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Ne Abstract								
This report reviews 47 survivable or partly survivable accidents investigated since								
1973 by personnel from the Civil Aeromedical Institute. The accidents were reviewed for a number of features of crashworthiness and, in particular, for injuries to occu-								
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pants in relation to the s		-	-					
restraint systems. Opinions were rendered by trained crash injury investigators as								
to the role or expected ro.	le in seats <mark>a</mark> n	d upper torso re	estraints in add	ling to or				
lessening the injuries.								
The data support the generation	al concepts th	at nonoccupiable	e portions of th	ne aircraft				
receive greater physical d								
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torso restraints, in the f	ew instances u	sed, were benef	icial, and had t	they been used				
by all occupants, would have	ve significant	ly reduced the	injuries.	-				
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CRASHWORTHINESS STUDIES: CABIN, SEAT, RESTRAINT, AND INJURY FINDINGS IN SFLECTED GENERAL AVIATION ACCIDENTS

I. INTRODUCTION.

The prime goal of aviation safety is to prevent injuries, loss of life, and loss of property. Of course, this is best done by keeping accidents from happening; the greatest efforts rightfully should be and are directed toward prevention.

However, accidents do happen and, based on past experience, they do occur with a certain predictability. Indeed, data gathered by the National Transportation Safety Board for a recent 6-year period (1973 through 1978) record a yearly average of 3,911 "small fixed-wing aircraft" (under 12,500 lb) in accidents. Of these, 663 (or 16.7 percent) resulted in one or more occupants being killed, with 1,303 being killed, as an average, or, statistically, two persons per fatal accident. In addition, there was untold injury, pain, suffering, and permanent disability in persons who survived the 663 (yearly average) fatal accidents or who were occupants in the 3,248 (yearly average) aircraft in nonfatal accidents.

Studies have shown that the human can withstand rather large impacts if the forces are properly distributed to the body. Such tolerances to decelerative forces have been amply demonstrated by a number of controlled studies using human subjects (1) and by findings in vehicular and other accidents. The tolerances, (withstanding decelerative forces without incurring permanent debilitation) are derived from evaluating impacts in relation to dynamic considerations such as rate of onset and duration of decelerative force acting upon the body. Besides varying with the rate of onset and duration, human tolerances are variable with other factors such as height, weight, and age of the individual; the type of restraint used; the application of the restraint to the body; etc. The crashworthiness load requirements applicable to seats and restraint systems specified in the Federal Aviation Regulations (2) are based on ultimate aircraft airworthiness load requirements met under static loading conditions. Although human tolerances to short duration dynamic loading appear to exceed several-fold the static loads applicable to seats and restraint systems, dynamic and static loading are not directly comparable. Specification of meaningful impact attenuating standards for seats and restraints will require definition of the dynamic components of crashes.

One of the greatest challenges to aviation safety in the coming years will be to make aircraft more crashworthy, i.e., to build and equip aircraft so that when a crash occurs the aircraft itself provides greater opportunity (within practical limitations) for reduced injury to occupants. Many of the developments in crashworthiness research are aimed at better cushioning of occupants against the decelerative forces of the crash. The most fruitful and practical means of doing this is by applying previously advocated packaging principles (3), and especially by improving seats and restraint systems (4).

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It is also helpful to analyze accidents to estimate the severity of the crash, noting the integrity of the structure, analyzing the performance of the restraint systems, and reviewing injuries received by occupants. Findings in accidents can be confirmed under controlled conditions in the laboratory.

For over a decade an ongoing biomedical and crash injury field investigation research program has been conducted at the FAA Civil Aeromedical Institute (CAMI). In this program, accidents were investigated to reveal any of a wide range of human factors such as: previous illnesses in the crew; medications or drugs taken by the crew; fatigue; physical stresses; psychological stresses; types of injuries received; causes of impact injuries; emergency egress from aircraft; smoke and fire as related to survivability; other environmental conditions such as water, ice, and snow, as related to postcrash survival; and a number of other biomedical factors that may have contributed to the crash or related to occupant injury or survival. Findings as related to survival of the impact have been a prominent feature of these investigations. Although each investigation was not undertaken specifically to investigate crashworthiness, certain such aspects came forth in many investigations. These included features such as the deformation of aircraft cockpit and cabin structures; the state of integrity and probable function of seats and restraint systems; probable impact of occupants against aircraft structures and the correlation of injuries with the direction and severity of impacts. The function and adequacy of seats and restraints have been of particular concern (5) because modifications of these systems, to give greater protection to occupants, often can be made at less expense to manufacturers or aircraft owners, than modification of the airframe. Indeed, some specific changes made by manufacturers, as a result of these investigative activities (6), have improved the crashworthiness of the respective aircraft and have saved lives.

For this report, we have surveyed a number of general aviation accidents for an overall assessment of findings, particularly as they relate to the function of the restraint system--seats, lapbelts, and shoulder harnesses. Elements of these data have been used in other reports (6).

II. METHODS.

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For this analysis we reviewed the reports of all general aviation accidents investigated by CAMI personnel from 1973 to and including most of 1979. Accidents investigated from CAMI prior to 1973 were previously reviewed (4). The current group of accidents was reviewed for a number of features of crashworthiness and, in particular, for the injuries to the occupants in relation to apparent severity of the impact and the adequacy of the function of the cabin and restraint systems. All aerial application aircraft accidents, accidents in which all occupants were killed, or where fire or water precluded a reasonable evaluation, were eliminated from the series. In all, 47 of a greater number of accidents were deemed worthy of more intensive review and tabulation, in that there was meaningful information in the accident reports or investigators were familiar enough with the particulars of the accidents

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to provide details. Trained crash injury investigators, who had personally investigated a number of these accidents or participated in the program at the time the accidents were investigated, reviewed all records and extracted data. In addition, these investigators, based on the information at hand, were asked to make judgments as to whether seats, lapbelts, upper torso restraints, or cabin structures were involved in producing or intensifying injuries in occupants. From these data a number of tables were derived in an attempt to answer certain questions pertaining to crashworthiness.

III. RESULTS.

The findings in the 47 accidents are shown in Table I (appended). Accompanying the table is the legend to codes used for representing the findings.

These 47 accidents involved 138 persons (including 2 lap-held children). There were 47 pilots, 40 occupants of the copilot seat, and 49 additional passengers (in seats other than the pilot and copilot seats). It was estimated that the major impact force was forward in 40 accidents, forward and left in 3, and forward and right in 1, both forward and vertical in 2 and only vertical in 1.

One aircraft crashed inverted and another cartwheeled. The remainder crashed on a straight or turning (coded as forward-turning) heading. Fortytwo accidents were judged to be survivable and the remaining five only partially survivable.

Survival of an aircraft accident depends to a great extent on providing a crash-resistant container for the occupants; that is, an occupiable area that will withstand crash forces without crushing, collapsing, or disintegrating. The accidents were judged on the basis of overall damage indices for nonoccupiable and occupiable areas. This crash severity index has been used at CAMI for a number of years. It is inadequate to describe fully what an investigator may observe but serves as a means of estimating damage so that accidents generally may be compared. Such a comparison is shown in Table II.

Damage, as assessed by this method, confirms what one would expect, that the nonoccupiable structures of wings, tail, and engine, sustain greater destructive damage than the more capsulized cabin. Indeed, the crumpling and breaking away of these exterior structures, to some extent, cushions the fuselage against the forces of the impact.

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Damage Index	Nonoccupiable (# of Accidents)	Occupiable (# of accidents)
Minor	None	8
Moderate	13	15
Moderately Severe.	10	13
Severe	13	7
Extremely Severe	5	1
Extreme	5	3
Unclassified	1	
Damage to:		# of Accidents
Nonoccupiable A	rea > Occupiable Area	30
Nonoccupiable A	rea = Occupiable Area	14
Nonoccupiable A	rea < Occupiable Area	2

TABLE II. Damage Indices (See Table I)

The results of a comparison between the damage to the cockpit area and the remainder of the cabin in 29 of the accidents (where such comparison was meaningful) are presented in Table III. Damage to the cockpit area was tabulated to be significantly greater in 13 of the accidents and equal in the remaining 16. In no instance was damage to the remainder of the cabin greater than to the cockpit area. In many individual accidents the differences in fore and aft damage in the occupiable areas were extreme.

TABLE III. Cockpit/Cabin Integrity in Accidents

	Cockpit (# of accidents)	Remainder of Cabin (# of accidents)
Intact	9	16
Distorted	5	6
Partly Collapsed	12	6
Collapsed	2	1
Burned	0	0
Disintegrated	1	0
Structural Damage to:		# of Accidents
Cockpit > Remaind	er of Cabin	13
Cockpit = Remaind	er of Cabin	16
Cockpit < Remaind	er of Cabin	0

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Who receives the worst injuries when both pilot and copilot positions are occupied? To explore this, the severity of injuries to the occupants of the pilot position (left front) and occupants of the copilot position (right front) was recorded. Of the 39 accidents, in which both positions were occupied, injuries to occupants of the pilot and copilot positions were greater in the pilot position in 10, greater in the copilot position in 10, and equal in the remaining 19. Of course, injuries are probably a function of which side of the aircraft impacts first. There were six fatalities at the pilot position and seven at the copilot position. These data suggest there is no difference between these two positions in regard to the severity of injuries received.

Is one likely to receive more serious injury when occupying the cockpit (pilot or copilot position) or a position behind the cockpit? Table IV presents data on 23 accidents in which there were occupants in passenger seats as well as the cockpit. The most serious injury of an occupant in passenger rows other than the first is included for completeness. The injuries listed represent only the worst injury an occupant or occupants received in their position in the aircraft. There were three accidents that involved a fatality in the cockpit. Of these three accidents, the most severe injury to other occupants in the aircraft was a "serious" injury. There were 16 accidents in which the most severe injury in the cockpit position was "serious," yet, in three of these, there was at least one fatality in the first row of passenger seats. There were four accidents in which injury to an occupant in the cockpit was minor/none; occupants in the first passenger row received "serious" in one accident and minor or no injuries in the other three accidents. In two accidents, the most severe injuries were in the second row of passenger seats. With some notable exceptions, such as case #27 in which occupants of the pilot and copilot seats survived but both occupants behind them received fatal injuries, these data tend to confirm the accumulated observational experiences of general aviation crash-injury investigators that persons in the pilot and copilot positions are subjected to greater impact forces and thus receive more severe injuries than occupants in rearward positions in the aircraft. There appears to be a cabin damage gradient in the occupiable areas, greater forward and diminishing rearward, and similarly there appears to be an occupant injury gradient, greater forward and diminishing rearward. The two are obviously correlated.

Cockpi	t		assen irst	-		assend			issenger hird Row
(Pilot-Cop Position		FAT	SER	MINOR NONE	FAT	SER	MINOR NONE	FAT	MINOR SER NONE
Fatal	3		2	1			1		
Serious	16	3	6	7		1	2		1
Minor/None	e 4		1	3		2			
Total	23	3	ò	11	<u>L</u>	3	3	1	1
*Figures a injury fo	-							-	sons) and worst

TABLE IV. Comparison of Injuries in Cockpit Area With Those Received in Other Locations in Aircraft*

Since the seat is an integral part of the aircraft occupant protection system, how did the seats function in these accidents and did seat failures or loss of adequate seat support add to the severity of the injuries received in the accidents analyzed?

Aircraft were found to have varying degrees of failures of the seats. Failures, to a great extent, varied with the design, installation, and position in the aircraft. For example, seats were found to fail at the attachment by sliding forward on the seat track, and to partially or completely detach from the track. Legs or seat pedestals were found to break, or break and the broken parts separate. For the most part, bending of legs and pedestals was considered beneficial to occupant protection. There were some failures of seat pans and seat backs. The data covered 136 seats. Of these, seat-to-track/floor attachments failed in 48, legs/pedestals failed in 25, and backs in 6. The distribution of these failures is represented by the data in Table V.

Seating Position	Attach	ments		Legs/Ped	lestals		Back		
	Failures	Total #	8	Failures	Total #	0,0	Failures	Total #	0 ⁰
Pilot	19	44	43	10	46	21	2	44	5
Copilot	16	39	41	9	38	24	4	39	10
1A	4	16	25	2	16	13	0	15	
1B	6	17	35	2	17	12	0	16	
2A	2	6	33	1	6	17	1	6	17
2B	1	6	17	1	6	17	1	6	17
3A	0	1		0	1		0	1	
3B	0	1		О	1		0	1	

TABLE V. Incidence of Seat Failures

Here again one can see a gradient of failure from forward to aft. From these data and the general experience of investigators, the greatest failures are in the pilot and copilot seats with the seat to track/floor attachments failing in approximately 40 percent of the accidents. In 20-25 percent of the accidents there was some breaking of the seat leg or pedestal. Other seats appeared to fare better but still there were enough failures to warrant concern.

For improved crashworthiness, seats should provide support for the occupants and attenuate both forward and vertical impact forces. Abrupt failure such as sliding forward, separating from the attachment to the floor of the aircraft, or breaking of the undersupport (legs/pedestals) allows occupants to impact against the floor, instrument panels, and other occupants or structures so that the decelerative forces are greater and injuries are incurred. Similarly, in some respects, a seat that is rigid and unyielding may intensify injuries. There are no FAA requirements for seats to attenuate decelerative forces. The accidents were reviewed with the question in mind that, from practical considerations, did the seats contribute to the severity of the injuries? Such data are tabulated in Table VI.

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TABLE	VI.	Contribution	of	Seats	to	Severity	of	Injury
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Seat Position	Seat Contributed To Severity	Seat Did Not Contribute To Injury	Undetermined
Pilot	16	26	5
Copilot	11	26	3
Passengers			_1
Total Percent	41 30	86 63	9 7

In 30 percent of the accidents, malfunction of a seat component (some factor in the seat), fracture of legs, separation from the seat-track, etc., contributed to injuries of occupants over and above what would have been expected from $in_{\rm PD}$ forces. In some accidents it was obvious that a factor in the seat design was a contributor to injuries.

Almost all seats were forward-facing but there were, in these aircraft, six aft-facing and three fixed side-facing seats that were occupied. Two occupants of side-facing seats received only minor injuries, (Case ± 25). In another (Case ± 37), the only occupant to receive greater than minor injuries was in a side-facing seat. This occupant had serious abdominal injuries related to seatbelt compression of internal organs.

The tubular frame of one of two aft-facing seats in Case #21 broke, allowing the occupant to come forward and strike the pilot from behind, adding to the pilot's injuries, as he was more forcefully driven into the instrument panel. Only minor injuries were incurred in two aft-facing seats in Cases #25 and #37. Injuries occurred to occupants of aft-facing seats in Case #47, but both seats were loosened by severe cabin and floor damage and occupants in their seats were thrown out of the aircraft.

The standard method of restraining occupants in an aircraft is by means of a lapbelt. In only two accidents were there well-documented lapbelt failures. In one (Case #10), the lapbelt attachment to the floor of the aircraft failed, allowing the pilot to be hurled out of the cabin and receive fatal injuries. In Case #11, a severe impact, both lapbelts failed and the occupants were thrown free of the aircraft. Both occupants survived.

An upper torso restraint (UTR) (or other adequate head protection in accidents) has been mandated in some aircraft by the Federal Aviation Regulations (8,9). In accidents reviewed, 57 occupants had the availability of a UTR. Of these, seven were used and held. For six occupants the use and function of a UTR was unknown. The remainder (44) did not use the available UTR.

Based on their familiarity with the accident or their experience as crash-injury investigators, the reviewers correlated the injuries in each accident with the apparent dynamic scenario of the crash. For each occupant of each aircraft they then estimated whether or not, in their opinion, a UTR would have been of value in reducing injuries in this selected series of accidents. These estimates along with the occupiable area severity damage are shown in Table VII.

Among these accidents there are rare examples in which a UTR was used and greatly aided in survivability of the occupant. Unfortunately, most of the occupants of the aircraft did not have the advantage of having a UTR available and, for the most part, those who had them available did not use them. Among pilots, an estimated 43 would have benefited from a UTR, versus 4 who would not have benefited. Among copilots 36 would have benefited as compared with 4 who would not have benefited. Similarly, among passengers, 42 would have benefited as compared with 11 who would not have benefited. It is apparent from these selected accidents and these estimates, that UTR's would have reduced the severity of injuries to aircraft occupants in all positions. These findings and experienced opinions are consistent with other field investigative findings, laboratory dynamic studies, and FAA requirement that general aviation aircraft manufactured after July 18, 1978 have UTR's installed for each front seat.

Injuries to aircraft occupants by seat position are shown in Table VIII. There were 17 fatalities, mostly in the pilot and copilot positions. Those injuries classified as serious with 10 percent or more residual disability, such as the loss of an eye, an extremity, or the impaired ability to work, all occurred in persons in pilot and copilot positions.

The known types of serious injuries received are shown in Table IX. Pilots and copilots received roughly a third of their injuries to the head and face, a third to the chest and a third to the spine. Spinal injuries appeared to predominate in passengers although about one-fourth of injuries were to the head and face. A further look at spinal injuries comes from Table X in which known spinal injuries and compression fractures of vertebrae are tabulated. These figures show that the majority of serious spinal injuries in aircraft accident victims is compression fractures.

IV. DISCUSSION.

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The data in this retrospective study, like much accident data, were not collected under a protocol that forced investigators to document specific findings such as attachments of all seats or precise review of hospital records on each occupant for exact details of injuries. Even so, the data recorded, findings familiar to the investigator, and the photographs allow a reasonably good overall evaluation of each accident.

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TABLE VII

			Occupiabl	Occupiable Area Damage Index	ge Index		
				Modcrately Extremely	Extremely	-	
Number of:	Minimum	Minimum Moderate	Severe	Severe	Severe	Extreme	Total
Accidents	ω	15	12	7	Ħ	Ф	47
Persons	23	41	31	22	4	15	136
Pilots would have been helped	٢	13	12	Q	Г	4	43
Pilots would not have been helped	Т		Ч	, 1		Ч	4
Copilots would have been helped	Q	10	IO	Ŷ		m	36
Copilots would not have been helped	Т		ч	ч		Ţ	4
Passengers would have been helped	æ	16	ω	ŝ		5	42
Passengers would not have been helped		Ц		4		2	7

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Seat Position	Fatal	Serious With Residual	Serious	Minor	No Significant Abnormalities/ NONE	Unknown
Pilots	6	4	24	12	0	1
Copilots	7	2	20	9	2	0
1A	1		8	5	2	1
1B	3		6	5	3	
2A			3	2	1	
2B			3	2	1	
2C				3	1	
3B						
			 -	—	<u> </u>	
Totals	17	6	64	37	* 10	2
Percent	12.5	4.4	47.0	27.2	7.4	1.5

TABLE VIII. Injuries to Aircraft Occupants

TABLE IX. Distribution of Major Injuries

	Total #	Head a	and Face	Cł	nest	Abo	domen	S	pine
Position	Tabulated	#	<u></u>	#		#		#	*
Pilot	37	11	30	12	32	1		13	35
Copilot	37	11	30	10	27	3	8	13	42
Passengers	26	6	23	6	23	1		13	46

TABLE X. Spinal Injuries

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	Spinal	Injuries #	Compressio #	on Fractures
Pilot Copilot		13 13	9 5	69 38
Passengers		13	10	77

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The accidents reviewed here confirm what is apparent to aircraft accident investigators, that:

1. The nonoccupiable portions of the aircraft receive greater physical damage than the occupiable areas.

2. If occupants are to survive the accident, the cockpit/cabin should remain reasonably intact and not collapse upon the occupants.

3. The greatest damage to the occupiable area is to the forward portion of the cockpit/cabin.

4. Impact forces on the aircraft, for the most part, cause greater injuries to occupants seated in the forward position of the cockpit/cabin than those stated more rearward.

What is not always apparent to general aviation accident investigators is that, in specific accidents, injuries and even overall survivability of the impact may be related to a lack of incorporation of crishworthiness features of the aircraft. Investigators intent on determining the cause of the accident may overlook the fact that occupants may have survived the accident had some feature not been present, had a seat not failed, or had a shoulder restraint been used. Also, they may not take cognizance of the fact that a properly restrained occupant in some crashes may withstand impact forces that would severely damage the integrity of the aircraft. Each of the accidents reviewed was survivable or partly survivable from the standpoint of what a well-restrained occupant can withstand.

A basic principle of occupant survivability is that the container (the cockpit/cabin) remain intact and not crush in upon the occupants. Experience reveals that in most accidents the forward portion of the aircraft, the landing gear and the underside receive the brunt of the impact forces. Generally, crushing is from forward to aft in such a way that the pilot and copilot are subjected to more longitudinal force than occupants seated behind them. There appears to be no difference of injury potential between the pilot and the copilot positions. Passengers have the advantages of more bending, crushing, and deformation of aircraft structures forward of them so that they are spared the full impact forces experienced in the pilot and copilot positions. This is brought out even in this limited data.

To withstand inpact forces, occupants should be adequately restrained. The seat is an integral part of any restraint system and the optimum design should cushion the occupant against forces, particularly forward and vertical forces, which are greatest in almost all accidents. Ideally, a seat should initially resist impact forces and then bend and deform in a controlled and progressive manner so as to attenuate and keep forces below a level that would cause serious injuries to the occupant. A rigid nonyielding or hard seat can lead to high peak loads on the occupant causing serious injuries. A frangible seat, one in which the attachments or seat parts break during impact, can lead to high peak impact forces on the occupants during secondary impacts with aircraft floors, panels or other structures. Seat placement (over main spar, near the floor or on or near other nonyielding structures) or seat failures of one degree or another were judged to have intensified the injuries of occupants in at least 30 percent of the accidents reviewed. Common findings were: failure of latching pins to restrain seats from traveling forward on seat tracks; detachment from seat tracks, usually by breaking of either the track or the track-attachment mechanism; and fracture of seat legs and pedestals. These and other findings (6) in which seats and seating placement appeared to be a factor raise the question of the crashworthiness suitability of seats in general aviation aircraft. In view of current FAA regulations prescribing minimum seat strength based on static testing (2), the data and observations in this report, along with other accident data, indicate that an area for improvement in occupant survivability is in providing seats that attenuate impact force to levels that can be tolerated. Additional documentation of seats as related to injuries in general aviation accidents is the subject of an ongoing accident investigation protocol in the FAA.

Except for lap-held infants and children, lapbelts were used by all occupants of the aircraft reviewed. Only a few lapbelt failures were noted and these primarily were due to failure at the attachment rather than the webbing. These findings support the general impression that if the aircraft impact is in any way survivable, the belt webbing rarely fails unless it is severely weathered and frayed, as seen in some aerial application aircraft, or it is configured so that the force of impact causes the fitting to cut the fabric. The weakest portion of the lapbelt system appears to be its attachments to the floor or aircraft structures.

Aircraft occupants use the lapbelt restraint but, for the most part, do not use the UTR. The value of restraining the upper torso cannot be overemphasized. For example, a seated passenger is restrained by a lapbelt and his/her upper torso may weigh as much as 120 lb. In an accident, the lapbelt holds the pelvis and acts as a fulcrum about which the upper torso rotates under the force of deceleration. If the deceleration is low, 2 G's, the upper torso will have an apparent weight of 240 lb, so that the occupant can barely resist the forward thrust. At 10 G's, well within the survivability envelope, the apparent weight of the upper torso will be 1,200 lb and it will swing forward with great velocity, possibly hitting the head on the instrument panel and the chest against the control wheel. Based on the velocity of the upper torso and head and the stopping distance, a force of several hundred G's may be exerted on the skull or chest. This rationale is supported by the finding that about 70 percent of general aviation accident fatalities have fractures of the skull (7). Crushing of the chest is common. These observations were made before UTR's were mandatory in aircraft.

Thus, for years it has been known that UTR's would be lifesaving to aircraft occupants in accidents. The double shoulder harness worn by aerial application pilots has saved hundreds of lives. Unfortunately there are few findings of other general aviation aircraft occupants wearing a UTR at time of impact. Of the 57 occupants of aircraft in this report who had a UTR available, only 7 used them and the UTR appeared to have lessened injuries. An outstanding example of the value of a UTR is Case #33 where the occupant in the copilot seat, an FAA employee, was estimated to have survived only because he had on the single shoulder harness.

Estimates based on accident investigation experience, as reflected in Table VII, show that of the 136 persons evaluated in the 47 accidents, 121 persons would have benefited by a UTR; the remainder would not have benefited.

The FAA has taken steps which should lead to improved occupant protection in survivable aircraft accidents. The Federal Aviation Regulations (FAR) have been changed so that since July 18, 1977, all new typecertificated airplanes must be equipped with UTR's in the front seats. For a pilot to operate a small civil airplane manufactured after July 18, 1978, the airplane must have, for each front seat, a shoulder harness designed to protect the occupant from serious head injury when the occupant experiences the ultimate inertia forces specified in other parts of the FAR (9). In addition, the FAR mandate that UTR's be worn on all takeoffs and landings by each required flight crewmember of a civil airplane, if the airplane is equipped with a shoulder harness and if the shoulder harness does not interfere with performance of duties (10). There is no provision that, in new type-certificated or newly manufactured aircraft, other seating positions (except for additional crew positions) be equipped with a means of restraining the upper torso. Neither is there provision that aircraft manufactured before the stated date be retrofitted with UTR's in any position. Crash injury experiences in other vehicles, decelerative testing under laboratory conditions, general aviation accident experience, and the experience and data in this report, all indicate that general aviation aircraft occupants under condition of impact, would benefit from wearing The FAA's requirement of a UTR in certain airplanes and other a UTR. crashworthiness improvements such as removal of sharp objects, installation of padding, etc., should reduce injuries and improve survivability.

The figures in Table IX indicate that in roughly a third of the occupants, severe injuries are to the head and face, a third to the chest, and a third to the spine. For the most part, in accidents where the cockpit/cabin retains its integrity and is not crushed upon its occupants, the severe head and face injuries probably result from the unrestrained torso traveling forward against aircraft structures. For the pilot and copilot positions this is most frequently the instrument panel or structural members. For other occupants, head and face injuries, usually less severe than for pilot and copilot positions, are received as they flex forward into the seats in front of them or move laterally into aircraft structures. Chest injuries in the pilot and copilot position frequently

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result from impact with the control wheel or by forward flexure onto one's own legs. Seats that travel forward, or that partially or fully detach, add to head and chest injuries. Crushed chests are less frequent in passenger positions, but can result from flexing forward and striking one's own knees. Both types of injuries would appear to be lessened by restraining the upper torso.

Spinal injuries are usually attributed to severe downloading. Overly rigid seats, seats that break and let the occupant "bottom out" on the floor, or seats that are positioned over solid structures or other unyielding structures, add to the severity of spinal injuries. Compression fractures of lower thoracic or lumbar vertebrae were conspicuous in the accidents reviewed.

This type of injury probably results from downloading on the spine or forward flexion over the lapbelt. The seat and restraint as an integrated system is apparent when one considers how a UTR may work. Restraint of forward motion and maintenance of the body in an upright position by the UTR in many instances will increase downloading on the spine--and on the seat. Increased loading on the spine should intensify injuries. It is thus apparent that the seat should be designed to attenuate this increased downloading so as to lessen injuries. The value of a seat that can attenuate these and other forces on the occupants cannot be overemphasized. The specifics of spinal injuries and seat failures should be given special emphasis in aircraft accident investigations as UTR's become more widely used. The overall and specific functioning of UTR's in general aviation accidents is the subject of an accident investigation protocol within the FAA.

The data from the 47 accidents in this report suggest that, although variable with the specific airplane, the greatest crash protection for the occupants of general aviation aircraft can be offered by providing each with a UTR (with strong attachments) and a well-anchored impact attenuating seat. This can only be accomplished though at a significant cost for newly manufactured airplanes and a major cost as a retrofit item.

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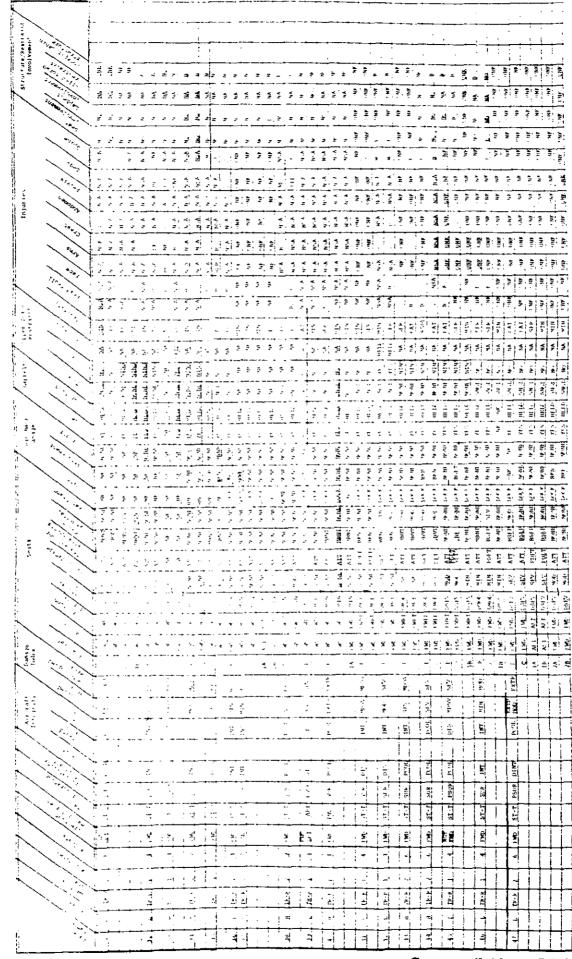
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CODING FOR CRASHWORTHINESS DATA

		XI. SBARS (Cor'L)	XI. <u>SEVIS</u> (ton't)
• •			h. Pan/France:
11.	5° NTM	1. I ALLEY.	And the second
	- 1.0×	PWD - Cotward	HOME - NO GEFORGIOTUL VI VIIVIT
	H - Mugh	APT - AKT SMVL - SMLVEL	MOD - moderate
	e numeros	EXSD - FIXEd side factors	SEV - severe of selving. . not wool cable
• • •		BEIC - bench alde factry	1917 - Jukhowe
	¶f - tijsycle ™ma - railwhorel	· DIFECTION OF POLANCE-LAIDIN ALL FOIL	
	I - fixed	char - Januard	
	R - retractable	PWPT - forward and right	a Tourt (type of attachment of the ch
	5.413 JUL 2	PGT right	27442/T 13
		NALT STUDIE AFTER AFT	st - fixed, rigid, not noveable
	1, 2, 3, 41 .	IAFT TRET AND THE	$\frac{1}{100}$ [nck=ble
	styles a	1177 - 1117 174114 - 1 1144 1151	1, x, k → ninged and torkable
	1, 2, 1, atc.		The statements
•	ATTICTION SF MAJOR FURCES	2.	(sther in writing)
			h Atto Demonst.
	PAD forward		Larage of seat thick attectaunt to
	AFT - TOARWARD LAT - Left	active provide a state	
	RGT - right	11, 2011 11, 201	byagi - ro damago osoèn , ⊸rocer bendrou at attachuent
	VERT - VERTION SPEER OF COMPLIATION OF the BLAVE		SHAT - SEVER DERALLIG AL ALLACTMENT
11	CRASH CONTLUSION UND	الإيلان من عن من المالية ، من المن الأراد المنافق من	
	CART · cartwheeled	التوالية المراجعة المعالية المنابعة المراجعة المراجعة المراجع المعالية المراجع المعالية المراجعة المراجع المراج المراجعة المراجعة الم	
	INV - inverted ST-T - straight of turning upright	and rate rates and rates being the second	Istradit Damage to teatract
		بند تابين "لينديا لا تريابا الإليانية المواهد من بند. ````````````````````````````````````	
VI11.	SURVIVABLUITY OF CHPACT	دور المراجع المراجع المراجع من المراجع المراجع المساحلة المراجع المراجع المراجع المراجع المراجع المراجع المراجع	
	<u>508</u> - survivable	בן, יצידיין ז'ידיין איניייני ונפרוויין ייני אורייים, אימייין ייניואיי	
	<u> 2908 - partiy turvivable</u> C <u>908</u> - murvivable by chance	Any/You attached	
	NSUR - noneurvivable	JRP	(hack from pan) SBMT - geveraly bant forward
ĽΧ	COCKPIT/CAMIN INTEGRITY	, join, stary for T (/ ton the seas	(toward pen) cours = excernic lent hav kværd
			(back from part)
	L. Forward Cabin	ATT syent tematured attack of hub sold	BKN - frame broken but not seperated
	c. Aft Cabin d. Occupied Area	POET - cost fattially be when from	HAEFY - ITALION DIOXOF and supplemented NA - not applicable
		DET - such completely left for the first of the second sec	UNK - Unkream
	<u>ijir</u> - intact		SU132 UV1 1111/
	<u>010</u> - distorted PCOL - partly collapsed	a Suma light first for the fort in the sum of	A, USA:
	COL - collapted	HORE - No SHAE deformation	
	DIN - distrated	pigner - munor bending of deformation of seat ley ana/or prodotal	NINS - not installed
х.	DANAGE INDEX (See below)	<u>Sairt - revert bending ur teformatium</u> But seat logs abd/or potestale	<u>1017</u> - Inscented Not view
КI	54.975	not broken 	h. Function.
;		BAN - 1043 AIR/UL JENEFICE - 1042	
	6. RON/SEAL	BARP - Lega and/or pedestal broken 10 several naces and merated	H <u>eid</u> - heid with no problem R ^{eig} t - complete failure at right
	<u>p</u> - pilot <u>c</u> - copilot	uA - not applicable UNX - unknown	attachment LFLR - complete failure at left
	ROW 1 A D C etc.		attachment CFLR - complete failure both
	Row 2 A B C etc. Row 3 A B C etc.		attachments OFLR - failure of hardware other
			than attachment

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CODING FOR CRASHWORTHINESS DATA

VONE - no damate UNE - number bendtangte (teamated paril (teamated paril) ULLA from paril (tack from paril (tack from paril SMM - teame broker and apported BAL - noknown UNE - unknown yayı - re-Jamauyo - yayı - ser-Jamauyo - Seyre - bendising at attachaent #Yy: - zroken bar noi fetarlard #Y: - unkreahi - unkreami j∡į - fixed, rugid, net muveable 10<u>1</u> - inged, muveable lat no interestic 1,χ,γ - ingel and uskable 101, net applicable 111, - net applicable 111, - net applicable HELD - held with no problem R⁻LR - complete failure at right attachment OFLR - failure of hardware other than attachment 192411: - no deformation or damage 1411: - antor 1420: - amoderato 1524: - awere or reparted 1824: - not applicable 1824: - not applicable a lourt ... (type of attachment of tack Atta INNERT Larage of seat brok attastaunt to seat Juni: LTLR - complete failure at left attachment JE 12 AM DAMAJE CO SEALESCH USE 12 AM (COTHER TEAM) SET SCHWART CO 2401 failure both not vsed (other in writing) <u>YFS</u> - used <u>NINS</u> - not installed <u>NO</u> - installed not use <u>UNK</u> - unknown CFLR - complete ALL SEAT BACK DAMAGE h. Function. h. Pan/Frame: XI. SEATS (Con't) /111. LAP BELTS ---With the foregraphic of the block constrained of the strain of the block constraints on agric to the strain of the block on agric to the strain of the block of the block of the block of the strained of the block of the block of the strained of the block of adds, about the strained of the block ATT - weat remained (Litzhend for That for T SLID - weat remained attached but sold DDT - weat completely (Portsfer for the sold for The - weat completely (Portsfer for the sold for QLK - addression) <u>1895</u> - 1894 ani/or vedestal larokor in several places and separated VA - not applicable UNK - unknowr DIFECTION OF FORMAND-LADORAL-ALL FOL. Saat legs and/or fedestil Damage Sear The Track (Let A Street Track DALF LUMINET MULTURE DALE LUMINET CONTENT DELE LUMINET CONTENT DELE LUMINET CONTENT NULL LUMINET CONTENT SEC. QUART SUCCE P.W. - Corvard M.T. - aft S.M.L. - aktvel S.M.L. - bartel S.M.C. - barch alde fartro B.S.T. - barch alde fartro eventer Part forward and right Pert right Part right and aft AFT sti LAST किट्टे कानी किं LET 1.10€ PadLast rwattarel कि tan e not hβlls delle Jar e sekrowe. Varting 1 Purces THE WORLD XI. SUARS (Cor't) h. Englig--, 7 • FMA: Forward 127 - nostward 1277 - nostward 1277 - ruch 1277 - rucht 1277 - ruchtia 1277 - ructical CART - Cartwheeled <u>INV</u> - inverted <u>ST-</u>T - straight of turning upright <u>1987</u> - intact <u>D10</u> - distorted <u>PCOL</u> - purtly collapsed <u>NCL</u> - collapsed <u>NUR</u> - burned <u>D111</u> - disintegrated <u>568</u> - survivable <u>PSGR</u> - partly survivable c<u>SGUR</u> - murvivable by chance <u>MSUR</u> - nonsurvivable X. DAMAGE INDEX (See below) Row -- 1 A B C etc. Row -- 2 A B C etc. Row -- 3 A B C etc. DIFILITION OF MAJOR FURCES VIII. SURVIVABILITY OF LAPAGT TX COCKPET/CAMER INTEGRETY Aft Cabin Occupted Area Mon-Occupted Area CRASH_CONTIGURATION Cockpit Forward Cabin TF - LicyLe TW - milwheel F - fixed R - retractable <u>2</u> - pilot <u>C</u> - copilet A. ROM/Seat 1, 2, 1, 400 1.2.3. 477 11011 111 нын - Н STV35 • · r.. Evines 1. 2ASL * XI. BEATS 11. <u>HIN 5</u> 479.94.9 ;; ٦

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CCDING (Continued)

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X111	(1, uo,) <u>Slije</u> der	COURS (Con't)
	CMEH - COMPLEAS FAILURE OF WEDDING	
		Å - blunt trauma with serious fracturus.
	(OUNT IN WILLING)	maxilla and/or other factal hones # = Mine account other material
4	UPPER TURE RESTINIED ("TRO- (SHO-LDEB PARA.")	g - Diunt traiada with modelately seriotis fractures, such an mandubly,
	idX; p	C - blunt trauma with minor fractures. manufale
	•	
		E - penetrating trauma with permanent
		r provisions and lacerations rectures 7 - abrasions and lacerations appears
	10 Car	
	rits - and	to moderate NCR - no attactivent attochmative
	•	UNK - UNKnown
		l Arma Designate 5 for left A for right for extreme ride.
	· Function	
		APPULATION
	REAM - AGAINATE SU PRADUCT AFGA - SOMPLATE FALLIES AT AFS STATIONS	
	٠	t - เอคระเมาส่^รณฑตรเทนted ใช้สุดขับระค
	•	LOWER ALM
	19659 - affiai failife of wather: Chate - commiste destruction of satistic	i fracture/fractures upper arm
	NA - PORT NOT DATE AND	I TRACTORE/TRACTORES (QSGI A.2) · · Tractores (state
	(other write su)	
		* staticated stoulder * - Atalocated structure
XV.	CODES FOR ENCORTEM AND INDUST CAUSES	
		ζ.
		Nor - no stanticat abrotatica Dat - sobrosse
	FATL - fatal	
		chest.
	dteathalltty excentral 140 - aerione	A . blane second side
	MIM - MINOR	s, start transmo with truthing of opening of chest revity, heart and human
	NONE - DOPA	B - Munt traine with fractures of sternum
	UNX - WKUNN	and/or ribe - not lethal if
	b. Haad, Shuil, and Brain	Le penetrating wound in chest, 2 or more
	or partial decapitation	c. SMERTEASING WOUND IN CREAT, JAMA Chan 2 INCHAR IN dismarks
	g - blunt traumme with lenser degree of skull	
		tearing or rupture
	• •	H - Curruston of heart - non-jethal I - combination of lights - sociation.
	K - penetrating trains without fractures	closed pneumothorax
	r - abrasions and lacerations - severe	K - bleeding into plaural cavitica
	La e altreatone and langerations e mundr fo motarate s e letata lanatation diete frankingen	<u>1985</u> - No alguificant almosmalifican UNK - unknown
		Aldomen
	「「 」 mg1d to moderate brain contumion and/ot いっついろ	t a thickness and an or a factor of the second s
	uttertany 1441 - no stanificant shrortailtist	
		2 - Internal bleeding severe E - Internal blueding moderate

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CODING (Continued)

XVI. CAUSES OF INJURIES

- Seat Involvement 9
- No caused no injury
- I = distorted and cushioned impactB = Partially broke, not adding to injuryC = partially broke, adding to injuryD = failed badly, did not add to injuryE = failed badly, added to injuriesMA = does not applyUNX = unknown
- Lapbelt Involvement <u>م</u>

- M9 caused no injury
 A left abrasions and contusions on pelvis, abdomen
 B apparently rode high and compressed abdomen
 C without serious injury
 C with internal injuries
 M1 internal injuries
 MM unknown
 Fig. failure

- Upper Torse Restraint (shoulder barness) З

- MD caused no injury
 A = left abrastions and market on chest without injury
 B = abrastions and centusion of chest
 C = contuston and fracture of chest
 C = contuston and fracture of chest
 MA = doss not apply
 UNK = unknown
- Cockpit/Cabin Structure Involvement ÷

- MD caused no injury
 A = struck instament panel
 a struck back of scart
 c = struck bartition or divider
 B = struck by flying object
 F = impact with ruder pedals
 f = impact with windscreen or windows
 f = struck cockpit/cabin structural member, post, etc.
 d = struck overhead

X. DAMAGE INDEX

< 1	DAMAGE SEVERITY **** STRUCTURE OF FIN	OCCUPINALE AREA	SCORE
	Intact	7	
	Distorted		
	Bent/Partially Collapsed	~	
	Collapsed/Buckled	ę	
	Torn-Free Disintegrated	÷	
	SCORE DEGREE OF DAMAGE	, 'IOTAL	
	0~25 Minor		MIN
	Muderato		dow
	Moderately Severe		ASGM
	Jijj Savere		SKV
	1742) Extremely Severe		EXSV -
	21436 Extreme		ENTR
	Unable to Classify		UCLA
i,		STATUTE AND A DESCRIPTION OF A DESCRIPTI	

DAMAGE SEVENTY **** NON OCCUPINELE AREA

		NOSE TATL	RZMG LZWG	GRADE	SCORE	ж
Intact	ct			-		
Dist	Distorted/Wrinkled			2		
Bent. Coll	Bent/Partially Collapsed			3		
Buckled/ Crumpled	led/ pled			4		
Broken, Col Lepi	Broken/ collapsed			5		· ·
Torn Dist	Torn-Free Disintegrated			ر ب		
SCORF	DEGREE OF DAMAGE	MGE		TOTAL		
4 -0					NIM	
5-8	Moderate				MOR	
9-12	Moderately Severe	evere			VSOM	2
13-16	Severe				SEV	
7-20	Extremely Severe	vere			L EXSV	N
21-24	Extreme				EXTR	ĸ
	Unable to Classify	asaify			Lucta	Y

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