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# Predicting Accident Rates From General Aviation Pilot Total Flight Hours

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**Final Report** 

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# PREDICTING ACCIDENT RATES FROM GENERAL AVIATION PILOT TOTAL FLIGHT HOURS

Is there a range of pilot flight hours over which general aviation (GA) pilots are at greatest risk? More broadly, can we predict accident rates, given a pilot's total flight hours (TFH)? Many GA research studies implicitly assume that accident rates are a linear function of TFH when, in fact, that relation appears nonlinear. This work explores the ability of a nonlinear gamma-based function ( $\Gamma_{mut}$ ) to predict GA accident rates from noisy TFH data. Two sets of National Transportation Safety Board (NTSB)/Federal Aviation Administration (FAA) data, parsed by pilot instrument rating, produced weighted goodness-of-fit ( $R^2_{w}$ ) estimates of .654 and .775 for non-instrument-rated (non-IR) and instrument-rated pilots (IR), respectively. This model class would be useful in direct prediction of GA accident rates, and as a statistical covariate to factor in flight risk during other types of modeling. Applied to FAA data, these models show that the range for relatively high risk may be far broader than first imagined, and may extend well beyond the 2,000-hour mark before leveling off to a baseline rate.

# INTRODUCTION

Total flight hours has long been considered one of the risk factors in aviation, and is often used to represent either pilot flight experience or cumulative risk exposure (e.g., Dionne, Gagné & Vanasse, 1992; Guohua, Baker, Grabowski, Qiang, McCarthy & Rebok, 2011; Mills, 2005; Sherman, 1997). TFH has served as both an independent variable in its own right, as well as a statistical covariate, to control for the effects of experience or risk.

Investigators have often unwittingly assumed a linear relation between TFH and accident frequency and/or rate, that is, a straight-line prediction function  $\hat{y}=a+bx$ , with *a* as *y*-intercept and *b* as slope. However, evidence is emerging that such relations are actually nonlinear. For instance, Bazargan & Guzhva (2007) reported that the logarithmic transform *log(TFH)* significantly predicted GA fatalities in a logistic regression model. More recently, Knecht & Smith (2012) reported that a risk covariate starting with *log(TFH)*, followed by a gamma transform, significantly predicted GA fatalities in a log-linear model.

In his 2001 book, *The Killing Zone*, Paul Craig presented early evidence that GA pilot fatalities might relate nonlinearly to TFH. Craig showed that fatalities occur most frequently at a middle range of TFH ( $\approx$ 50-350), and hypothesized that this band of time may be one in which pilots are at greatest risk due to overconfidence at having mastered flying the aircraft, combined with lack of actual experience and skill in dealing with rare, challenging events. He supported this hypothesis with histograms of GA accident frequencies, although not with a formal model. In group after group, a similar pattern emerged in his data, one of a "skewed camel hump" with a long tail at higher TFH.

Following Craig's findings, I investigated a nonlinear model for accident frequency counts (Knecht, 2012). Derived from NTSB and FAA data, Figure 1 illustrates this basic class of relations between TFH and fatal accident counts for a group of 831 U.S. instrument-rated GA pilots from 1983-2011.

The "skewed camel-hump" relation appears as we focus on the range TFH<5000. It appears quite durable, persisting whether we break out different data categories (e.g., IR vs. non-IR or

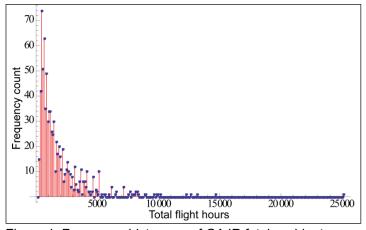


Figure 1. Frequency histogram of GA IR fatal accident counts (y-axis) as a function of total flight hours (x-axis, bin width=100 TFH).

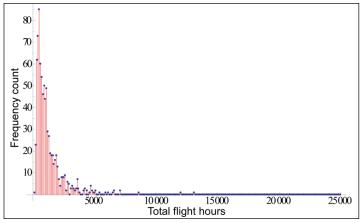


Figure 2. Frequency histogram of GA IR SERIOUS + FATAL accident counts as a function of TFH (bin width=100 TFH).

serious vs. fatal accidents) or even combine them. Figure 2 shows GA accident data for 832 pilots from 2003-2007 for NTSB SERIOUS+FATAL accident categories that were combined to boost frequency counts.

So, frequency count histograms do support the notion of a nonlinear "killing zone" somewhere in the lower range of TFH.

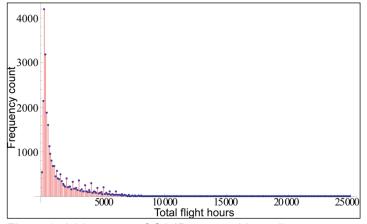


Figure 3. A histogram of GA IR non-accident pilot counts from the same data cohort as Fig. 2, showing how numbers of pilots decrease rapidly as TFH increase (bin width=100).

And, we have successfully modeled that relation with a gamma probability density function (pdf).

But, this raises a deeper question: How can we be certain that this "relation" is not an artifact, merely a side-effect of the fact that there are simply fewer pilots at higher TFH, hence fewer accidents? Figure 3 shows how frequency counts for a matched sample of *non*-accident pilots also drop off markedly as TFH increase.

Figures 2 and 3 look suspiciously similar. In fact, the Pearson correlation between these datasets is .845 (p<.001). This implies that *most* ( $r^2 \approx .71$ ) of the "killing zone" seen in accident frequency counts can be explained simply by the fact that the numbers of pilots tend to decrease as TFH increases.

With that in mind, what support remains for the hypothesis of an actual killing zone? And, if it exists, what shape does it truly assume?

We can address these questions by analyzing accident *rate* histograms, in which the  $i^{th}$  data bin is

$$rate_{i} = \frac{accidents_{i}}{accidents_{i} + nonaccidents_{i}}$$
(1)

Rates will control for the fact that our frequency counts decrease as TFH increases, and uniformly sized bins will control for the effect of exposure. Therefore, if a killing zone continues to persist with binned rates, then the empirical evidence for it would be much stronger.

# The Problem With Rates

Unfortunately, analyzing accident rates is easier said than done. Rate data can be quite noisy, as Figure 4 depicts.

This "noise" represents sampling error—random fluctuations in rate from one bin to the next. It is a vexing problem at higher TFH where Eq. 1's denominator is small compared to its numerator, and even small changes in frequency count can cause substantial variations in accident rate. Despite having tens of thousands of cases in the "hump," bins in the high-TFH tail may, by pure chance, sometimes consist of one or two individuals, resulting in alarmingly high-rate bins right next to zero-rate bins.

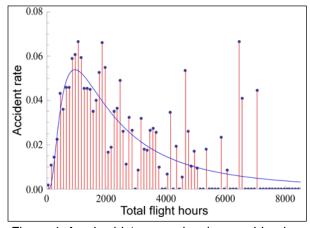


Figure 4. A noisy histogram showing combined "serious"+"fatal" accident rates for all GA pilots as a function of TFH (bin width=100).<sup>1</sup>

It is not hard to appreciate how useful a modeling function could be here. A good model would smooth the noise in the data, allowing investigators to better predict rates, given TFH. This would be directly useful in areas such as determination of insurance premiums, allocation of resources for pilot training, accident investigation staffing, and public relations. To safety researchers, it would be useful as an improved statistical covariate, to either factor in or out risk as a function of flight experience. This would have broad application in numerous types of aviation research.

With this in mind, the present work moves from fitting GA accident frequency *count* data to fitting accident *rates* as a function of GA pilot TFH. The calculation of binned accident rates is a key step in risk analysis because uniformly sized bins will control for risk exposure while rates control for the actual number of individuals within a given data bin.

### **METHOD**

#### The Modeling Function

Ideally, modeling functions should be motivated by theory involving causal processes inherent to the data. Unfortunately, in aviation accidents, causes tend to be numerous and hard to disentangle, making theory-based modeling difficult.

We therefore proceed with the less ambitious goal of simply trying to fit a well-behaved, continuous mathematical function to empirical GA accident rate data. Standard techniques are used, namely minimization of least-squares residuals between a modeling function and empirical data.

The model itself begins with a versatile probability density function  $\Gamma_{pdf}$  (Spanier & Oldham, 1987). Its *x*-axis represents TFH. The *y*-axis can represent either frequency count or proportion (here, we will focus on proportions). The model contains a *shape* parameter,  $\alpha > 0$ , and a *scale* parameter,  $\beta > 0$ .

$$\Gamma_{pdf,c}(x;\alpha,\beta) = \frac{e^{-x/\beta} x^{\alpha-1} \beta^{-\alpha}}{\Gamma(\alpha)}$$
(2)

<sup>&</sup>lt;sup>1</sup> Data from Knecht & Smith (2012).

The gamma function  $\Gamma(\alpha)$  itself is described as the Euler integral, defined for  $\alpha > 0$ 

$$\Gamma(\alpha) = \int_{0}^{\infty} t^{\alpha - 1} e^{-t} dt$$
(3)

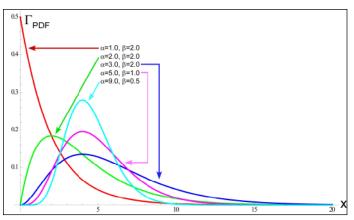


Figure 5.  $\Gamma_{ndf}$  with various values of  $\alpha$  and  $\beta$ .

Shown in Figure 5,  $\Gamma_{pdf}$  has the attractive feature of being able to represent overdispersion (variance>mean) or underdispersion (variance<mean), as well as having calculable confidence intervals around the function itself.

Gamma pdfs have been used to model a wide variety of processes, including the size of insurance claims (Hogg & Klugman, 1984), amounts of rainfall (Chiew, Srikanthan, Frost, & Payne, 2005), waiting times and mean-time-to failure (Winkelman, 2008), where it represents time until the  $a^{th}$  event in a constanthazard model, and distributions of microburst wind velocity (Mackey, 1998).

Prior experience with GA accident frequency data shows that this canonical version of  $\Gamma_{pdf}$  is unable to deal with critical features of data such as amplitude, non-zero *x*-axis (TFH) start values, and the long right-hand tail inherent to real-world data. Therefore, a more practical model is proposed (Eq. 4), which we can call  $\Gamma_{nate}$ . This includes an amplitude term *A* (which can "stretch" the entire function up or down on the *y*-axis), a location parameter  $\delta$  (which can shift it left or right), an *x*-axis log-transform (which compresses larger values more than smaller ones), and a base-rate term *b* (which sets the pdf on a thin, long rectangle capable of moving up or down on the *y*-axis), added to reflect the notion that all flights harbor some constant risk, no matter how experienced the pilot.

$$\Gamma_{rate} = b + A \frac{e^{-(\ln(x) - \delta)/\beta} (\ln(x) - \delta)^{\alpha - 1} \beta^{-\alpha}}{\Gamma(\alpha)}$$
(4)

### The Empirical Data

Models need values for their parameters. The current work parameterizes  $\Gamma_{mte}$  on four U.S. GA pilot datasets collected during another project (Knecht & Smith, 2012). Those data span the time period 1/1/2003-8/26/2007 (4.65 yr.), and represent GA pilots licensed after 1995, matched on private-pilot school type and examiner type for the purposes of that particular study.<sup>2</sup> To boost frequency counts, two official categories of injury (Serious+Fatal) are now combined into a single category. The data are parsed by pilot instrument rating (IR vs. non-IR) and accident status (Accident vs. Non-accident). Table 1 shows set sizes.

These accident data come from the NTSB accident database, by way of FAA's Accident Investigation and Prevention Division (AVP-210). Non-accident data come from the FAA Comprehensive Airman Information System (CAIS, pronounced "CASS"), managed by the Flight Standards Service (AFS-760), supplemented with pilot TFH from the FAA Document Imaging Workflow System (DIWS), managed by the Aerospace Medical Certification Division (AAM-300).

Each of Table 1's four datasets is aggregated into *x*-axis bins 100 TFH wide. Experience shows that a bin width of 100 produces a reasonable balance between the number of bins and the population of each, given datasets in Table 1's size range. The bins span a range from 0-32,500 TFH, with each bin's *x*-value centered mid-way into its data range. Student pilots are excluded, so the actual data range for the first bin is 45-100. Consequently,  $x_1=77.5$ ,  $x_2=150$ ,  $x_3=250...x_{i>1}=100i-50$ . Bin accident rates are then calculated by Eq. 1 to produce one dataset for IR pilots and a second for non-IR pilots.

#### **Important Considerations**

First, it is important to normalize  $\Gamma_{nate}$ . This requires some explanation. Accidents (which form our rate numerators in each TFH bin) accumulate over time, whereas the corresponding number of *non*-accident pilots (which forms each bin's

Table 1. Pilot statistics in the four datasets modeled.								
	Accident status			Est.	Ann acc			
Pilot rating	Accs <sup>A</sup>	Non-accs	Totals	Annual TFH <sup>B</sup>	rate per 100 FH			
IR	n <sub>11</sub> = 832	n <sub>12</sub> =27,528	28,360	7,805,950	.00229			
non-IR	<i>n</i> <sub>21</sub> =1,036	<i>n</i> <sub>22</sub> =38,291	39,327	2,630,452	.00847			
Totals	1,868	65,819	67,687	10,436,402	.00385			
ABased on 4.65 years' NTSB accident data.								
<sup>B</sup> Based on 2 times the reported past-6-month FH from DIWS database.								

 $<sup>^2</sup>$  Currently, NTSB accident data beyond 2007 are not readily available to the FAA, for reasons unknown.

denominator) is based on a "snapshot" in time and is therefore considered a constant. Consequently, we should normalize our bin accident rates to reflect a standard unit time such as one year. In that event, each bin's initial rate should be divided by the number of years over which accident data were accumulated (here, 4.65 years) to represent estimated accident rate per year, a.k.a. *annualized rate*.

Second, we need to clearly understand that, unlike accident *frequency* histograms, a distribution of *rates* is not technically a pdf, even though we may try to curve-fit binned rate data with a pdf-like function (e.g., Eq. 4).<sup>3</sup> More specifically, to cumulate accident frequencies over a range of TFH, we would compute a pdf's definite integral over that range. In contrast, to cumulate the *probability of having an accident* over a range of TFH requires a different method (described later in Eq. 12).

# **Parameter Estimation**

The *NonlinearModelFit* function of *Mathematica* 7.0 (Wolfram, 2008) is used to estimate parameters for Eq. 4. For unconstrained parameters, *NonlinearModelFit* offers a range of standard numerical methods (e.g., Newton-Gauss, quasi-Newton, Levenberg-Marquardt). For constrained parameters (the method used here), where starting and/or final parameter values are forced to lie within some range  $p_{min} < p_r < p_{max}$ , the Karush-Kuhn-Tucker (KKT) method was used.

# The Problem of Noise

Noisy data such as Figure 4's pose great difficulty for standard least-squares minimization. Parameters often fail to converge in rugged ratescapes, or may converge to a local minimum rather than a global one.

To confront the noise, we first tried the simplest approach of using wider data bins. Surprisingly, this had almost no effect. Accidents within these particular data tended to bunch up in several adjacent bins, leaving no way to smooth them without resorting to arbitrary "bin widths of convenience," which would negate our attempt to standardize risk exposure by having equal bin widths.

We next tried another standard tactic, namely dropping outliers. Outliers can be defined in various ways, but a rule of thumb is to drop all data points (here, rates) greater than 2.5-3 standard deviations (s) from the group sample mean  $\bar{y}$ 

$$s = \frac{1}{n-1} \sqrt{\sum_{i=1}^{n} (y_i - \overline{y})^2}$$
(5)

This method was tested but abandoned for several reasons. Not only was it based on an (here invalid) assumption of distributional normality, but useful information was destroyed, and the extremely long tail of the fitting function resulted in an *s* so small that we end up dropping very few of the obviously noisy values in the tail that threatened our curve-fit. A third standard approach was next tried, involving datasmoothing using moving averages. Unfortunately, this also had the effect of destroying information by smearing, greatly altering the "camel hump" representing the majority of the data.

A fourth approach was then considered. Simulated annealing (Kirkpatrick, Gelatt, & Vecchi, 1983) would involve adding noise to each parameter's starting value before engaging in the gradient-descent, residual error-minimization process. Then, as that process progressed, we would "cool the noise," allowing parameter estimates to bounce around and out of local minima until a global near-minimum could be found.

Nonetheless, in the case of noisy rate data, even such an elegant approach unfortunately would still leave us one major logical problem: All our data bins—including the noisy—would have equal opportunity to influence the residual sum-of-squares during the data-fit. Any "one bin, one vote" approach would implicitly allow all bins equal influence, no matter how many individuals each bin represented or what that bin's expected variance was. This would confer inordinately large influence to sparsely populated bins, as well as to larger bins having inherently low reliability.

After being frustrated by the problems outlined above, we finally settled on the standard procedure of weighting empty bins by 0 while weighting non-empty bin's accident rate  $r_i$  by the inverse of its sample variance  $(1/s_i^2)$ . This is easily done in *Mathematica* by supplying a vector *Weights* $\rightarrow w$  to *Nonlinear-ModelFit*, where  $w = \{w_1, w_2, \dots, w_i\}$ .

Since our data are rates are proportions,  $0 \le r_i \le 1$ , the *i*<sup>th</sup> bin weight  $w_i$  becomes

$$w_i = \frac{1}{s_i^2} = \frac{n_i - 1}{r_i (1 - r_i) (1 - (n_i / N_i))}$$
(6)

 $n_i$  being the *i*<sup>th</sup> bin's total frequency count (accidents+non-accidents) used to derive  $r_i$ , and  $N_i$  being the number of pilots in the general population with that range of TFH.

In this particular instance, we do not know  $N_i$ , so we shall assume a constant sample ratio  $n/N_i$ , making the term  $1-(n/N_i)$ constant for all bins. Since weights only need to be expressed relative to each other during curve-fitting, we can eliminate this constant term, leading to a functional weight of

$$w_{i} = \frac{n_{i} - 1}{r_{i}(1 - r_{i})}$$
(7)

Unfortunately, Eq. 7 is ill-behaved when  $r_i=0$  or 1, leading to division by zero. This can arise when either TFH bins are too narrow or when TFH is large, making the bin sample size  $n_i$  too small, since few pilots have high TFH. Figure 6 (top) illustrates the problem.

<sup>&</sup>lt;sup>3</sup>The area under a pdf is always constrained to equal 1.0. The area under a distribution of rates is unconstrained.

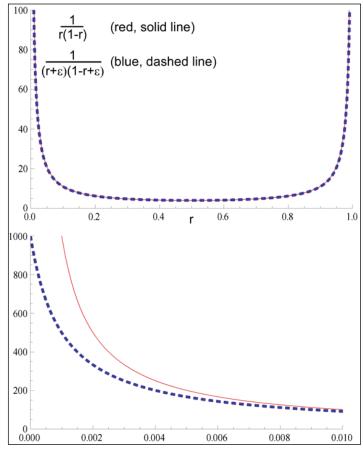


Figure 6. (red, solid lines) With Eq. 7, as  $r \rightarrow 0$  or 1, 1/(r (1-r)) $\rightarrow \infty$ ; (top, dashed, blue line) With  $\varepsilon$  =.001, Eq. 8 gives a good approximation across most of r, from around .006<r<.994., but gracefully self-limits at r=0 and 1.

To address this issue, Eq. 7 is modified slightly by adding a term  $\varepsilon$  designed to constrain behavior at 0 and 1:

As Figure 6 shows, when  $\varepsilon$  is small, Eq. 8 closely approximates Eq. 7 over most of its range but self-limits gracefully at  $r_i=0$  and 1.

#### Parameter Start Values and Evolution

In multidimensional parameter spaces with unknown local minima, the choice of start values can be critical. A hybrid approach was used here. Start values were first selected that roughly approximated the shape of the binned data, namely A=1,  $\alpha=6$ ,  $\beta=0.5$ ,  $\delta=b=0$ . Figure 7 shows the shape of that plot.

To allow parameters to evolve in a manner emulating annealing, *NonlinearModelFit* was set up to use weighted binned data and to run inside a For[i=1,  $\gamma > 1.0$ , i=i+1] loop. Inside the For[] loop, to emulate noise injected at each iteration, the value of each parameter  $p_j$  was multiplied by a random real number drawn from the range  $1.0\pm\gamma$ . with  $\gamma$  starting at 0.02.

During each (*i*<sup>th</sup>) iteration of the For[] loop, *NonlinearModelFit* was itself allowed to run for *i* iterations, after which *i* increased by 1 while  $\gamma$  decreased by a small fixed amount (.0002).  $\gamma$  thus emulated the cooling rate. Eventually,  $\gamma$  fell below 1.0, halting the injection of parameter noise and terminating the For[] loop after one final gradient descent.

Running this method repeatedly on a given dataset, we could broadly sample the parameter space. After inspection

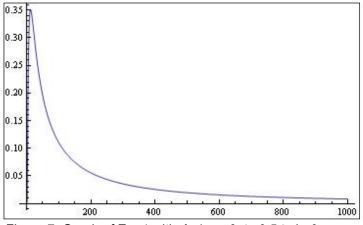


Figure 7. Graph of Eq. 4 with A=1,  $\alpha=6$ ,  $\beta=0.5$ , $\delta=b=0$ .

of the graphic data-fit plots verified a good fit, and multiple runs showed parameters similar to three significant Figures, we could be confident that those parameter values represented a near-optimal data-fit.

### RESULTS

#### Modeling the Accident Rate Histograms

Below are the curve-fits for the empirical data, our two accident rate histograms. Figure 8 represents non-IR pilots, Figure 9, IR pilots.  $\Gamma_{rate}$  models are overlaid and parameter estimates given beneath.

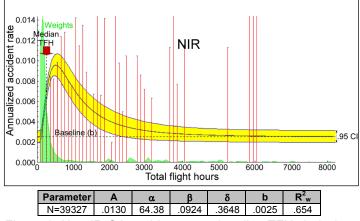


Figure 8. Non-IR GA accident rates (median TFH=250.5).

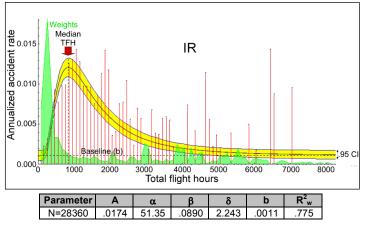


Figure 9. IR GA accident rates (median TFH=823.5).

Each Figure shows accident rate (*y*-axis) plotted by TFH-ofpilot-at-the-time-of-accident (*x*-axis, bin width = 100). Each bin's raw number of accidents (based on 4.65 years' data) was divided by the total number of pilots (accident+non-accident) in the same range of TFH to produce a bin rate for that 100-hr range.<sup>4</sup> The resulting rates were then divided by 4.65 to represent one average year's rates before starting the curve-fitting process. Keep in mind that the height of the y-axis represents the probability of having an accident over a 100-FH span,  $x\pm50$ .

 $\Gamma_{nate}$  is based on Eq. 4 with data weighted by Eq. 8 ( $\epsilon = .001$ ). Relative weights are drawn in green and have been scaled to fit within the Figure. Empty bins ( $n_i=0$ ) were weighted 0 to prevent them from influencing the curve-fit. The yellow band surrounding  $\Gamma_{rate}$  represents its 95% confidence interval (CI).

# **Model Evaluation**

A model's ability to predict empirical data can be expressed by a variety of metrics. One of the simplest measures of goodnessof-fit is the coefficient of determination,  $R^2$ .  $R^2$  varies between 0 and 1 and estimates the proportion of explained variance. The weighted form of  $R^2$  is nominally

$$R_{w}^{2} = 1 - \frac{SS_{w,error}}{SS_{w,total}} = 1 - \frac{\sum_{i=1}^{n} w_{i} (y_{i} - f_{i})^{2}}{\sum_{i=1}^{n} w_{i} (y_{i} - \overline{y}_{w})^{2}}$$
(9)

with weighted means, for instance  $\bar{y}_{w}$ , of the form

$$\overline{y}_{w} = \frac{\sum_{i}^{n} w_{i} y_{i}}{\sum_{i}^{n} w_{i}}$$
(10)

However, Eq. 9 can throw values < 0 with nonlinear data, so we opt for the robust form of  $R^2_{w} = r^2_{w}$ , the square of the weighted correlation coefficient

$$r_{w} = \frac{\sum_{i}^{n} w_{i}(x_{i} - \bar{x}_{w})(y_{i} - \bar{y}_{w})}{\sqrt{\left(\sum_{i}^{n} w_{i}(x_{i} - \bar{x}_{w})^{2}\right)\left(\sum_{i}^{n} w_{i}(y_{i} - \bar{y}_{w})^{2}\right)}}$$
(11)

With these data, Eq. 11 yields  $r^2_{w-non-IR} = .654$  and  $r^2_{w-IR} = .775$ , both in the "moderate" range of fit. This is surprisingly high, given the noisy data. The explanation is straightforward. We see from the distribution of weights and the .95CIs that these relatively high  $r^2_w$  values are due to the weighting function (Eq. 8) assigning large weights mostly to low values of TFH. The large majority of pilots have low TFH (median values  $x^{-}_{IIR}$ =823.5). Large *ns* increase the numerator of Eq. 8, thus bin weight. In practical terms, weights are simply reliability estimates that are higher when based on larger numbers coupled with proportions close to 0 or 1. The .95CI reflects this. As weights shrink, .95CI widens.

The downside of high reliability at low TFH is lower reliability at high TFH. Figures 8 and 9 show the penalty imposed by rate noise. We can predict fairly reliably within the low-end range of about 45-250 TFH for non-IR (representing about 50% of non-IR pilots) and about 45-500 TFH for IR pilots ( $\approx$ 30%). More accurate prediction awaits the arrival of larger datasets, an important point we will revisit in the Discussion section.

# **USING** $\Gamma_{rate}$

 $\Gamma_{rate}$  can either be used as a point-estimate of relative accident risk, or as a measure of cumulative risk over a known range of TFH. The latter is preferable but requires accurate values of TFH for each pilot at known start and end times. These are rarely available. Therefore, both estimators are defined.

# A Point-Estimate of GA Flight Risk

To estimate the relative accident rate for a given value of THF=*x*, (meaning the range from *x*-50 to *x*+50, flown over the course of 1 year), simply populate Eq. 4's parameters given the instrument rating of the pilot, and insert *x*. A variety of mathematical and statistical programs will handle the computation, including estimating  $\Gamma(\alpha)$ —*Mathematica*, MATLAB, SPSS, SAS, and *Excel*, to name a few.

To illustrate, for non-IR pilots with the median value of TFH $x_{xyp}$ =250.5, Eq. 4 becomes

$$.0025 + .013 \frac{e^{\frac{\ln(250.5) - .3648}{.0924}} (\ln(250.5) - .3648)^{64.38-1} .0924^{-64.38}}{\Gamma(64.38)}$$

With  $\Gamma(64.38) \approx 9.611 \times 10^{87}$ ,  $\Gamma_{rate} = .0068$  ( $\approx 1$  in 150) expected serious-to-fatal accidents per 100 TFH, which we can see is correct from Figure 8.<sup>5</sup>

This point-estimate can be used as a statistical covariate to control for estimated relative flight risk when the data contain only a single value of TFH for each pilot. What the point-estimate essentially embodies is an estimated average probability of having an accident while flying 100 flight hours over the course of one year, given that value of TFH=x as the midpoint (i.e.,  $x\pm 50$ ).

What we need to keep in mind while using such a pointestimate is the implicit assumption that a) all pilots fly the same number of hours per year, and b) their accident rate does not change during that time. Naturally, both assumptions are false for any given pilot and can only be safely ignored where large sample sizes tend to dampen statistical randomness.

In spite of these deficiencies, the point estimate should mark an improvement over a linear assumption of flight risk because it embodies a nonlinear relation derived from a reasonably large sample of GA pilots.

We now turn to a second, more comprehensive estimate of flight risk.

 $<sup>^4</sup>$  Since each rate bin is 100 TFH wide, the per-FH rate is simply the height of the y-axis divided by 100.

<sup>&</sup>lt;sup>5</sup>Gamma can be easily determined using the Microsoft *Excel*<sup>®</sup> function =EXP(GAMMALN( $\alpha$ ))

# A Cumulative Estimate of GA Flight Risk

In the fortunate circumstance where our data contain two separate, time-stamped values of TFH for each pilot,<sup>6</sup> we can represent each pilot's annualized accident probability as

$$\frac{1}{t_2 - t_1} \left( 1 - \lim_{dx \to 0} \prod_{TFH = x_1}^{x_2} \left( 1 - \frac{\Gamma_{rate}(x) \, dx}{100} \right) \right)$$
(12)

where  $t_2 t_1$  is the number of years over which the number of hours  $(TFH_2 - TFH_1)$  were flown. The term  $\Gamma_{rate}(x)$  is the value of Eq. 4, appropriate to the pilot's instrument rating, at TFH=x. Bin width is dx, which we could ideally shrink to zero to find the limit of the definite product  $\prod$ . Note that the constant 1/100 corrects for the original bin width of 100 TFH that we first used during curve-fitting. That is,  $\Gamma_{rate}$  is now renormalized to an hourly rate, and the total number of bins used to calculate the limit of Eq. 12 would be  $(TFH_2 - TFH_1)/dx$ .

Eq. 12 represents an idealized process of finding the overall probability of having an accident over a specified period of time. Each separate bin's probability  $p_i$  of having an accident is first transformed into the probability  $1-p_i$  of *not* having an accident over the small period dx. These separate  $1-p_i$ s are then multiplied together to let  $\Pi$  represent the probability  $p_{noAcc}$  of not having any accident across  $TFH_2-TFH_1$ . This then makes  $1-p_{noAcc}$  the probability of having at least one accident during  $TFH_2-TFH_1$ . Finally,  $(1-p_{noAcc})/(t_2-t_1)$  represents that as an annualized rate.

Finding a closed solution for the limit of Eq. 12 is elusive and unnecessary in practice. We can let dx=1.0 and get solutions accurate to about four decimal places—precise enough for our noisy empirical data. Eq. 12 then reduces to

$$\frac{1}{t_2 - t_1} \left( 1 - \prod_{TFH=x_1}^{x_2} \left( 1 - \frac{\Gamma_{rate}(x)}{100} \right) \right)$$
(13)

Eq. 13 is easily computed by a variety of common programs, using a simple For[] loop. *Excel* will do this as a macro, SPSS will do it with syntax, and so forth.

To illustrate, suppose an IR pilot has 347 TFH reported on 3 Feb, 2012, and 633 TFH on 13 Jun, 2014 (i.e., over 861 days). With parameters from Figure 9, and  $\Gamma(51.35)=1.202\cdot10^{65}$ ,  $\Gamma_{rate}$  becomes

$$\Gamma_{rate} = .0011 + .0174 \frac{e^{-(\ln(x) - 2.243)/.089} (\ln(x) - 2.243)^{50.35} .089^{-51.35}}{1.202 \cdot 10^{65}}$$
  
and Eq. 13 is

$$\frac{365}{861} \left( 1 - \prod_{x_1=347}^{633} \left( 1 - \frac{\Gamma_{rate}(x)}{100} \right) \right)$$

Pseudocode for  $\prod$  will resemble myPi=1.0; For[x=347 to 633; myPi = myPi \*(1-( $\Gamma_{rate}(x)/100$ )); x=x+1 ]; Final=(365/861)\*(1- myPi);

Here, Eq. 13 evaluates to about 0.0100 effective annual probability of having a serious-to-fatal accident, given the number of hours flown.

#### DISCUSSION

Can total flight hours predict general aviation accident rates? If so, what does that relation look like? Is there a "killing zone"—a range of TFH over which GA pilots are at greatest risk? These questions interest pilots, aviation policy makers, and insurance underwriters alike.

Craig (2001) proposed that such a killing zone does indeed exist, and that it spans the range of approximately 50 to 350 total flight hours. Unfortunately, his analysis relied solely on accident *frequency counts*, which fails to control for the number of *non*accident pilots having equivalent TFH. The use of accident *rates* solves that problem by dividing the number of accident pilots by the number of non-accident pilots in each bin of our TFH frequency histograms.

Now, given this more proper methodology, if such a killing zone truly exists, what does it look like? Many aviation research studies implicitly assume a straight-line relation between accident rates and TFH. They merely assume that risk decreases as pilots get more experienced. In fact, that relation appears markedly nonlinear.

The present work uses a nonlinear, gamma-based function  $(\Gamma_{nate})$  to predict GA accident rates, even from extremely noisy TFH data. Two log-transformed sets of 2003-2007 NTSB/FAA data produced weighted goodness-of-fit  $(R_{w}^{2})$  of .654 and .775 for non-instrument-rated and instrument-rated pilots, respectively, considered to be "moderately good" by convention.

<sup>&</sup>lt;sup>6</sup> All raw data should be checked to ensure that  $t_2 > t_1$  and  $TFH_2 > TFH_1$ . Experience with self-reported FAA data proves that this is usually (but not always) the case.

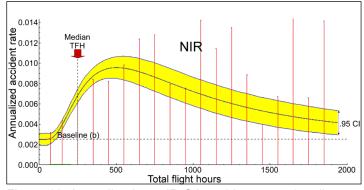


Figure 10. Annualized non-IR GA accident rates (median TFH=250.5).

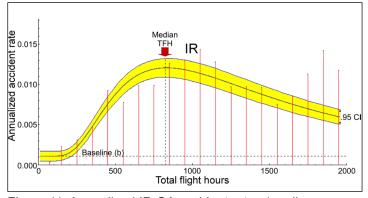


Figure 11. Annualized IR GA accident rates (median TFH=823.5).

Figures 10 and 11 show the rate data with the modeling function superimposed. The yellow region surrounding each modeling curve represents the .95 confidence interval.

Consistent with our intuition and the frequency count studies, these models suggest that a "killing zone" indeed exists. Accident rates seem to increase for GA pilots early in their post-certification careers, reaching a peak, and then declining with greater flight experience.

However, that zone may be far broader than earlier imagined. Relatively high risk for an individual pilot may extend well beyond the 2,000-hour mark before leveling off to a baseline rate.

For now, we should consider these conclusions tentative. First, the 67,687 pilots in these datasets constitute just over 10% of those currently licensed in the U.S. While this might represent an excellent dataset size in other venues, the relatively small number of pilots at high TFH leads to considerable noise in high-TFH rates. Data-weighting can overcome part of the problem but leads to wide confidence intervals at high TFH, reflecting the low statistical reliability of high-TFH data.

The net effect is that, at this point in time, we have high confidence in our data only at relatively low values of TFH. Better results await improved cross-communication between the FAA and NTSB databases, which will result in much larger sample sizes and much greater reliability across the full spectrum of TFH. We end with an appeal to the FAA and NTSB. The data needed to compute accident rates such as these come from multiple sources that are hard to access in the U.S. These include the FAA CAIS and DIWS databases, as well as the NTSB Aviation Accident Database. These databases do not intercommunicate well, and could be greatly improved by sharing a common pilot identifier code. The FAA has such an identifier called *UniqueID*, and sharing that with the NTSB would allow researchers to conduct studies with far greater sample sizes and reliability than is currently possible.

Nonetheless, we have explored a methodology here that will hopefully prove useful, either in its own right, or as a stimulus to further research.

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