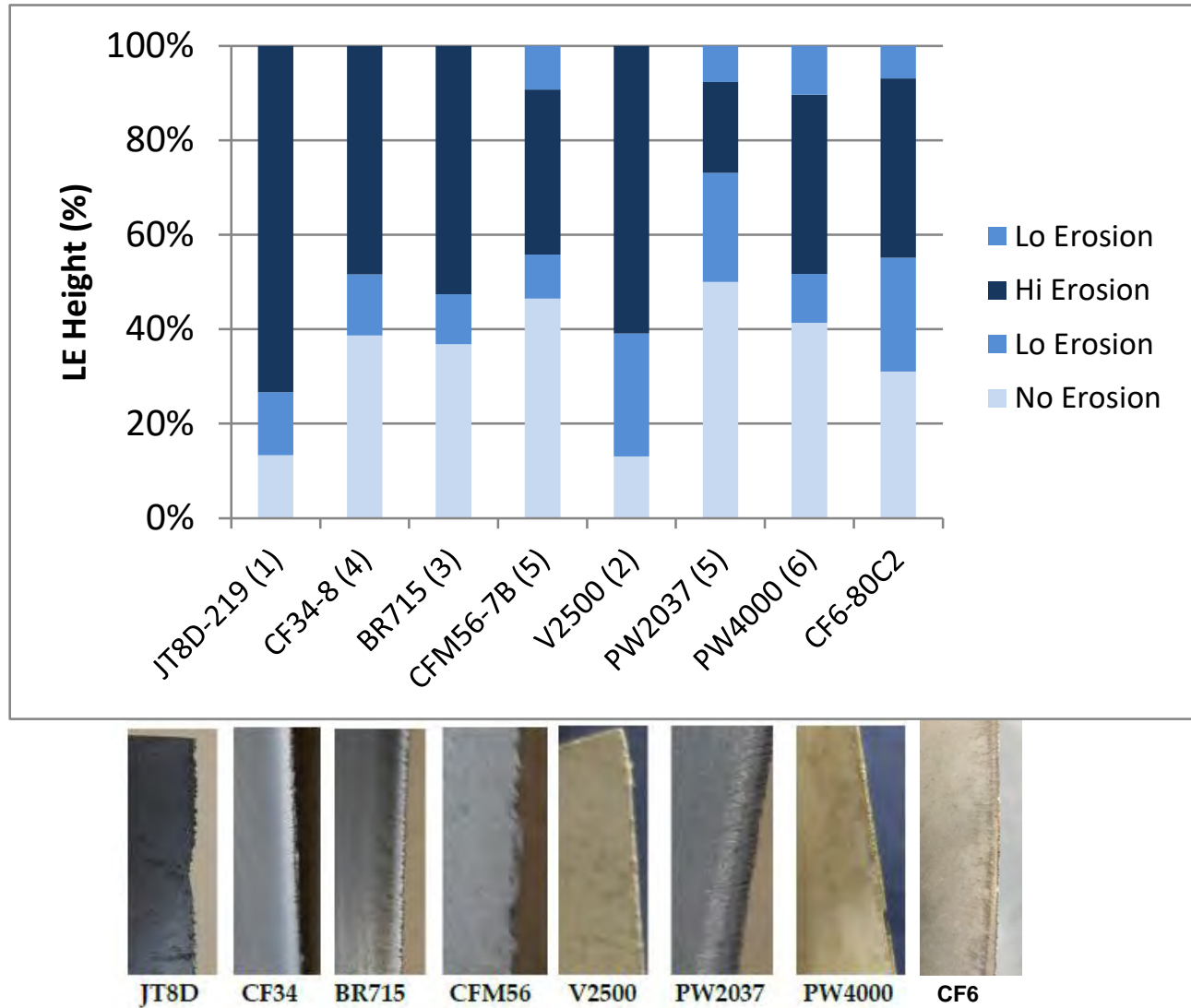
Engine Test Data



CFD Analysis

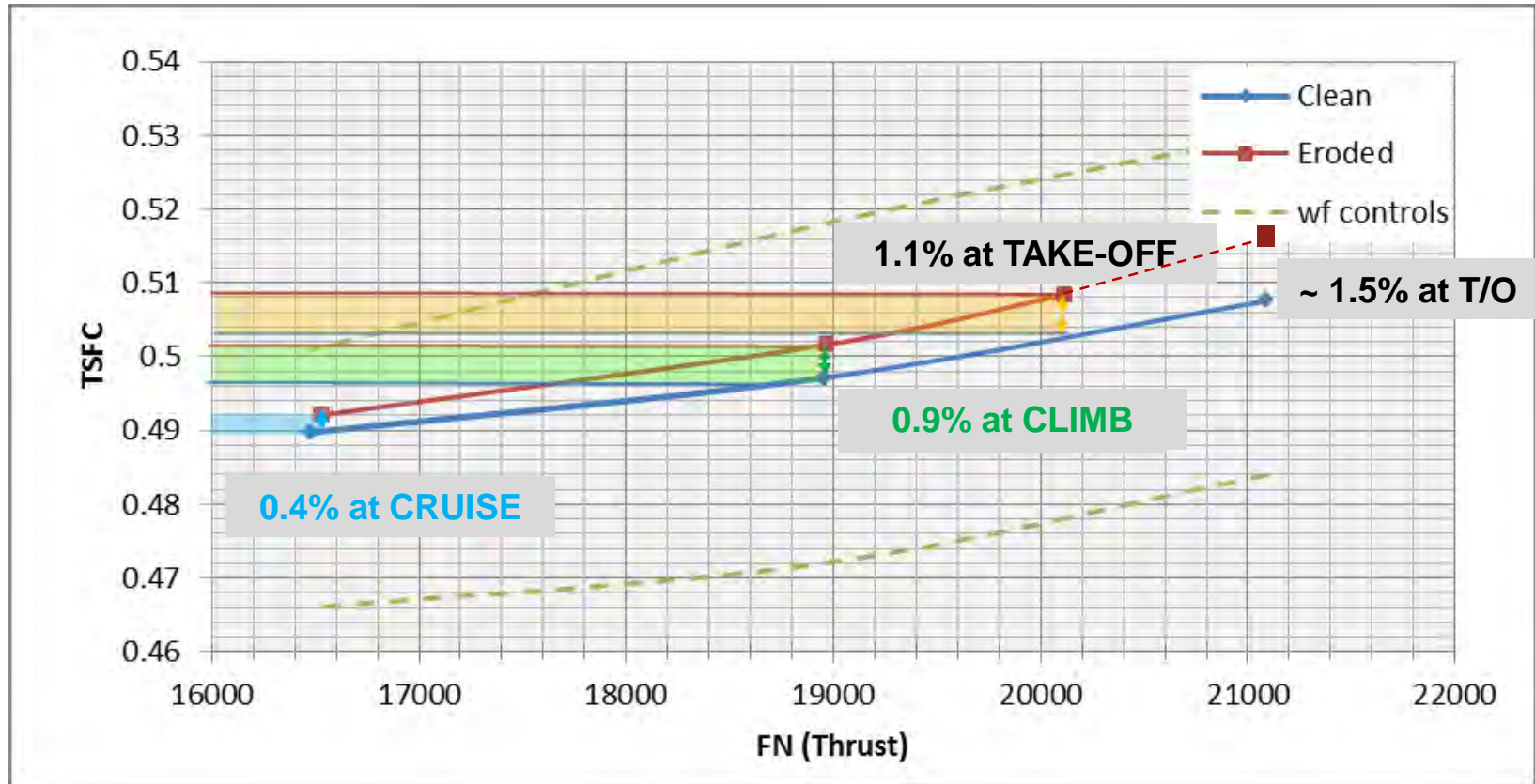
- Following 1<sup>st</sup> stage fan blades inspected and analyzed:
  - JT8D
  - BR715
  - CFM56
  - PW4000
  - V2500
  - CF34
  - PW2037
  - CF6
- Engine test completed on inducted JT8D engine with:
  - existing 1<sup>st</sup> stage fan blades
  - serviceable condition 1<sup>st</sup> stage fan blades
- CFD Analysis completed on serviceable and used blades at following conditions:
  - Take-off
  - Cruise

# Blade Condition Analysis



# Phase I – Engine Test

## Thrust Specific Fuel Consumption (TSFC) Comparison Eroded vs. Serviceable Fan blades



— Clean = repaired, serviceable fan blades

[JT8D Engine S/N 726044](#)

Inducted May 2015

TSO = 3126 hours

Tests / Engine Service Manual:  
PN773128



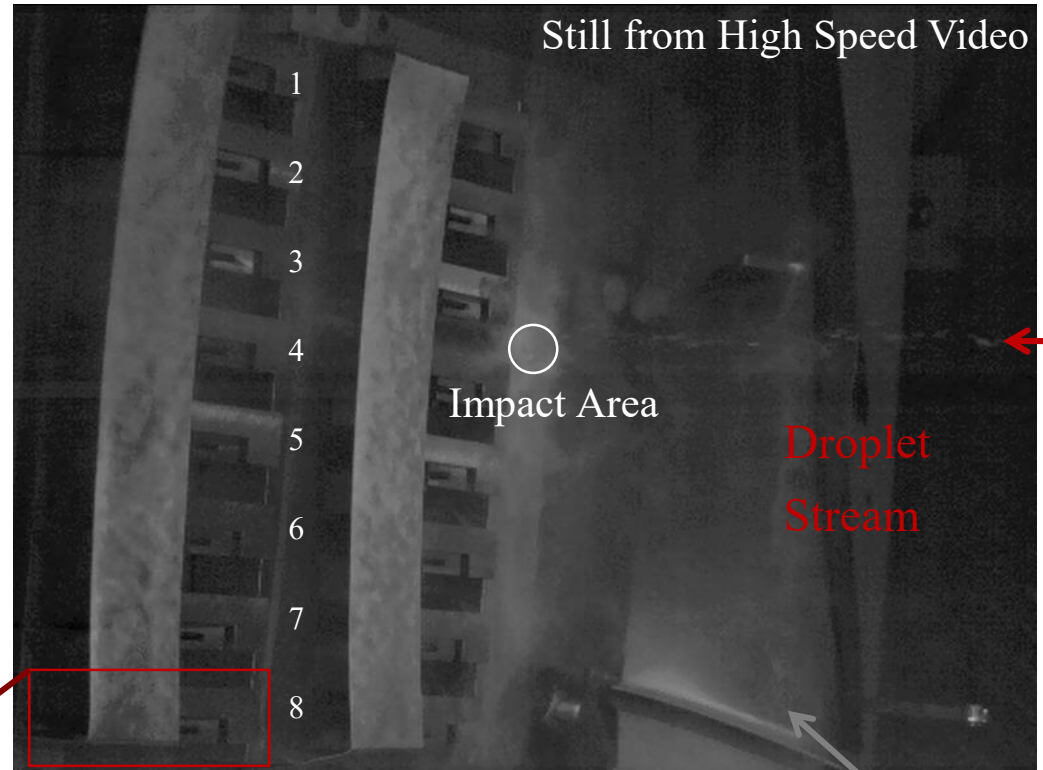
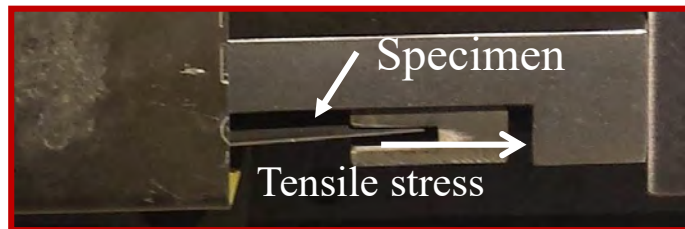
# Phase II – Fluid Erosion Test

## @ AFRL – Supersonic Rain Erosion (SuRE) Rig

### Specimen Preparation



### Specimen Tooling



**BlackGold®** Coated  
Blade Specimen

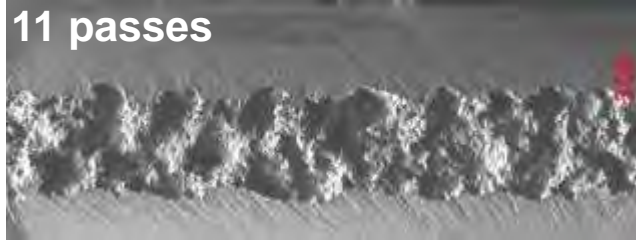
Droplet  
Stream

Coated Blade

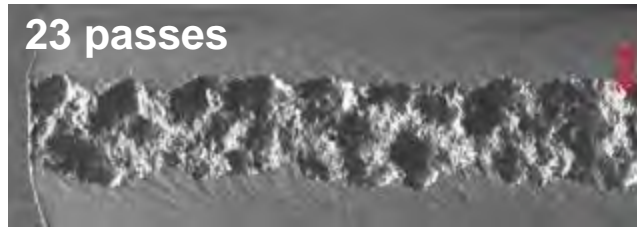
# Fluid Erosion Tests @ AFRL

## Phase IIA - February 2017 Tests

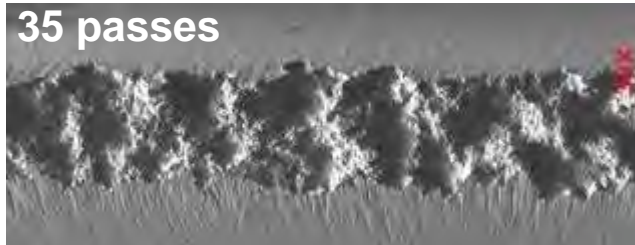
**Uncoated PW2000**



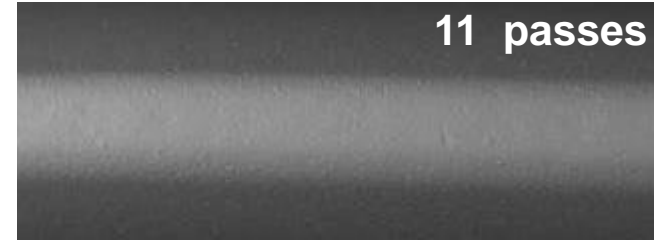
23 passes



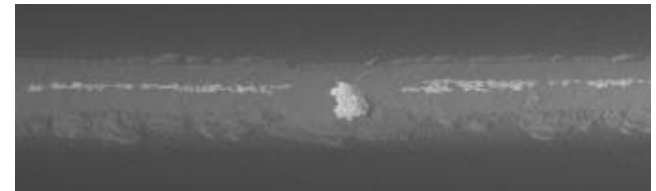
35 passes



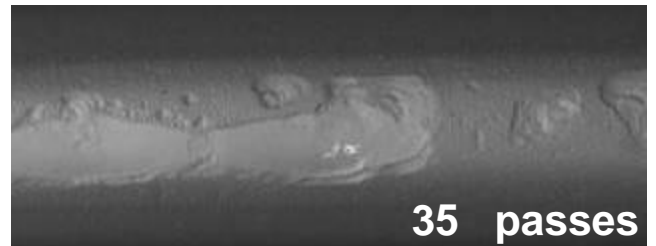
**BlackGold® Coated PW2000**



23 passes



35 passes



*First substrate damage noticed after 23 passes*

# Phase III – Air Worthiness Certification Tasks

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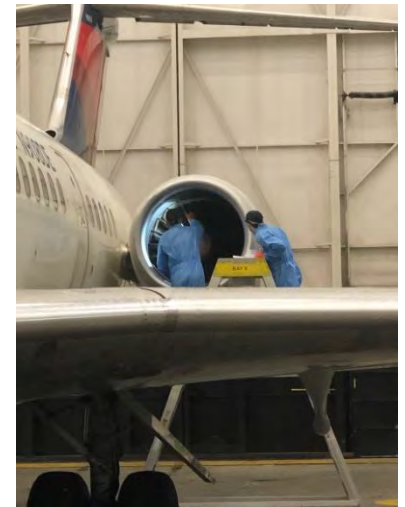
- FAA Certification Plan Approval
- FAA Test Plan Approval
- Weight Analysis
- Metallographic Analysis
- Stress Analysis
- Frequency Analysis
- High Cycle Fatigue Tests
- Mechanical Property Tests
- Impact (Jelly Ball) Tests
- Ice Adhesion Analysis
- Compressor Wash Analysis



# Phase IV – Flight Demonstration

## Flight Demonstration, Field Engine Op Status

### Field Engine Op Status



No.	Engine S/N	Status*	Notes
1	718045	2,378 hrs	4 Coated Blades. Currently off-wing awaiting installation. Estimated date early March 2020.
2	725536	2,245 hrs	4 Coated Blades. Removed, in-storage in Birmingham. Not expected to be re-installed. Scheduled for final inspection.
3	725558	284 hrs	4 Coated blades removed. Engine experienced non-coating related in-flight shutdown.
4	No S/N ID yet	0 hrs	2 Coated blades. Engine awaiting on-wing installation. Expected installation late Jan 2020.**
5	718150	1,489 hrs	2 Coated blades installed and flying**

\* As of 18 November 2019

\*\* Engines 4 and 5 have only two (2) coated blades in order to maintain four (4) flight demo aircraft

# Phase IV – Flight Demonstration

## Phase IV – Flight Demonstration Fan Blade Inspections

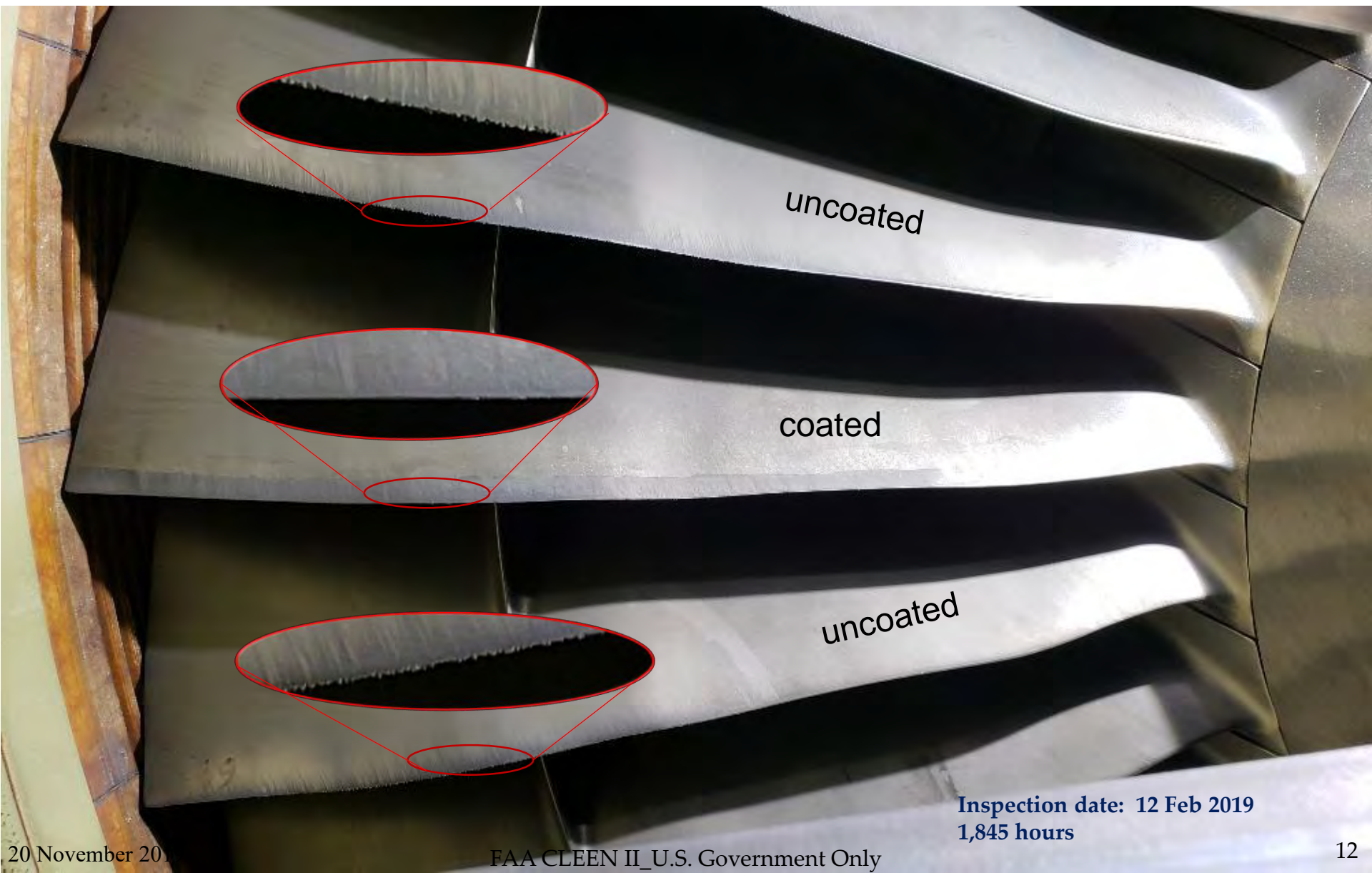
- **BlackGold®** coated 1<sup>st</sup> stage turbofan blades installed on four (4) JT8D engines for flight operations on MD88 aircraft
  - Inspections  $\approx$  every 250 to 500 hours

### JT8D 1<sup>st</sup> Stage Fan Blades



# Phase IV – Flight Demonstration

Flight Demonstration, Field Engine Op Status



Inspection date: 12 Feb 2019  
1,845 hours

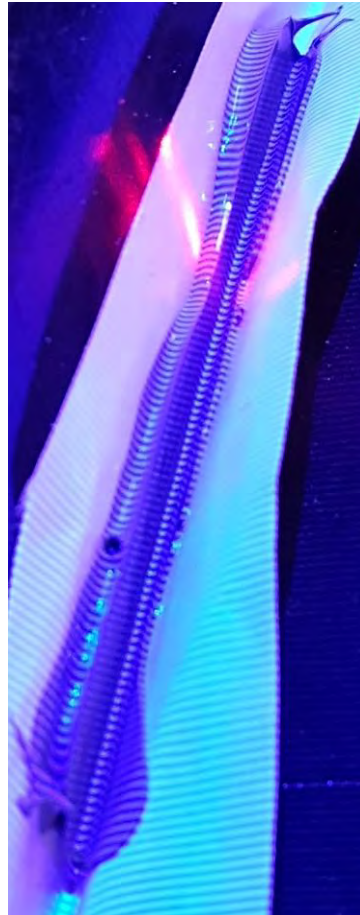


# Phase IV – Flight Demonstration

## 3D Scans of “Dental Mold” LE Blade Specimens



JT8D Fan Blades with Molds

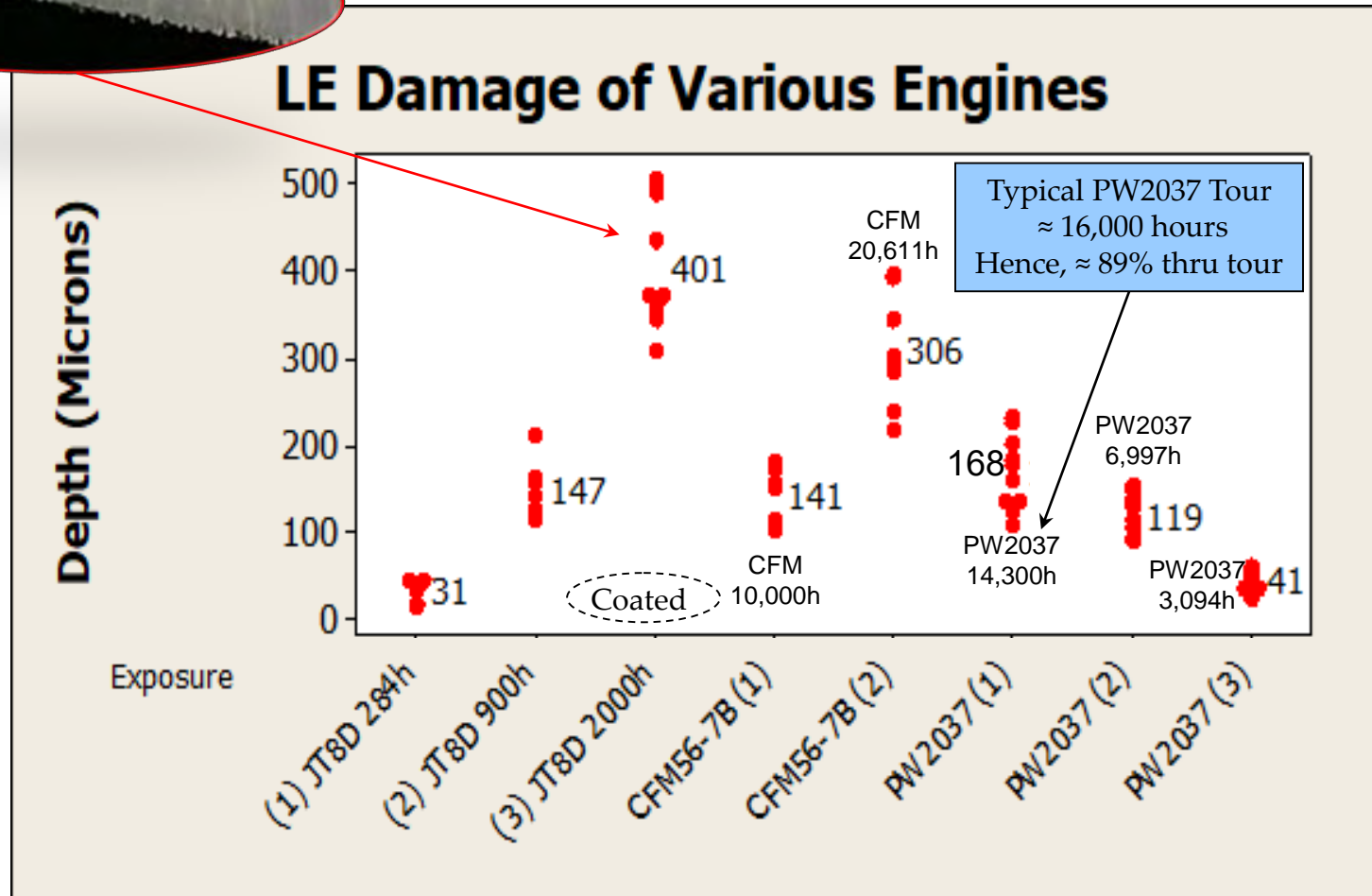


3D mold scans

- LE molds placed in erosion area from tip towards root.
- Molds scanned with white-light 3D scanner.
- Damage depth measured along LE from scanned image and processed with appropriate software.

# LE Depth Measurements

Uncoated @ 1845h





# Phase IV Schedule

## Estimated remaining inspection schedule

- **Engine # 1 @ 2,378 op hours. Status – on-wing by March 2020**
  - 2,500 + hours by April 2020
  - 3,000 + hours by July 2020
  - 4,000 + hours by December 2020
- **Engine # 4 @ 0 hours. Status – awaiting installation Jan 2020.**
  - 500 + hours by March 2020
  - 1,000 + hours by May 2020
  - 2,000 + hours by October 2020
  - 2,500+ hours by end-of-Dec 2020
- **Engine # 5 @ 1,489 op hours. Status – currently flying.**
  - 1,700 + hours by January 2020
  - 2,000 + hours by March 2020
  - 3,000 + hours by July 2020
  - 4,000 hours by end-of-Dec 2020
- **Engine # 2 – completed flying @ 2,245 hours**
- **Engine # 3 – completed flying @ 284 hours**

**Total Op Hours as of 18 Nov 2019  
= 6,396 hours**

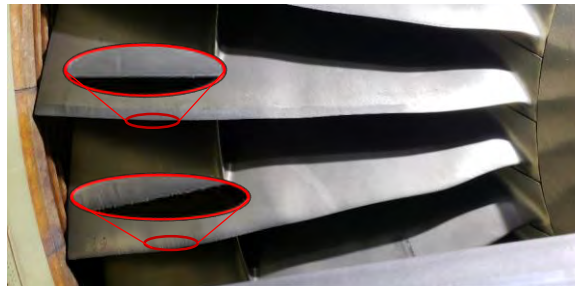
**Delta's MD-88 fleet scheduled for sunset at the  
end of CY20**

**Estimate > 12,000 total op hours by end of CY20**

**Estimates based on 50 hours / week average**

# CLEEN II Program Summary

- **Blade Condition and Operational Analysis complete:**
  - LE erosion documented
  - Engine test confirmed 1.1%+ TSFC increase
- **Coating Component Level Tests complete:**
  - For JT8D, PW2037, CF6 and Ti strips
- **Flight Certification Tests Complete**
- **Flight Demonstration Engines**
  - Over 5,000 op hours
  - Visual and measured results confirms coating protecting LE
  - Project over 12,000 op hours by end of CY20
- **Installing up to four (4) completely coated engines to track fuel savings benefits**





AMERICA'S PHENIX



# THANK YOU



Federal Aviation  
Administration

